

# Building with Martian Regolith

Experimental Research on Efficient and Sustainable Production Process  
and Construction Method

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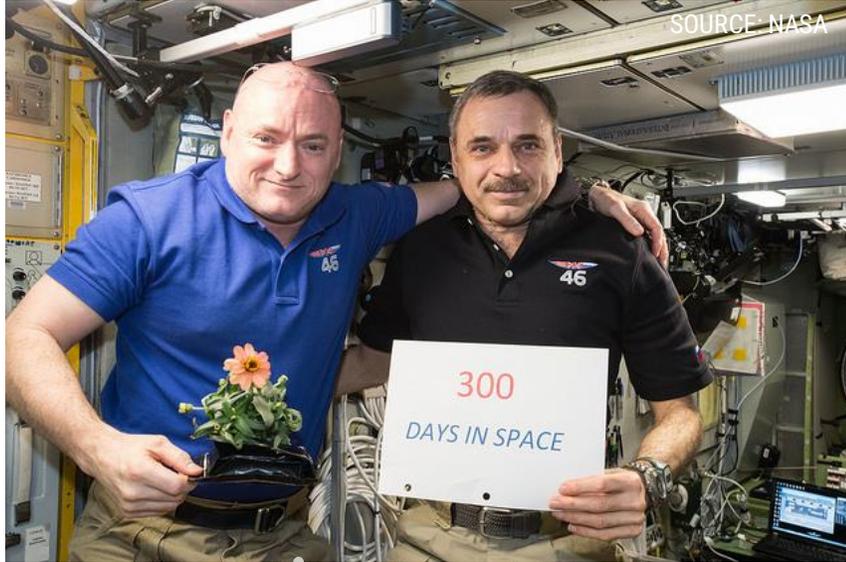
**Graduation Studio:** Sustainable Design Graduation Studio

**Mentors:** Fred Veer (1<sup>st</sup> mentor), Oguzhan Copuroglu (2<sup>nd</sup> mentor), David Peck (3<sup>rd</sup> mentor)

**Supervisors:** Layla van Ellen, Fernando Franca de Mendonca Filho



1961



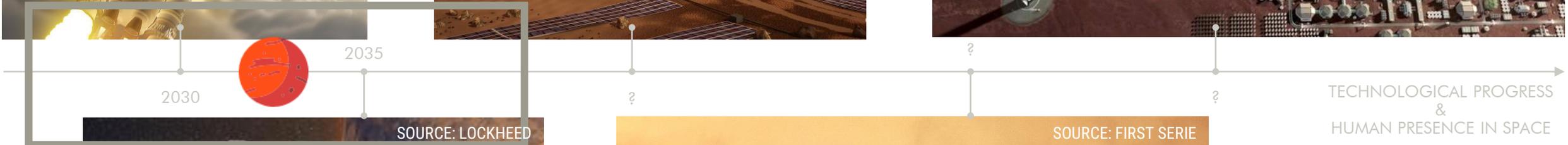
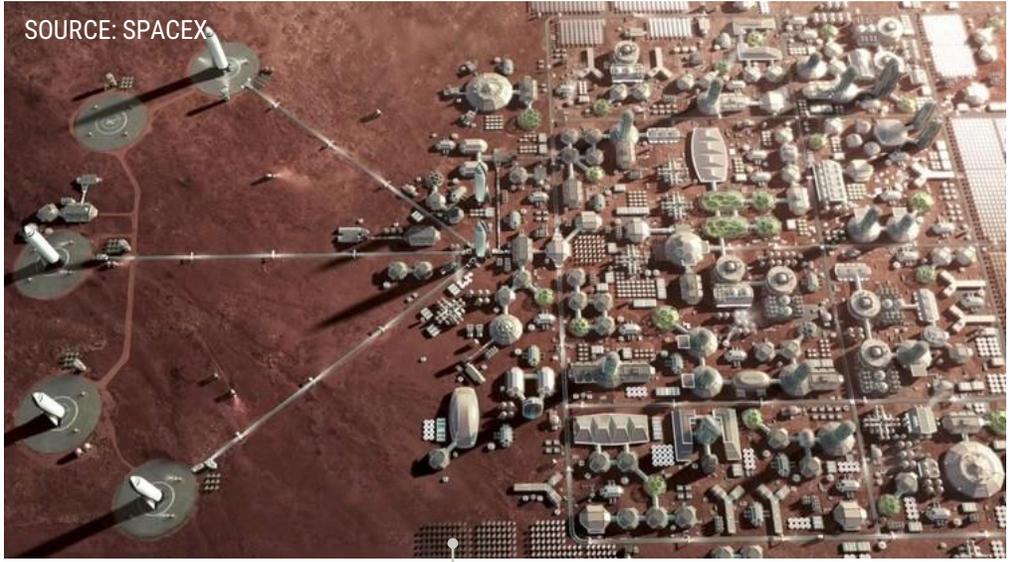
2016



2025-30

# Humans presence in space

# CONTEXT



## Context

1<sup>ST</sup> MISSIONS – 2030-35

TECHNOLOGY  
AVAILABILITY

LONG-TERM  
SETTLEMENT

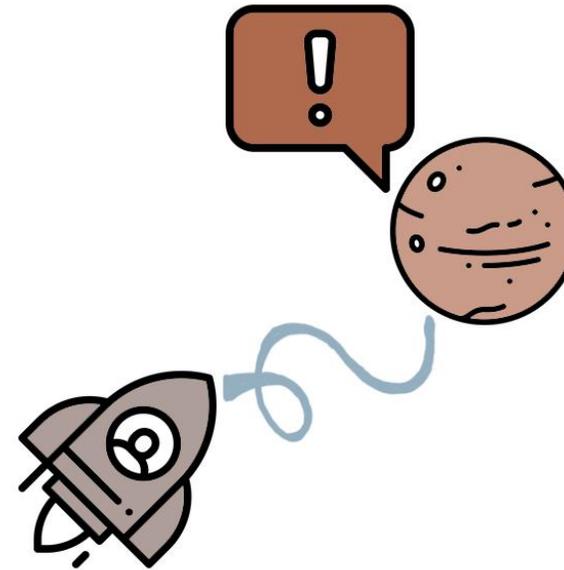


## Issues related to the topic

MISSION CONDITIONS

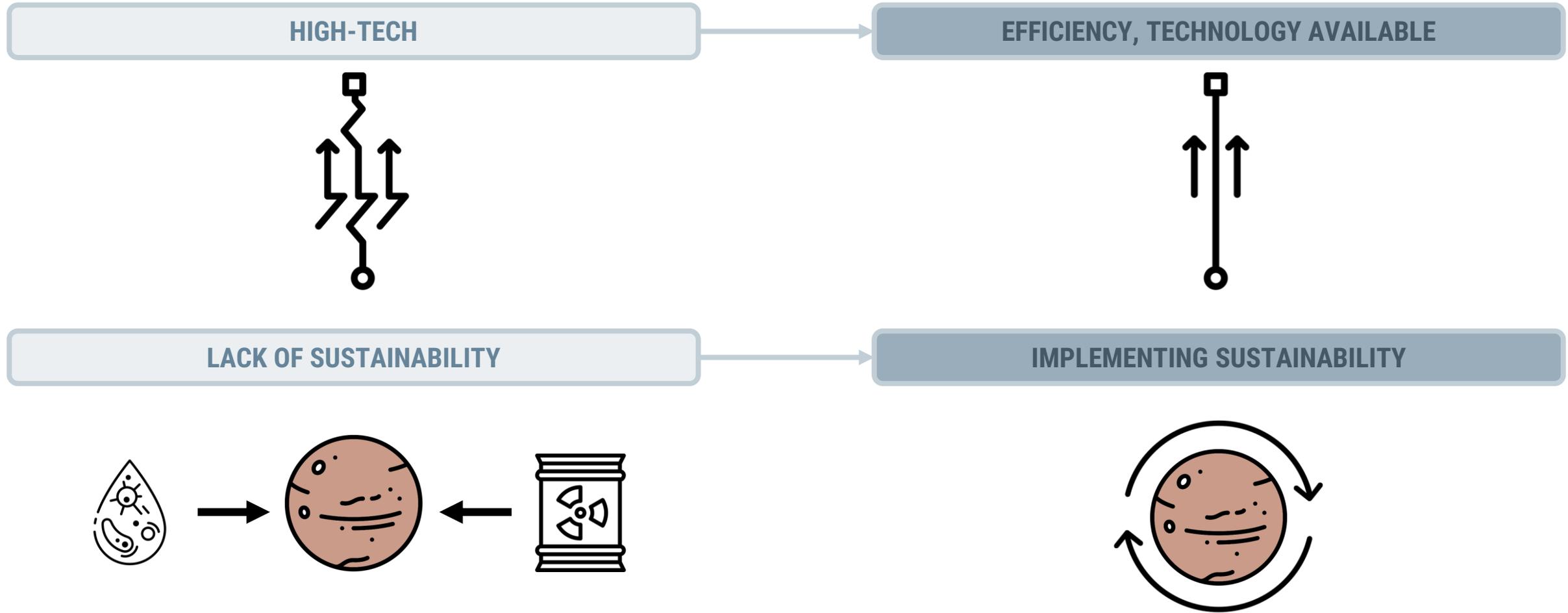
MARTIAN CONDITIONS

COMMON APPROACH

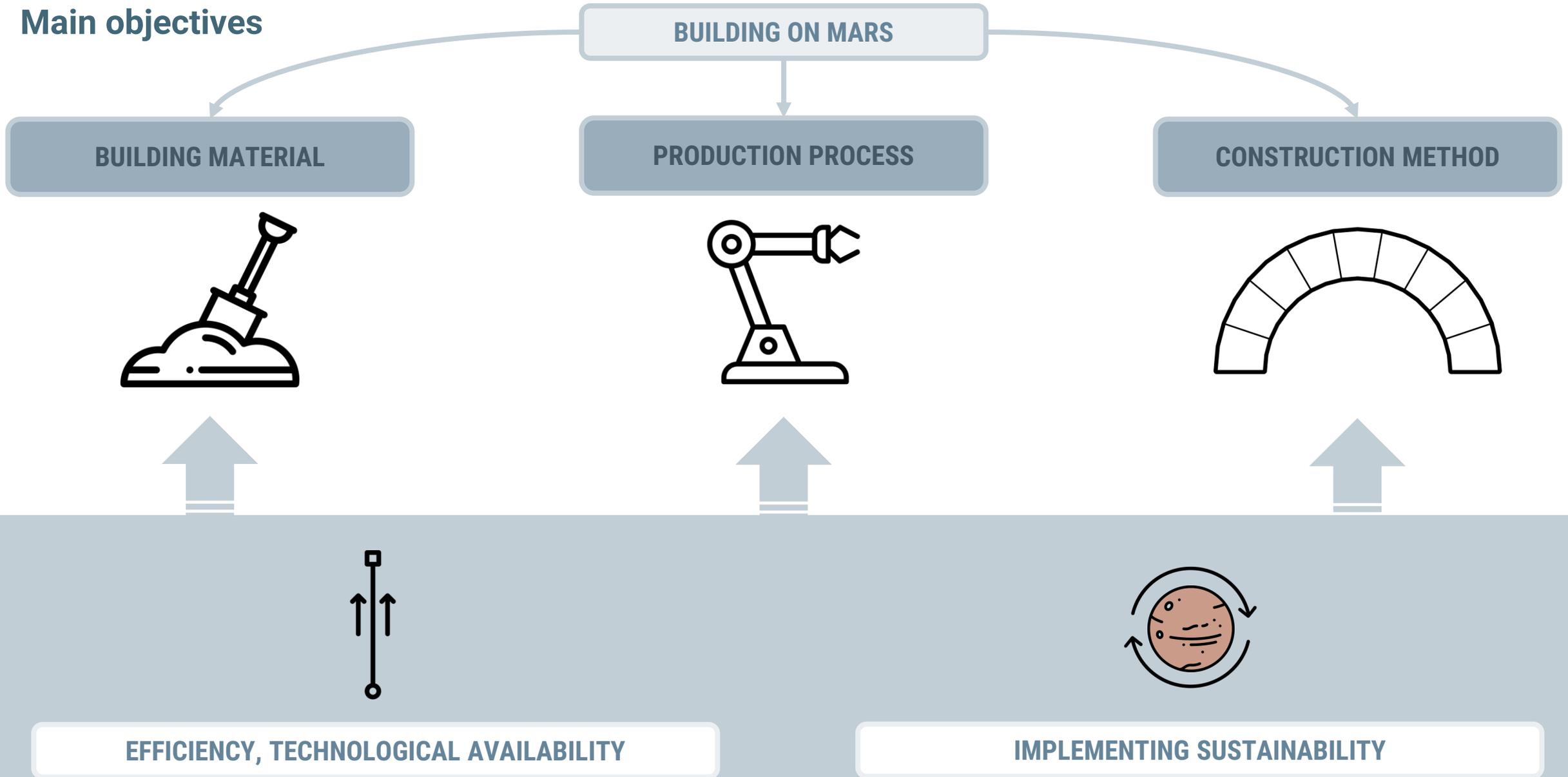


## Common approach

## Research approach

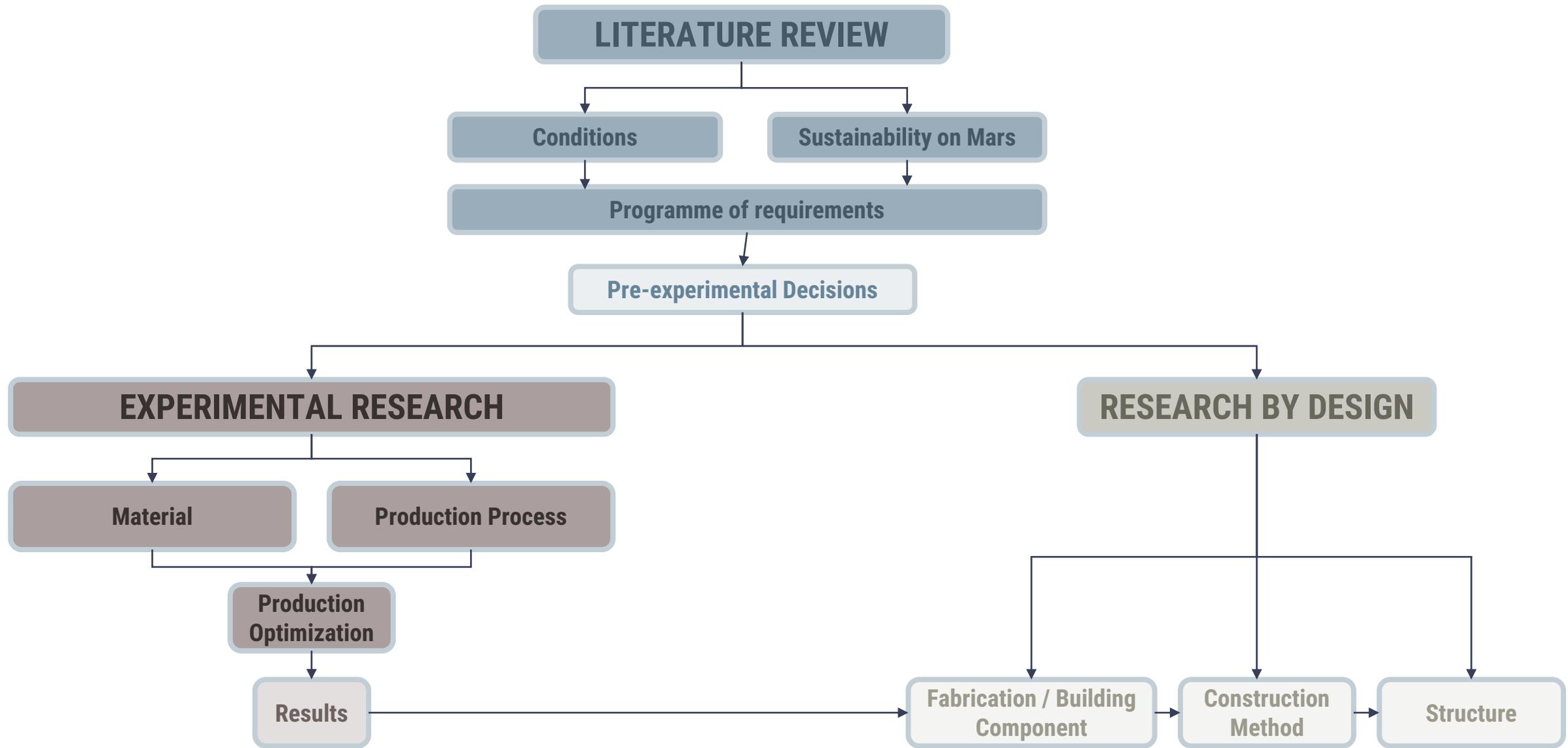


## Main objectives



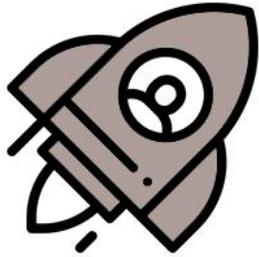
## Research question

**HOW TO SUSTAINABLY BUILD WITH MARTIAN REGOLITH USING ENERGY EFFICIENT IN-SITU PRODUCTION PROCESS AND CONSTRUCTION METHOD ?**



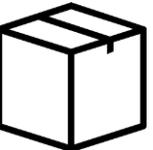


## TRAVEL/TRANSPORT

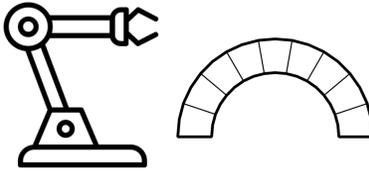


### ISSUE

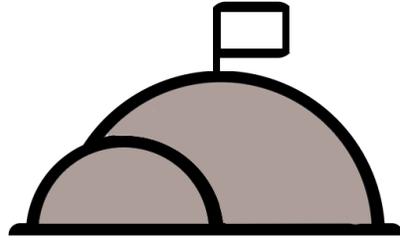
### EFFECT

- launch windows - 26 months 
- payload - 16 800 kg 
- multiple missions 



-  autonomous / independent
-  ISRU resources
-  adaptable building

## ARCHITECTURE

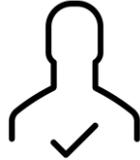


### ISSUE

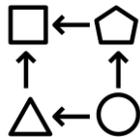
crew size - 6x



crew safety



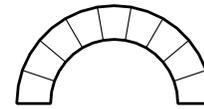
adaptability



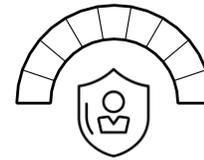
mission-functional



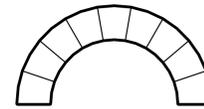
### EFFECT



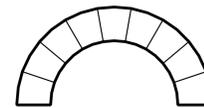
150m<sup>3</sup> – habitat volume



protection against hazardous conditions



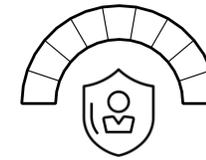
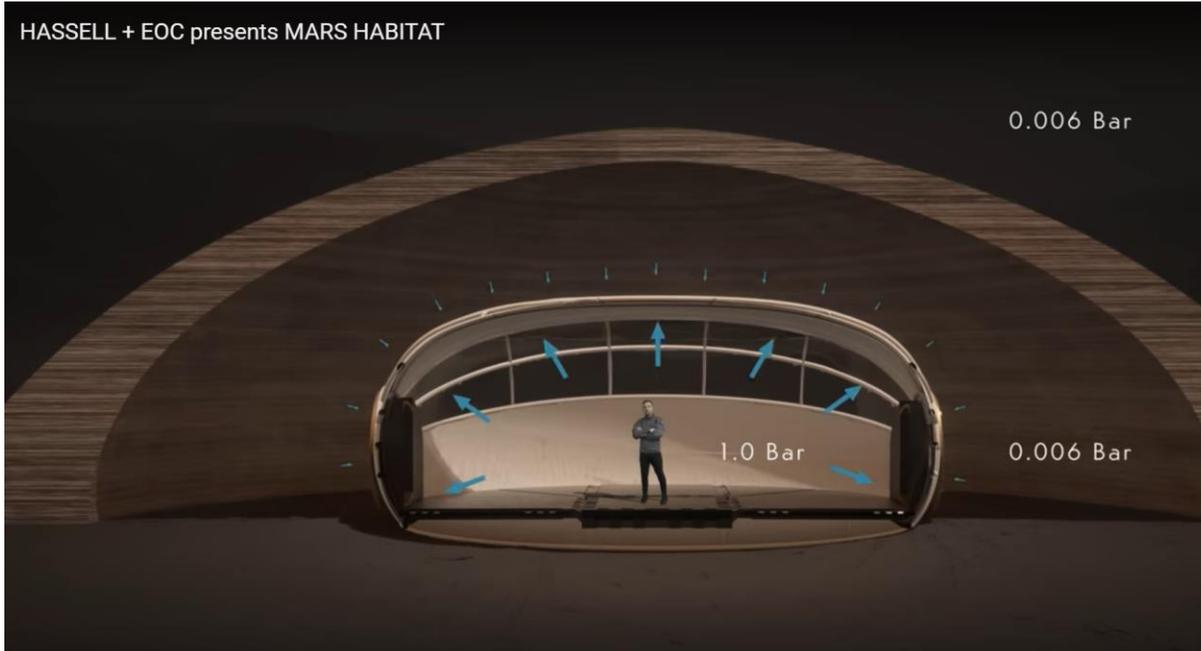
modular structure



modular configuration - zoning



## ARCHITECTURE



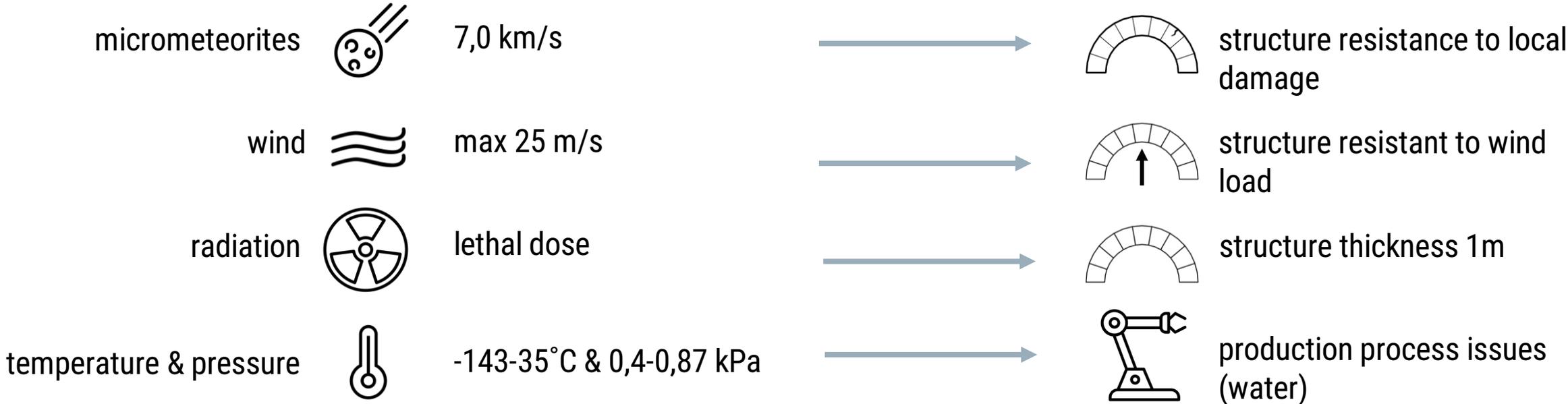
protection against  
hazardous conditions

## HAZARDS



### ISSUE

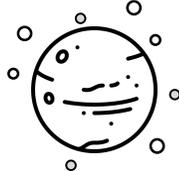
### EFFECT



## RESOURCES

### ISSUE

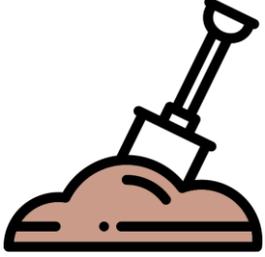
dust form homogenous



fine grains



### RESOURCES



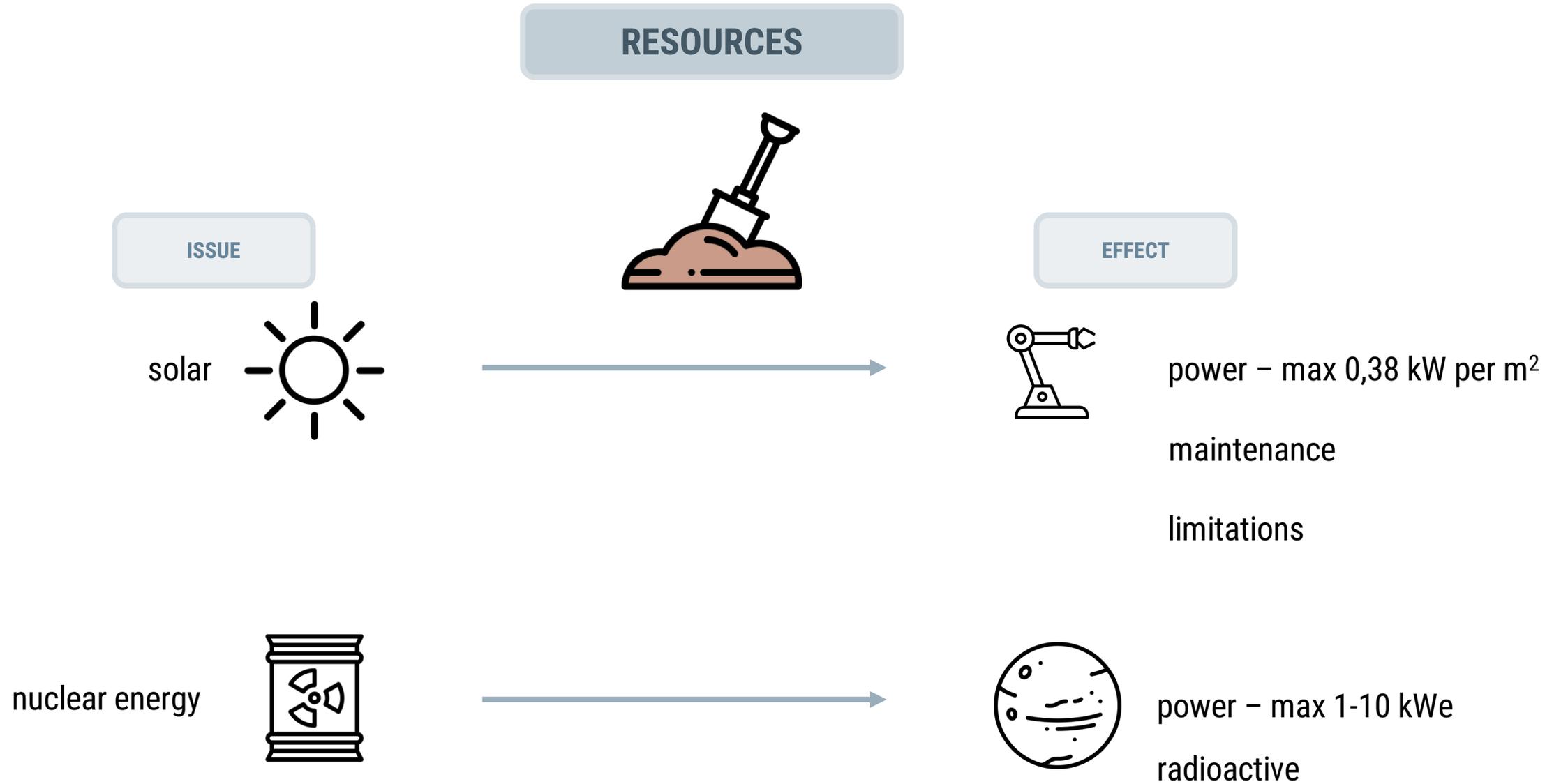
### EFFECT



adaptable to different regions



direct usage

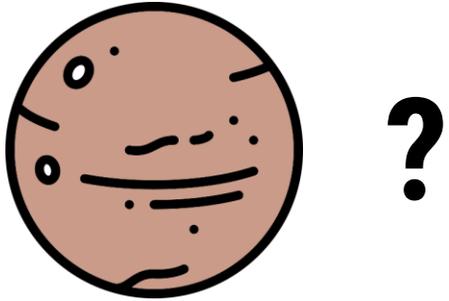




WHAT IS SUSTAINABILITY

HOW TO IMPLEMENT

CONCLUSIONS

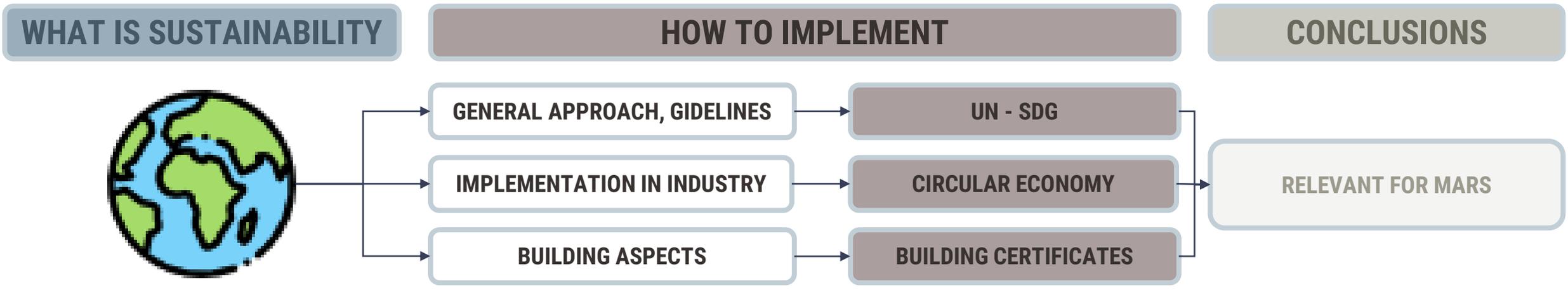


## WHAT IS SUSTAINABILITY

1



**Brundtland Commission Report** – „sustainable development is a development which meets the needs of the present without compromising the ability of future generations to meet their own needs



## CONCLUSIONS

RELEVANT FOR MARS

environment protection

renewable and clean energy

sustainable management and efficient use  
of resources

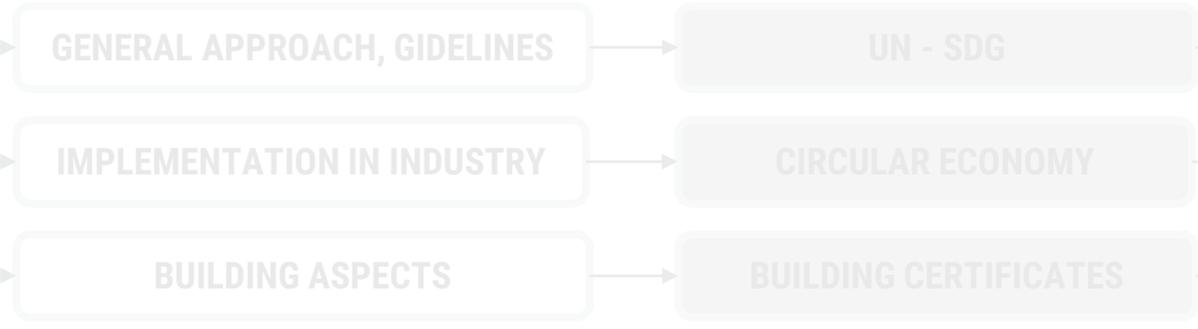
minimize waste

circularity

## WHAT IS SUSTAINABILITY

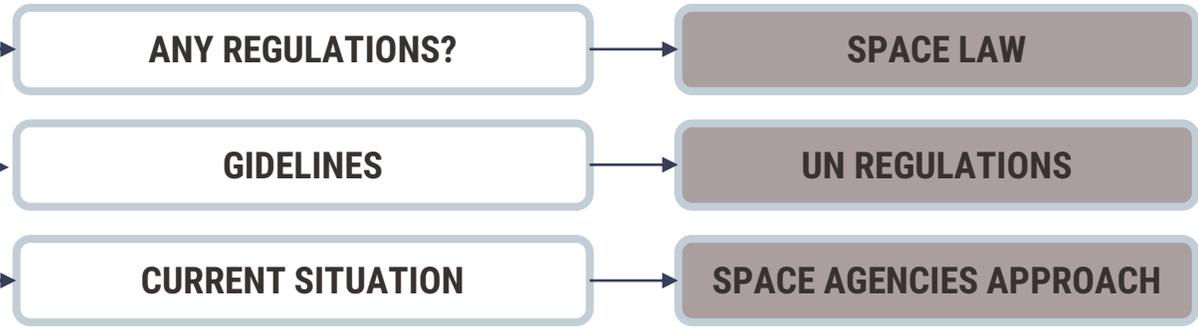
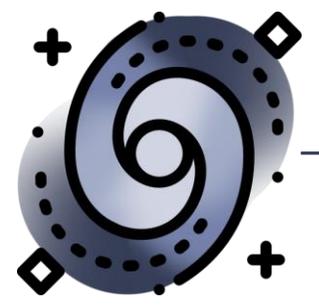


## HOW TO IMPLEMENT



## CONCLUSIONS

RELEVANT FOR MARS



WHAT IS MISSING?

## CONCLUSIONS

WHAT IS RELEVANT?

environment protection

sustainable management and efficient use  
of resources

circularity

renewable and clean energy

minimize waste

WHAT IS MISSING

environmental impacts /  
sustainable materials

long-term issue

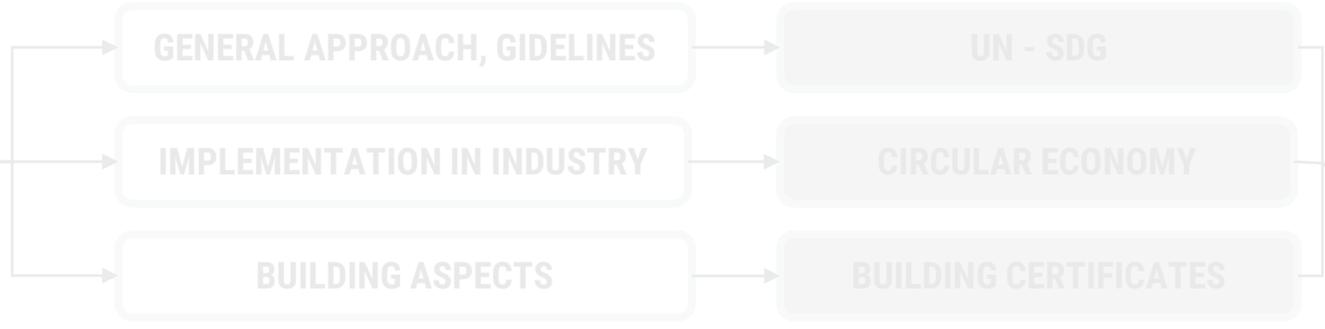
big scale

commercial exploration

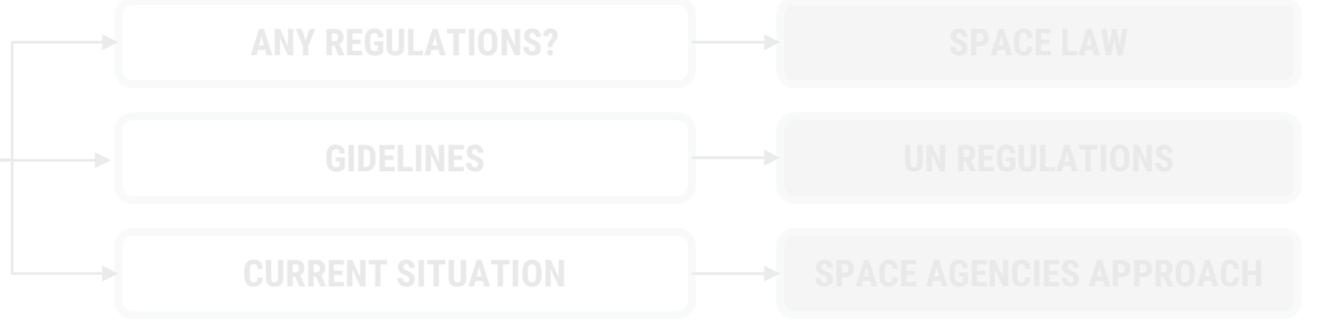
## WHAT IS SUSTAINABILITY

## HOW TO IMPLEMENT

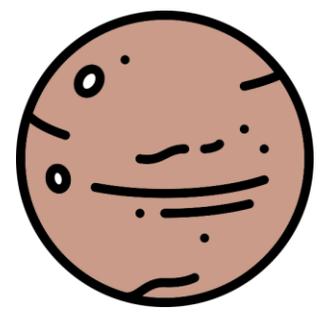
## CONCLUSIONS



RELEVANCE FOR MARS



WHAT IS MISSING?



IMPLEMENTED IN FIRST MISSIONS?

IMPLEMENTED IN RESEARCH

## CONCLUSIONS

WHAT IS RELEVANT?

environment protection

sustainable management and efficient use of resources

circularity

renewable and clean energy

minimize waste

**HOW TO SUSTAINABLY BUILD WITH MARTIAN REGOLITH USING ENERGY EFFICIENT IN-SITU PRODUCTION PROCESS AND CONSTRUCTION METHOD ?**

IMPLEMENTED IN THE FIRST MISSIONS?

test approach

analyse impacts

test technology

establish collaboration and regulations

IMPLEMENTED IN RESEARCH

material and energy efficient strategy

avoid waste production

solar + limited nuclear energy

material 100% in-situ

recyclable structure / material

weathers when abandoned

# LITERATURE REVIEW

Conditions

Sustainability on Mars

Programme of requirements

## Pre-experimental Decisions

### EXPERIMENTAL RESEARCH

### RESEARCH BY DESIGN

Material

Production Process

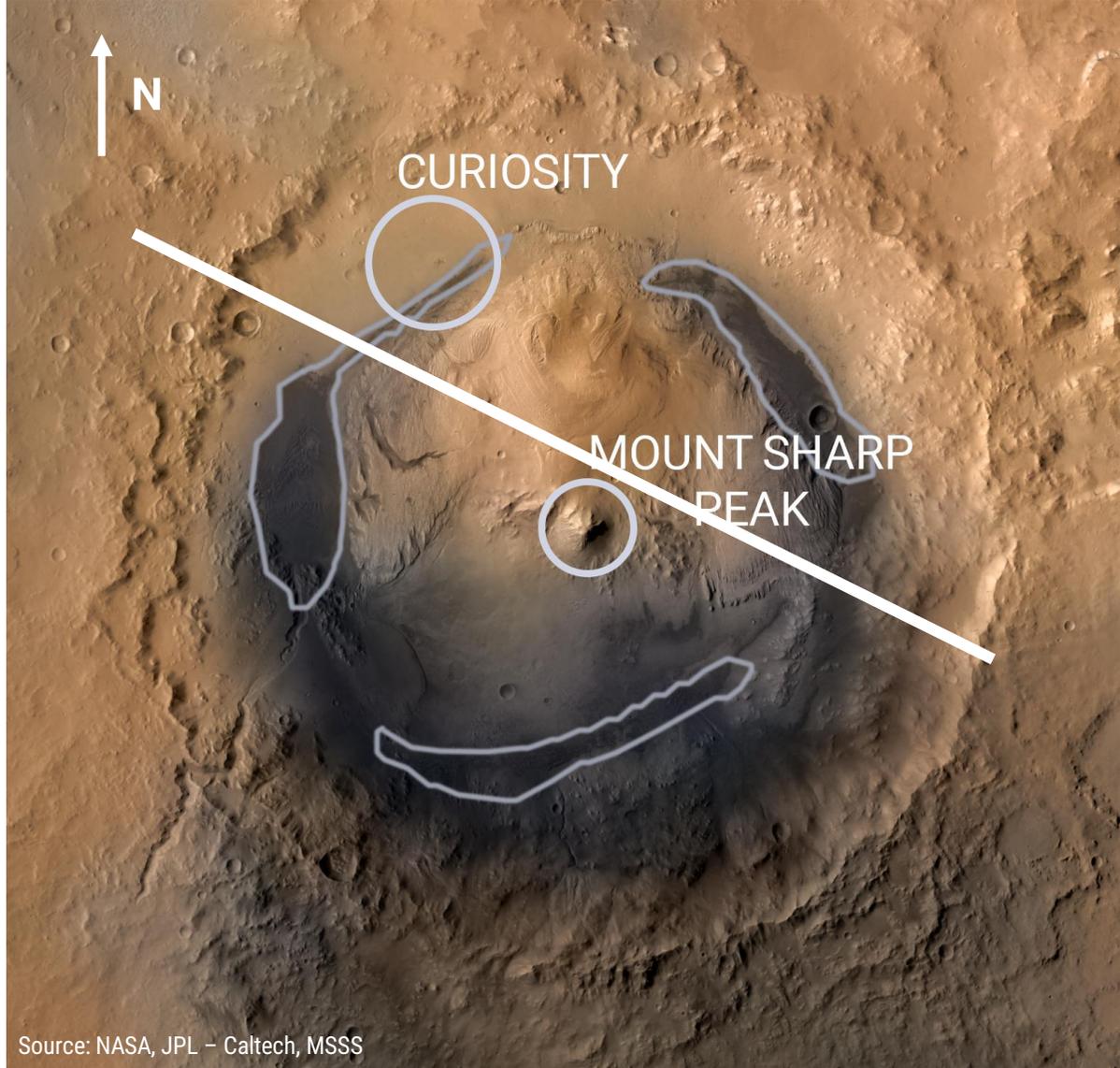
Production Optimization

Results

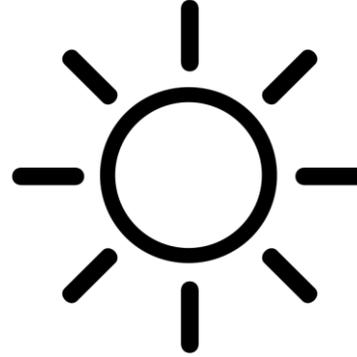
Fabrication / Building Component

Construction Method

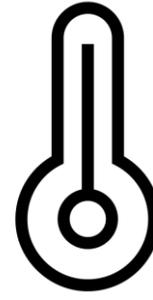
Structure



Source: NASA, JPL - Caltech, MSSS



- daily solar power output:  
1034 W/m<sup>2</sup> - 1470 W/m<sup>2</sup>



- average temperature:  
-70°C - 0°C



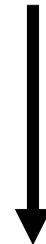
- wind speed:  
4m/s - 25m/s

Simulant preparation



ROCKNEST

NAMIB



available on Earth



SIMULANT, MGS-1	
Mineral	Weight %
70%	
Sodic Plagioclase	41,6
Highly Forsteritic Olivine	21
single bronzite-variety pyroxene	31

Magnetite (as black iron oxide)	2,9
Anhydrite	1,3

Hematite (as red iron oxide)	1,7
------------------------------	-----

30%	
Ferrihydrite	4
Basaltic Glass	65
Hydrated Silica (as diatomaceous earth)	14
ferric-sufate (iron (III) sulfate pentahydrate)	20

Fe-carbonate (siderite)	4
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Source: Cannon K. et.al., 2019

Cannon, et.al. (2019). Mars global simulant MGS-1: A Rocknest based open standard for basaltic martian regolith simulants.

## Addition of water on Earth

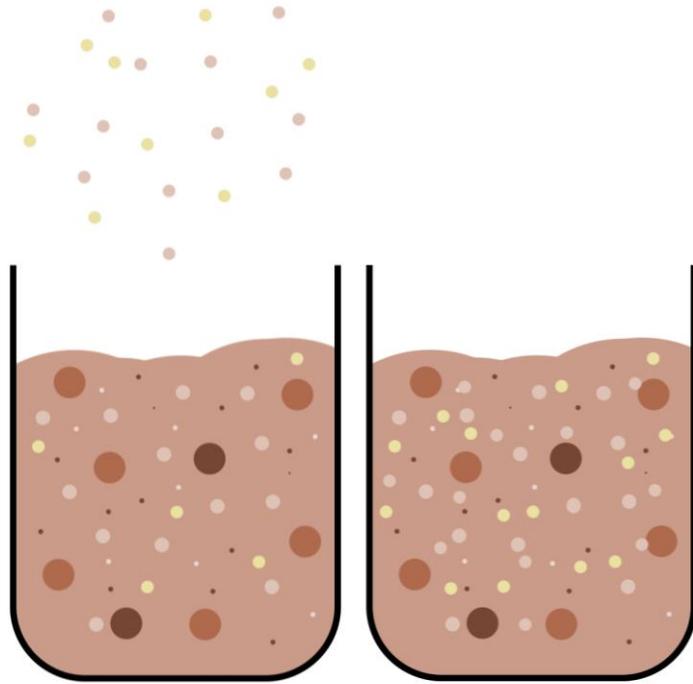


## What could be the binder on Mars ?



## Potential binders

1

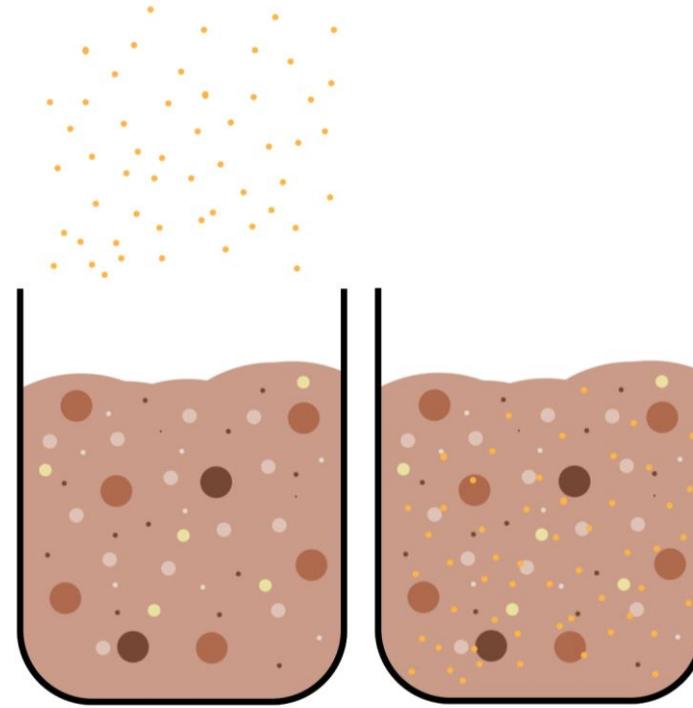


**lowest melting point**

Plagioclase – 600-800°C

Ferric-sulfate – 480°C

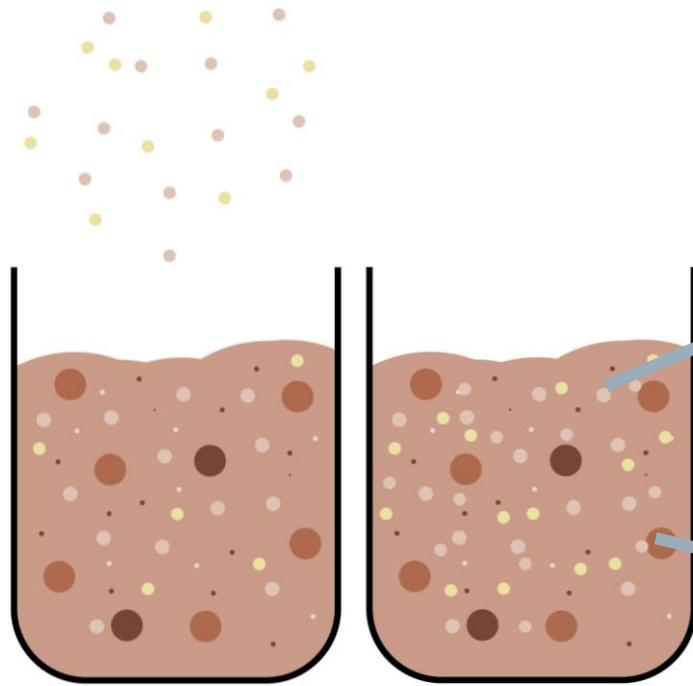
2



**molten sulfur**

Sulfur – 120°C

## Potential binders

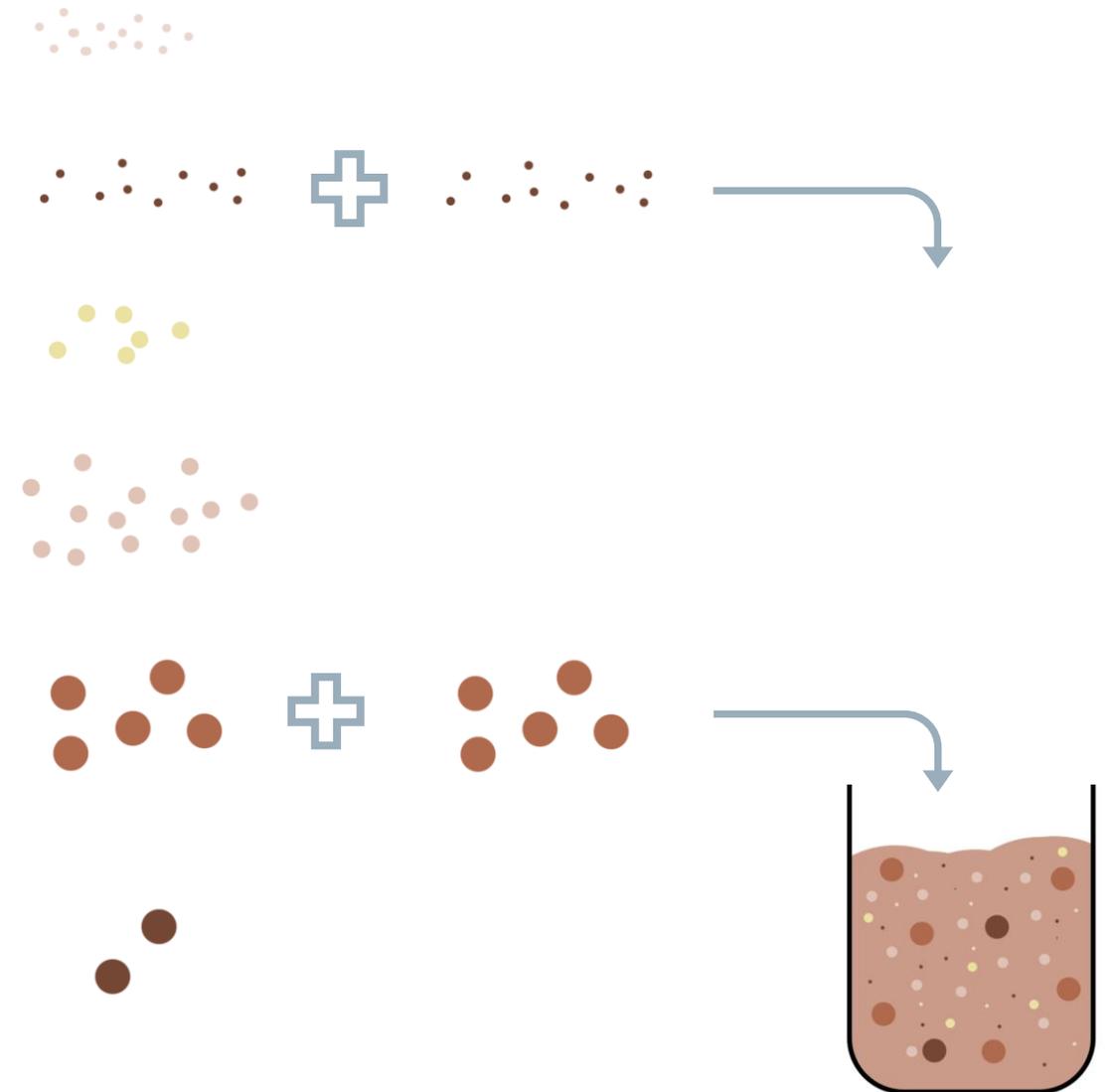


### lowest melting point

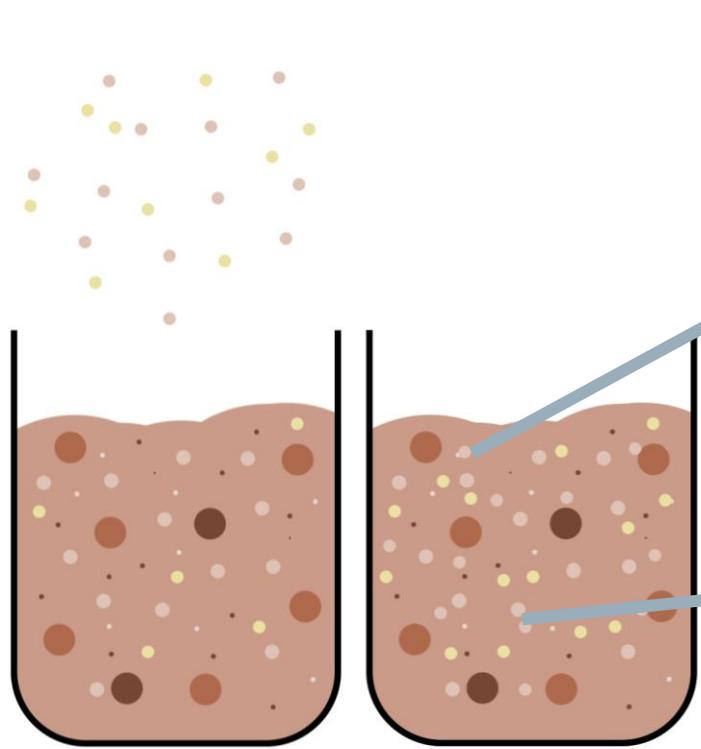
Plagioclase – 600-800°C

Ferric-sulfate – 480°C

## Grain size - sieving

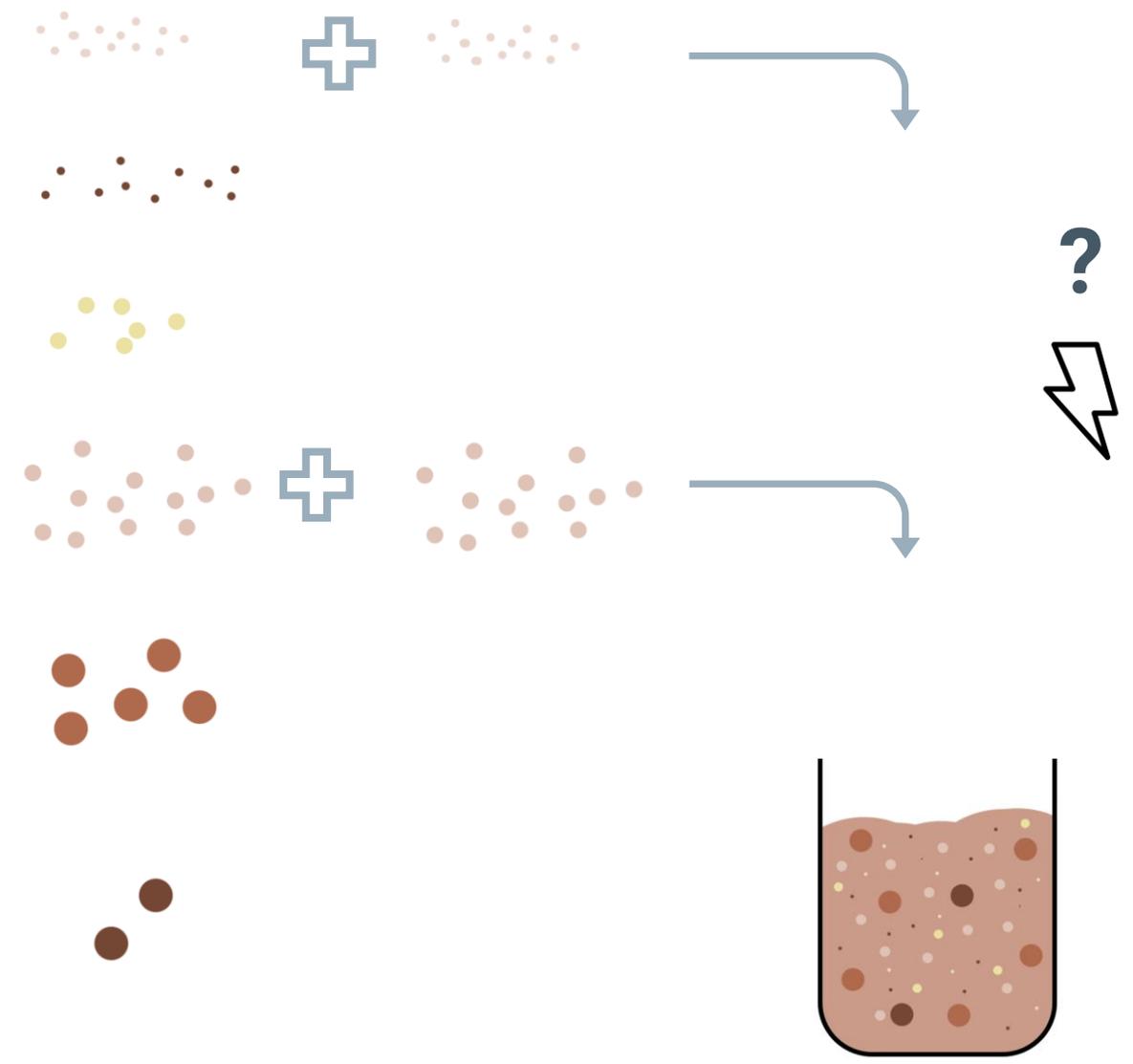


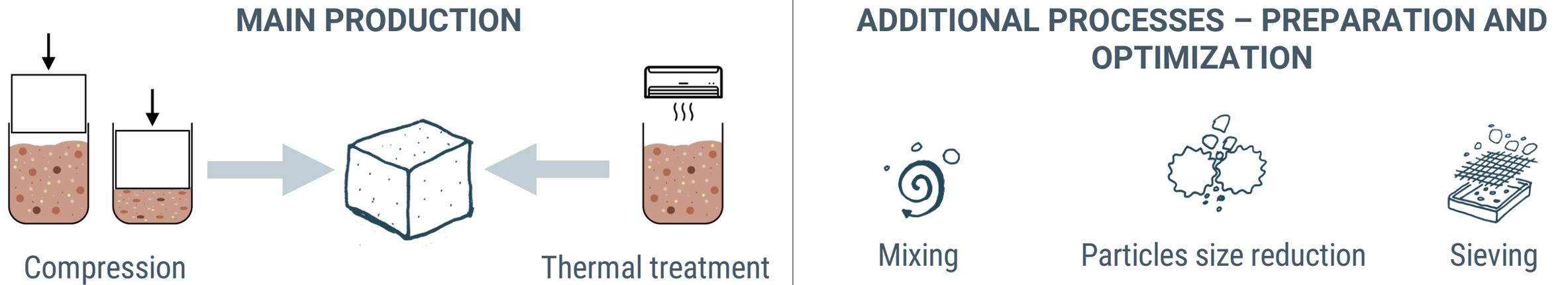
### Potential optimization



Other minerals

### Grain size sieving

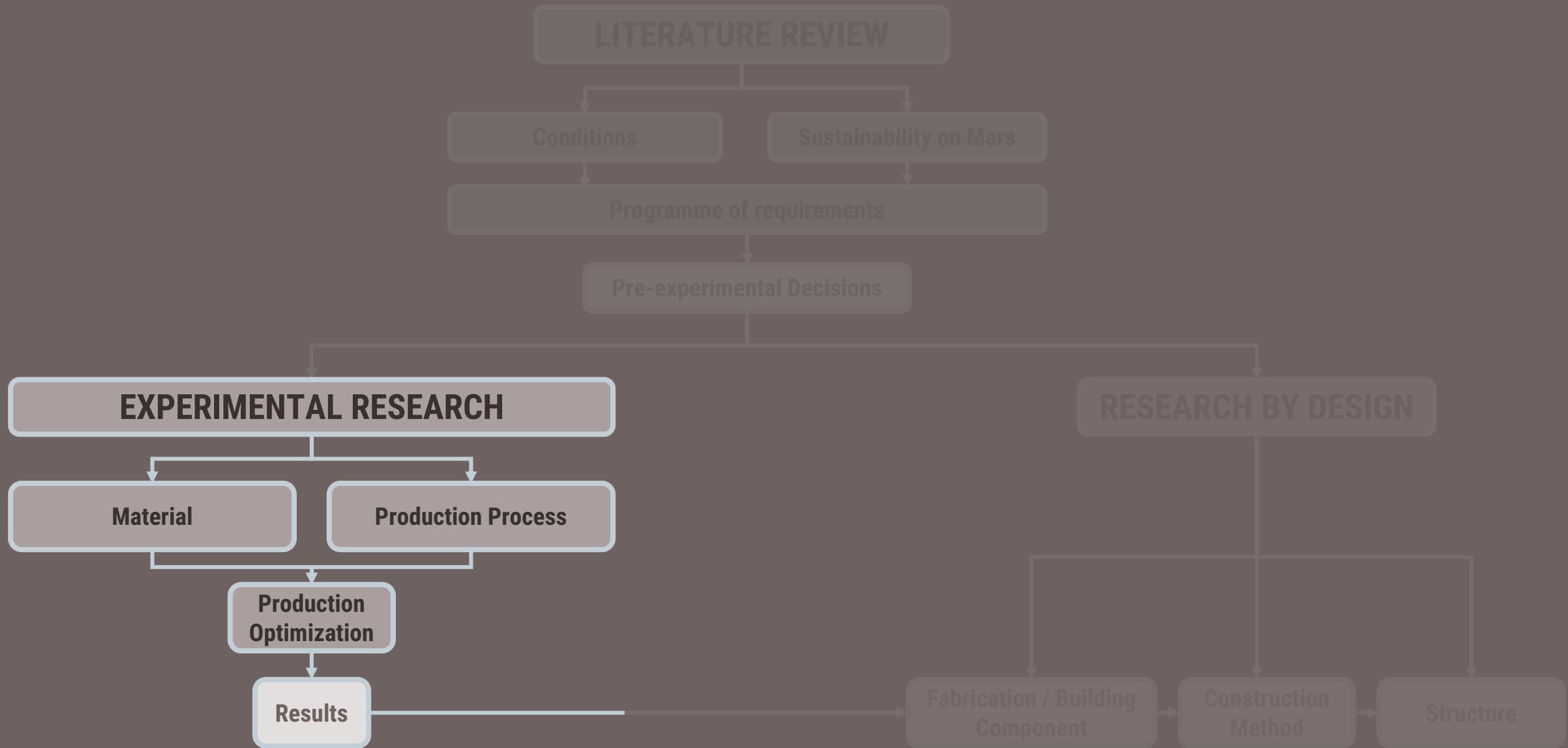




## BUILDING COMPONENT



- Compressive strength : 1,5 – 2 MPa



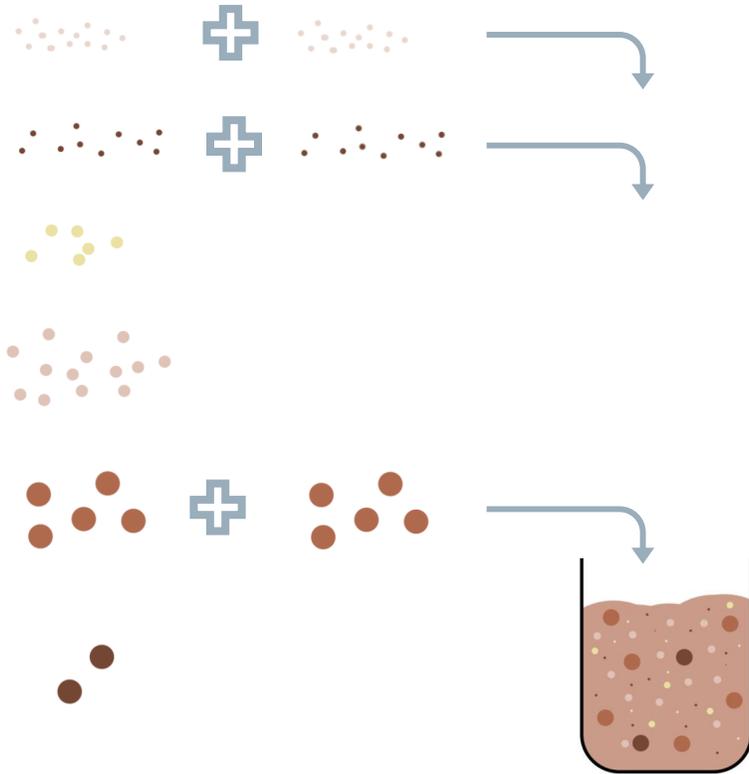
STEP

## PREPARATION

GRAIN SIZES

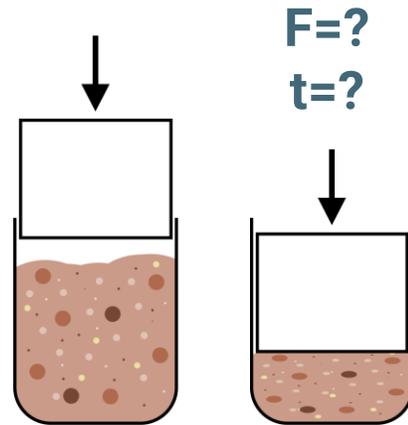
MIXING

BATCHES



## PILOT TEST

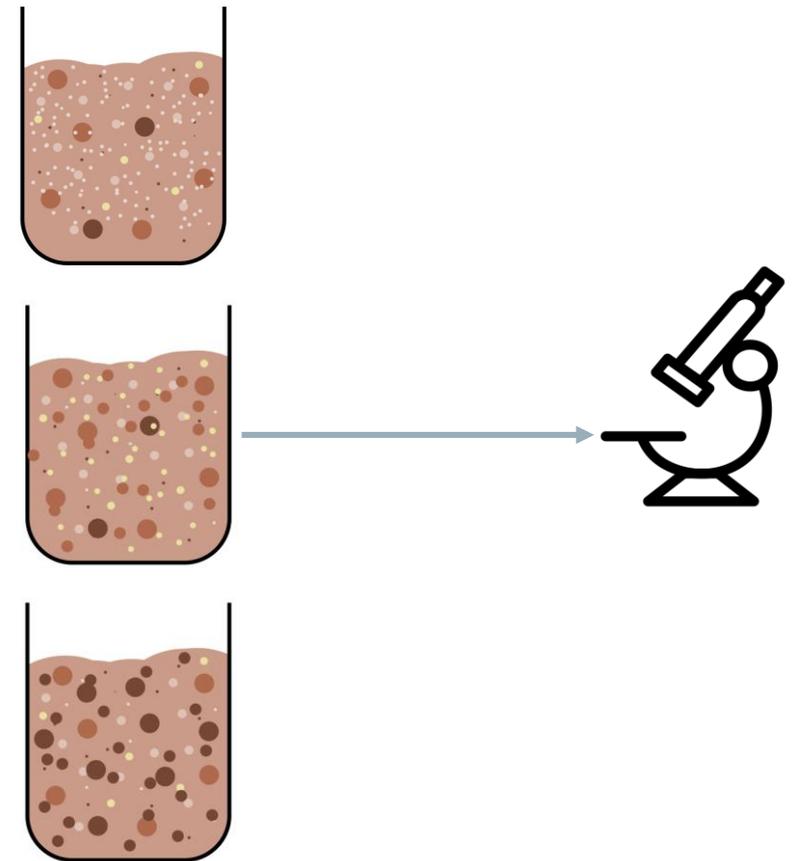
COMPRESSION



## OPTIMISATION

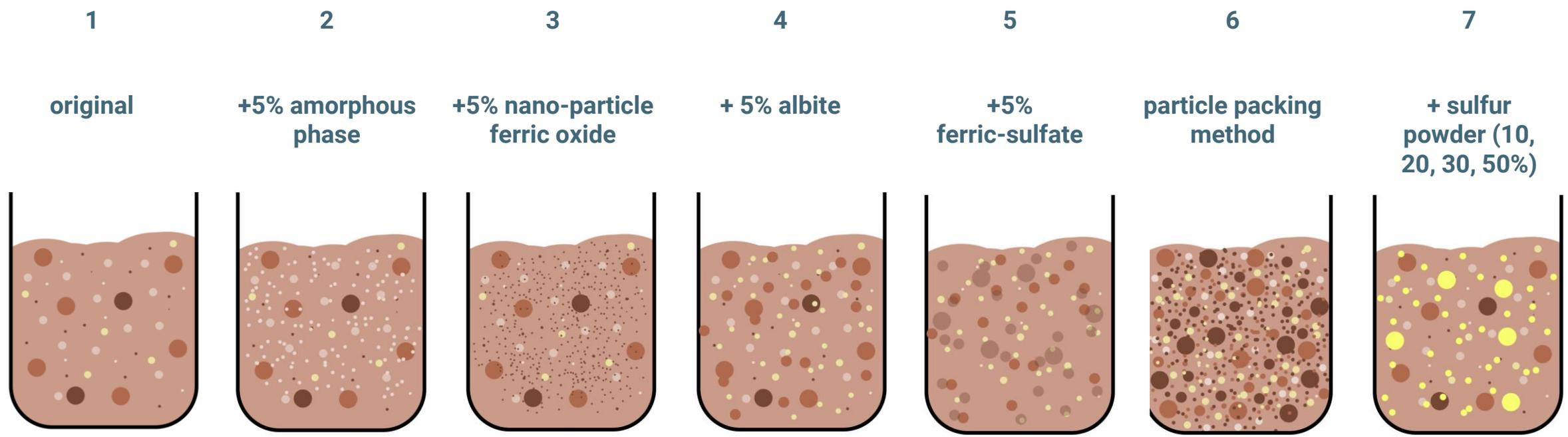
DIFFERENT BATCHES

MICROSCOPIC ANALYSIS





## Compositions



comparison

improved results of the compression process

minerals with lowest melting point

more compact material

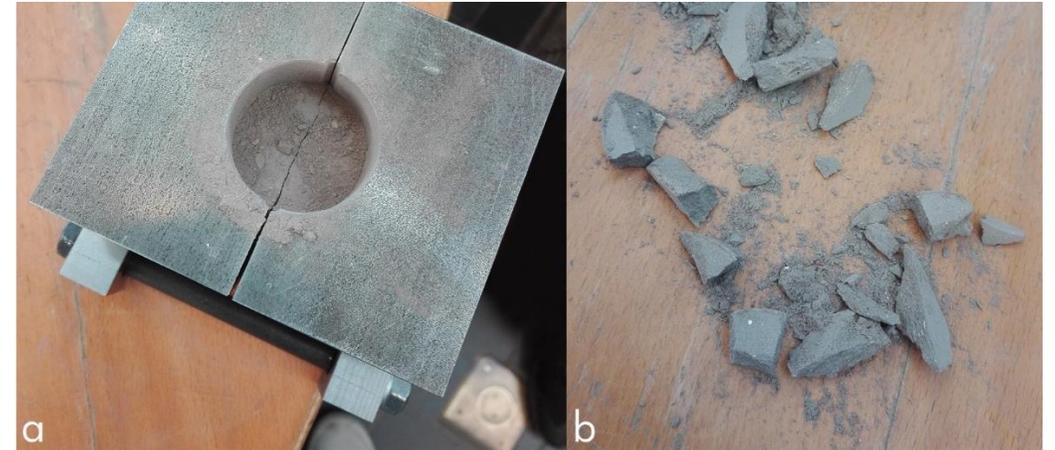
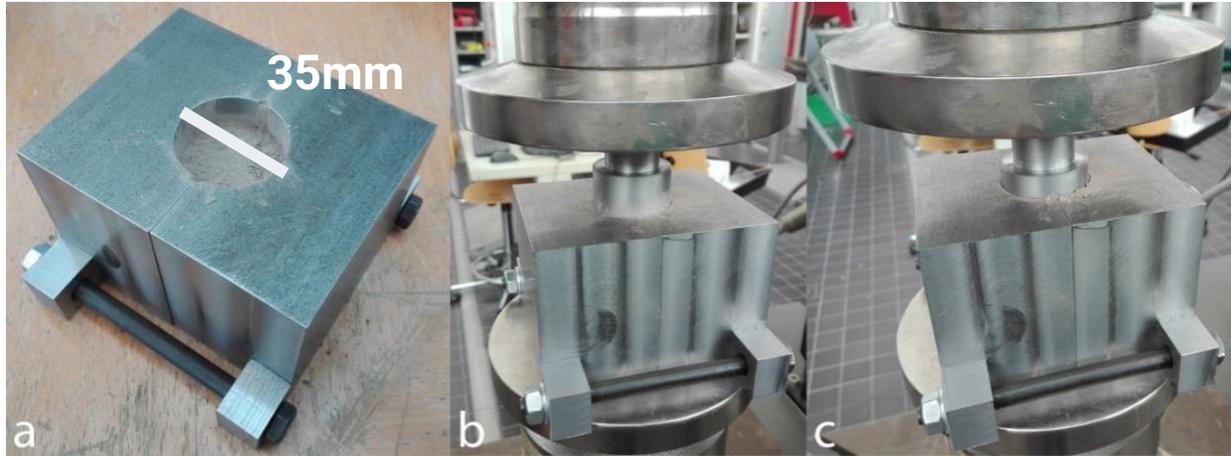
molten sulfur as a binder

Chow, et al., 2017, "Direct formation of structural components using a martian soil simulat"

Experiments



# 1. PILOT TEST

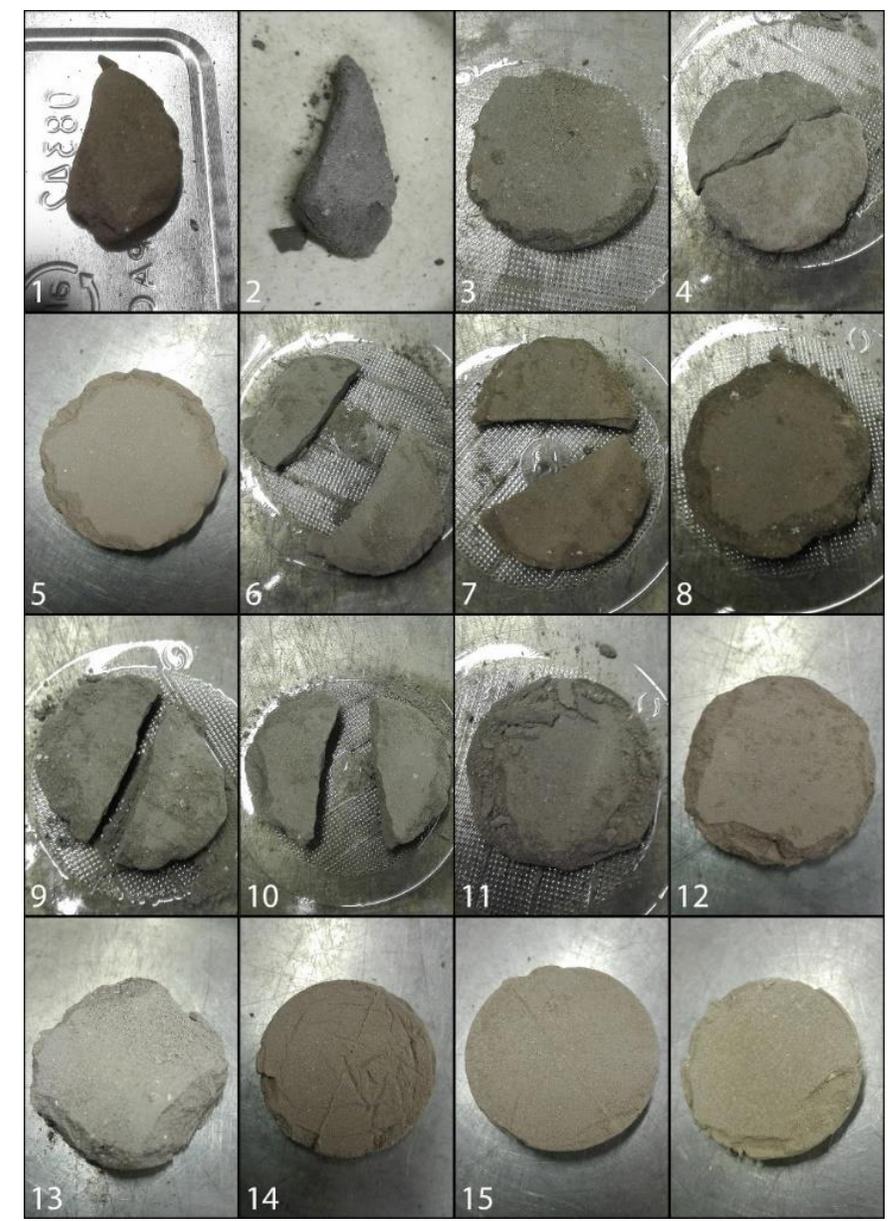
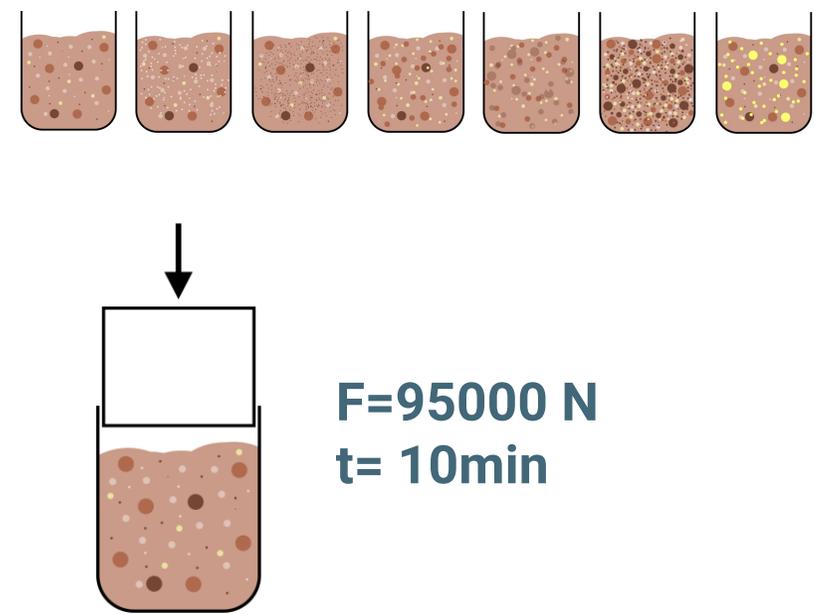


**$F = 9,5 \text{ kN}$**   
 **$t = 10 \text{ min (later 5)}$**



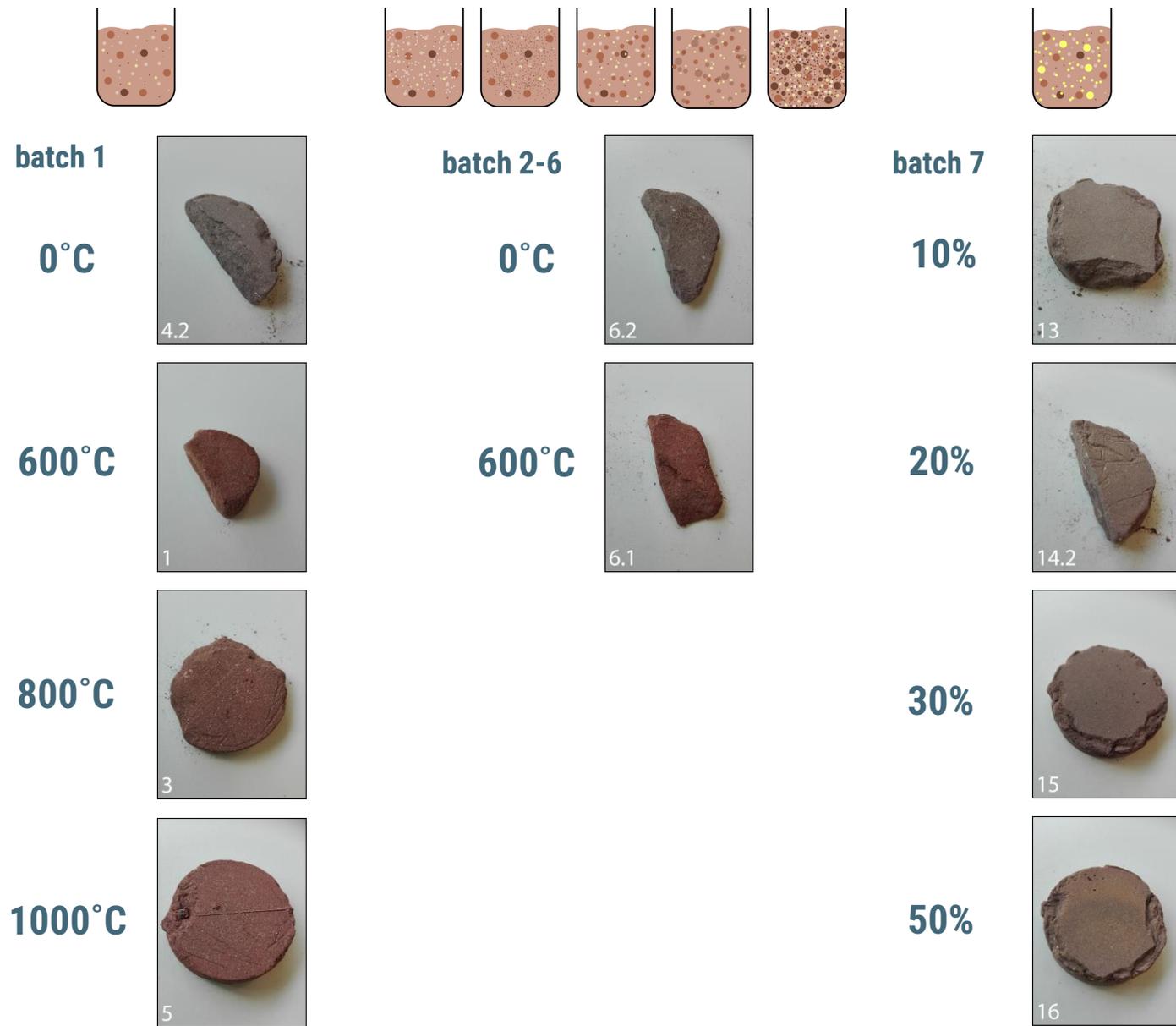
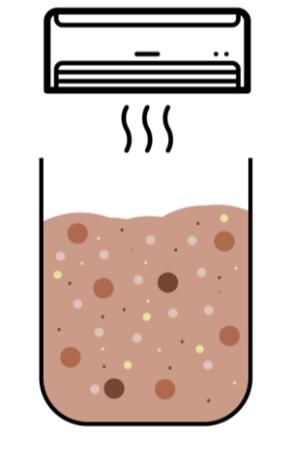
# EXPERIMENTS - DIFFERENT BATCHES

## COMPRESSION PROCESS



# EXPERIMENTS - DIFFERENT BATCHES

## THERMAL TREATMENT

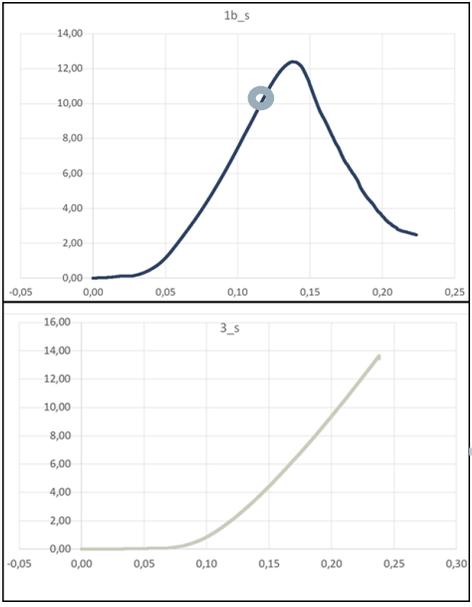


## COMPRESSION TEST

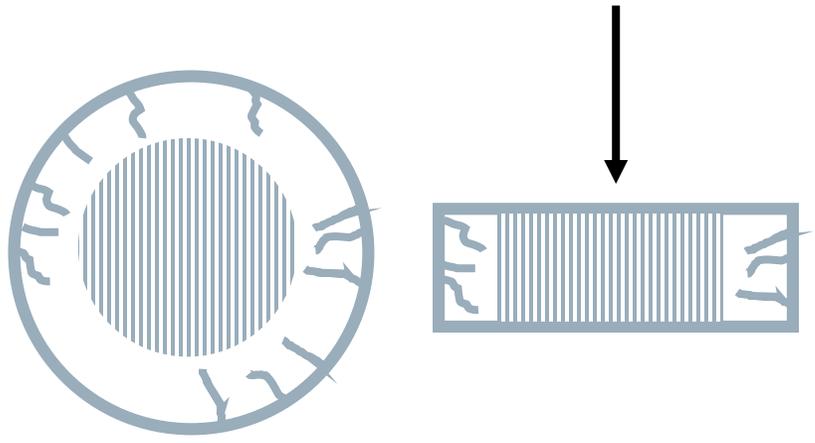


x 2 load settings  
impact

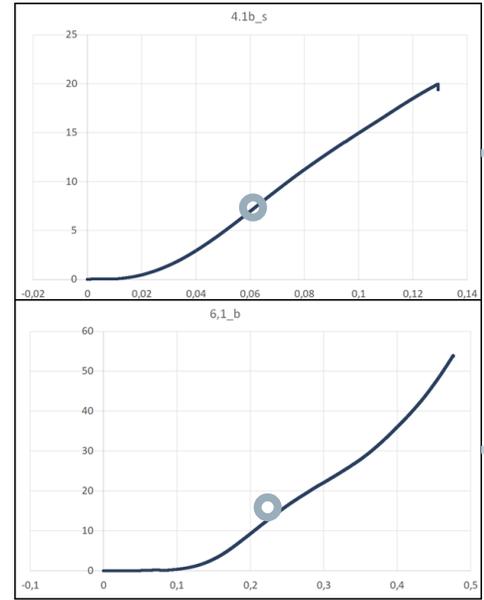
## STRESS/STRAIN GRAPH



no break in second loading



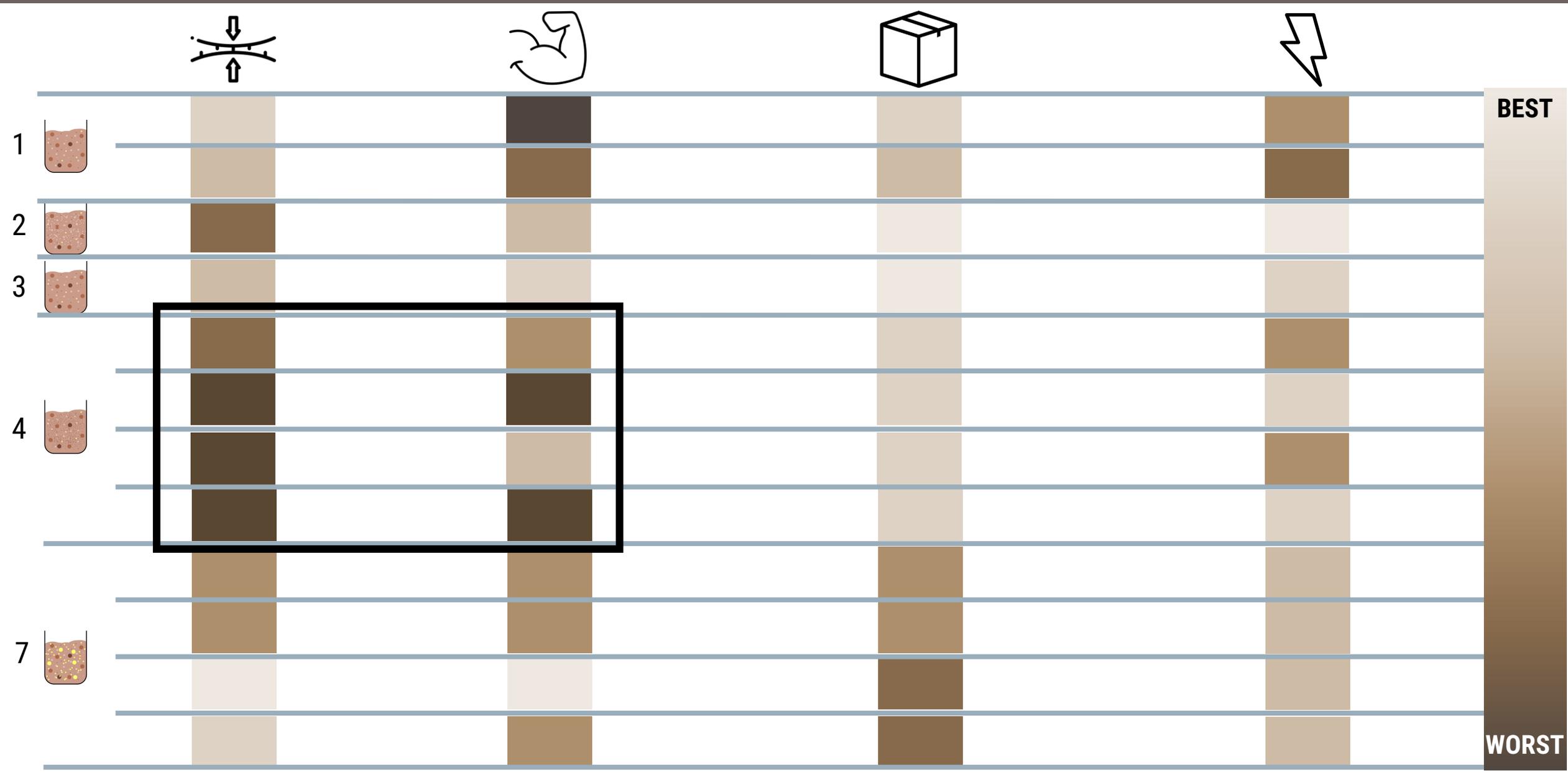
shape impact



first changes in graph – edges cracked

edges cracked  
central hardened

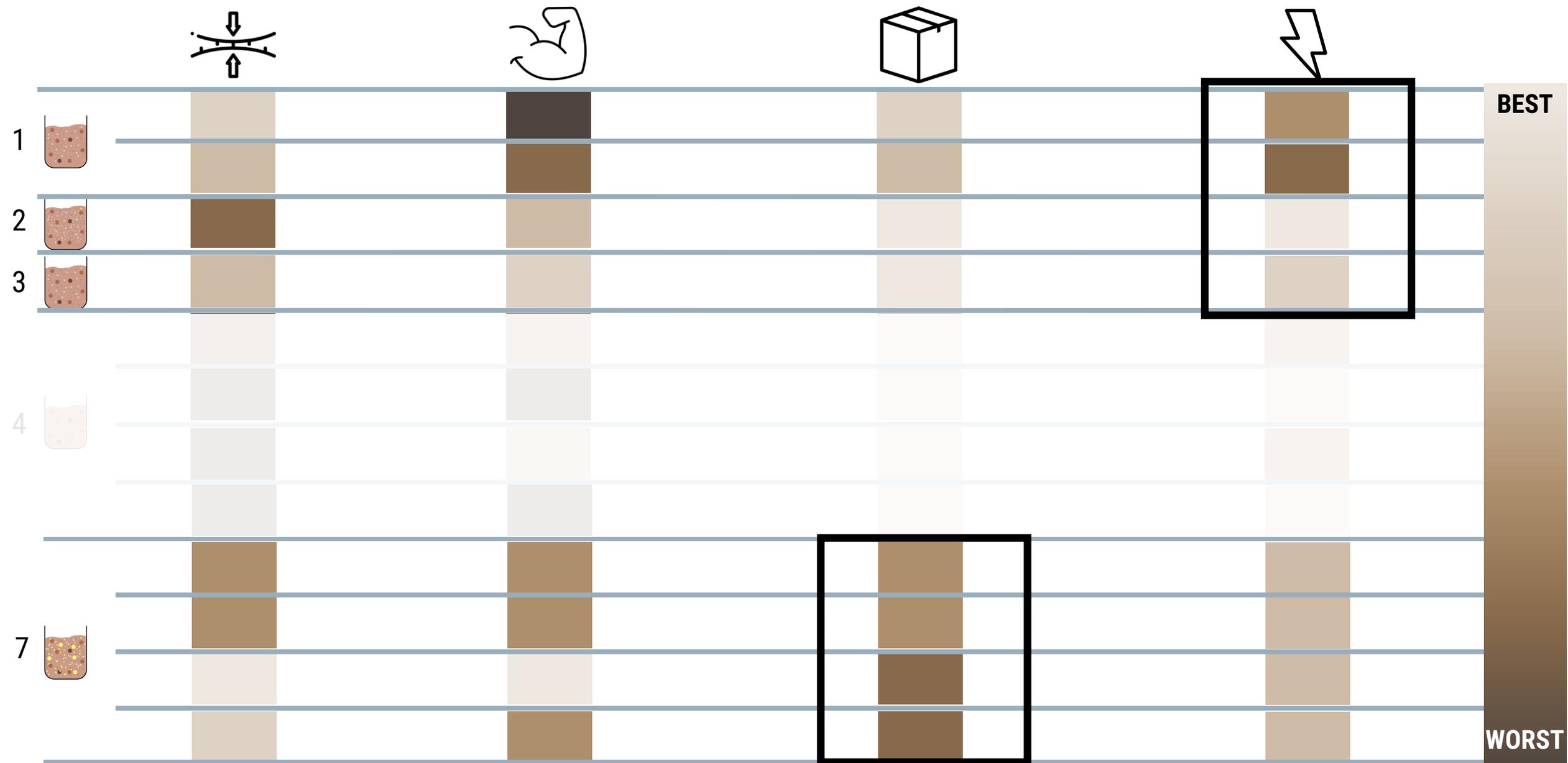
# EXPERIMENTS - DIFFERENT BATCHES



Experiments

43

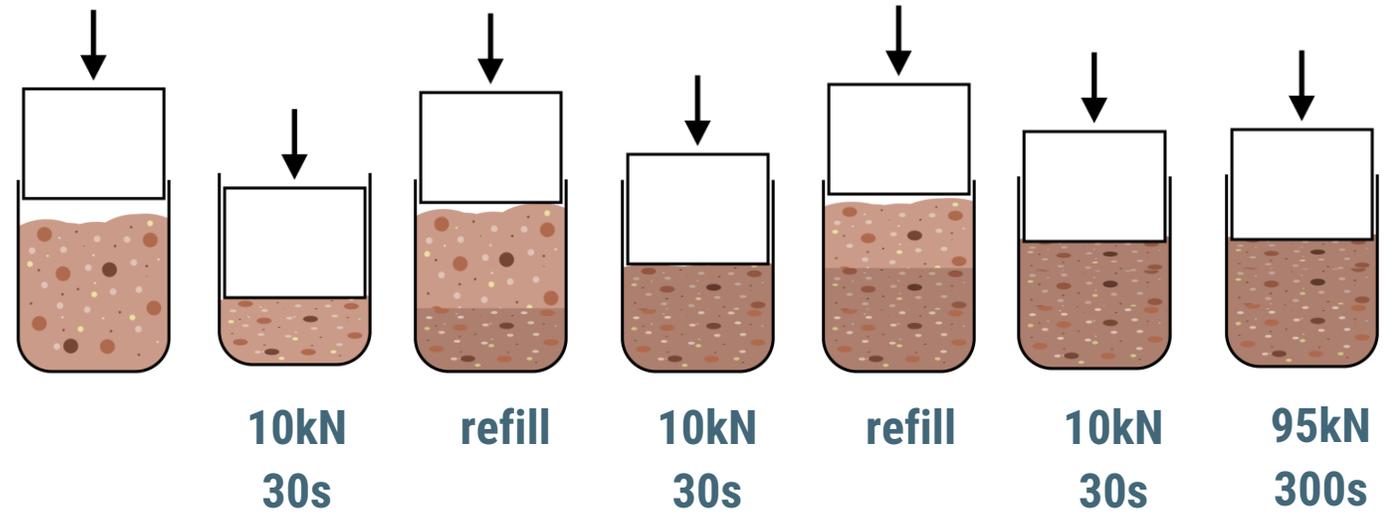
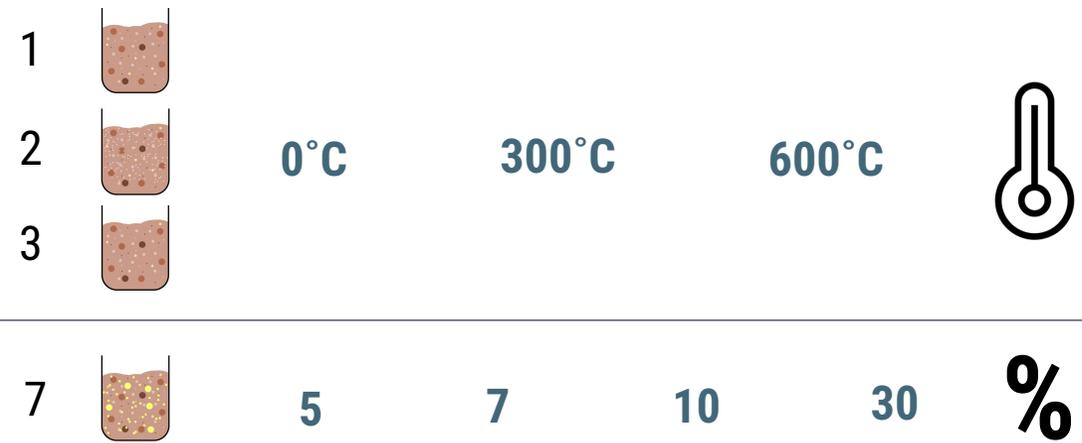
# EXPERIMENTS - DIFFERENT BATCHES



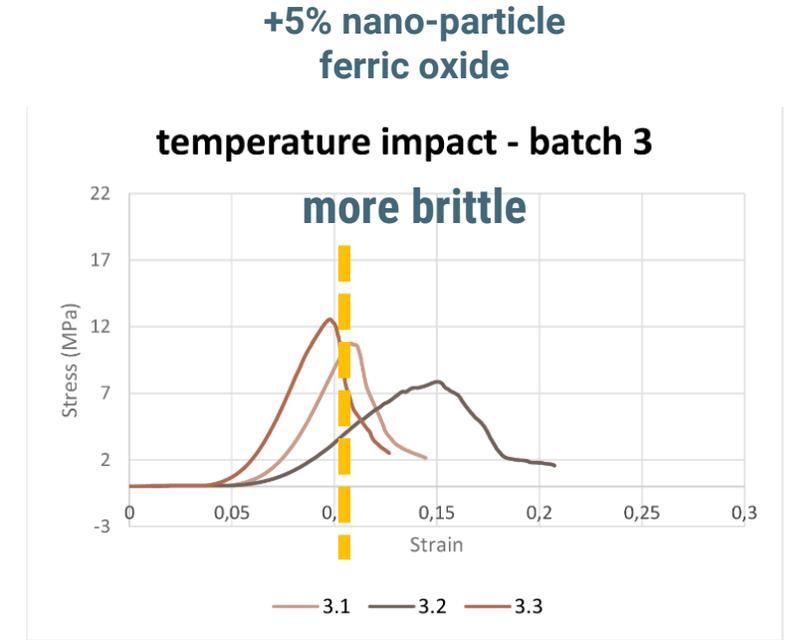
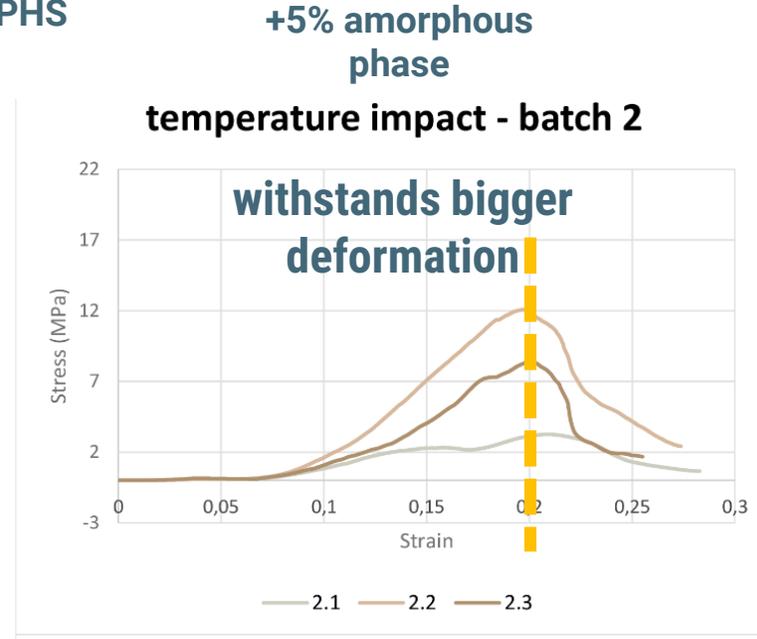
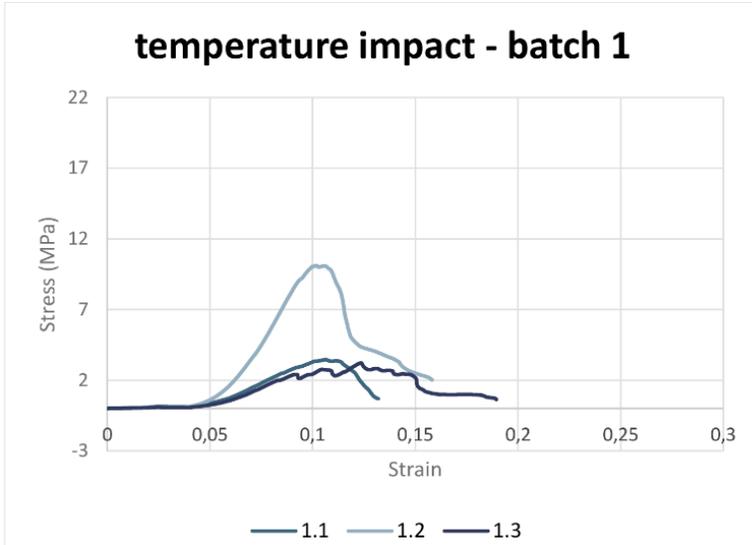
# EXPERIMENTS - SCALING UP

SAMPLES

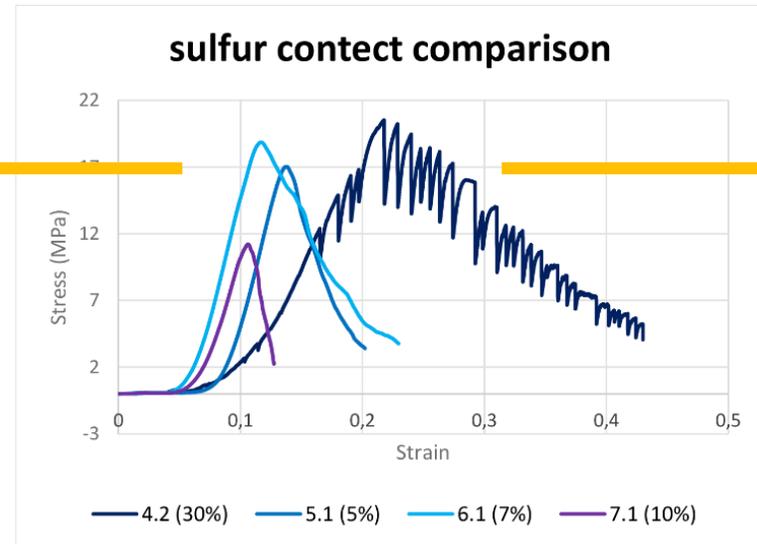
LOADING SETTINGS



## COMPRESSION TEST – STRESS/STRAIN GRAPHS

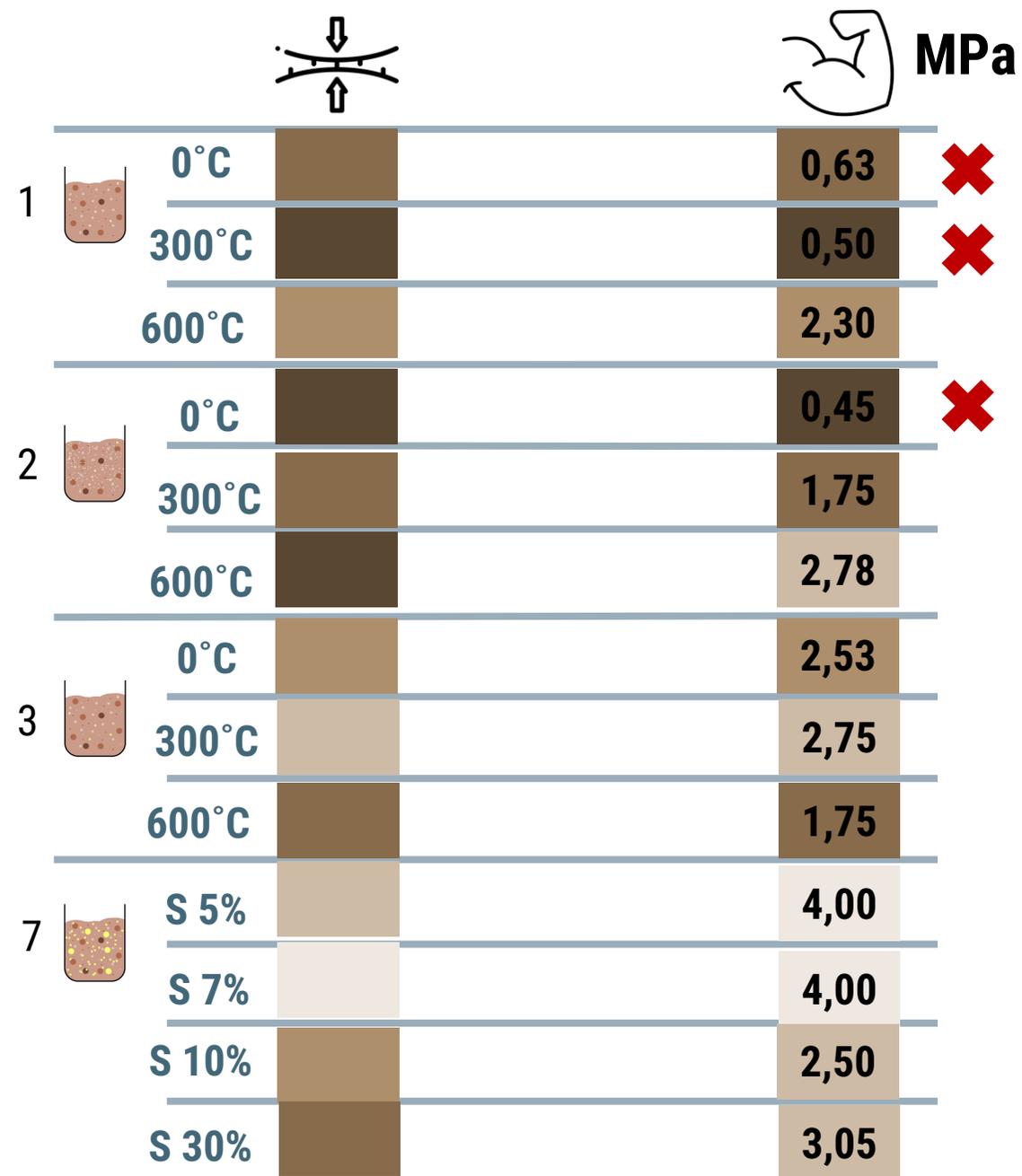


**highest peaks**



**withstands small cracks**

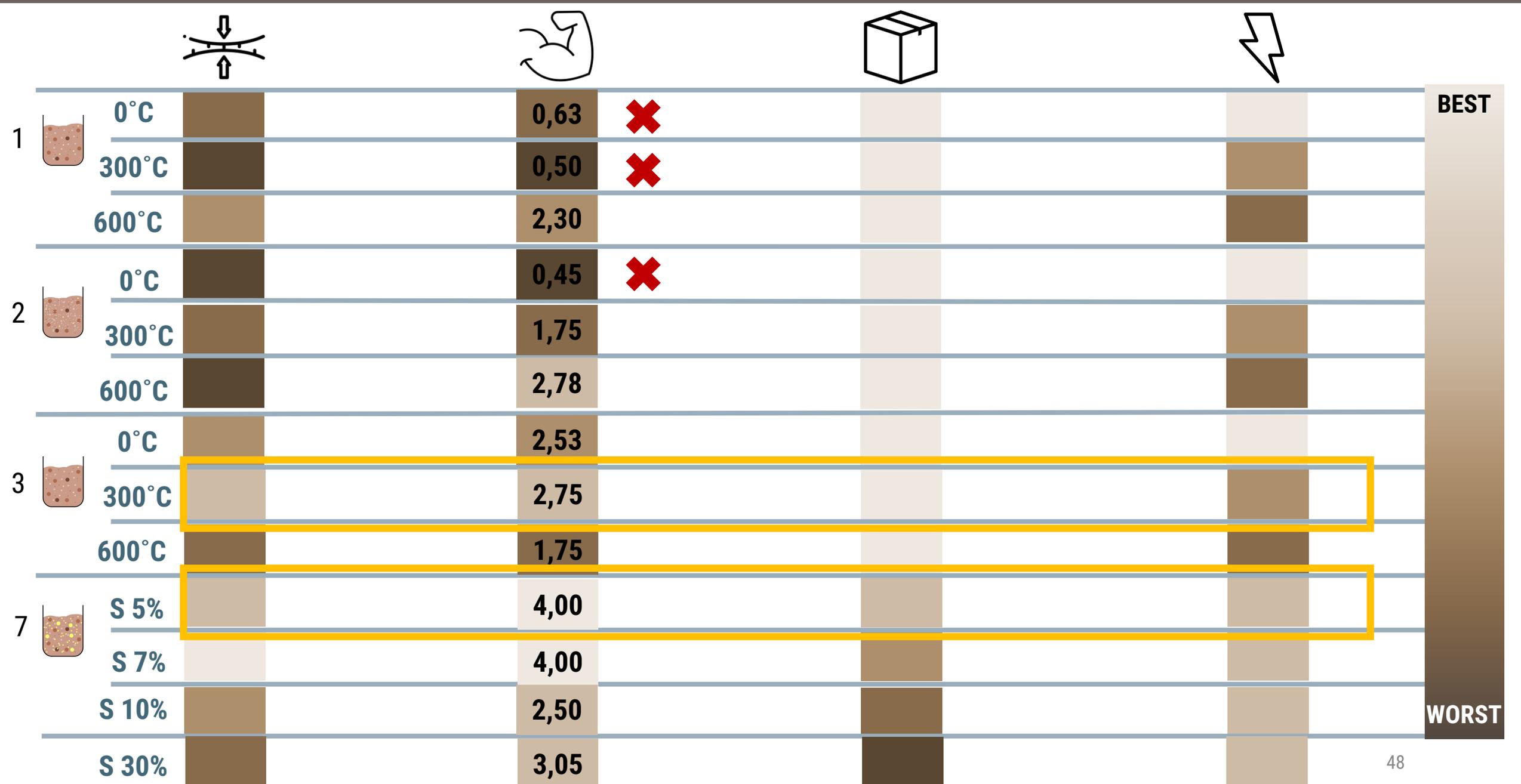
**+ sulfur powder (10, 20, 30, 50%)**



compressive strength 1,5 – 2 MPa



# EXPERIMENTS - SCALING UP – MECHANICAL TESTS - RESULTS

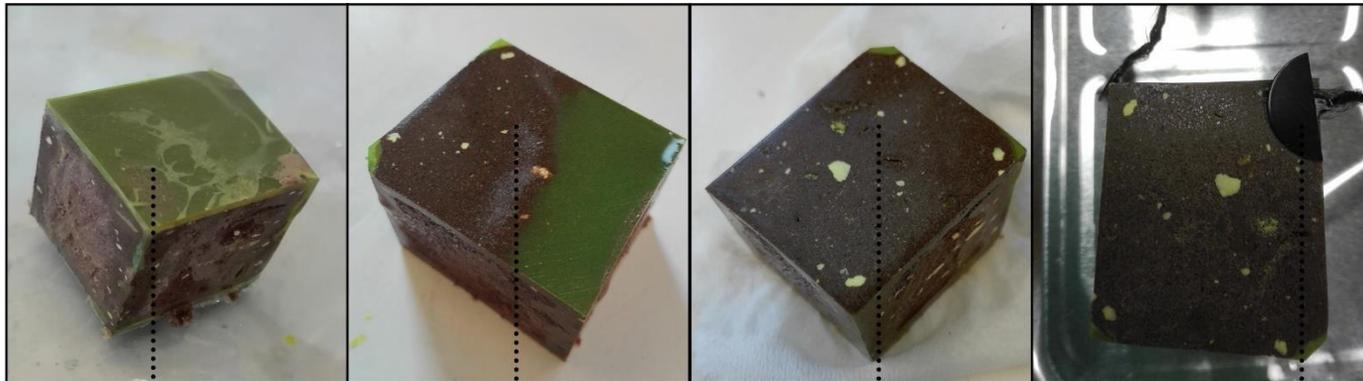


## ESEM – ENVIRONMENTAL SCANNING ELECTRON MICROSCOPE

## BSE – BACKSCATTERING ELECTRON IMAGING

**DENSER MATERIAL  
= BRIGHTER COLOUR**

### SAMPLE PREPARATION STEPS

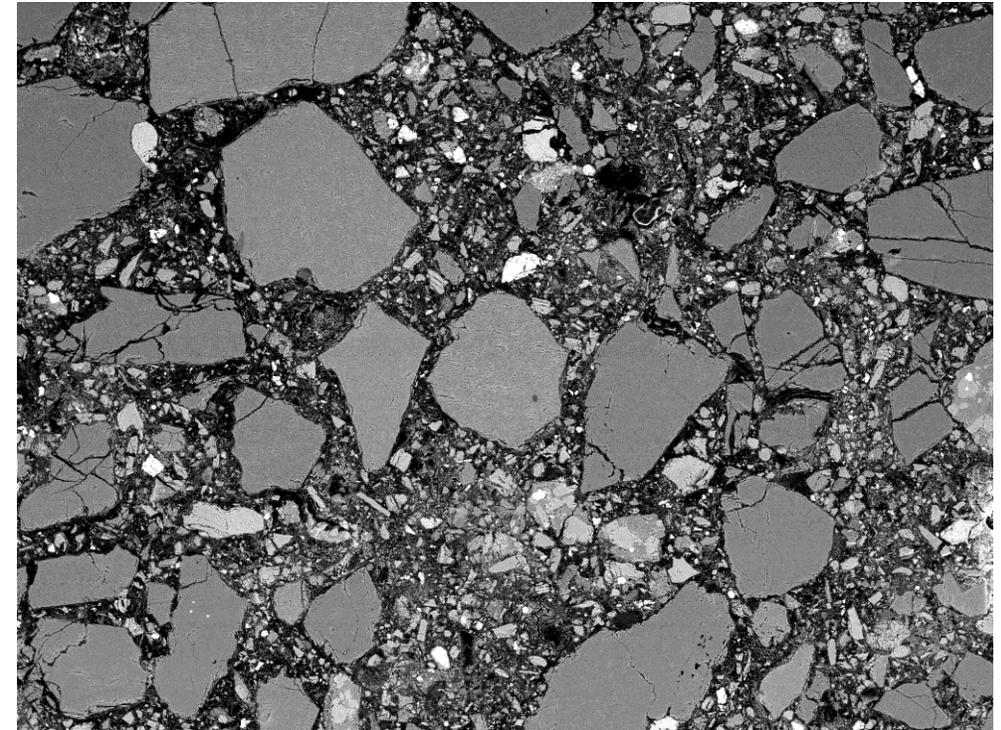


epoxy layer after impregnation

grinding and polishing

placing micro layer of conductive carbon

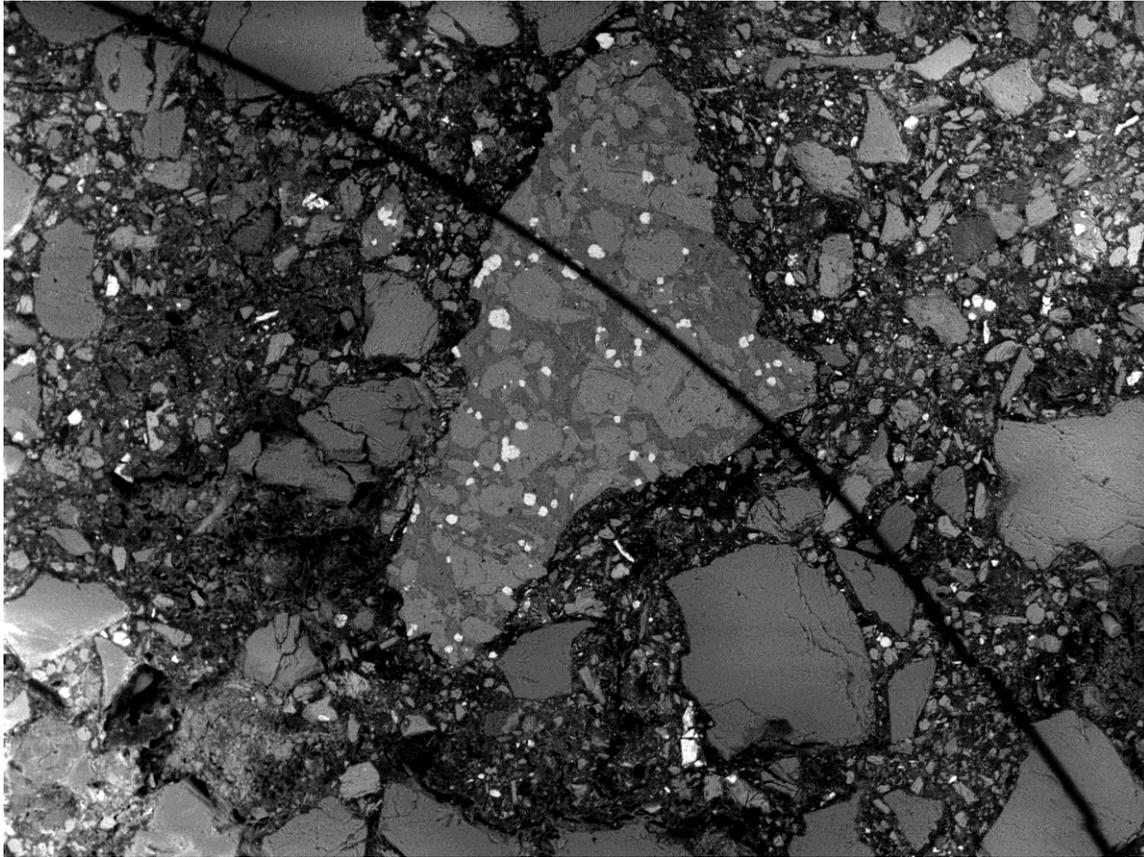
prepared sample for microscopic analysis, earth connection



1

NANO-PARTICLE FERRIC OXIDE

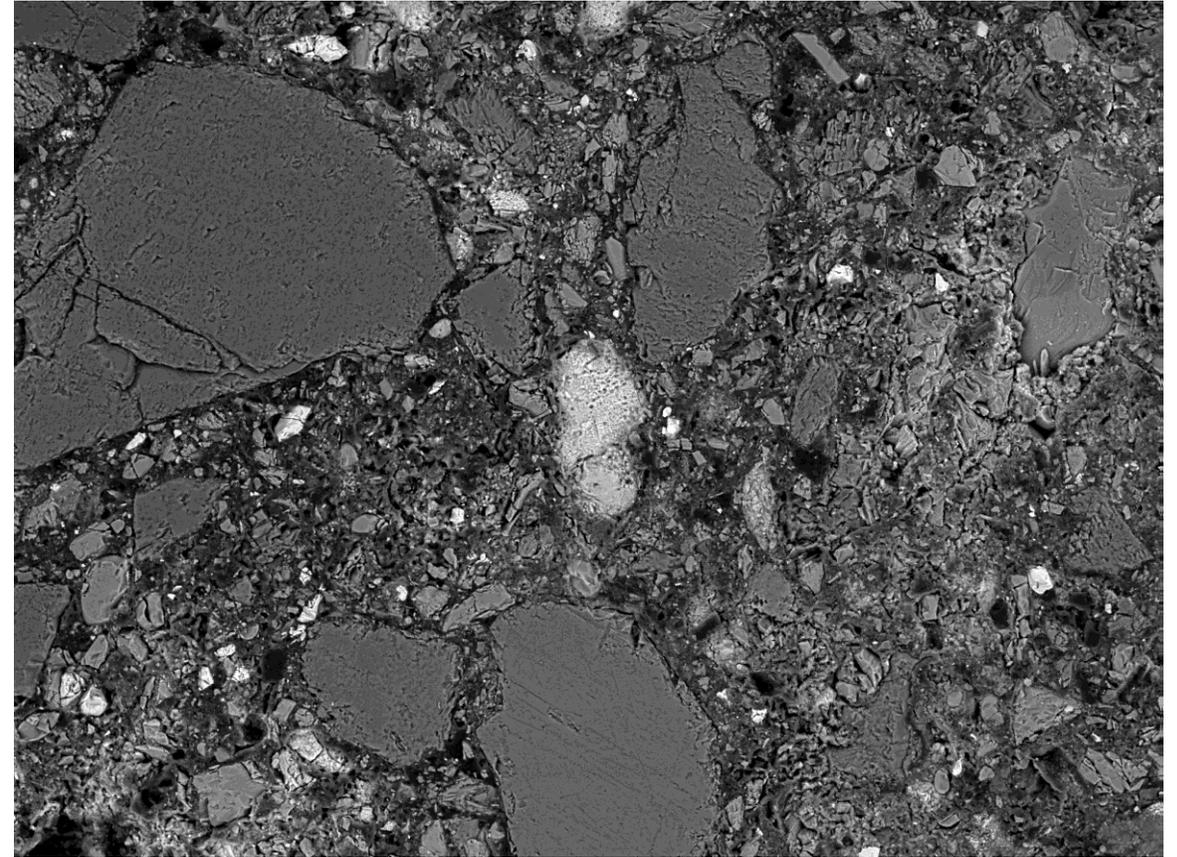
**HOMOGENOUS COMPOSITION**



2

5% SULFUR POWDER

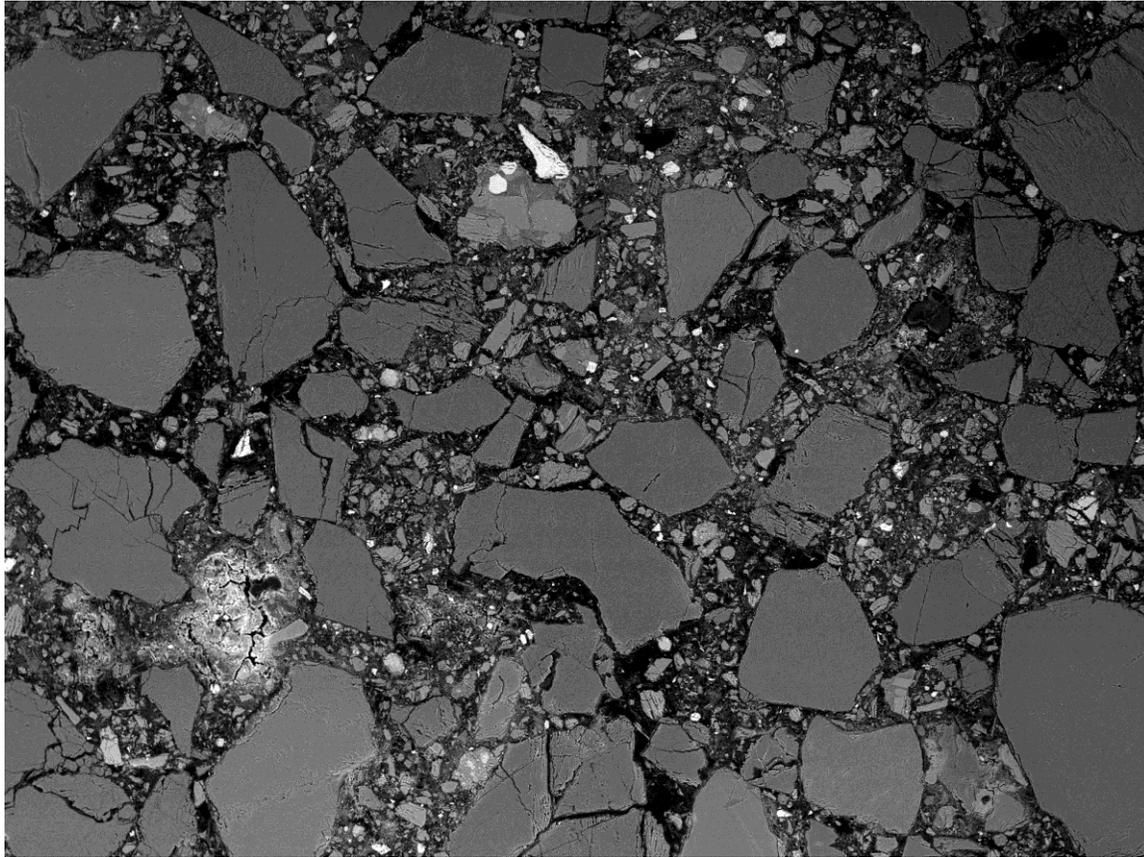
**HOMOGENOUS COMPOSITION**



1

NANO-PARTICLE FERRIC OXIDE

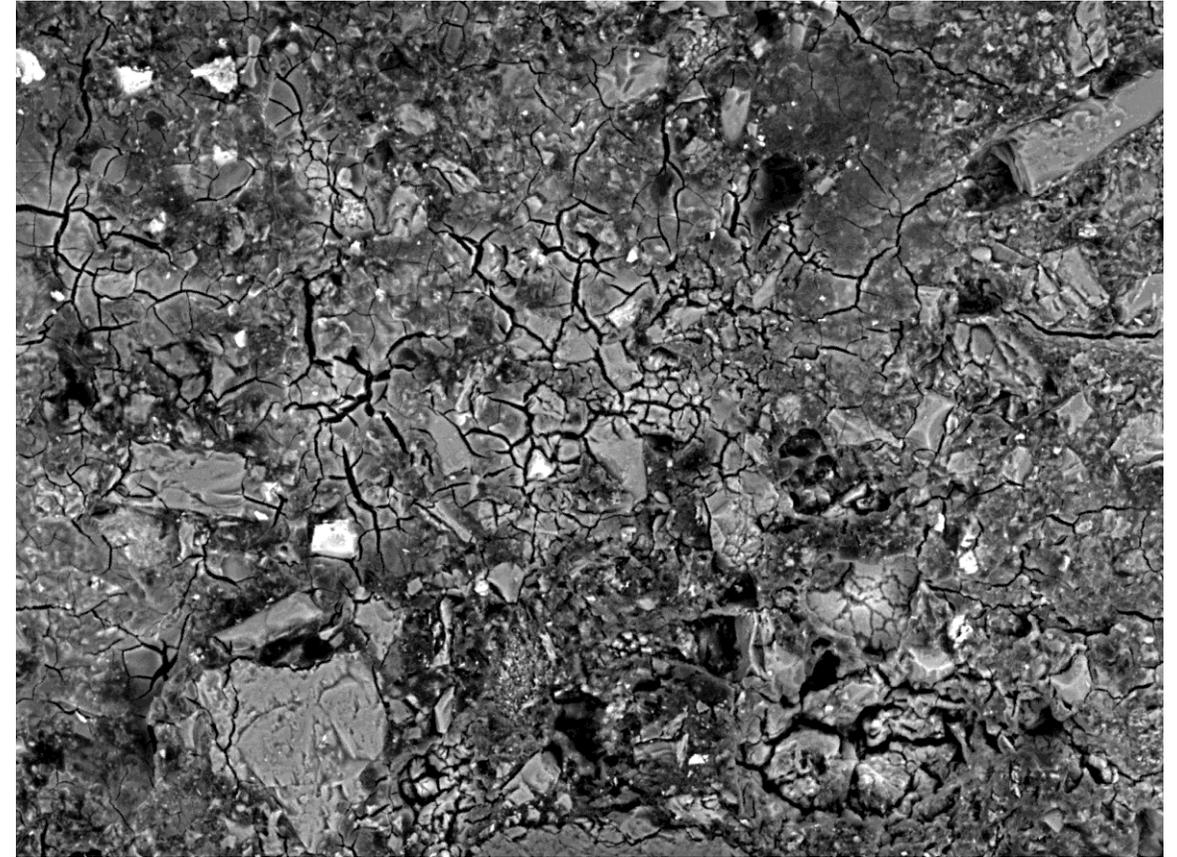
**CRACKS**



2

5% SULFUR POWDER

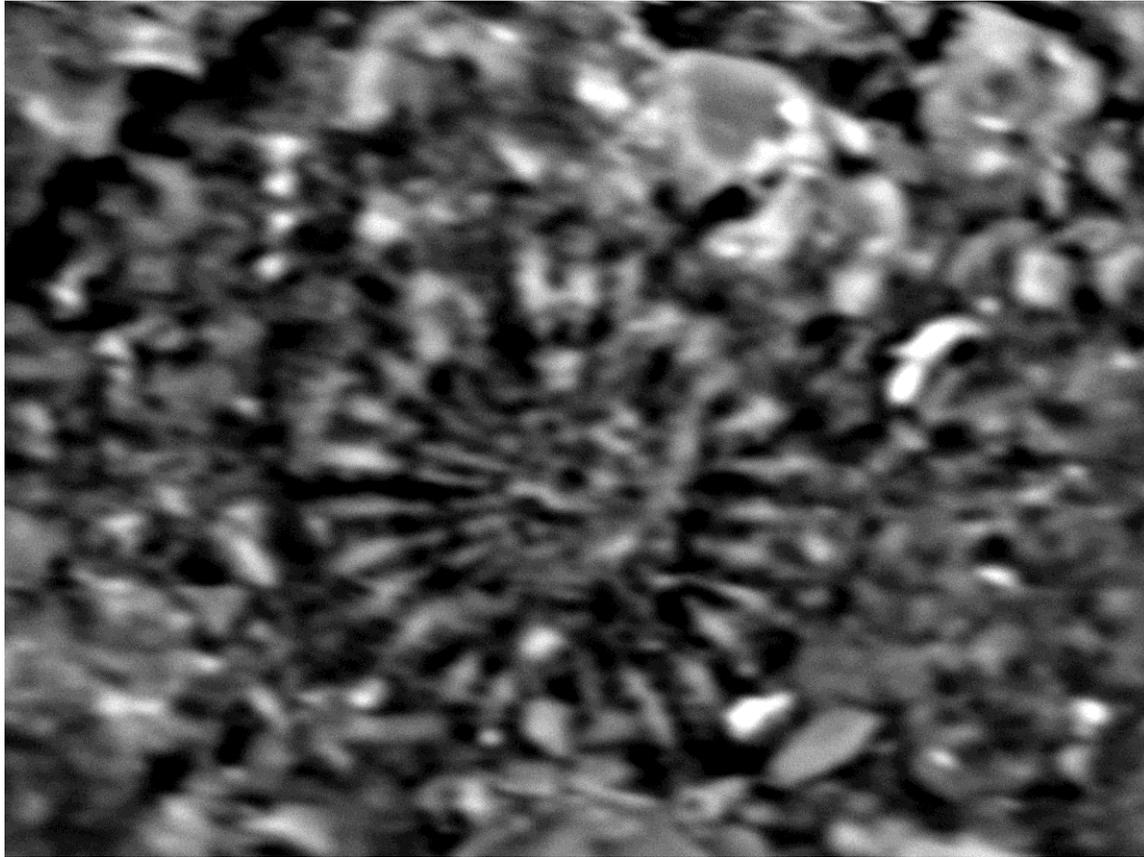
**CRACKS**



1

NANO-PARTICLE FERRIC OXIDE

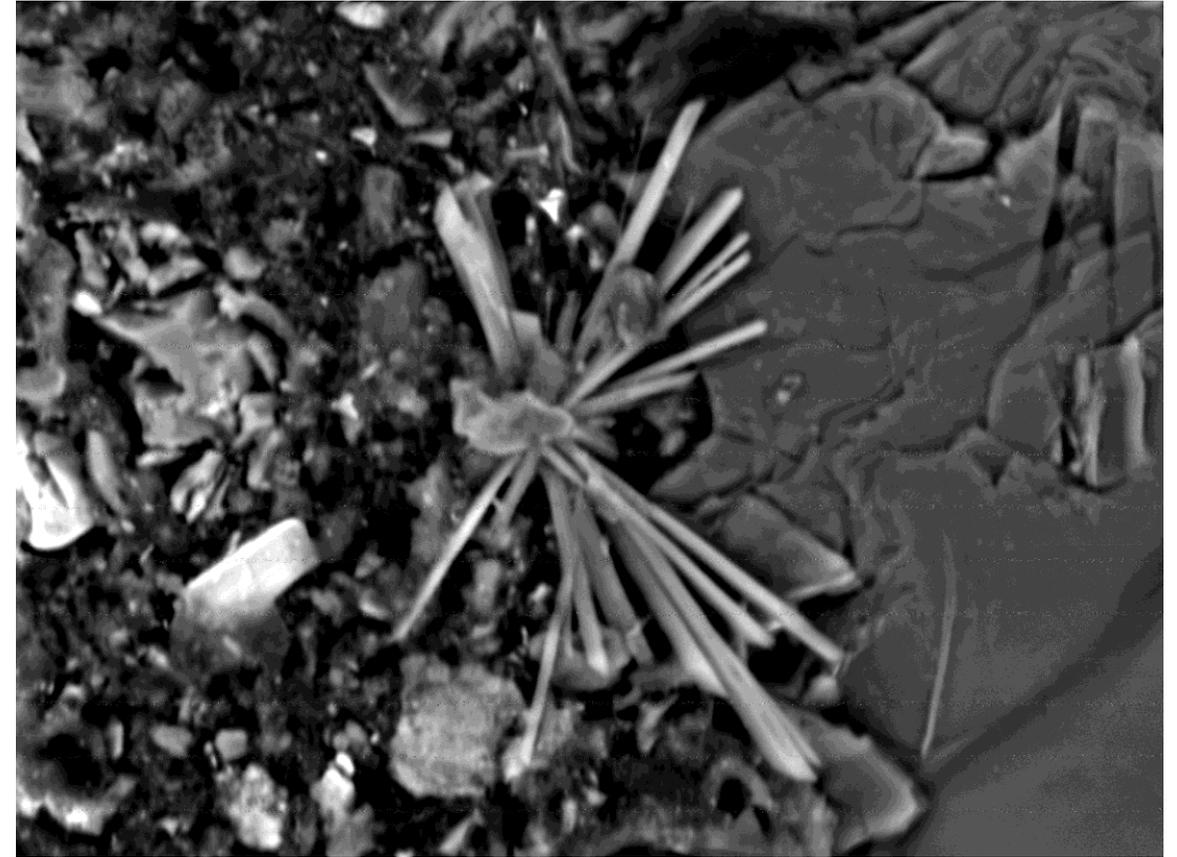
## CHARACTERISTIC FORMS



2

5% SULFUR POWDER

## CHARACTERISTIC FORMS



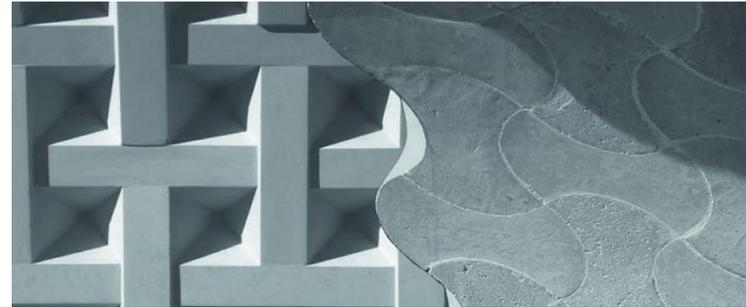


## COMPONENT



- Compressive strength : 1,5 – 2 MPa

## CONSTRUCTION METHOD



- Brittle material
- Constructing curved shapes
- Minimized payload, energy demand

## STRUCTURE



- Compressive only - vault
- Resistant to loads
- Radiation protection
- Dimensions required for habitable space

## OPTIONS

### INTERLOCKING



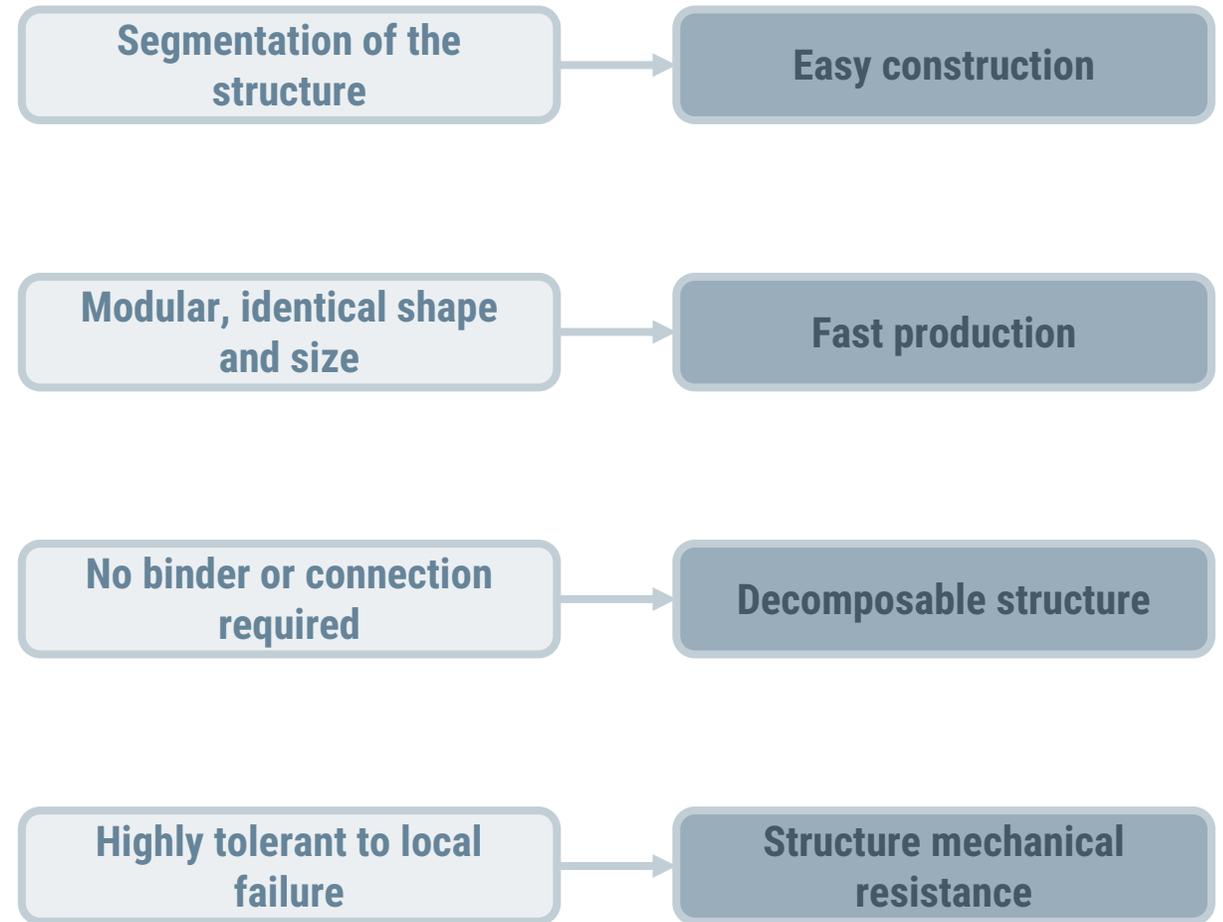
### MORTAR



## OPTIONS



## ADVANTAGES



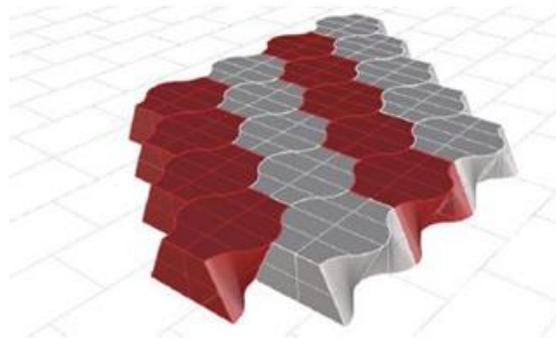
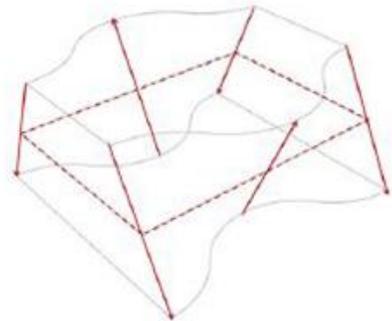
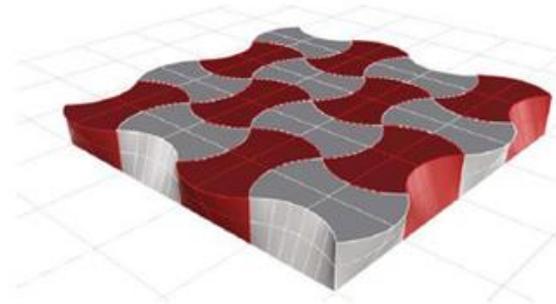
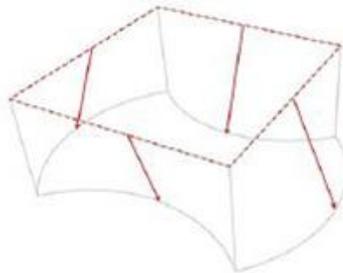
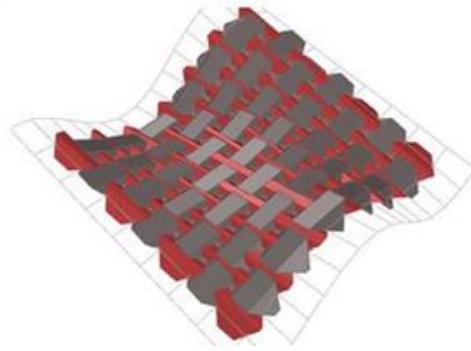
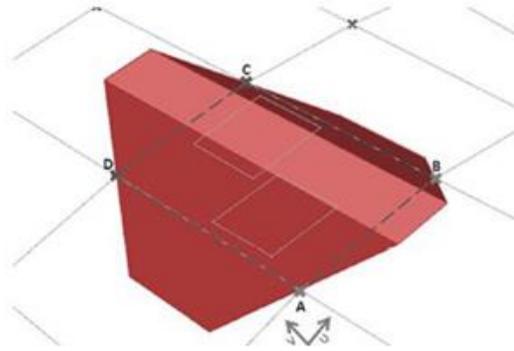
## Factors

Brittle material

Construct solid forms

Simple shapes

## Analysed examples



Design



Source: Fallacaras reinterpretation of Truchet vault

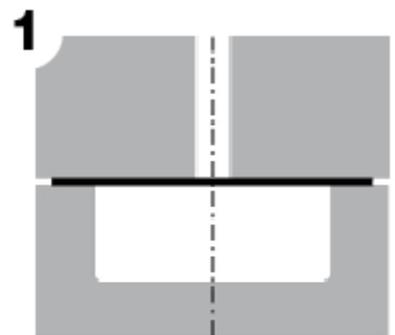


Source: VECO, Marmomacc

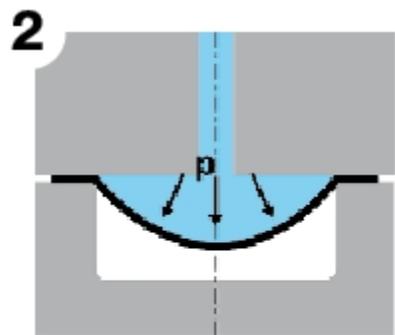
## FABRICATION PROCESS



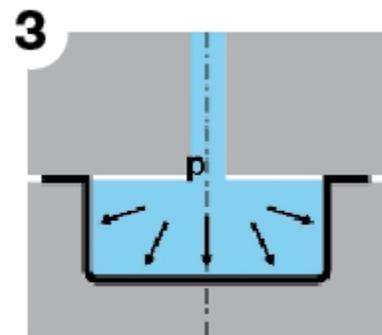
## ALTERNATIVES FOR FURTHER RESEARCH – RUBBER POD FORMING, HYDROFORMING



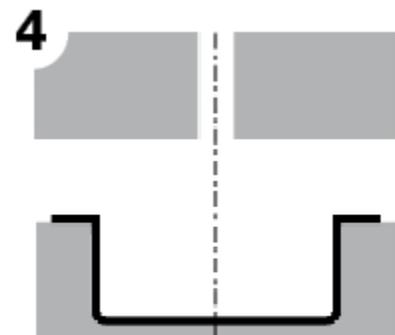
1  
Positioning of metal sheet



2  
Fluid enters



3  
Forming metal sheet



4  
Extraction of metal part

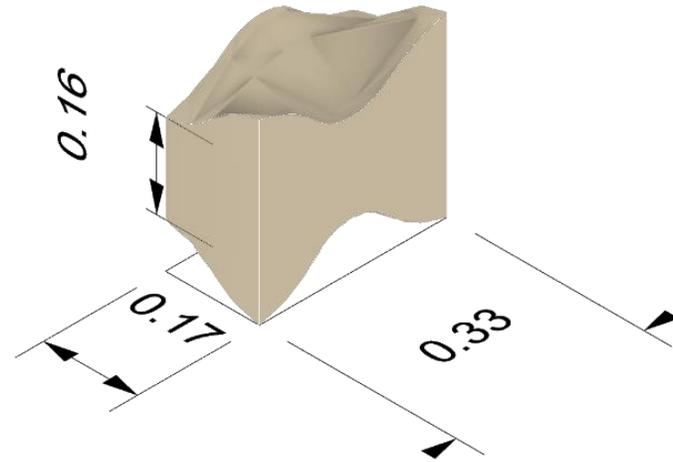
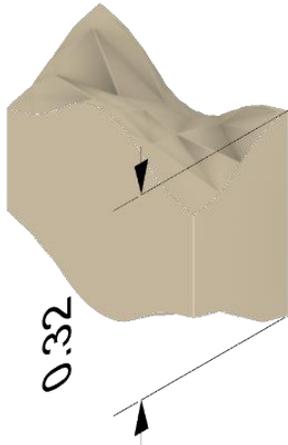
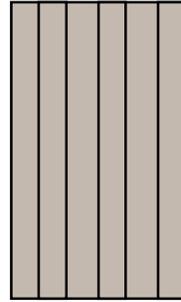
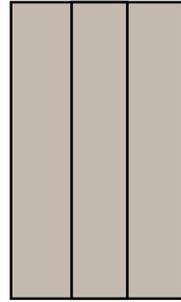
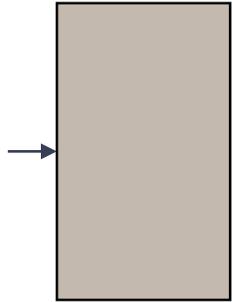


## SIZE

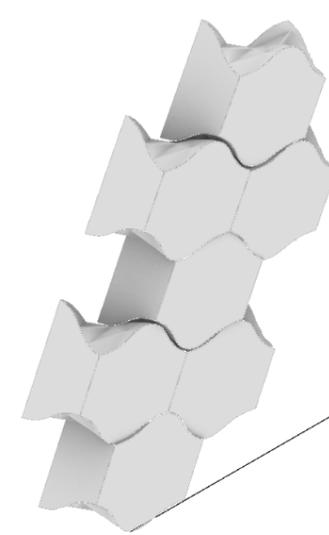
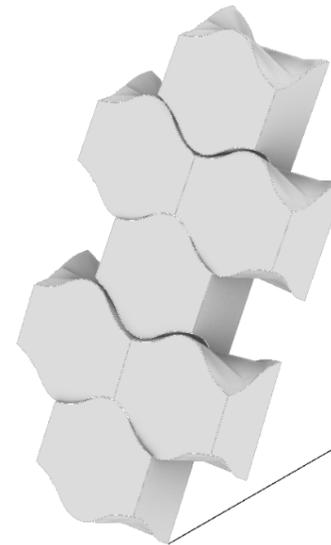
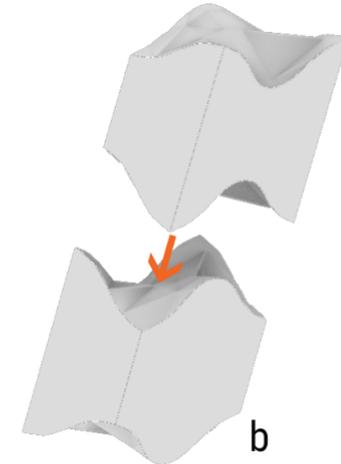
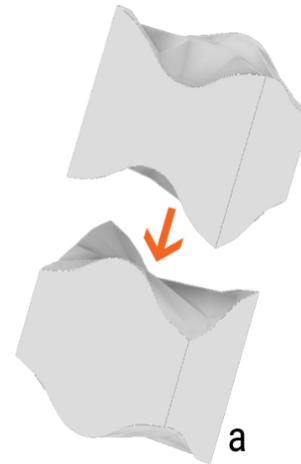
1m

3x33cm

6x17cm

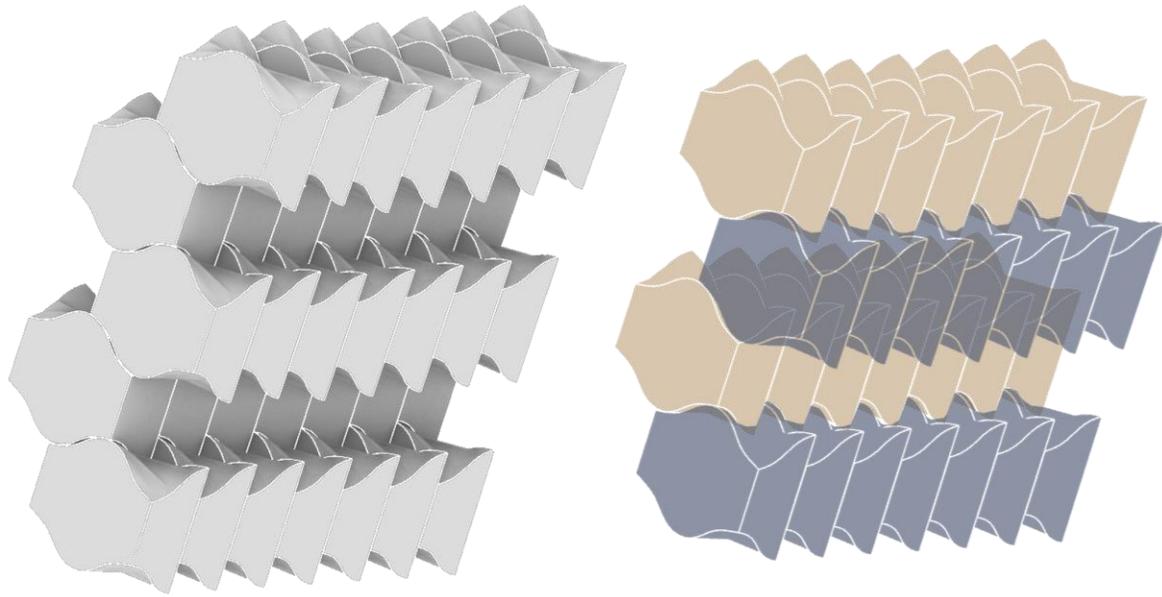


## ARCH BUILDING ORIENTATION

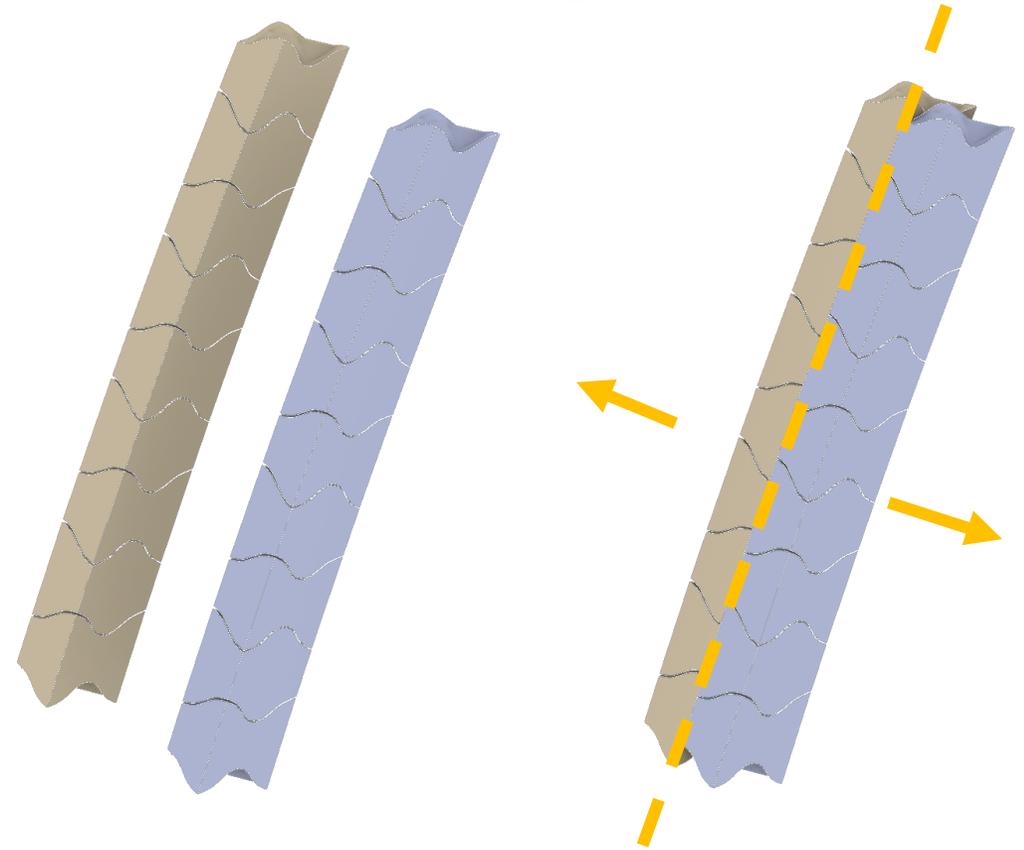


# BRICK ORIENTATION OPTIONS

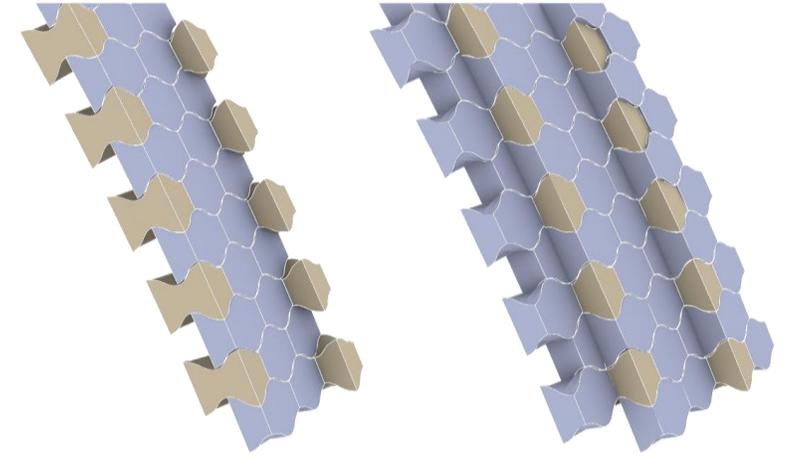
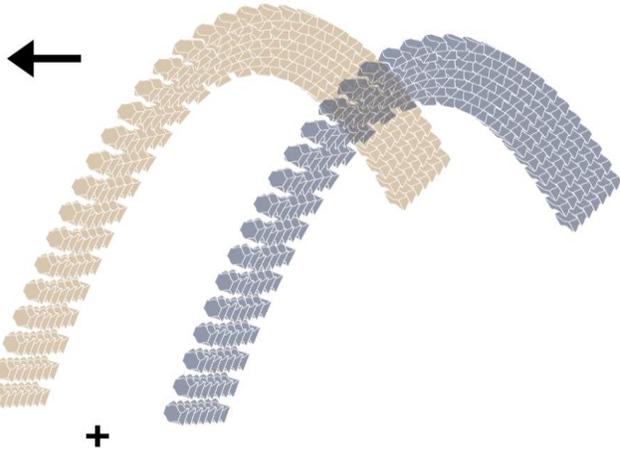
**a**



**b**



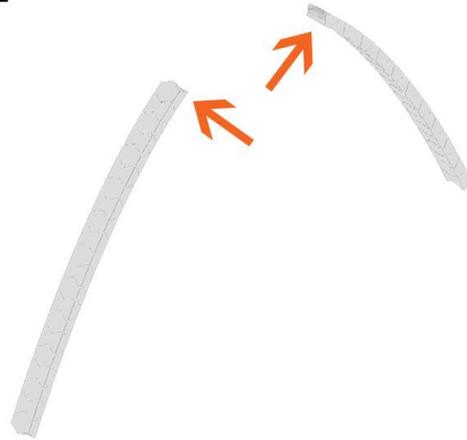
**a**



1



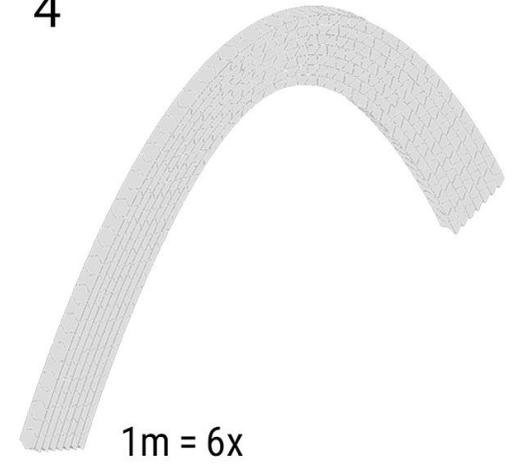
2



3

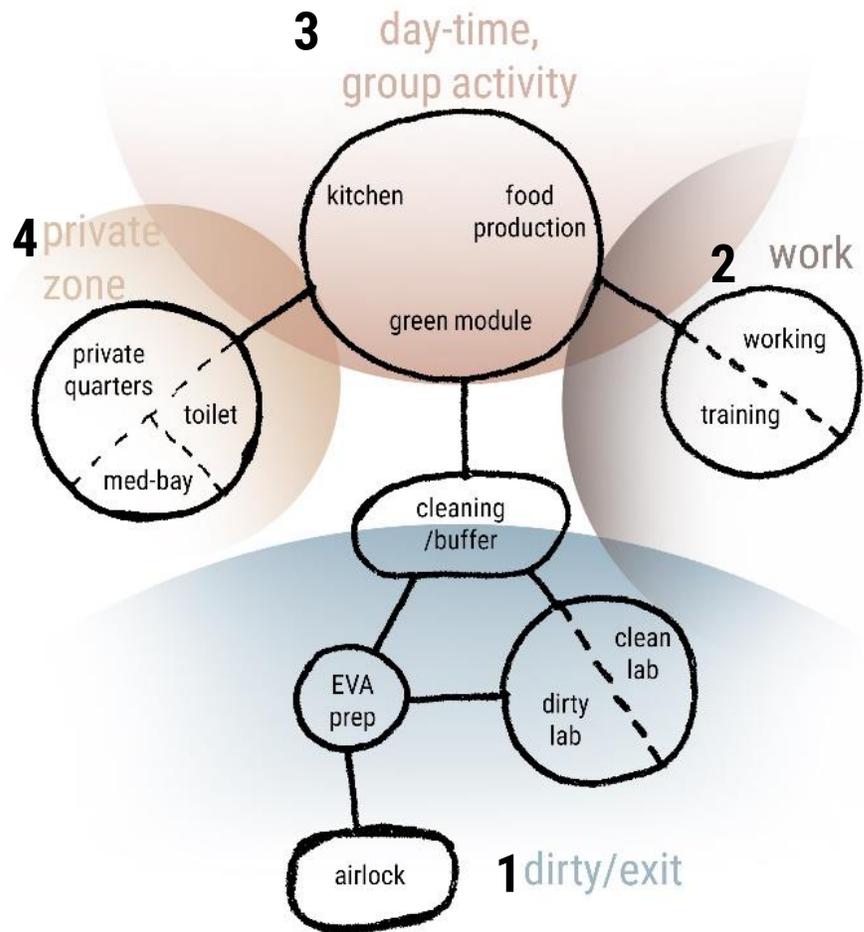


4

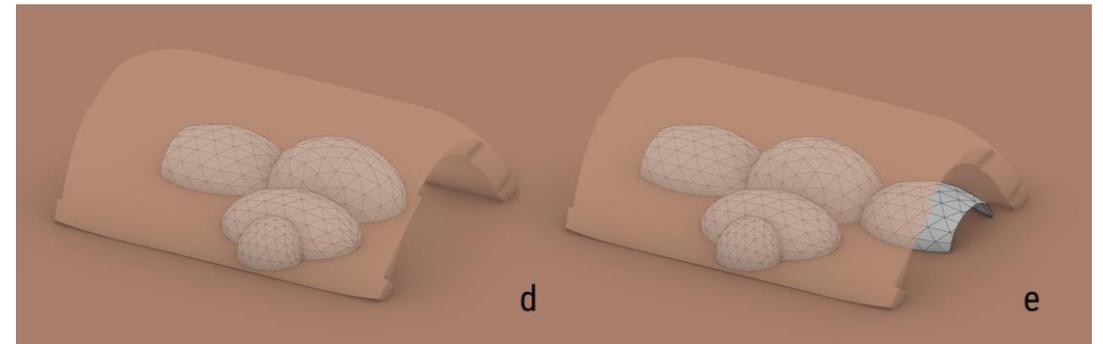
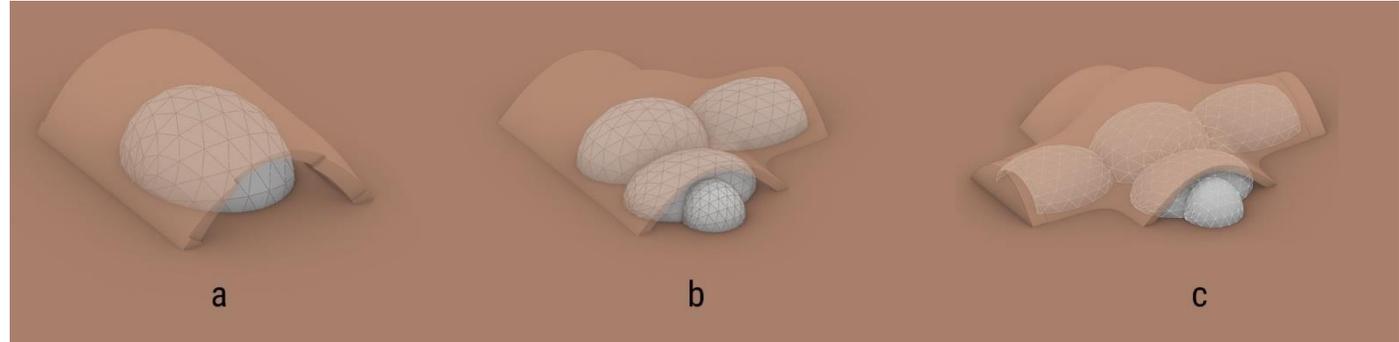




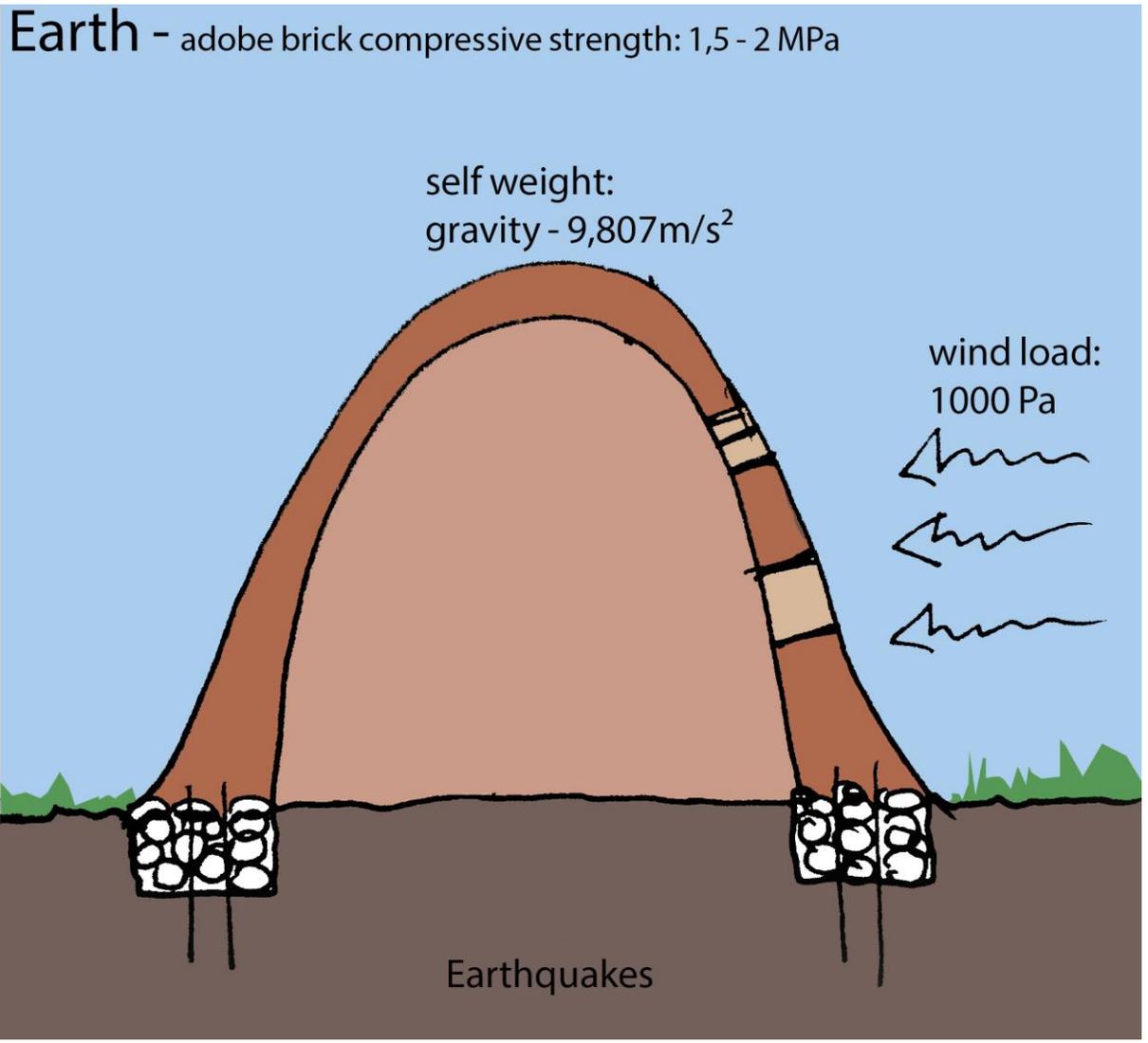
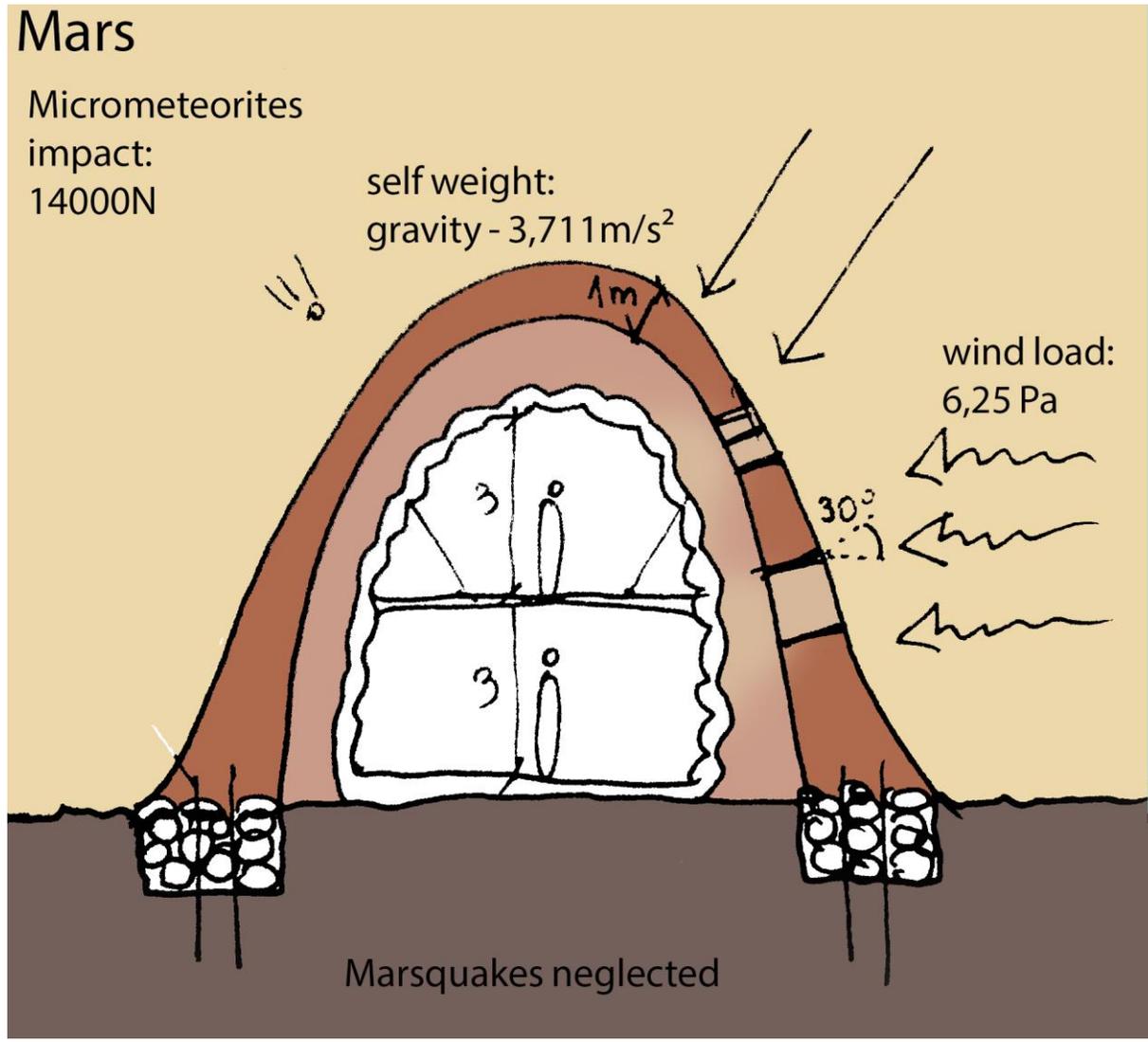
## Ideal configuration - zoning



## Ideal form – possible to expand and adapt to mission

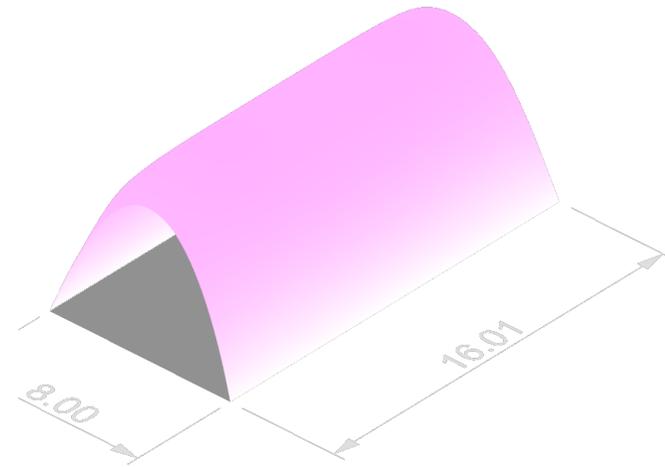
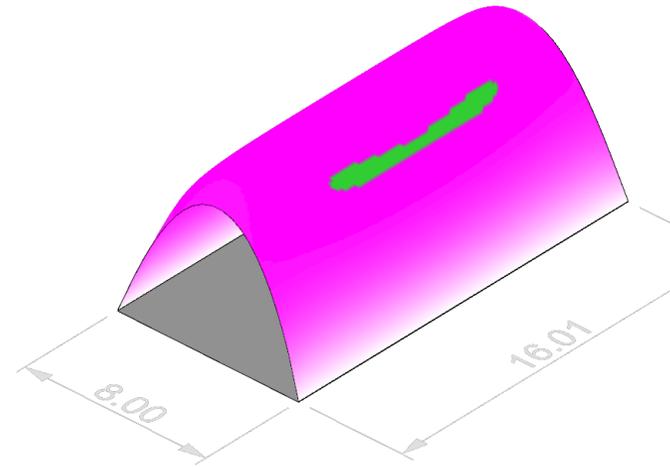
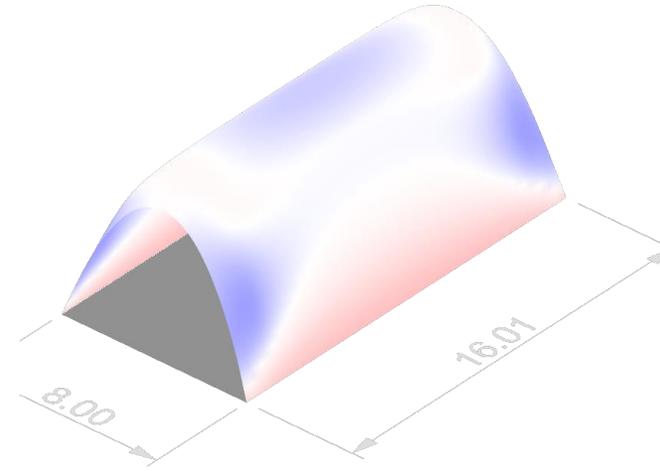
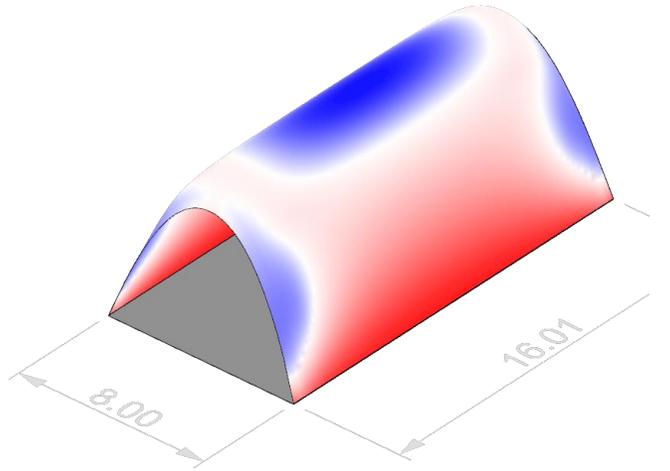
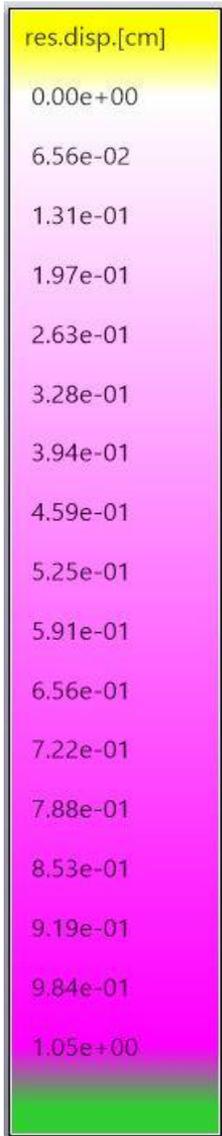


# Structure - requirements

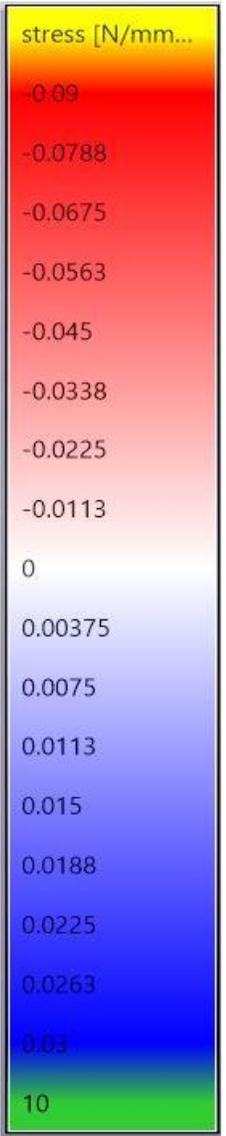


# COMPARISON EARTH VS MARS

## EARTH



## MARS



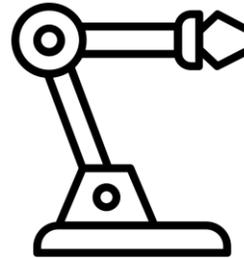
## MINIMUM 300 DAYS AVAILABLE

### Preparation



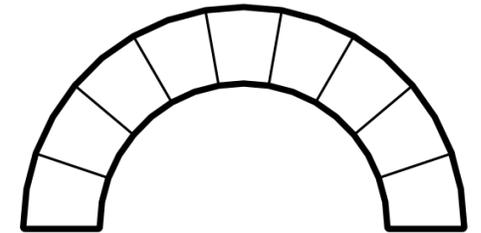
95 DAYS

### Production Process



180 DAYS

### Construction

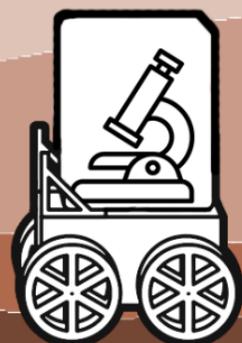


35 DAYS

**X 12 HOURS**

PREPARATION

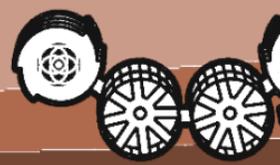
95 DAYS



CHARACTERISATION



EXCAVATION



COLLECTION



SIEVING/GRINDING



240 HOURS

240 HOURS

540 HOURS

total 180 kW

0,6 kW/h



Source: NASA, JPL

	Unit	Value
Regolith Payload	kg	80
Power usage per kg	W	4
Structure Mass	kg	45000
Regolith volume	m <sup>3</sup>	~27
Volume per day (min)	m <sup>3</sup>	2,7

FABRICATION

180 DAYS



ADJUSTING COMPOSITION



COMPACTION



THERMAL TREATMENT

Batch type 3.3	Unit	Value
Compaction of 1 brick (Work = F x d)	kW	0,7
Number of bricks required		1740
Heating time	h	4
	s	14400
Thermal treatment of one brick ( $\Delta E$ )	J	276800
	W	19

720 HOURS

total 1195 kW

1,63 kW/h

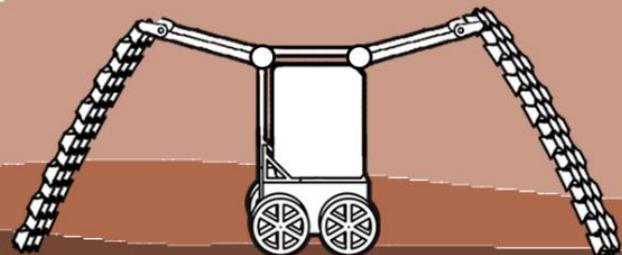
1440 HOURS

total 33447 kW

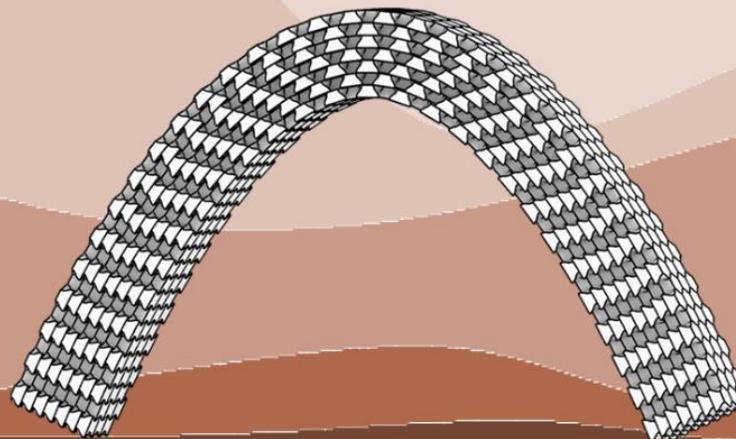
23 kW/h

**CONSTRUCTION**

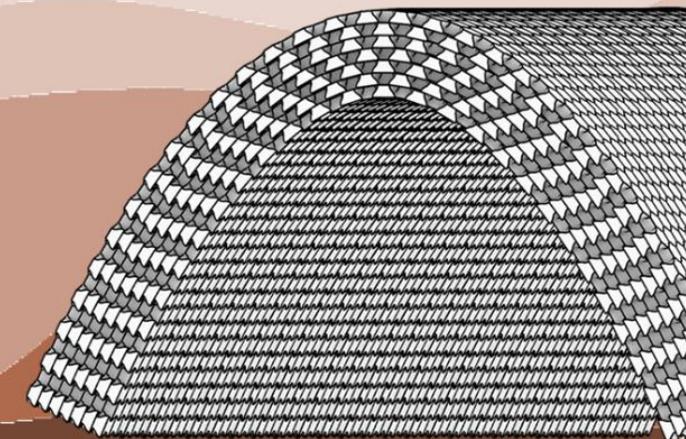
**35 DAYS**



**ROBOTIC ASSEMBLY**



**MODULE FINISHING**



**CONTINUING CONSTRUCTION**

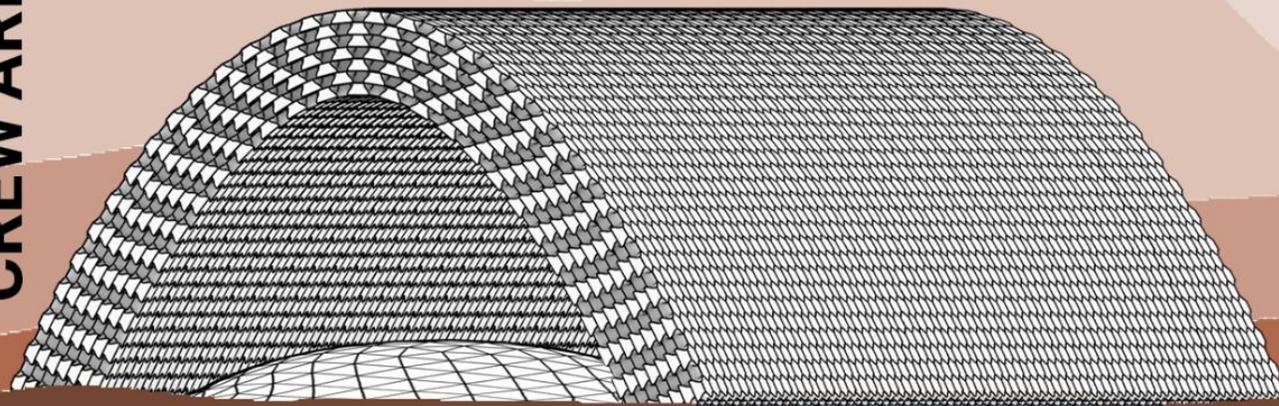
**420 HOURS**

**0,285 1195 kW**

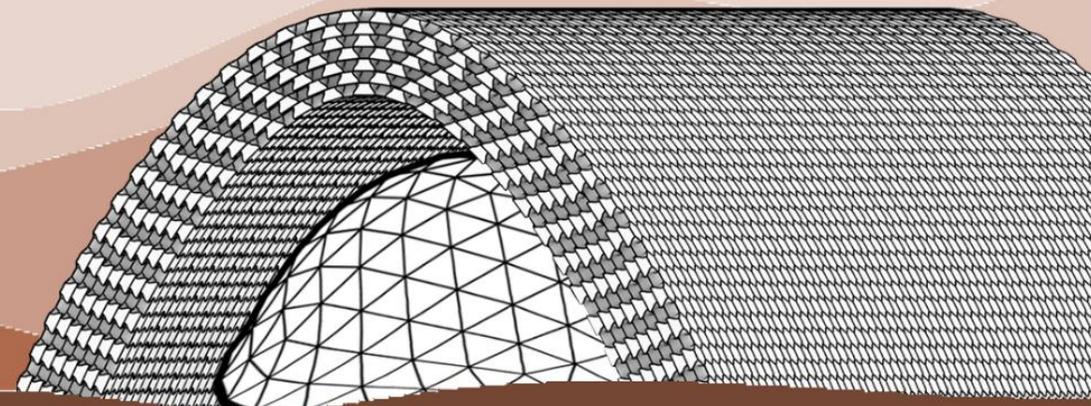
**0,00068 kW/h**

	Unit	Value
<b>Brick mass</b>	kg	25,86
<b>gravity</b>	m/s <sup>2</sup>	3,71
<b>Height 1</b>	m	6
<b>Energy 1</b>	J	514
<b>Power 1</b>	W	0,30
<b>Power 1 for all bricks</b>	W	171
<b>Height 2</b>	m	3
<b>Energy 2</b>	J	257
<b>Power 2</b>	W	0,15
<b>Pwer2 for all bricks</b>	W	86
<b>Height 3</b>	m	1
<b>Energy 3</b>	J	86
<b>Power 3</b>	W	0,05
<b>Power 3 for all bricks</b>	W	29

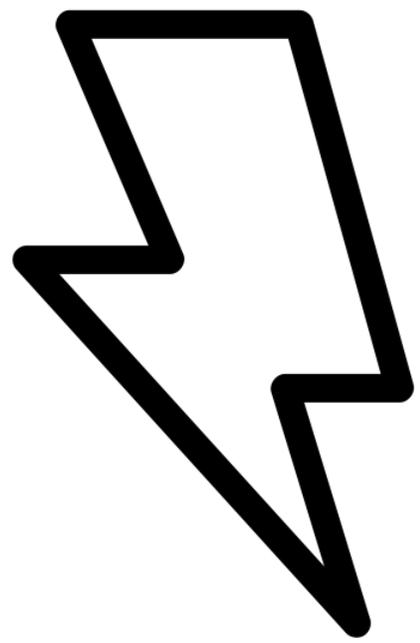
**CREW ARRIVAL**



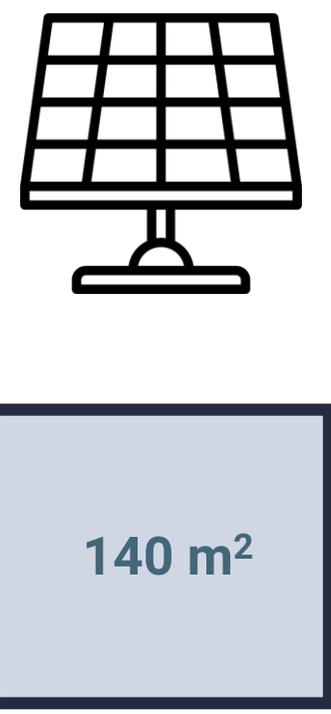
PLACING HABITABLE MODULE



INFLATING MODULE

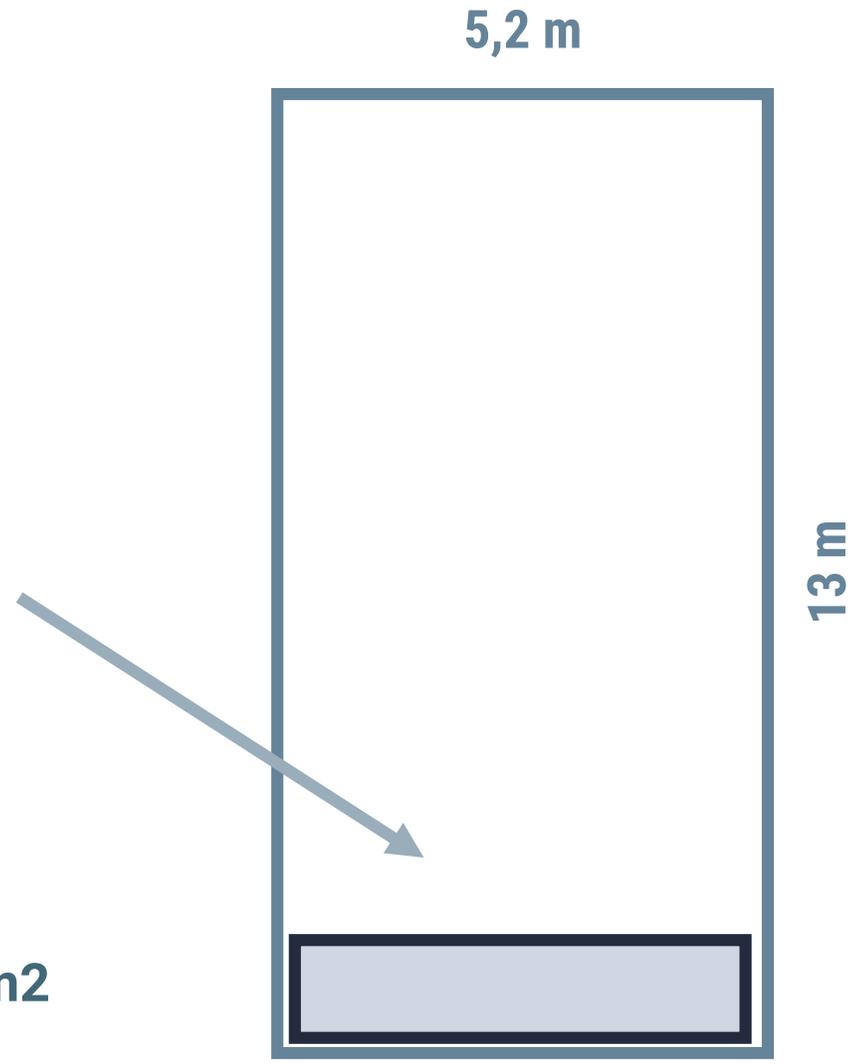


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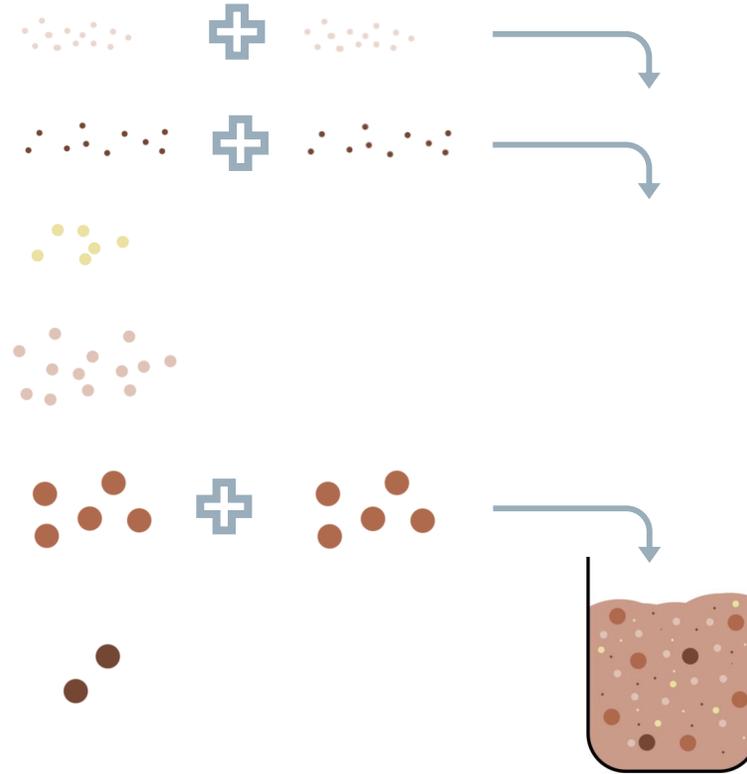
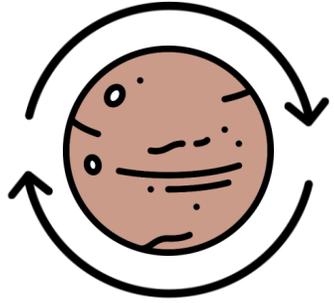
TOTAL ~ 40MW  
144 kW per sol  
144 000 W per sol

1034 W per sol per 1m2



Falcon Heavy Rocket

# CONCLUSIONS



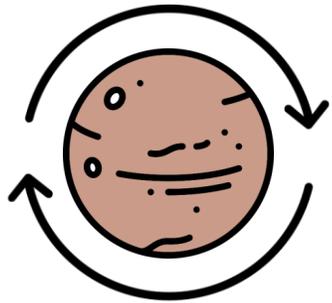
**DISCUSSION & TESTING**

**INNOVATIVE OPTIMISATION**

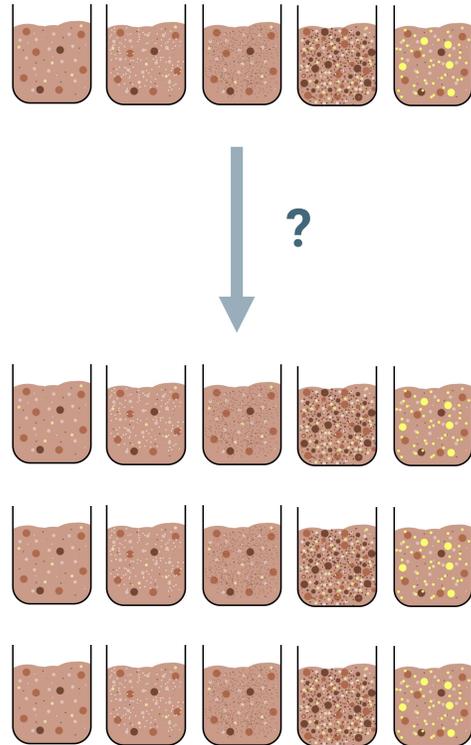
**AVAILABLE TECHNOLOGY, EFFICIENCY**

## Future Research

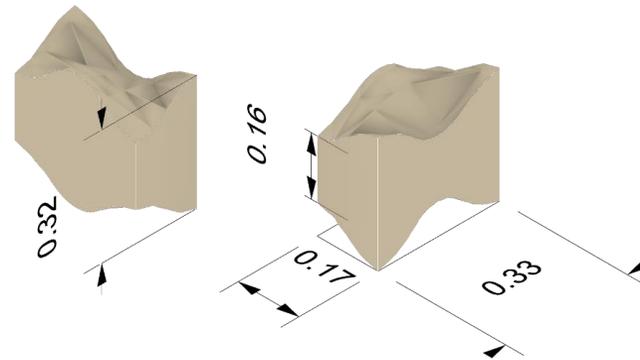
Sustainability on Mars



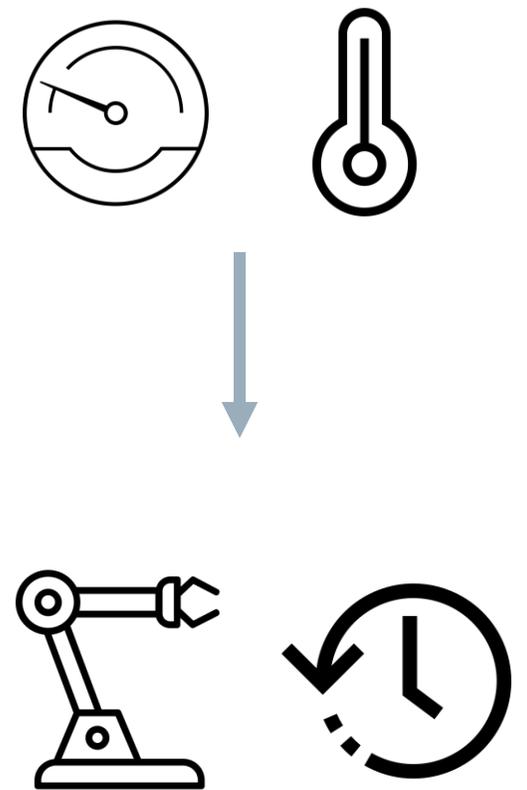
Composition adjustment options study



Construction & Fabrication Method



Conditions Simulation - IMPACT



## Limitations

**small amount of samples**

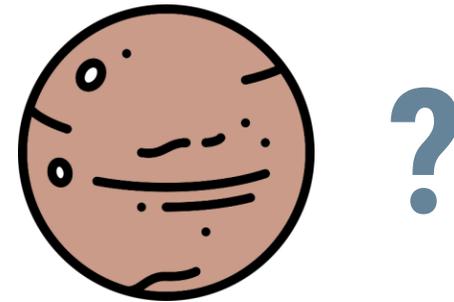
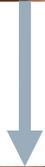
**often estimations instead of calculations**

**limited time for experimental research resulted in inaccuracy**

**malfunctions of equipment**



GAP



# THANK YOU!

## QUESTIONS / FEEDBACK