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# Material-Structure Integrated Design Optimization of GFRP Bridge Deck on Steel Girder

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## Abstract:

Design optimization of fiber-reinforced polymeric (FRP) composite products is essential to facilitate their applications in engineering structures. For bridge structures, the main design optimization goals are the reduction of FRP material consumption and the structure weight, which aim to reduce the initial construction cost and achieve a longer bridge span. Compared with conventional steel-concrete composite bridges, FRP-steel composite bridges possess more design variables and more complex design process, which necessitate the simplified optimization models. This paper aims to propose a two-scale design optimization method for FRP bridge deck on the steel girder. The macro behavior of the pultruded FRP composite bridge deck is analyzed. Regarding the micro level, the equivalent properties of pultruded GFRP lamination are calculated by combining micromechanics and classical lamination theory (CLT). The above-mentioned macro pultruded GFRP bridge level and the micro fiber/resin level were bridged based on the assumption that the micro-component effective homogenized strain equals to the corresponding macro strain. The two-scale lamination optimization of pultruded GFRP bridge deck is finally achieved by finding optimized two-scale design variables that can achieve the minimum bridge weight or the lowest initial construction cost with all listed constraint requirements satisfied. A pultruded FRP deck supported on equally-spaced steel girders was selected as a case study to show how to obtain

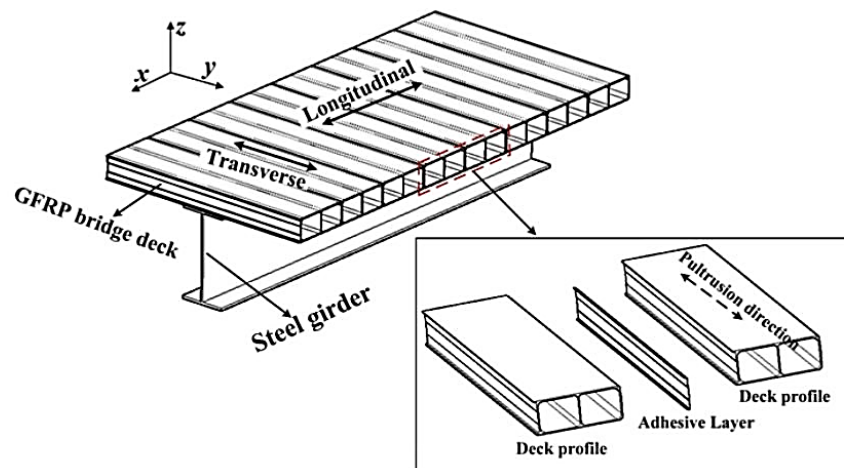
26 the optimized two-scale parameters by using this proposed optimization method. The optimized results of  
27 the top flange thickness,  $t_u$ , the bottom flange thickness,  $t_l$ , the web height,  $h_w$ , and the web thickness per  
28 meter,  $t_w$ , are 46.02 mm, 45.86 mm, 300.0 mm and 37.42 mm, respectively. Results also showed that the  
29 optimized ratio of the  $0^\circ$ -lamina,  $45^\circ$ -lamina, and the  $90^\circ$ -lamina are 77.9%, 17.1%, 5.0%. The optimized  
30 fiber volume fraction is 65.2%.

31 **Keywords:** Composite bridge girder; Pultruded GFRP bridge deck; Laminations; Multiscale  
32 optimization.

33

## 34 1. Introduction

35 Fiber-reinforced polymer (FRP) composites have been greatly developed worldwide and have  
36 become one of the most popular construction materials for repair and rehabilitation and new construction  
37 [1–10]. Pultruded glass fiber reinforced polymer (GFRP) composites are great candidates for newly  
38 constructed bridges decks. A variety of GFRP bridge deck applications are presented in [11]. Figure 1  
39 shows a commonly used composite girder system which consists of the pultruded GFRP bridge deck and  
40 the supporting steel girders. Noted that the the steel girder with a corrugated web [12] is also an  
41 interesting surrogate. The pultruded GFRP bridge decks and steel girders can be connected using  
42 adhesives or bolts [13].



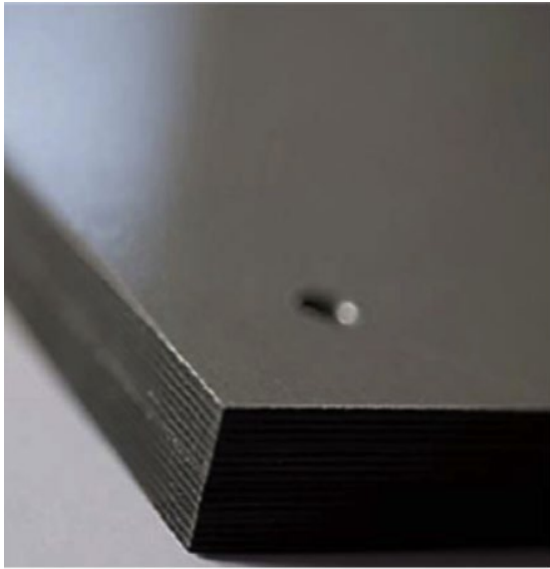
43

44 **Figure 1.** Pultruded GFRP bridge deck and steel girder system [3]

45 Different from conventional isotropic construction materials like reinforced concrete and steel,  
46 GFRP composites are inhomogeneous and anisotropic, which require to be analyzed and designed on  
47 different scales, namely, the micro-scale and macro-scale. The importance of multi-scale analysis to  
48 determine the mechanical properties of GFRP materials has been pointed out in previous studies[14,15].

49 During the design stage of a GFRP bridge deck, engineers are not only interested in fulfilling the  
50 strength and serviceability requirements, which are the top design priorities, but also in satisfying these  
51 requirements with the least possible amount of materials that will result in a weight reduction of the  
52 structure and further achieve lower initial construction cost. Thus optimization techniques is very  
53 important in obtaining the best use of FRP material in bridge decks. The optimization tasks involve  
54 determining the optimal ratio of fiber reinforcements, the optimum fiber volume fractions and geometric  
55 variables in order to achieve the best design in both material and structure scales. In addition, the  
56 complexity of general pultruded GFRP bridge decks necessitates the development of simplified  
57 optimization models.

58 Most of the previous optimization work in the design of composite structures [16–20] focused on  
59 aerospace structures, but pultruded GFRP composites, commonly used in bridge decks, are quite different  
60 in nature with the composites used in aerospace structures [15], as can be reflected in Figure 2. These  
61 differences include: (i) the pultruded FRP laminations have a relatively poor quality, and (ii) the roving  
62 content is larger than fabrics, leading to an increase in the thickness of the unidirectional lamina  
63 ( $0^\circ$ -lamina) of up to 5–15 times the laminas with other orientations.



(a) High-quality



(b) Pultruded

**Figure 2.** Difference in quality and accuracy of stacking sequence of composite laminates[15]

64 A pilot investigation related to material-structure integrated design is presented in this paper. The  
65 macro behavior of the pultruded FRP composite bridge deck is analyzed. Regarding the micro level, the  
66 equivalent properties of pultruded GFRP lamination are calculated by combining micromechanics and  
67 classical lamination theory (CLT). The above-mentioned macro pultruded GFRP bridge level and the  
68 micro fiber/resin level were bridged based on the assumption that the micro-component effective  
69 homogenized strain equals to the corresponding macro strain. The two-scale lamination optimization of  
70 pultruded GFRP bridge deck is finally achieved by finding optimized two-scale design variables that can  
71 achieve the minimum bridge weight or the lowest initial construction cost with all listed constraint  
72 requirements satisfied. Also, a case study was presented to show how to obtain the optimized two-scale  
73 parameters by adopting the proposed optimization method in the last part of this paper.

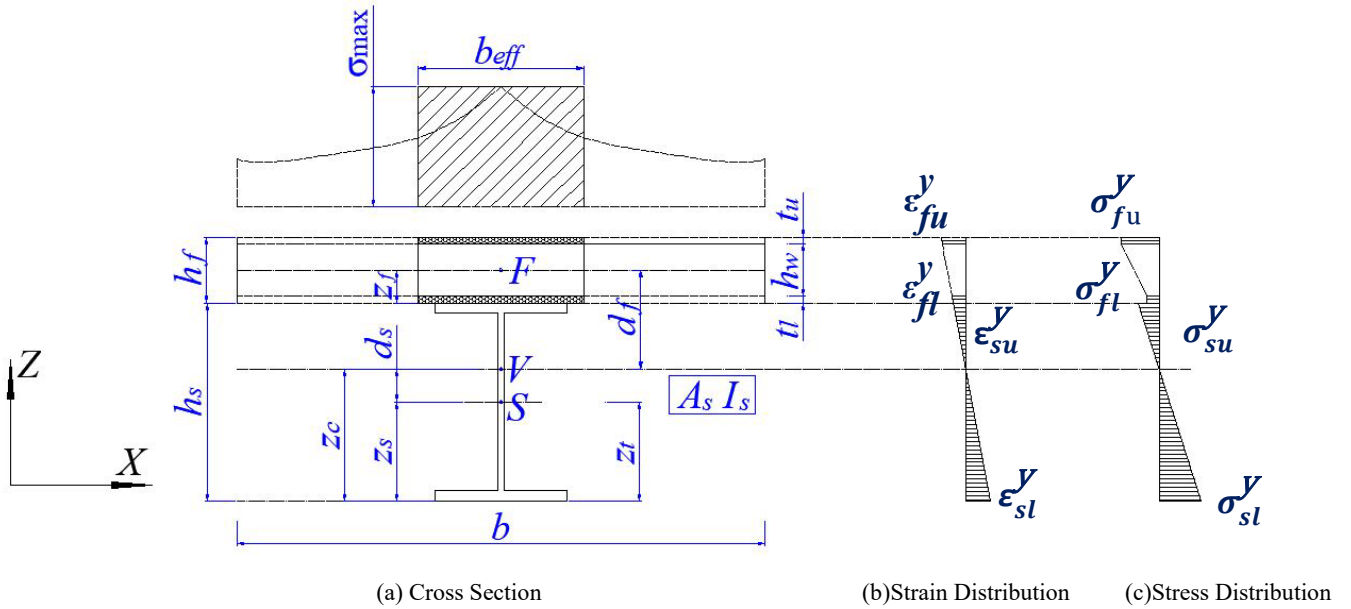
## 74 **2. Macro Behaviour of the Pultruded GFRP Composite Bridge Deck**

75 GFRP composite bridge decks, together with the supporting steel girders, were subjected to  
76 longitudinal bending moment ( $M^L$ ) and shear force ( $Q^L$ ), as well as transverse bending moment ( $M^T$ )  
77 and shear force ( $Q^T$ ). The following sections would describe the mechanical behaviors of bridge deck in

78 both the longitudinal and transverse directions under corresponding bending moment and shear force.

## 79 2.1 Macro Behavior in the Longitudinal Direction

80 Following assumptions were made to analyze the mechanical behavior of the composite girder along  
 81 longitudinal direction: (i) the shear connection stiffness is sufficient to ensure a full composite action  
 82 between the GFRP bridge deck and the supporting steel girder; (ii) the longitudinal shear forces are fully  
 83 resisted by the steel webs; (iii) the macro longitudinal stresses are uniformly distributed along the flange  
 84 thickness considering the fact that the laminate thickness dimension is quite small relative to the total  
 85 height of the steel girder; (iv) the flexural and shear resistances provided by discontinuous web along the  
 86 longitudinal direction are neglected.



88 **Figure 3.** Schematic of composite cross section

89 Due to the in-plane shear flexibility of the GFRP composite deck, the normal stress along the width  
 90 of the deck is non-uniformly distributed, see Figure 3. The maximum stress in the deck occurs in the  
 91 centerline of the web and stresses in the bridge deck away from the web lag behind [21]. Thus, the  
 92 effective flange width,  $b_{eff}$ , is introduced in design practice to simplify the analytical procedure, as  
 93 denoted in Figure 3. The effective flange width,  $b_{eff}$ , is defined as a reduced width of the deck over which  
 94 the normal stresses are assumed to be uniformly distributed, and it is calculated [22] based on the premise

95 that the stress resultant over the effective width should be equal to the stress resultant over the  
 96 actual flange width, as defined in Eq. (1).

$$97 \quad b_{eff} = \frac{\int_0^b \sigma_f^L(x) dx}{(\sigma_f^L)_{max}} \quad (1)$$

98 where:  $\sigma_f^L(x)$  is longitudinal normal stress in the flange of GFRP bridge deck;  $(\sigma_f^L)_{max}$  is the maximum  
 99 longitudinal normal stress in the flange of GFRP bridge deck, and  $b$  is the center-to-center spacing of  
 100 the steel girders.

101 The effective flange width of the GFRP bridge deck supported by the steel girders can be simply  
 102 predicted [22] by using Eqs. (2) and (3) as follows:

$$103 \quad b_{eff} = R b_{eff,s} \quad (2)$$

$$104 \quad R = 1.025(1 - 0.0244^g) \quad (3)$$

105 where:  $b_{eff,s}$  is the effective width suggested by highway bridges design specifications [30, 31], and  $g$  is  
 106 the degree of composite action between the GFRP composite bridge deck and the main girders. The  
 107 longitudinal normal stresses at the top flange,  $\sigma_{fu}^y$ , and the bottom flange,  $\sigma_{fl}^y$ , can be calculated by Eqs.  
 108 (4) and (5) as follows:

$$109 \quad \sigma_{fu}^y = -\frac{M^L z_{fu}^L}{nI_v} \quad (4)$$

$$110 \quad \sigma_{fl}^y = -\frac{M^L z_{fl}^L}{nI_v} \quad (5)$$

111 where:  $n$  is the elastic moduli ratio (modular ratio) between steel modulus ( $E_s$ ) and the longitudinal  
 112 modulus of the GFRP composites deck ( $E_f^y$ ) and is expressed by Eq. (6):

$$113 \quad n = \frac{E_s}{E_f^y} \quad (6)$$

114  $z_{fu}^L$ , and  $z_{fl}^L$  are the distances from the top and the bottom flanges of the GFRP deck to the neutral axis  
 115 of the GFRP/steel composite girder,  $z_c$ , respectively. Thus:

116 
$$z_{fu}^L = h_s + t_l + h_w + t_u / 2 - z_c \quad (7)$$

117 
$$z_{fl}^L = h_s + \frac{t_l}{2} - z_c \quad (8)$$

118 The distance between the neutral axis of the GFRP/steel composite girder and the bottom fiber of the  
 119 steel girder,  $z_c$ , is calculated by the following equation:

120 
$$z_c = \frac{2A_s E_s z_s + b_{eff} E_f^y (t_l^2 + t_u^2 + 2t_u h_w + 2t_u t_l + 2t_u h_s + 2h_s t_l)}{2(A_s E_s + b_{eff} t_u E_f^y + b_{eff} t_l E_f^y)} \quad (9)$$

121 The equivalent moment of inertia of the GFRP/steel composite girder  $I_v$  could be calculated by Eq.  
 122 (10).

123 
$$I_v = I_s + A_s (z_c - z_s)^2 + b_{eff} (t_u^3 + t_l^3) / (12n) + b_{eff} (t_u + t_l) (h_s + z_f - z_c)^2 / n \quad (10)$$

124 
$$z_f = \frac{t_l^2 + t_u^2 + 2t_u h_w + 2t_u t_l}{2(t_l + t_u)} \quad (11)$$

125 where:  $h_s$  is the height of the steel beam;  $t_l$  is the thickness of bottom flange;  $t_u$  is the thickness of the  
 126 top flange;  $h_w$  is the web height of pultruded GFRP bridge deck;  $A_s$  is the cross-sectional area of the  
 127 steel beam, and  $z_s$  is the distance between the neutral axis and the bottom fiber of the steel girder.

## 128 2.2 Macro Behavior in the Transverse Direction

129 The following assumptions were made to analyze the longitudinal mechanical behavior of the  
 130 GFRP-steel composite girder: (i) the transverse shear force is fully resisted by the web of GFRP bridge  
 131 deck; (ii) the transverse normal stress is uniformly distributed along with the top/bottom flange thickness.

132 The transverse normal stress in the top flange  $\sigma_{fu}^x$  and bottom flange  $\sigma_{fl}^x$ , as denoted in Figure 4,  
 133 could be calculated based on Eqs. (12)–(13).

134 
$$\sigma_{fu}^x = -\frac{M^T z_{fu}^T}{I_f^x} \quad (12)$$

135 
$$\sigma_{fl}^x = \frac{M^T z_{fl}^T}{I_f^x} \quad (13)$$

136 where: the transverse moment of inertia  $I_f^x$  of pultruded GFRP bridge deck is:



$$I_f^x = \frac{1}{12} (1000 t_u^3 + 1000 t_l^3 + t_w h_w^3) + 1000 t_u (z_{fu}^T)^2 + 1000 t_l (z_{fl}^T)^2 + t_w h_w (z_{fw}^T)^2 \quad (14)$$

Note positive and negative signs in Eqs. (12)–(13) represent tensile and compressive stresses, respectively.

The web thickness per meter  $t_w$  along longitudinal direction is calculated by Eq. (15).

$$t_w = \frac{1000}{a} \sum_i^n t_{w(i)} \quad (15)$$

where,  $a$  is the width of GFPP deck profile, and  $t_{w(i)}$  is the thickness of web in each GFRP deck profile.

$z_{fu}^T$  in Eq. (11) and  $z_{fl}^T$  in Eq. (12) respectively refers to the distances between the top/bottom flange of GFRP composite bridge deck and its neutral axis, and can be calculated by Eqs. (16) and (17), respectively:

$$z_{fu}^T = t_l + h_w + t_u - z_f^T \quad (16)$$

$$z_{fl}^T = z_f^T \quad (17)$$

where the height of the GFRP bridge deck neutral axis along the transverse direction,  $z_f^T$ , is given by Eq. (18):

$$z_f^T = \frac{1000(t_u t_l + t_u h_w) + h_w t_w (t_l + h_w / 2) + 500(t_l^2 + t_u^2)}{1000 t_u + h_w t_w + 1000 t_l} \quad (18)$$

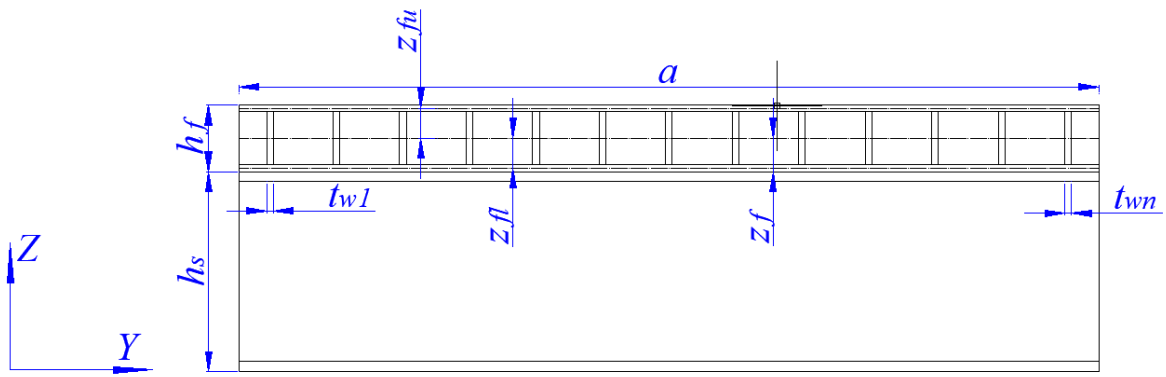
The shear stress,  $\tau_{fw}^{xy}$  in the web of the pultruded GFRP bridge deck is calculated by Eqs. (19):

$$\tau_{fw}^{xy} = \frac{Q^T}{t_w h_w} \quad (19)$$

In order to guarantee a safe design, the GFRP bridge deck is assumed simply supported by steel girder. The transverse deflection of pultruded GFRP bridge deck can be conservatively predicted using Timoshenko beam theory [25]:

$$\delta_f^{z(\max)} = \frac{5M^T b^2}{48E_f^x I_f^x} + \frac{Q^T b}{4t_w h_w G_f^{xy}} \quad (20)$$

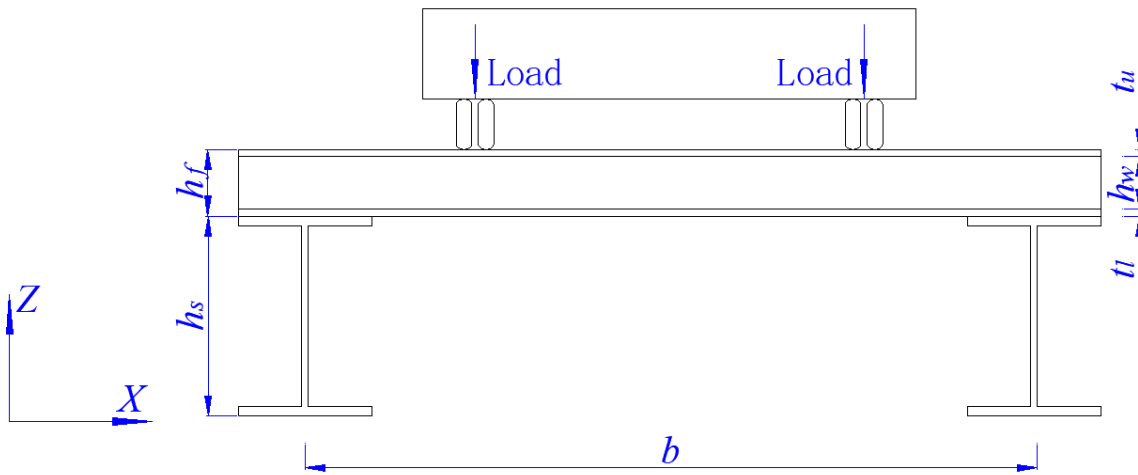
158 where:  $E_f^x$  and  $G_f^{xy}$  are the elastic and in-plane shear moduli of the GFRP composite bridge deck in the  
 159 transverse direction.



160

161

(a) Geometry symbols in YZ plane



162

163

(b) Geometry symbols in XZ plane

164

**Figure 4.** GFRP/Steel composite bridge girder parameters

### 165 **3. Micro behavior of pultruded lamination**

166 The reinforcements used for manufacturing the pultruded GFRP composite bridge deck described in  
 167 this paper are composed of (i) unidirectional E-glass roving, and (ii) non-crimp (multi-warp knitted)  
 168 fabrics [15]. In general, the laminations lay-up includes three different types of the lamina, namely,  
 169  $0^0$ -plies in the form of E-glass roving, and non-crimp E-glass fabrics with  $90^0$  and  $\pm 45^0$  orientations.  
 170 Based on the classical lamination theory [26], the effective modulus of the pultruded laminate could be

171 estimated using Eqs. (21)–(23), assuming that the ratio of  $0^\circ$ ,  $45^\circ$ , and  $90^\circ$  lamina to the total  
 172 lamination are  $\xi_0$ ,  $\xi_{45}$ , and  $\xi_{90}$ , respectively.

$$173 \quad E_f^x = \xi_0 \left[ \frac{E_1 - \nu_{12}^2 E_2}{1 - \nu_{12} \nu_{21}} \right] + \xi_{90} \left[ \frac{E_1 E_2 - \nu_{12}^2 E_2^2}{E_1 (1 - \nu_{12} \nu_{21})} \right] + \xi_{45} \frac{1}{(1 - \nu_{12} \nu_{21})} \left[ \frac{(E_1 + E_2 + 2\nu_{12} E_2 + 4(1 - \nu_{12} \nu_{21}) G_{12})^2 - 16(\nu_{12} E_2)^2}{4[E_1 + E_2 + 2\nu_{12} E_2 + 4(1 - \nu_{12} \nu_{21}) G_{12}]} \right] \quad (21)$$

$$174 \quad E_f^y = \xi_0 \left[ \frac{E_1 E_2 - \nu_{12}^2 E_2^2}{E_1 (1 - \nu_{12} \nu_{21})} \right] + \xi_{90} \left[ \frac{E_1 - \nu_{12}^2 E_2}{1 - \nu_{12} \nu_{21}} \right] + \xi_{45} \frac{1}{(1 - \nu_{12} \nu_{21})} \left[ \frac{(E_1 + E_2 + 2\nu_{12} E_2 + 4(1 - \nu_{12} \nu_{21}) G_{12})^2 - 16(\nu_{12} E_2)^2}{4[E_1 + E_2 + 2\nu_{12} E_2 + 4(1 - \nu_{12} \nu_{21}) G_{12}]} \right] \quad (22)$$

$$175 \quad G_f^{xy} = \xi_0 G_{12} + \xi_{90} G_{12} + \xi_{45} \left[ \frac{E_1 + E_2 - 2\nu_{12} E_2}{4(1 - \nu_{12} \nu_{21})} \right] \quad (23)$$

176 where:  $E_f^x$  is the effective elastic modulus of GFRP laminates in the longitudinal direction of the bridge;  
 177  $E_f^y$  is the effective elastic modulus of GFRP laminates in the transverse direction of the bridge;  $G_f^{xy}$  is  
 178 the effective in-plane shear modulus of GFRP laminates.

179 The longitudinal modulus,  $E_1$ , transverse modulus,  $E_2$ , shear modulus,  $G_{12}$ , and Poisson's ratio,  $\nu_{12}$  of  
 180 the lamina can be determined based on the modified mixture formulae [6]:

$$181 \quad E_1 = E_{f1} V_f + E_m V_m \quad (24)$$

$$182 \quad E_2 = \frac{E_{f2} E_m [V_f + \eta_{12} V_m]}{E_m V_f + E_{f2} \eta_{12} V_m} \quad (25)$$

$$183 \quad \eta_{12} = \frac{0.2}{1 - \nu_m} \left( 1.1 - \sqrt{\frac{E_m}{E_{f1}}} + \frac{3.5 E_m}{E_{f1}} \right) (1 + 0.22 V_f) \quad (26)$$

$$184 \quad G_{12} = \frac{G_f G_m (V_f + \eta_{12} V_m)}{G_m V_f + G_f \eta_{12} V_m} \quad (27)$$

$$185 \quad \eta_{12} = 0.28 + \sqrt{\frac{E_m}{E_f}} \quad (28)$$

$$186 \quad \nu_{12} = \nu_f V_f + \nu_m V_m \quad (29)$$

187 where:  $E_{f1}$  is the longitudinal elastic modulus of fiber,  $E_{f2}$  is the transverse elastic modulus of fiber,  $V_f$  is

188 the fiber volume fraction,  $\nu_f$  is the fibers' Poisson's ratio,  $E_m$  is the matrix elastic modulus,  $V_m$  is the resin  
 189 volume fraction,  $\nu_m$  is the matrix's Poisson's ratio,  $G_f$  is the shear modulus of fibers, and  $G_m$  is the resin  
 190 shear modulus.

191 The strength-based design method is accepted in many design practices, however, in this study, the  
 192 variation of elastic moduli and ultimate strength of each lamina complicates the lamination optimization  
 193 procedures. Thus, the strain-based design method is adopted in this paper.

194 By neglecting the curvature effects, the ultimate strain of each ply in the laminate is deemed to be  
 195 the same based on First-Ply-Failure (FPF) analytical method [26]. The ultimate strain of each lamina can  
 196 be obtained based on the micromechanics approach [6] using Eqs. (30)–(34).

$$197 \quad {}^t \varepsilon_1^u = \frac{X_T}{E_1} = \frac{X_{f1}}{E_{f1}} = {}^t \varepsilon_f^u \quad (30)$$

$$198 \quad {}^c \varepsilon_1^u = \frac{X_C}{E_1} = \frac{X_{fc}}{E_{f1}} = {}^c \varepsilon_f^u \quad (31)$$

$$199 \quad {}^t \varepsilon_2^u = \frac{Y_T}{E_2} = \frac{X_{mt}}{SCF} \frac{(E_m V_f + E_{f2} \eta_2 V_m)}{E_{f2} E_m (V_f + \eta_2 V_m)} \quad (32)$$

$$200 \quad {}^c \varepsilon_2^u = \frac{Y_C}{E_2} = \varepsilon_{mc} \left[ 1 - (4V_f / \pi) \left( 1 - \frac{E_m}{E_{f2}} \right) \right] \quad (33)$$

$$201 \quad {}^u \gamma_{12} = \frac{S}{G_{12}} = S_m \left[ 1 + (V_f - \sqrt{V_f}) \left( 1 - \frac{G_m}{G_f} \right) \right] \frac{(G_m V_f + G_f \eta_{12} V_m)}{G_f G_m (V_f + \eta_{12} V_m)} \quad (34)$$

202 When applying loads along the pultrusion direction, the ultimate strain of the  $0^0$ ,  $90^0$ ,  $\pm 45^0$  lamina is  
 203  $\varepsilon_1^u$ ,  $\varepsilon_2^u$ , and  ${}^u \gamma$ , respectively. When the loads are applied perpendicular to the pultrusion direction, the  
 204 ultimate strain of  $0^0$ ,  $90^0$ ,  $\pm 45^0$  lamina is  $\varepsilon_2^u$ ,  $\varepsilon_1^u$  and  ${}^u \gamma$ , respectively. Based on the “First-Ply-Failure”  
 205 failure criterion, the ultimate strain of each lamina can be calculated using Eqs. (35)–(37). The ultimate  
 206 strain variation as related to fiber volume fraction is shown in Fig. 5. These values were calculated using  
 207 Eqs. (31)–(37) with material properties listed in Tables 1 and 2 [27].

$$208 \quad {}^t \varepsilon^u = \min({}^t \varepsilon_1^u, {}^t \varepsilon_2^u, {}^u \gamma) = {}^t \varepsilon_2^u \quad (35)$$

209

$${}^c \varepsilon^u = \min({}^c \varepsilon_1^u, {}^c \varepsilon_2^u, {}^u \gamma) = {}^c \varepsilon_2^u \quad (36)$$

210

$$\gamma^u = \gamma_{12}^u \quad (37)$$

211

**Table 1.** Mechanical properties of E-glass fibers[15]

Longitudinal modulus ( $E_{f1}$ )	Transverse modulus ( $E_{f2}$ )	Poisson's ratio ( $\nu_f$ )	Shear modulus ( $G_f$ )	Tensile strength ( $X_{ft}$ )	Compressive strength ( $X_{fc}$ )	Density ( $\rho$ )
74.0 GPa	74.0 GPa	0.20	30.80 GPa	2150 MPa	1450 MPa	2560 kg/m <sup>3</sup>

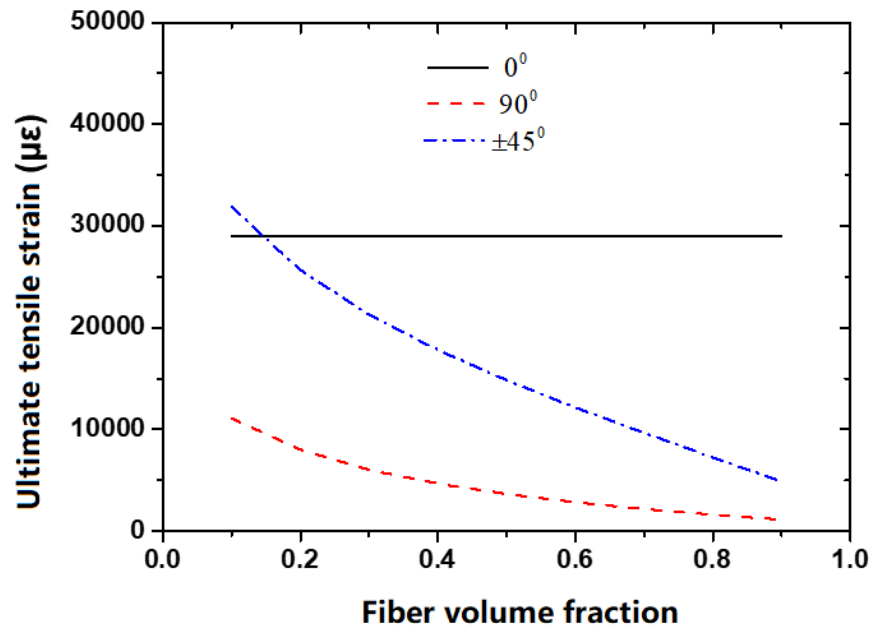
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**Table 2.** Mechanical properties of epoxy resin [15]

Modulus ( $E_m$ )	Poisson's ratio ( $\nu_m$ )	Shear modulus ( $G_m$ )	Tensile strength ( $X_{mt}$ )	Compressive strength ( $X_{mc}$ )	Shear strength ( $S_m$ )	Density ( $\rho$ )
3.35 GPa	0.35	1.24 GPa	80 MPa	120 MPa	75 MPa	1160 kg/m <sup>3</sup>

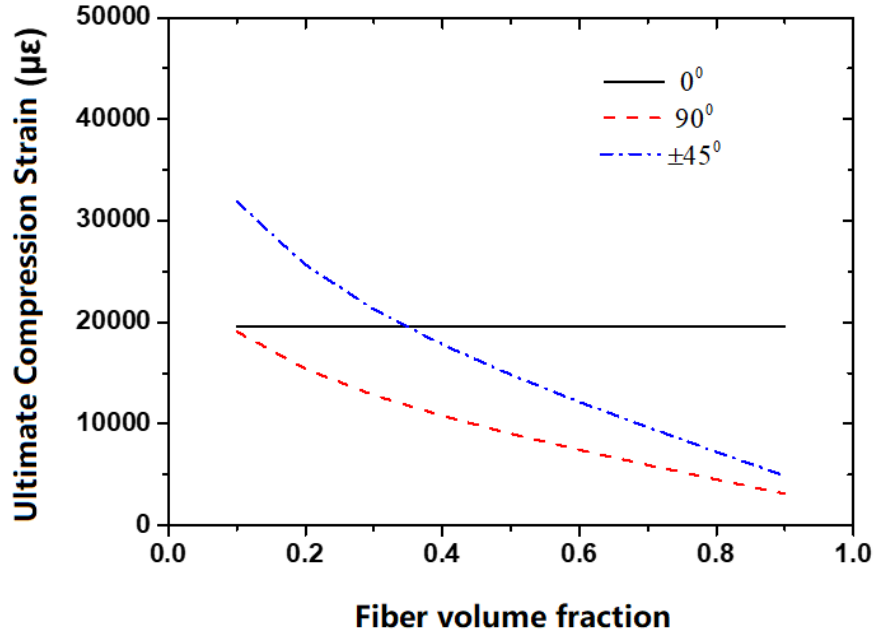
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215

216

(a) Tension



(b) Compression

Figure 5. Ultimate strain variation of FRP lamina in relation to fiber volume fraction

#### 4. Design values

In general, bridge structural members are exposed to harsh and changing environments such as moisture, salt-spray agents, freeze-thaw cycles, and large variations in both temperature and humidity[28–32]. Due to continuous exposure to such harsh environments, degradation in the mechanical properties of composites is expected to occur [28–32]. In this section, the assumption was made that the design strain equals to the product of ultimate strain and a reduction or a degradation coefficient. For the top flange of a pultruded GFRP composite bridge deck, we have:

$$\begin{cases} \left| \bar{\varepsilon}_{fu}^x \right| \leq {}^c \varepsilon^d \approx {}^c \chi_d {}^c \varepsilon^u \\ \left| \bar{\varepsilon}_{fu}^y \right| \leq {}^c \varepsilon^d \approx {}^c \chi_d {}^c \varepsilon^u \end{cases} \quad (38)$$

for the bottom flange of pultruded GFRP bridge deck, we have:

$$\begin{cases} \left| \bar{\varepsilon}_{fl}^x \right| \leq {}^t \varepsilon^d \approx {}^t \chi_d {}^t \varepsilon^u \\ \left| \bar{\varepsilon}_{fl}^y \right| \leq {}^c \varepsilon^d \approx {}^c \chi_d {}^c \varepsilon^u \end{cases} \quad (39)$$

and for the web of pultruded GFRP bridge deck, we have:

231 
$$\left| \overline{\gamma}_w^{xy} \right| \leq {}^s \gamma^d \approx {}^s \chi_d {}^s \gamma^u \quad (40)$$

232 where:  ${}^c \chi_d$ ,  ${}^t \chi_d$ ,  ${}^s \chi_d$  are the reduction (degradation) coefficients for GFRP materials in compression,  
 233 in tension, and in shear, respectively.

234 The Chinese Technical Code for Infrastructure Application of FRP Composites (GB 50608-2010)  
 235 [32] suggests that the design values are determined by dividing experimental ultimate strength by  
 236 appropriate partial safety factors that account for material type, and the surrounding environment. The  
 237 following equations can be used to calculate the reduction (degradation) coefficient:

238 
$$\chi_d = \frac{\mu^u - 1.645\sigma}{\mu^u} \frac{1}{\gamma_f \gamma_e} \quad (41)$$

239 where:  $\mu^u$  is the average material strength;  $\sigma$  is the standard derivation of the test number;  $\gamma_f$  is the  
 240 partial safety factor to account for material type;  $\gamma_e$  is the partial safety factor to account for  
 241 environmental exposure.

242 In addition, the transverse deflection of the pultruded GFRP bridge deck should always be smaller  
 243 than a limiting transverse deflection to ensure the stiffness requirement.

244 
$$\delta_f^{z(\max)} \leq \delta^u \quad (42)$$

245 where:  $\delta_f^{z(\max)}$  is the maximum transverse deflection of GFRP bridge deck under applied load,  $\delta^u$  is the  
 246 limited transverse deflection based on the design requirement.

## 247 **5. Bridging fiber/resin level to structure level**

248 In this section, the micro fiber/resin scale is bridged to the macro-the GFRP/steel composite girder  
 249 scale by assuming that the effective homogenized strain obtained from micro-component equals to  
 250 macro-strain. Linking micro and macro longitudinal and transverse strains at the top flange of a pultruded  
 251 GFRP bridge deck is achieved by using the following equations:

252 
$$\overline{\varepsilon}_{fu}^x = \frac{\overline{\sigma}_{fu}^x}{E_f^x} \approx - \frac{M^T z_{fu}^T}{E_f^x I_f^x} \quad (43)$$

253

$$\bar{\varepsilon}_{fu}^y = \frac{\bar{\sigma}_{fu}^y}{E_f^y} \approx -\frac{M^L z_{fu}^L}{nI_v E_f^y} \quad (44)$$

254

255

Similarly, linking both micro and macro longitudinal and transverse strains at the bottom flange of a pultruded GFRP bridge deck is achieved by the following equations:

256

$$\bar{\varepsilon}_{fl}^x = \frac{\bar{\sigma}_{fl}^x}{E_f^x} \approx \frac{M^T z_{fl}^T}{E_f^x I_f^x} \quad (45)$$

257

$$\bar{\varepsilon}_{fl}^y = -\frac{M^L z_{fl}^L}{nI_v E_f^y} \quad (46)$$

258

259

Eq. (48) shows how to link the micro and macro shear strains at the web of a pultruded GFRP bridge deck:

260

$$\bar{\gamma}_w^{xy} = \frac{\bar{\tau}_w^{xy}}{G_f^{xy}} = \frac{Q^T}{t_w h_w G_f^{xy}} \quad (47)$$

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## 6. Optimization equations for pultruded bridge decks

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The main goals of multiscale optimization of GFRP bridge decks towards material-structure integrated design are to achieve: (i) the lightest weight to increase the bridge span while satisfying all design and manufacturing requirements, or (ii) the lowest cost for the economy and constructional convenience. Mathematically speaking, the multiscale optimization of GFRP bridge decks is to seek a minimum value of cost or weight by optimizing multiscale design variables within given allowed constrained functions determined by design and manufacturing requirements. In this paper, the multiscale lamination optimization of a pultruded GFRP bridge deck is achieved by finding an optimized two-scale design variable vector,  $\mathbf{x}$ , that drive the objective weight function,  $\Phi_1$ , or the objective price function,  $\Phi_2$ , to its lowest values while satisfying all constraint functions ( $\Phi^1 \sim \Phi^6$ ). The design variables, objective functions, and constraint functions will be explained in the following sections.

272

273

**(1) Design Variables:** Eq. (48) describes the two-scale optimization design variable vector,  $\mathbf{x}$ , including the thickness of the top flange,  $t_u$ , the thickness of the bottom flange,  $t_l$ , the height of the web,



274  $h_w$ , the thickness per meter of the web,  $t_w$ , the ratio of  $0^\circ$ ,  $45^\circ$ ,  $90^\circ$  lamina to the total laminate are  $\xi_0$ ,  
 275  $\xi_{45}$ , and  $\xi_{90}$ , respectively, and the fiber volume fraction  $V_f$ .

$$276 \quad x = [t_u, t_l, h_w, t_w, \xi_0, \xi_{90}, \xi_{45}, V_f]^T \quad (48)$$

277 **(2) Objective function:** The objective function  $\phi_1$  related to the optimizing weight is given as  
 278 follows:

$$279 \quad \phi_1 = 1000 [1000(t_u + t_l) + h_w t_w] (\rho_f V_f + \rho_m V_m) \quad (49)$$

280 where:  $\rho_f$  is fiber density, and  $\rho_m$  is the resin density.

281 The objective function,  $\phi_2$ , related to the optimizing cost is given in Eqn. (50). It should be noted that  
 282 the manufacturing cost is not included in this expression due to the fact that different manufacturers have  
 283 different selling prices.

$$284 \quad \phi_2 = 1000 [1000(t_u + t_l) + h_w t_w] (\eta_f \rho_f V_f + \eta_m \rho_m V_m) \quad (50)$$

285 where:  $\eta_f$  is the price of the fibers, and  $\eta_m$  is the price of the matrix.

286 **(3) Constraint functions:** In this study, a total of six constraint functions were specified as follows.

287 (i) Constraint function  $\Phi^1$  (strength requirement of the top flange):

288 The longitudinal and transverse normal strains at the top flange of the GFRP deck should be smaller  
 289 than corresponding design values of normal strains, i.e.,

$$290 \quad \Phi^1 = \begin{cases} |\bar{\epsilon}_{fu}^x| \leq {}^t \epsilon^d \\ |\bar{\epsilon}_{fu}^y| \leq {}^c \epsilon^d \end{cases} \quad (51)$$

291 (ii) Constraint function  $\Phi^2$  (strength requirement of the bottom flange):

292 The longitudinal and transverse normal strains at the bottom flange of the GFRP deck should be  
 293 smaller than corresponding allowable maximum normal strains, i.e.,

$$294 \quad \Phi^2 = \begin{cases} |\bar{\epsilon}_{fl}^x| \leq {}^t \epsilon^d \\ |\bar{\epsilon}_{fl}^y| \leq {}^c \epsilon^d \end{cases} \quad (52)$$

295 (iii) Constraint function  $\Phi^3$  (strength requirement of the web):

296 The shear strain at the web of the GFRP deck should be smaller than allowable maximum shear  
297 strain, i.e.,

$$298 \quad \Phi^3 = \left| \bar{\gamma}_w^{xy} \right| \leq {}^s \gamma^d \quad (53)$$

299 (iv) Constraint function  $\Phi^4$  (stiffness requirement):

300 The transverse displacement of the GFRP deck should be smaller than the specified deflection, i.e.:

$$301 \quad \Phi^4 = \delta_f^{z(\max)} \leq \delta^u \quad (54)$$

302 (v) Constraint function  $\Phi^5$  (manufacturing requirement):

303 The fractions of different types of laminates should be within the specified ranges, which are  
304 determined by the pultrusion manufacture, i.e.:

$$305 \quad \Phi^5 = \begin{cases} 0.25 \leq V_f \leq 0.75 \\ \xi_0^l \leq \xi_0 \leq \xi_0^h \\ \xi_{45}^l \leq \xi_{45} \leq \xi_{45}^h \\ \xi_{90}^l \leq \xi_{90} \leq \xi_{90}^h \end{cases} \quad (55)$$

306 (vi) Constraint function  $\Phi^6$  (geometrical requirement):

307 The thickness of the plates should be within the specified ranges to avoid local buckling occurring in  
308 the excessive thin plate, and to meet the manufacturing capabilities since each manufacturer can only  
309 produce the GFRP plate within the specific range of the thickness, i.e.:

$$310 \quad \Phi^6 = \begin{cases} (t_u)^l \leq t_u \leq (t_u)^u \\ (t_l)^l \leq t_l \leq (t_l)^u \\ (t_w)^l \leq t_w \leq (t_w)^u \\ (h_w)^l \leq h_w \leq (h_w)^u \end{cases} \quad (56)$$

## 311 **7. Application to composite bridge girder**

312 A composite bridge girder with a main span of 20.0 meters was selected for a case study. This bridge  
313 girder consists of GFRP bridge decks and I-shaped steel girders with equal center-to-center spacing of 3.0

314 meters. The GFRP composite deck is connected to steel girders using the bolted connector, and the degree  
 315 of composite action between GFRP bridge deck and steel girder,  $\beta$  is specified as 0.72. The total height  
 316 of the I-shaped steel girder is 1000 mm, the thickness of the top flange, bottom flange, and the web, are,  
 317 20.0mm, 25.0mm, and 20.0mm, respectively, and the width of both the top and bottom flanges is 400.0  
 318 mm. According to the Chinese bridge specifications [24], the design load was calculated as:

$$319 \quad S_{ud} = \sum_{i=1}^m \gamma_{Gi} S_{Gik} + \gamma_{Q1} S_{Q1k} + \varphi_c \sum_{j=2}^n \gamma_{Qj} S_{Qjk} \quad (57)$$

320 where:  $\gamma_{Gi}, \gamma_{Q1}, \gamma_{Qj}$  is the partial safety factor to dead load, vehicle load and live load excluding vehicle  
 321 load;  $S_{Gik}, S_{Q1k}, S_{Qjk}$  represent the load effects, resulting from the dead load, vehicle load, and live load  
 322 excluding vehicle load, respectively; and  $\varphi_c$  is the combination reduction parameter for the load effect  
 323 resulting from the live load excluding vehicle load. Note that the design load  $S_{ud}$  can refer to different  
 324 types of load effects, such as bending moment or shear force. In this study, the design loads include  
 325 longitudinal bending moment  $M^L$ , longitudinal shear force,  $Q^L$ , transverse bending moment,  $M^T$ , and  
 326 transverse shear force,  $Q^T$ , and they were computed based below equation:

$$327 \quad M^L = \frac{1.1 \left[ 1.2 (q_{s1} + {}^L q_{f1} + b q_{G2}) + 1.4 (1 + \mu) q_{Q1} + 1.12 b q_{Q2} \right] L^2}{8} + \frac{(1 + \mu) {}^L P_{Q1} L}{4} \quad (58)$$

$$328 \quad Q^L = \frac{1.1 \left[ 1.2 (q_{s1} + {}^L q_{f1} + b q_{G2}) + 1.4 (1 + \mu) q_{Q1} + 1.12 b q_{Q2} \right] L}{2} + \frac{1.2 (1 + \mu) {}^L P_{Q1}}{2} \quad (59)$$

$$329 \quad M^T = \frac{1.1 \left[ 1.2 ({}^T q_{f1} + 1000 q_{G2}) + 1120 q_{Q2} \right] b^2}{8} + \frac{1.4 (1 + \mu) {}^T P_{Q1} b}{4} \quad (60)$$

$$330 \quad Q^T = \frac{1.1 \left[ 1.2 ({}^T q_{f1} + 1000 q_{G2}) + 1120 q_{Q2} \right] b}{2} + \frac{1.4 (1 + \mu) {}^T P_{Q1}}{2} \quad (61)$$

331 Where  ${}^L q_{f1}$  and  ${}^T q_{f1}$  is the self-weight of GFRP deck along the longitudinal and transverse direction  
 332 respectively;  $q_{s1}$  is the self-weight of steel girder;  $q_{G2}$  is the self-weight of paving, defined as 5 kN/m<sup>3</sup>;  
 333  $q_{Q1}$  is the line load of the vehicle, defined as 10.5 kN/m \* b/3000;  $P_{Q1}$  is the concentration load of the  
 334 vehicle, defined as 280kN\*b/3000;  $\mu$  is impact coefficient, defined as 0.3.

335 The objective function of weight,  $\phi_1$ , is specified as:

$$336 \quad \phi_1 = 1000 \left[ 1000(t_u + t_l) + h_w t_w \right] \left[ 1.28 \times 10^{-5} V_f + 3.25 \times 10^{-5} (1 - V_f) \right] \quad (62)$$

337 while the objective function of price,  $\phi_2$ , is specified as:

$$338 \quad \phi_2 = 1000 \left[ 1000(t_u + t_l) + h_w t_w \right] \left[ 2.56 \times 10^{-6} V_f + 1.16 \times 10^{-6} (1 - V_f) \right] \quad (63)$$

339 The reduction (degradation) coefficient is specified as 0.43 based on Eqn. (41) as well as on  
 340 experimental results of several durability tests [28–32]. The constraint functions for strength requirement  
 341  $\Phi^1 \sim \Phi^3$  thus can be presented as Eqs. (65)–(67).

$$342 \quad \Phi^1 = \begin{cases} \left| \bar{\varepsilon}_{fl}^x \right| \leq 0.43^c \varepsilon^u \\ \left| \bar{\varepsilon}_{fl}^y \right| \leq 0.43^c \varepsilon^u \end{cases} \quad (64)$$

$$343 \quad \Phi^2 = \begin{cases} \left| \bar{\varepsilon}_{fl}^x \right| \leq 0.43^t \varepsilon^u \\ \left| \bar{\varepsilon}_{fl}^y \right| \leq 0.43^c \varepsilon^u \end{cases} \quad (65)$$

$$344 \quad \Phi^3 = \left| \bar{\gamma}_w^{xy} \right| \leq 0.43^s \gamma^u \quad (66)$$

345 The Chinese design specifications of highway bridges [24] recommended that the bridge deck  
 346 transverse deflection should be smaller than the girder's span ( $b$ ) divided by 400 (i.e.  $b/400$ ). The  
 347 constraint functions for stiffness requirement thus should be expressed as:

$$348 \quad \Phi^4 = \delta_f^{z(\max)} \leq b / 400 \quad (67)$$

349 The  $0^\circ$ -lamina of pultruded GFRP laminates is in the form of E-glass roving, while both the  $90^\circ$ - and  
 350  $\pm 45^\circ$ -laminates are in the form of stitched E-glass fabrics. Due to the limitation of pultrusion  
 351 manufacturing process, the contents of roving are much larger than fabrics for guaranteeing necessary  
 352 pultrusion traction, making the content of  $0^\circ$  lamina is much larger than the laminas with other angle  
 353 orientations[15]. The minimum ratio of  $0^\circ$  lamina is specified as 50% to guarantee necessary pultrusion  
 354 traction, and the maximum ratio of  $90^\circ$  and  $\pm 45^\circ$  lamina is set as 20% considering manufacture  
 355 difficulties with larger fabric content. Then constrain functions for pultrusion manufacture requirement is

356 specified by Eq. (68).

$$357 \quad \Phi^5 = \begin{cases} 0.25 \leq V_f \leq 0.70 \\ 0.5 \leq \xi_0 \leq 0.95 \\ 0.05 \leq \xi_{45} \leq 0.2 \\ 0.05 \leq \xi_{90} \leq 0.2 \end{cases} \quad (68)$$

358 To avoid local buckling and consider manufacturing capabilities and limitations, a maximum height  
359 of the GFRP bridge deck is set to 300 mm, the maximum flange thickness is set as 50 mm, and the  
360 maximum web thickness per meter is assumed as 250 mm. The constraint functions for geometry  
361 requirements are specified as in Eq. (69) as follow:

$$362 \quad \Phi^6 = \begin{cases} 5 \leq t_u \leq 50 \\ 5 \leq t_l \leq 50 \\ 5 \leq t_w \leq 250 \\ 50 \leq h_w \leq 300 \end{cases} \quad (69)$$

363 The optimization process was achieved by minimizing  $\phi_1$  or  $\phi_2$  under the constraint  $\Phi^1 - \Phi^6$  using  
364 constrained nonlinear minimization (fmincon) function in the MATLAB™ software [33]. The optimized  
365 two-scale parameters of this case study are listed in Table 3. It can be seen that the weight objective  
366 function  $\phi_1$  and the price objective function  $\phi_2$  also calculate the same results. This is mainly because  
367 that the stiffness requirement (constrain function  $\Phi^4$ ) is most strict based on the specification of steel or  
368 concrete deck among all the constrained functions. The optimized top flange thickness  $t_u$ , bottom flange  
369 thickness  $t_l$ , web height  $h_w$ , web thickness per meter  $t_w$  are 46.02 mm, 45.86 mm, 300 mm and 37.42 mm.  
370 Also, the optimized ratio of the  $0^0$ -lamina, the  $45^0$ -lamina, and the  $90^0$ -lamina are 77.9%, 17.1%, 5.0%.  
371 The optimized fiber volume fraction is 65.2%. The optimized parameters are the same in terms of price  
372 and weight optimization because the governing factor is the web height.

373 **Table 3.** Optimized two-scale parameters of case study

Item	Unit	Price Optimization	Weight Optimization
Top flange thickness $t_u$	mm	46.02	46.02

Bottom flange thickness $t_l$	<i>mm</i>	45.86	45.86
Web height, $h_w$	<i>mm</i>	300.0	300.0
Web thickness per meter, $t_w$	<i>mm</i>	37.42	37.42
Ratio of 0 <sup>0</sup> lamina, $\xi_0$	--	0.779	0.779
Ratio of 45 <sup>0</sup> lamina, $\xi_{45}$	--	0.171	0.171
Ratio of 90 <sup>0</sup> lamina $\xi_{90}$	--	0.050	0.050
Fiber volume fraction, $V_f$	--	0.652	0.652
Price per square meter, $\varphi_1$	<i>RMB</i>	2025.9	2025.9
Weight per square meter, $\varphi_2$	<i>kg</i>	213.7	213.7

## 8. Conclusions

The optimization process described in this paper involves identifying the optimal ratio of reinforcements (roving and/or fabric), fiber volume fractions, in conjunction with geometrical variables in order to achieve the optimum design in both material and structure scales. In this paper, the macro behaviors of pultruded FRP bridge deck are analyzed based on the design specification of the highway bridge. The equivalent properties of the pultruded GFRP lamination are calculated by combining both micromechanics and classical lamination theory. The micro fiber/resin level is bridged to macro pultruded GFRP bridge level by assuming the effective strain homogenized from micro component equals to macro strain. The multiscale lamination optimization is achieved by finding optimized two-scale design parameters for minimizing bridge weight and/or materials and construction cost while satisfying all design parameters for the pultruded composite deck. The optimized two-scale parameters were obtained by solving the proposed multiscale optimization model, for a bridge with a main span of 20.0 meter and steel girders equal spacing of 3.0 meters. The optimized values of the top flange thickness,  $t_u$ , the bottom flange thickness,  $t_l$ , the web height,  $h_w$ , and the web thickness per meter,  $t_w$ , are 46.02 mm, 45.86 mm,

388 300.0 mm and 37.42 mm, respectively. Results also showed that the optimized ratio of the 0°-lamina,  
389 45°-lamina, and the 90°-lamina are 77.9%, 17.1%, 5.0%. The optimized fiber volume fraction is 65.2%.

390

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