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Assimilating aircraft-based measurements to improve Forecast Accuracy of Volcanic Ash Transport

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Abstract

The 2010 Eyjafjallajökull volcano eruption had serious consequences to civil aviation. This has initiated a lot of research on volcanic ash transport forecast in recent years. For forecasting the volcanic ash transport after eruption onset, a volcanic ash transport and diffusion model (VATDM) needs to be run with Eruption Source Parameters (ESP) such as plume height and mass eruption rate as input, and with data assimilation techniques to continuously improve the initial conditions of the forecast. Reliable and accurate ash measurements are crucial for providing a successful ash clouds advice. In this paper, simulated aircraft-based measurements, as one type of volcanic ash measurements, will be assimilated into a transport model to identify the potential benefit of this kind of observations in an assimilation system. The results show assimilating aircraft-based measurements can significantly improve the state of ash clouds, and further providing an improved forecast

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as aviation advice. We also show that for advice of aeroplane flying level, aircraft-based measurements should be preferably taken from this level to obtain the best performance on it. Furthermore it is shown that in order to make an acceptable advice for aviation decision makers, accurate knowledge about uncertainties of ESPs and measurements is of great importance. *Keywords:* volcanic ash forecast, aircraft-based measurements, data assimilation, uncertainties

1 1. Introduction

The volcanic activity of Eyjafjallajökull in Iceland in 2010 (Figure 1) has 2 revealed that air traffic is highly vulnerable to volcanic eruptions. Aerosols 3 and ash from eruptions not only reduce visibility for visual navigation, but microscopic glass-rich debris accumulates and melts in the heat of aircraft turbine engines, which eventually leads to engine failure. Most of the volcanic 6 ash from the Evjafjallajökull eruption on April 14, 2010 and the following days was carried by prevailing winds in South-East direction, provoking a 8 dramatic collapse of European air traffic during several days. Preliminary 9 estimates of the direct and indirect costs that can be attributed to disruption 10 of air travel in Europe as a consequence of this volcanic eruption are in the 11 range of several billion Euros. 12

As a result of the serious consequences on civil aviation, more than 50 volcanologists, meteorologists, atmospheric dispersion modellers, and space and ground-based monitoring specialists from 12 different countries (including representatives from 6 Volcanic Ash Advisory Centers and related institutions) gathered at the Weather Meteorology Organization (WMO) head-



Figure 1: Eyjafjallajökull volcano eruption plume.

quarters in Geneva, addressing some important research priorities related to Volcanic Ash Transport Forecast problems (Bonadonna et al., 2012). One of the priorities they have identified is to use data assimilation, which refers to the (quasi-) continuous use of observational data to create initial conditions for sequences of model runs. In each assimilation step, a forecast from the previous model run is used as a first guess which is then modified to be in (better) agreement with the observations (Zehner, 2010).

For the purpose of using a data assimilation system to improve the initial 25 conditions of the the ash load, the volcanic ash measurements must be avail-26 able near-real-time. Flemming and Inness (2013) assimilated for example 27 satellite retrievals in a four-dimensional variation (4D-var) approach. Be-28 sides satellite observations of volcanic ashes, many other different scientific 29 measurement campaigns were performed in order to get information about 30 the ash plume, such as using lidars, ceilometers, balloon sondes etc. Among 31 these, also aircraft-based measurements were obtained close to the eruption 32

plume, which are probably the most direct observations possible. In this 33 study, the potential benefit of these kind of observations in an assimilation 34 system is studied. The experiments consist of so-called twin-experiments, 35 where observations are simulated from model simulations and fed into an 36 assimilation system using the same model. In this setup it is a first step 37 towards assimilation of real observations, to obtain a first idea on how to use 38 this kind of observations and what their impact is in an assimilation system. 39 These aircraft-based data has some advantages compared with satellite 40 data: (1) The aircraft measurement is frequently obtained from the optical 41 particle counters which are equipped on the aircraft, thus the particle con-42 centration observation is real-time and directly detected and it has a higher 43 accuracy. With an error estimate of about 10 percent, which can be achieved 44 by well calibrated instruments (Weber et al., 2012), the accuracy of these 45 observations is high compared to for example satellite data, for which errors 46 50-60 percent are reported (Zehner, 2010). (2) The aircraft measurement is 47 in-situ which is suitable to be compared directly to a 3-dimensional model 48 state, whereas some other measurements such as satellite data and LIDAR 40 data observe optical properties being accumulated into a single value per 50 vertical column which cannot be compared directly to a 3D model state. (3) 51 An aircraft can decide the route in the sky to follow the ash cloud to always 52 get an appropriate ash concentration. And it can also decide to fly at differ-53 ent altitudes, e.g., if we mainly care about the intercontinental commercial 54 aircraft safety, we can choose to fly at 9 km with a suitable research aircraft 55 (which is commonly the lowest height level for intercontinental commercial 56 planes) to perform measurements. Note that most national and maybe some 57

⁵⁸ continental passenger flights are below 9 km altitude, while intercontinental
⁵⁹ flights are at 9 km and higher altitude.

Currently aircraft-based measurements are only used for validation pur-60 pose of volcanic ash clouds (Weber et al., 2012), not yet involved in data 61 assimilation systems. This paper will study the use of aircraft-based mea-62 surements in data assimilation with an Ensemble Kalman filter (EnKF) algo-63 rithm. This study aims at (1) investigating the performance of aircraft-based 64 measurements in data assimilation systems; (2) study the impact of measure-65 ments from different flight altitudes on forecasts at aviation level; (3) discuss 66 the influence of uncertainties in the ESPs and measurements. 67

This paper is organized as follows. Section 2 gives a brief introduction 68 of volcanic eruption models and introduces the LOTOS-EUROS model used 69 in this study. The validation of LOTOS-EUROS as a VATDM is specified 70 in Section 3. Section 4 gives an introduction of the aircraft-based measure-71 ments used in our assimilation experiments. Sequential data assimilation 72 methodology including the stochastic environment and the ensemble-based 73 filter algorithm is presented in Section 5. Section 6 contains the assimilation 74 results and the discussion on the results. Finally, the last section summarizes 75 the concluding remarks of our research. 76

77 2. Volcanic eruption models and the LOTOS-EUROS model as 78 VATDM

⁷⁹ Numerous volcanic ash transport and dispersion models are available
⁸⁰ worldwide, and in recent efforts a comparison report among these existing
⁸¹ models has been compiled (Bonadonna et al., 2012, 2014). These models

are usually off-line coupled to a meteorological model, which require that 82 numerical weather prediction data should be generated first such that the 83 VATD model could use these data. For the transport either Lagrangian or 84 Eulerian approach are used. Some models run quickly such as PUFF (Searcy 85 et al., 1998) can run within minutes and others require many hours to run, 86 such as ATHAM (Oberhuber et al., 1998) requires several days. Several 87 VATD models are used in operational settings, like NAME (Jones et al., 88 2007) and HYSPLIT (Draxler and Hess, 1998) and therefore are designed 89 to produce volcanic ash simulations quickly for the corresponding volcanic 90 ash advisory center. Inter-comparisons between volcanic ash transport and 91 dispersion models, volcanic ash real-time advisories as well as to the satellite 92 observations have been reported by multiple authors, such as (Witham et al., 93 2007) and (Webley et al., 2009b). 94

In our study, the LOTOS-EUROS model is used (Schaap et al., 2008) 95 with model version 1.10. This model is an operational air-quality model, 96 used for daily air quality forecasts over Europe (Curier et al., 2012), fo-97 cussing on ozone, nitrogen oxides, and particular matter. In addition, it 98 could be configured to simulate transport of tracers in other regions of the 99 world. The model uses the off-line approach and is driven by meteorological 100 data produced by European Centre for Medium-Range Weather Forecasts 101 (ECMWF). The model is used in a tracer mode to produce volcanic ash 102 simulations in a timely and useful manner for forecasting. 103

To describe a volcanic eruption in LOTOS-EUROS model, Eruption Source Parameters (ESP) such as Plume Height (PH), Mass Eruption Rate (MER), Particle Size Distribution (PSD) and Vertical Mass Distribution (VMD) are

6

needed. In (Mastin et al., 2009) ESPs for different volcanoes are provided 107 as a look up table. LOTOS-EUROS through the add-on initial plume mod-108 ule uses the ESP type data as volcanic emission information for the model 109 forecasting. The LOTOS-EUROS model with volcanic ash configuration has 110 been used to simulate the April 2010 period of activity from Eyjafjallajökull. 111 The input parameter PH in LOTOS-EUROS is referred from hourly based 112 Icelandic Meteorological Office (IMO) plume height detection (see Figure 2) 113 and usually the uncertainty of PH is taken as 20 % (Bonadonna and Costa, 114 2013). 115

For VMD, large explosive volcanic plumes have a typical 'umbrella' shaped vertical distribution (Sparks et al., 1997) and as such this 'umbrella' shaped VMD is adapted into LOTOS-EUROS in this paper, see Figure 3.

The PSD in LOTOS-EUROS is defined in the ESP type S2 as defined by Mastin et al. (2009), in which the mass fraction of erupted debris finer than $63 \ \mu m$ is 0.4. For the S2 type eruption, Durant and Rose (2009) provides the base for the PSD from their analysis of the 1992 Crater Peak, Mount Spurr event. Hence, Table 1 provides the ash distribution based on their analysis and is used by LOTOS-EUROS for its 6 volcanic ash bins.

Another input parameter MER is very hard to measure for an explosive onsetting volcano. Usually it is calculated from plume height. Mastin et al. (2009) did some studies on the parameter relationship and concluded that an empirical relationship between plume height H(km) and eruption rate is

$$H = 2.00V^{0.241},\tag{1}$$

¹²⁹ in which the MER is converted to volumetric flow rate $V(m^3/s)$. For the ¹³⁰ S2 type eruption, the relationship between volumetric flow rate and MER $_{131}$ (kg/s) is (Mastin et al., 2009):

$$\frac{V}{MER} = \frac{1.5e^3}{4.0e^6}.$$
 (2)

Through PH, Eq. (1) and Eq. (2), MER can be approximately calculated,
see Table 2 where PH is specified as that used by Webley et al. (2012) for the
WRF-Chem model. Mastin et al. (2009) estimated the uncertainty of MER
through this calculation is about 50%.



Figure 2: Icelandic Meteorological Office [IMO] plume height detection from April to May, 2010. Courtesy from IMO on-line database.

For the study of Eyjafjallajökull events the model is configured on a 136 domain from 45° to 70° North and 30° West to 15° East covering Iceland 137 and North Europe (Figure 4). The grid resolution is 0.5° longitude $\times 0.25^{\circ}$ 138 latitude, approximately 25×25 km. In the vertical the model version used 139 has 12 vertical layers. On top of a surface layer of 25 m, three dynamic layers 140 are present, where the lowest dynamic layer represents the variable mixing 141 layer with the height obtained from the meteorological input, and the upper 142 two dynamic layers are reservoir layers with equal thickness; the top of the 143

 Table 1: Volcanic Ash Particle Size Distribution and ash bins property for LOTOS-EUROS

 model simulation.

Bins	Particle Diameter	Percent of Mass	Average Particle Size (μm)
$vash_1$	250 to 2000 $\mu {\rm m}$	29	1125.00
$vash_2$	63 to 250 $\mu {\rm m}$	31	156.50
vash_3	30 to 63 $\mu {\rm m}$	12	46.50
$vash_4$	10 to 30 $\mu {\rm m}$	18	20.00
$vash_5$	2.5 to 10 $\mu {\rm m}$	8	6.25
vash_6	0.0 to 2.5 $\mu {\rm m}$	2	1.25

Table 2: Plume height and Eruption rate in LOTOS-EUROS Model Simulation for April 14-18, 2010. Courtesy from (Webley et al., 2012).

Start Time – End Time	Height ASL (km)	Eruption Rate (kg/s)
$4/14 09{:}00 - 4/14 19{:}00$	9	5.71E + 05
4/14 19:00 - 4/15 04:00	5.5	3.87E + 04
$4/15\ 04{:}00-4/16\ 19{:}00$	6	6.44E + 04
$4/16 19{:}00 - 4/18 00{:}00$	8.25	3.65E + 05



Figure 3: The 'umbrella' vertical mass distribution (VMD) of ash cloud in LOTOS-EUROS. Shown in this case is the vertical profile of an eruption with 10 km plume height.

dynamic layers is set to 3.5 km. The remaining 6 layers have fixed altitudes
with equal thickness of 1 km, which set the top to 11.5 km in total.

The volcanic ash concentration is described by 6 aerosol tracers as men-146 tioned above. The physical processes that are relevant for volcanic ash are 147 similar as those that apply for mineral dust, e.g., advective transport and 148 diffusion, deposition, coagulation, sedimentation, and resuspension (Lang-149 mann, 2013). Where the transport is determined by the wind fields that 150 could be regarded as rather well known, the other processes deposition and 151 sedimentation processes are rather uncertain. The parameterizations for the 152 later processes involve assumptions on the particle shape for example, which 153 is difficult to summarize in a few numbers. These processes act on the dis-154 tribution of the total ash mass over the modes (particle sizes) and the total 155 mass load; one could therefore state that almost everything in the descrip-156 tion of an ash cloud is uncertain, except for its shape and position. The 157 processes included in this study are transport, sedimentation, and wet- and 158

dry-deposition, where the relevant properties such as average particle size (Table 1) are implemented following Zhang (2001). Processes that are missing yet are for example coagulation, evaporation, and resuspension, which might be considered in future when appropriate observations are available to constrain them, for example sedimentation amounts.

¹⁶⁴ 3. Validation for Eyjafjallajökull volcanic ash simulation

Based on the input parameters settings described above, validation with 165 the LOTOS-EUROS model has been made. For the validation experiment, 166 the time period of April 14-18, 2010 is chosen. Figure 4 shows examples 167 of ash plumes simulated by LOTOS-EUROS as well as two other models 168 at two time snapshot 00:00 (UTC) April 15 and 00:00 (UTC) on April 17, 169 2010. The WRF-Chem results are taken from (Webley et al., 2012) where 170 the WRF-Chem model has been validated as a proper VATDM. For both 171 LOTOS-EUROS and WRF-Chem the figures represent ash mass loadings 172 at selected time, thus the total mass measures over all aerosol modes per 173 area. Figure 4(e) and Figure 4(f) show simulations provided by the Volcanic 174 Ash Advisory Center (VAAC) based on the NAME model. VAAC are set 175 up by the International Civil Aviation Organization (ICAO), to provide in-176 formation to the aviation community through timely volcanic ash advisories 177 (VAA). For the NAME model the figures show ash cloud locations at specific 178 altitude bounds and ash cloud boundaries in the figures are corresponding 179 to 200 $\mu g m^{-3}$ which is a very low value by today's standard for aircraft 180 operations (Zehner, 2010). Comparison of NAME model to LOTOS-EUROS 181 and WRF-Chem is made by comparing superposition of ash cloud locations 182

¹⁸³ over all the altitude bounds to the boundaries of ash mass loadings.

Table 3: Comparison of Total Mass in KT (10^6 kg) between the LOTOS-EUROS model and the WRF-Chem model simulation of eruption of Eyjafjallajökull volcano in 2010.

Time	LOTOS-EUROS	WRF-Chem
00:00 (UTC) 15 April	11315.45	10648.4
00:00 (UTC) 17 April	5738.63	6729.2

The LOTOS-EUROS simulations showed that on April 15, 2010 at 00:00 184 (UTC), (Figure 4(a)) wind patterns advected the modeled ash cloud in South-185 East direction toward continental North-West Europe. This closely matches 186 the WRF-Chem simulation also from 00:00 (UTC) April 15 (Figure 4(c)). 187 This is South-East advection of the ash cloud during April 15, 2010 and until 188 April 17, 2010 at 00:00 (UTC) (Figure 4(b)), when the modeled ash cloud 189 is advected toward continental Europe. The LOTOS-EUROS model simu-190 lation also shows a good match to the VAAC volcanic ash advisory (VAA) 191 generated operationally at the time of the eruption. The VAA's (Figure 4(e)192 and 4(f) showed ash being forecasted across continental Europe as well as 193 from surface to FL200 (approximately 6 km above sea level as flight levels 194 are on a pressure based coordinate system) ash across the United Kingdom 195 (Figure 4(e)). These similarities among different model simulations are from 196 the similar ESP definitions and also possibly from the same or similar wind 197 fields driving data. Table 3 is the comparison of the total mass calculated 198 from LOTOS-EUROS model and WRF-Chem model. Although the values 190 from two models are not same, they are of same magnitude and not different 200



Figure 4: Volcanic ash simulations with different models (a)(b) LOTOS-EUROS, (c)(d) WRF-Chem and (e)(f) NAME at two time snapshot (a)(c)(e) 00:00 (UTC) April 15 and (b)(d)(f) 00:00 (UTC) on April 17, 2010.

too much, which means the LOTOS-EUROS model can produce reasonable simulation results as WRF-Chem does. Note that the differences might
be caused by the difference in simulation of two models such as advection
scheme, deposition scheme, etc.

Figure 5 is the decomposition of Figure 4(b) with different ash bins de-205 fined in Table 1. From Figure 5, we can see the coarse bins vash_1 and 206 vash_2 only remain in the plume near the source, which is because of pro-207 cesses of sedimentation and deposition. In contrast, fine ash bins from vash_3 208 to vash 6 (particles finer than 63 μm) are transported along the plume to 209 continental Europe. This result fits the fact that after several days in conti-210 nental Europe only finer ash bins were detected (Webley et al., 2012), so that 211 LOTOS-EUROS simulation can be considered as realistic. It shows to us that 212 along the plume only these finer ash bins are those which can be measured. 213 Comparison of the total mass from the LOTOS-EUROS model with the full 214 extent of all three levels in the VAA and the total mass from WRF-Chem, 215 show that the LOTOS-EUROS model matches well to the NAME model and 216 WRF-Chem model simulations. 217

As shown above, LOTOS-EUROS model is capable of modeling volcanic 218 ash transport problem. Table 3 implies that different VATDMs will provide 219 different forecast values because different models have different details, there-220 fore only relying on VATDM to make forecast is not sufficient, that is also 221 one of the motivations for using assimilation to correct VATDM to improve 222 the forecast accuracy. In the following, a data assimilation technique will be 223 introduced and used to combine LOTOS-EUROS model and measurement 224 information to improve the ash transport forecast accuracy. 225



Figure 5: LOTOS-EUROS simulation with different bins. (a) vash_1, (b) vash_2, (c) vash_3, (d) vash_4, (e) vash_5 and (f) vash_6.

226 4. Aircraft-based Measurements

227 4.1. Measurements description

During the period of eruption of the Icelandic volcano Eyjafjallajökull, a 228 large number of different scientific measurement campaigns were performed 229 to gather information about the nature and occurrence of the ash plume. The 230 measurements comprised for example LIDAR measurements (Pappalardo 231 et al., 2010; Tesche et al., 2010; Groß et al., 2010; Miffre et al., 2010; Flentje 232 et al., 2010), satellite observations (Stohl et al., 2011), groundbased in-situ 233 measurements (Schäfer et al., 2010; Emeis et al., 2011), as well as balloon 234 (Flentje et al., 2010) and aircraft based measurements (Weber et al., 2010; 235 Schumann et al., 2011; Bukowiecki et al., 2011; Eliasson et al., 2011; Lolli 236 et al., 2010). Aircraft-based measurements are of special interest, because 237 they allow sampling of the ash plume with a high temporal and spatial res-238

olution. Another beneficial aspect of aircraft measurements is that they are more flexible than other type measurements, because usually the aircraft is operated as a "state aircraft" enabling operations in otherwise closed air space where the pilots are able to change the flight plan in-flight in direct contact with air traffic control or wisely decide to follow the downwind ash trend to obtain the best observations.

During the period of that eruption, the outskirts of the eruption plume 245 were entered directly by research flights (Figure 6(a)), delivering most direct 246 measurements within the eruption plume during this eruptive event. All of 247 the measurement flights were equipped with optical particle counters for in-248 situ measurements. Real-time monitoring of the particle concentrations was 249 possible during the flights and In-situ measurements from the eruption plume 250 were obtained with high time- and spatial-resolution. It has been proven that 251 by entering the outskirts of the plume directly the research aircraft can detect 252 ash concentrations of up to 2000 $\mu g m^{-3}$ (Weber et al., 2012). That used to be 253 considered as the highest concentration an aircraft can endure at that time. 254 because areas with ash concentration higher than 2000 $\mu g m^{-3}$ were classified 255 as No Fly Zone (NFZ) (Zehner, 2010), which means the aircraft flying in these 256 areas can crash. However, recently 2000-4000 $\mu g m^{-3}$ is classified as medium 257 level concentration (EASA, 2011). Many airlines are certified to operate in 258 this regime based on the application of Safety Risk Assessment. Therefore 259 now the highest concentration an aircraft can endure is updated to be 4000 260 $\mu g m^{-3}$ instead of 2000 $\mu g m^{-3}$. 261

Optical particle counters (OPC) were used for in-situ ash concentration measurements. The principle of OPC's can be summarized as follows: Ash



Figure 6: (a) Example of aircrafts used for volcanic ash measurements and (b) optical particle counter OPC equipped on aircrafts. Courtecy from (Weber et al., 2012) and (Weber et al., 2010), respectively.

contaminated air is pumped through the OPC where the particles cross a 264 continuous laser beam. Every single particle causes a scattering/diffraction 265 of the laser beam. This is recorded by a detector that counts the particles, 266 see Figure 6(b). Moreover, scattering/diffraction intensity of the laser beam 267 is a measure for the size of the particles. From that, the mass can be calcu-268 lated, provided the density of the particles is known. A mean mass density 269 of 2.65 qcm^{-3} (Heim et al., 2008) for the coarse mode ash particles is rec-270 ommended to use by European Facility for Airborne Research (EUFAR) for 271 Evjafjallajökull volcano ash. 272

In our study, the most interesting thing is how accurate OPC's measurements are, because the knowledge of uncertainties is crucial for a successful data assimilation. Through a direct laboratory calibration experiment, in which the mass concentration obtained with the OPC was compared with the absolute mass concentration gathered on a gravimetric filter, the deviation between the gravimetric measurement and the OPC was about 10%

(Weber et al., 2010) which can be taken as the instrumental error for this type of measurements in well calibrated cases.

281 4.2. Model representation error

For assimilation of measurements with a simulation model, it is necessary 282 to quantify the model representation error. The model representation error 283 is the difference between the quantity that instrument tries to observe, and 284 what the model could represent in terms of its state. This does not include 285 instrumental errors as defined above, or model deficiencies such as inaccurate 286 input parameters, but only the difference due the model being defined on a 287 discrete grid with finite resolution and simulations valid for discrete time 288 steps. 289

As shown in Section 2, the spatial resolution of the model used in this 290 paper is around 25 km \times 25 km \times 1 km, therefore the volume of one grid-291 box is about 625 km^3 . Through model processing, the concentration of one 292 grid-box represents an average value for this grid-box, while one aircraft-293 based measurement is a sample (a point value) in a 3 dimensional field. In 294 this paper, we choose the in-situ measurement corresponding to the grid-box 295 average value. This approximation makes sense only when two assimilated 296 measurements are positioned in two different grid-boxes. This requires that 297 the assimilation frequency is not too high, so that the measurements used in 298 two sequential assimilation steps are in different grid-boxes. Moreover, the 299 assimilation frequency should also not be too low because a measuring air-300 craft usually can work in the sky for less than 10 hours continuously (Weber 301 et al., 2012; Schumann et al., 2011). If an hourly assimilation frequency is 302 chosen, then along the whole route, only less than 10 measurements will be 303

used, which is a waste for other continuous measurements. Therefore based 304 on the analysis above and also considering the aircraft speed of 100-200 km/h 305 (Weber et al., 2012) and the LOTOS-EUROS horizontal resolution, a 15 min-306 utes assimilation frequency is chosen in this study. Within 15 minutes, the 307 aircraft can fly over about 2 grid-boxes in the model, which guarantees dif-308 ferent assimilated measurements are in different grid-boxes. The model time 309 step cannot be over the assimilation time step, hence in this paper, 15 min-310 utes is also chosen to be one model time step, without loss of generality. Note 311 that if the research aircraft is faster or the horizontal resolution is higher, 312 the assimilation frequency can be chosen smaller than 15 minutes (e.g., 10 313 or 5 minutes which can be considered sufficient). 314

Through the settings defined above, the observation almost corresponds to one model state in a grid-box, which means the representation error of the model is probably small. For the moment we will there not explicitly specify a model representation error, but implicitly assume that it is zero. Therefore, the total observation representation error, defined as the sum of the instrumental error and the model representation error, is taken as 10% in this paper.

Since the knowledge about the uncertainties and representation errors of aircraft measurements are known, data assimilation can now be used to combine observations with the model to get an improved estimate of the ash load.

³²⁶ 5. Sequential data assimilation methodology

327 5.1. Stochastic state space representation

For application of the filter algorithm to the LOTOS-EUROS model, a 328 stochastic representation must be defined for the model error. The specifi-320 cation of uncertainties is crucial for a successful data assimilation. Using a 330 stochastic model for several uncertain parameters, an assimilation scheme is 331 able to produce an optimal estimate of the state and parameters given the 332 observations. For application of the filter algorithms to a dynamical model, 333 a stochastic representation should be written in a state-space form according 334 to: 335

$$x(k) = M_{k-1}(x(k-1)) + w(k-1), \quad w(k-1) \sim N(0, Q(k-1))$$
(3)

The state-space operator M_{k-1} describes the time evolution from the time k-1 to k of the state vector x. In this paper, x contains the ash concentrations in the model grid boxes for the 6 size modes as described in Section 2. The random forcing term w is drawn from the normal distribution with zero mean and covariance matrix Q. The definitions of w and Q will be different per experiment, and are discussed in detail in the coming sections.

342 5.2. Observational operator

The state of the observational network is defined by the observation operator H that maps state vector x to observation space y:

$$y(k) = H_k(x(k)) + v(k), \quad v(k) \sim N(0, R),$$
(4)

where the observation representation error v is drawn from Gaussian distribution with mean 0 and covariance matrix R. This error accounts for the instrumental error as well as for the model representation error (Section 4.2). With the later assumed to be zero, the value of v is solely the instrumental error, which has been estimated to be 10% of the measured values y.

Here, y contains aircraft in-situ concentration, the states are updated ev-350 ery time step of 15 minutes (see Section 4.2) given the new instantaneous 351 concentration. The operator H then simply selects the grid cell in x that 352 corresponds to the observation location. In this paper, for the purpose of 353 investigating the performance of aircraft-based measurements in data assim-354 ilation system, only one aircraft is considered to provide measurements. In 355 this scenario, at a fixed time only one measurement is obtained and the ob-356 servation location keeps changing with the time because aircraft is movable. 357

358 5.3. Ensemble Kalman Filter

The assimilation technique used in this study is an Ensemble Kalman 359 Filter technique (EnKF). Apart from the original formulation in (Evensen, 360 1994), other formulations have been introduced such as the Ensemble Kalman 361 Smoother (EnKS) (Evensen and van Leeuwen, 2000), Ensemble Square Root 362 Filter (EnSR) (Evensen, 2004), Reduced Rank Square Root Filter (RRSQRT) 363 (Verlaan and Heemink, 1997), etc. Ensemble-based assimilation is easy to im-364 plement, suitable for real-time estimation of concentrations and allows a very 365 general statistical description as Eq. (3). Different methods have different 366 advantages and disadvantages. This paper aims to compare the performance 367 of aircraft-based measurements in data assimilation systems, not to compare 368 performance of different data assimilation schemes. Therefore, in this paper, 369 we choose the commonly used method EnKF in our data assimilation system. 370 The EnKF essentially is a Monte Carlo sequential method (Evensen, 371

2003), based on the representation of the probability density of the state estimate in an ensemble of N states, $\xi_1, \xi_2, \dots, \xi_N$. Each ensemble member is assumed to be a single sample out of a distribution of the true state.

In the first step of this algorithm an ensemble of N states $\xi^{a}(0)$ is generated to represent the uncertainty in x(0). In the second step, the *forecast* step, the stochastic model propagates the ensemble members from the time k - 1 to k:

$$\xi_j^f(k) = \tilde{M}_{k-1}(\xi_j^a(k-1)) + Gw_j(k-1),$$
(5)

when w_j represents the realizations of a white noise process w. The filter state is a stochastic distribution with mean x^f and covariance P^f following:

$$x^{f}(k) = \frac{1}{N} \sum_{j=1}^{N} \xi_{j}^{f}(k).$$
(6)

$$L^{f}(k) = [\xi_{1}^{f}(k) - x^{f}(k), \cdots, \xi_{q}^{f}(k) - x^{f}(k)],$$
(7)

$$P^{f}(k) = \frac{1}{N-1} L^{f}(k) L^{f}(k)'.$$
(8)

When measurements become available, the ensemble members are updated in the *analysis* step using the Kalman gain:

$$K(k) = P^{f}(k)H(k)'[H(k)P^{f}(k)H(k)' + R]^{-1},$$
(9)

383

$$\xi_j^a(k) = \xi_j^f(k) + K(k)[y(k) - H(k)\xi_j^f(k) + v_j(k)],$$
(10)

where v_j represents realizations of the observation representation error v. Advantages of the ensemble formulation is that the dynamical model is not restricted to linearity and the implementation can be rather simple. The number of required ensemble members depends on the complexity of the probability density function (eps) to be captured, which is usually determined by the nonlinearity of the model and the description of the involved uncertainties. In general, an ensemble with 30–100 members is acceptable to keep computations feasible (Barbu et al., 2009).

³⁹² 6. Assimilation results and discussions

393 6.1. Experimental setup

The EnKF is applied to the stochastic version of LOTOS-EUROS. The 394 study comprises experiments with different settings for the uncertainty in 395 parameters such as plume height (PH), mass eruption rate (MER), particle 396 size distribution (PSD) and vertical mass distribution (VMD). In this paper, 397 the uncertainty of PH and MER are taken as 20 % and 50 %, respectively 398 (see Section 2). The stochastic version of the model is built by considering 399 these two uncertain parameters. The temporal correlation for a uncertain 400 model parameter defines how the value at current time is related to that at 401 prior time. However, due to volcano inner fierce and fast physical processes, 402 the PH and MER could change very fast, and therefore taking temporal 403 correlation into account is not necessary and realistic. Therefore, in this 404 paper, we consider PH and MER as temporal uncorrelated. Aircraft-based 405 measurements are used in the analysis step of the EnKF algorithm; the 406 uncertainty in the measurements has been investigated as a fixed standard 407 deviation of 10 %, see Section 4. This paper focuses on studying how aircraft-408 based measurement performs well in a data assimilation system, thus it is 400 not necessary to use real measurements. Therefore, the measurements in 410



Figure 7: (a) Aircraft-based Measurements for vash_5 and vash_6. (b) is the designed route at 9 km where the measuring aircraft enters the outskirts of ash plume, red and magenta lines represent different flying directions to Reykjavik airport.

this paper are designed based on the real aircraft-based measuring campaigns (Weber et al., 2012). In these campaigns, concentrations of ash with diameter 0 to 2.5 and 2.5 to 10 μ m were observed, which from Table 1 respectively corresponds to vash_5 and vash_6 in this paper.

From (Weber et al., 2012; Schumann et al., 2011), a measuring aircraft 415 can work in the sky for less than 10 hours continuously, so based on this 416 condition, a 10 hour aircraft measurement experiment is designed. Note 417 that in reality usually an aircraft measurement mission is 4 to 6 hours, 10 418 hours is not very realistic, but we use the duration of 10 hours in our twin 419 experiments to evaluate effect of assimilation over a longer time. The height 420 of interest in this paper is 9 km as mentioned in Section 1. The flight routes 421 are chosen at an altitude of 9 km, 7 km, 5 km, 3 km separately to study 422 which level measurements provide the best air traffic advisory for 9 km. The 423 start time of Evjafjallajökull eruption is set at 9:00 (UTC), 14 April 2010, 424 and the aircraft-based measurements are designed to start at 11:00 (UTC), 425 14 April 2010. The whole assimilation time is from 11:00 to 19:00 (UTC), 14 426 April 2010. The flying route is designed as shown in Figure 7(c) based on the 427 fact that the measuring aircraft can enter the plume outskirt where the ash 428 concentration is less than 4000 $\mu g m^{-3}$ (see Section 4.1). During this period, 429 measurements are taken every 15 minutes, see Figure 7(a) and Figure 7(b) 430 taken at 9 km for example. 431

From the start time to aircraft returning time, the simulation parameters are set as introduced in Section 2. For evaluating the performance of assimilation, twin experiments are designed with the Truth obtained as one realization of the stochastic model by adding uncertainty 20 %, 50 % to PH,

- ⁴³⁶ MER, separately. The measurements are obtained through combining the
- ⁴³⁷ Truth values with 10 % uncertainty, see Figure 7(a) and Figure 7(b).

Table 4: Plume height and Eruption rate used in LOTOS-EUROS to generate the Truth for April 14-18, 2010.

Start Time – End Time	Height ASL (km)	Eruption Rate (kg/s)
$4/14 \ 09:00 - 4/14 \ 11:00$	8.8	5.23E + 05
4/14 11:00 - 4/14 13:00	9.3	5.85E + 05
$4/14 13{:}00 - 4/14 15{:}00$	7.8	3.98E + 05
4/14 15:00 - 4/14 17:00	9.1	5.38E + 05
$4/14 17{:}00 - 4/14 19{:}00$	8.5	4.41E + 05
$4/14 \ 19:00 - 4/14 \ 22:00$	6.3	5.73E + 04
$4/14 \ 22:00 - 4/15 \ 01:00$	4.8	3.13E + 04
$4/15 01{:}00 - 4/15 04{:}00$	5.9	4.97E + 04
$4/15 \ 04{:}00 - 4/15 \ 17{:}00$	5.0	5.07E + 04
$4/15 17{:}00 - 4/16 06{:}00$	7.1	8.32E + 04
$4/16 06{:}00 - 4/16 19{:}00$	6.8	8.15E + 04
$4/16 \ 19:00 - 4/17 \ 01:00$	9.2	5.10E + 05
4/17 01:00 - 4/17 07:00	8.0	3.12E + 05
$4/17 \ 07{:}00 - 4/17 \ 13{:}00$	9.4	3.89E + 05
4/17 13:00 - 4/17 19:00	7.9	2.97E + 05
4/17 19:00 - 4/18 01:00	8.5	3.93E + 05

The experiment procedure can be briefly summarized by stating that the model run starts at 09:00 (UTC), 14 April, 2010 by considering the first ini-

tial condition as zero. With the model propagating, the model result from 440 previous time step is taken as the initial condition for the next time step. 441 When the model run arrives at 11:00 (UTC), 14 April, the initial condition 442 gets continuously modified by the data assimilation process through combin-443 ing all the aircraft-based measurements until the time 19:00 (UTC), 14 April. 444 Thus at this time, an analyzed state (which can be taken as an initial condi-445 tion for next model run) combining all aircraft measurements of 8 hours can 44F be obtained. Evaluation on this analyzed state will be given to invastigate 447 the possible improvement compared to simulation without assimilation. In 448 the remainder of the paper, AnaSta and SimSta are used to denote the an-449 alyzed state (obtained with assimilation) and the simulation state (obtained 450 without assimilation), respectively. Thus, AnaSta_{19:00(14)} denotes AnaSta at 451 19:00 (UTC), 14 April, 2010 and will be further used to forecast over multiple 452 days (typically one or two days according to NAME model forecast). 453

454 6.2. Assimilation experiments

Based on the setup above, an experiment is designed to test whether the analyzed state AnaSta_{19:00(14)} is improved through an 8 hours continuous assimilation of aircraft-based measurements. For this experiment, the measuring aircraft flies at the 9 km height and the ensemble size is chosen to be 50 in the EnKF system. Before we show the result of AnaSta_{19:00(14)}, first how data assimilation continuously works in this system is explained using Figure 8 and Figure 9.

Figure 8 is the result of specific measurements at one location on 12:00 (UTC), 14 April, 2010. At this time, we can see from Figure 7, the measuring aircraft location is (11.75°W, 65.625°N, 9 km). In Figure 8, the forecast of

the ash concentrations at this location in each of the 50 ensemble members 465 shown. The concentrations are distributed around the mean values (96.38 466 $\mu g m^{-3}$ for vash_5, 24.80 $\mu g m^{-3}$ for vash_6) indicated by the black circle. At 467 this time, the measurements of the concentrations are (126.69 μ g m⁻³, 32.93 468 $\mu g m^{-3}$) which is significantly different from the forecast mean. Through 469 assimilating these measurements at this time, analysis values of vash₋5 and 470 vash_6 are obtained as (123.61 $\mu g m^{-3}$, 32.54 $\mu g m^{-3}$) which are much closer 471 to the truth (135.88 $\mu g m^{-3}$, 33.88 $\mu g m^{-3}$) than the forecast mean. This 472 result illustrates that the assimilated state better approximates the truth 473 than that without assimilation. Moreover, spread in the analysis ensemble is 474 smaller than that of the forecast ensemble, that means the error variance of 475 analysis value is reduced through assimilation. 476



Figure 8: Assimilation at one location.



we cannot see the influence of the assimilation on the whole plume. It is 478 not clear whether measurements from one location can influence the whole 479 plume or not. Figure 9 is used to answer this question. In Figure 9, without 480 loss of generality the time 16:30 (UTC), 14 April, 2010 is chosen to show the 481 result for the whole plume at 9 km. $Tru_{16:30(14)}$ in Figure 9 is the truth state, 482 while $FC_{16:30(14)}$ is the forecast state and $AnaSta_{16:30(14)}$ is the analyzed state. 483 By comparing $FC_{16:30(14)}$ with AnaSta_{16:30(14)}, we can see that with assimi-484 lating aircraft-based measurements at one location, the difference between 485 them only appears in a local area (approximatly the red ellipse in Figure 486 9) around the measuring location, while the results outside this local area 487 are hardly changed. This means that the assimilation process doesn't influ-488 ence the entire plume, but only a local area around the measuring aircraft 489 location. Note that this is achieved without explicit enforcing of localization 490 as for example in (Houtekamer and Mitchell, 1998). In the chosen setup 491 without temporal correlation (see Section 6.1), the ensembles consist of pat-492 terns that arise from uncertainties during a single time step. The spatial 493 impact of observations from a single location is therefore bounded to an area 494 where ashes present have been emitted during a short period. Moreover, in 495 this local area, AnaSta_{16:30(14)} is much closer to $Tru_{16:30(14)}$ than FC_{16:30(14)}. 496 This shows that through each assimilation, the state within a local area is 497 improved. Therefore with a continuously assimilation using aircraft-based 498 measurements of changing locations, the states in a large area around the 499 measuring flight route will be improved, as shown in Figure 7(c). 500

Next, the experiment result of AnaSta_{19:00(14)} is shown in Figure 10. Tru_{19:00(14)} is the Truth at 19:00 (UTC), 14 April, 2010 which is imple-



Figure 9: Assimilation results during continuously assimilation (red ellipse represents a local area where the assimilation can influence). (a) $Tru_{16:30(14)}$, (b) $FC_{16:30(14)}$ and (c) $AnaSta_{16:30(14)}$.

mented based on Table 4. $SimSta_{19:00(14)}$ is the simulation result directly 503 implemented based on input parameters specified in Table 2 without assimi-504 lating aircraft-based measurements. Ana $Sta_{19:00(14)}$ is the assimilation result 505 at this time with assimilating aircraft-based measurements (detailed settings 506 are in Section 6.1). Big differences can be observed between Figure 10(a)507 and Figure 10(b). The difference is caused by implementing with different 508 PH and MER. In reality, $Tru_{19:00(14)}$ is unknown, thus $SimSta_{19:00(14)}$ is used 509 as the initial condition for the forecast over multiple days. AnaSta_{19:00(14)} is 510 that with continuously assimilating aircraft-based measurements. Compar-511 ing AnaSta_{19:00(14)} and SimSta_{19:00(14)}, we can see both of them overestimate 512 the truth, but $AnaSta_{19:00(14)}$ is much closer to $Tru_{19:00(14)}$ and the over-513 estimation is much lower than $SimSta_{19:00(14)}$. This means the state after 514 assimilating aircraft-based measurements is much more accurate than that 515 without assimilation. 516

Now we have verified that through continuously assimilating aircraft-517 based measurements, an improved state is obtained. There are two main 518 reasons that explain why it performs very well: (a) The measuring aircraft 519 always follows the ash flowing trend and enters the plume outskirt to mea-520 sure concentration. This movable aircraft-based measuring path makes the 521 measurements always informative and useful for data assimilation; (b) the 522 uncertainty knowledge of PH, MER and the measurements is known, which 523 is important for EnKF to generate proper ensembles. 524

AnaSta_{19:00(14)} can be used as an initial condition to do forecast over multiple days to see the possible improvement in advisories to aviation. Without loss of generality, the forecast at 00:00 (UTC) 15 April is chosen as illustra-



Figure 10: Comparison of results with and without assimilating aircraft-based measurements. (a) $Tru_{19:00(14)}$, (b) $SimSta_{19:00(14)}$ and (c) $AnaSta_{19:00(14)}$.

tion in Figure 11. $FC^n_{00:00(15)}$ and $FC^a_{00:00(15)}$ are used to represent the forecast initiated with SimSta_{19:00(14)} and the forecast with AnaSta_{19:00(14)} as an initial condition.



Figure 11: Comparison of Volcanic Ash Forecast with assimilating aircraft-based measurements and without assimilation. (a) $\text{Tru}_{00:00(15)}$, (b) $\text{FC}^{n}_{00:00(15)}$ and (c) $\text{FC}^{a}_{00:00(15)}$.

In Figure 11, at time 00:00 (UTC) 15 April, 2010, we can see $FC^a_{00:00(15)}$ better approximates $Tru_{00:00(15)}$ than $FC^n_{00:00(15)}$. The result shows the forecast accuracy is improved through assimilating aircraft-based measurements. Note that in Figure 11(a), the plume does not appear south of Iceland, it means in this area the ash plume is below the altitude of 9 km. In real life,

the truth is unknown, thus usually Figure 11(b) is used to provide advice 536 to decision makers. In this experiment, if we use $FC_{00:00(15)}^n$ for the advice, 537 then it will be that at 9 km ash concentrations in West-North areas outside 538 Norway are higher than 6000 $\mu g m^{-3}$. Whereas in fact this advice is inaccu-539 rate compared to the truth which shows in these areas ash concentrations are 540 lower than 4000 $\mu g m^{-3}$. This clearly shows only using simulation to provide 541 advice is not sufficient for decision makers. Figure 11(c) is the assimilation 542 forecast combining 8 hours continuous aircraft real-time measurements (Fig-543 ure 7). The only difference between Figure 11(c) and Figure 11(b) is that 544 Figure 11(c) assimilates aircraft-based measurements. From Figure 11(c), 545 we can get an accurate advice with a much closer to truth estimate at 00:00 546 (UTC) 15 April, where ash concentrations in all the areas at 9 km are lower 547 than 4000 μg m⁻³. This is a big improvement compared to Figure 11(b). 548

In this experiment, through initiation with $AnaSta_{19:00(14)}$, the forecast 549 of volcanic ash transport has been significantly improved. This tells us for 550 volcanic ash forecast, with a good state obtained from assimilating aircraft-551 based measurements, it can provide an improved advice for aviation. In the 552 following, two other experiments are designed to study (1) for the interested 553 advice at height level of 9 km, at which altitude the aircraft should fly to 554 give the best analyzed state $AnaSta_{19:00(14)}$? (2) how important is having a 555 good knowledge of uncertainties in parameters PH, MER and measurement? 556

557 6.3. Experiments with Different Flight Levels

In this experiment, measurements are simulated at different altitudes as 7 km, 5 km, 3 km, respectively. The interested level is still 9 km as in last experiments and Figure 12(a), Figure 12(b) and Figure 12(c) are the designed

aircraft measurement routes at different heights 3 km, 5 km, 7 km, respec-561 tively. The performance of assimilating these measurements compared with 562 the 9 km measurements is shown in Figure 13. We extend $AnaSta_{19:00(14)}$ to 563 AnaSta $^{3}_{19:00(14)}$, AnaSta $^{5}_{19:00(14)}$, AnaSta $^{7}_{19:00(14)}$ and AnaSta $^{9}_{19:00(14)}$ to repre-564 sent the analyzed state at 19:00 (UTC) 14 April, 2010 through assimilating 565 aircraft-based measurements from heights 3 km, 5 km, 7 km and 9 km, re-566 spectively. $Tru_{19:00(14)}$ is the truth and $SimSta_{19:00(14)}$ is the simulation result 567 without assimilation. From Figure 13(c) to Figure 13(f), we can see that all 568 cases with different altitude perform worse than the 9 km case, and some 569 of them (3 km case and 5 km case) are even worse than the case without 570 assimilation. 571

From these comparison, we can get that the locations (flight levels in this 572 paper) of aircraft-based measurements are crucial for providing a more accu-573 rate analyzed state on interested level. The best $AnaSta_{19:00(14)}$ is that assim-574 ilating aircraft-based measurements from the same flight level with interested 575 level. Furthermore, Figure 13(e) is also shown to perform an improvement 576 compared to the case without assimilation. Thus based on $AnaSta_{19:00(14)}^7$ we 577 can also obtain an improved advice where the overestimation of ash concen-578 trations has been reduced compared to $SimSta_{19:00(14)}$. 579

Through this experiment, two conclusions can be drawn that (1) in order to get the best analyzed state on the interested commercial aeroplane level with assimilation, the aircraft-based measurements should be preferably taken at the same level of height; (2) If this level measurements can not be provided, through assimilating measurements from close levels, an acceptable analyzed state can still be obtained.



Figure 12: Designed aircraft-based measurements on different flight levels (a) 3 km, (b) 5 km and (c) 7 km.

586 6.4. Uncertainties in PH, MER and measurement

This experiment is undertaken to investigate the importance of having 587 a good knowledge of uncertainties for assimilating aircraft-based measure-588 ments. In this paper, the uncertainties of PH, MER and measurements are 589 considered to be 20 %, 50 % and 10 % respectively as discussed in Section 590 6.1. However, if the uncertainty information can not be well estimated, how 591 will the assimilation perform? What are the consequences due to overesti-592 mation and underestimation of uncertainty? Which uncertainty information 593 is of most importance for ash forecast? To answer these three questions, we 594 modify the three uncertainties in the experiment one by one and evaluate 595 the performance. For evaluating influence of one uncertainty, we change its 596 uncertainty with underestimation and overestimation, separately, while, we 597 keep uncertainties of other two unchanged. The results are summarized in 598 Figure 14. 599

Figure 14(g) is the truth, Figure 14(h) is the analyzed state through assimilating aircraft-based measurements with the correct uncertainty information of PH, MER and measurement. Figure 14(a) and Figure 14(d) are the results with the wrong PH uncertainty of 10 % and 30 %, respectively.



Figure 13: AnaSta_{19:00(14)} Comparison with assimilating aircraft-based measurements taken from different flight levels. (a) $Tru_{19:00(14)}$, (b) $SimSta_{19:00(14)}$, (c) $AnaSta_{19:00(14)}^3$, (d) $AnaSta_{19:00(14)}^5$, (e) $AnaSta_{19:00(14)}^7$ and (f) $AnaSta_{19:00(14)}^9$.



Figure 14: AnaSta_{19:00(14)} implemented with different uncertainties of PH, MER and measurement. (a) AnaSta^{PH10}_{19:00(14)}, (b) AnaSta^{MER30}_{19:00(14)}, (c) AnaSta^{meas01}_{19:00(14)}, (d) AnaSta^{PH30}_{19:00(14)}, (e) AnaSta^{MER70}_{19:00(14)}, (f) AnaSta^{meas30}_{19:00(14)}, (g) Tru_{19:00(14)} and (h) AnaSta_{19:00(14)}.

Similarly, Figure 14(b) and Figure 14(e) are results with the wrong MER uncertainty of 30 %, 70 %; Figure 14(c) and Figure 14(f) are results with the wrong measurement uncertainty of 1 % and 30 % respectively. We extend a superscript in AnaSta_{19:00(14)} as PH10, PH30, MER30, MER70, meas01 and meas30 to represent these different cases, respectively.

The result in Figure 14 shows that $AnaSta_{19:00(14)}$ with the correct un-609 certainties has the best performance to approximate the truth. All the 610 other assimilation results are inferior to AnaSta_{19:00(14)}. Among them, some 611 have very strong overestimation such as $AnaSta_{19:00(14)}^{MER30}$, $AnaSta_{19:00(14)}^{MER70}$ and 612 AnaSta $_{19:00(14)}^{meas30}$. Based on these results, the answer to the first question is 613 that when the uncertainty information is not well estimated, the assimila-614 tion accuracy cannot be guaranteed. In order to get a better analyzed state 615 in data assimilation systems, the uncertainties of PH, MER and measure-616 ment should be obtained as accurate as possible, otherwise analyzed state 617 can become very unacceptable. 618

The second question is to investigate the consequences due to overesti-610 mation and underestimation of uncertainties. Figure 14(a), Figure 14(b) and 620 Figure 14(c) are forecast with underestimated uncertainty of PH, MER and 621 measurement, respectively. Whereas Figure 14(d), Figure 14(e) and Figure 622 14(f) are the cases with overestimated uncertainties. The result shows for un-623 certainties of PH or MER, overestimation can provide a better result, while 624 for uncertainties of measurement, underestimation is better. This knowl-625 edge is of practical importance because when we are not sure about some 626 parameter uncertainties in real-life assimilation, this investigation can give a 627 guidance for choosing reasonable initial uncertainties. 628

Finding the most important uncertainty is equal to investigate which one is the most sensitive to forecast accuracy. For this, we first introduce a measure for sensitivity of accuracy, which is defined as:

$$(SenAccu)_p = \frac{|(TotalMass)_p(i) - (TotalMass)_p(j)|}{|(Uncert)_p(i) - (Uncert)_p(j)|},$$
(11)

where p means one of the parameters PH, MER, measurement, i and j means 632 implementations with parameter uncertainties. Moreover, $(TotalMass)_p$ and 633 $(Uncert)_p$ represent total mass and parameter uncertainties corresponding 634 to some parameter p. Using Eq. (11), we can get sensitivities of PH, MER 635 and measurement as: $(SenAccu)_{PH} = 15.74, (SenAccu)_{MER} = 12.69$ and 636 $(SenAccu)_{measurement} = 25.52$. So the parameter sensitivity in descending 637 order is measurement, MER and PH. Thus, the most sensitive parameter 638 for assimilation is measurement whose uncertainty is the most important 639 uncertainty for achieving an accurate analyzed state. Therefore, we should be 640 very careful about defining measurement uncertainty, a slight overestimation 641 can already cause a big inaccuracy. 642

643 7. Conclusions

In this paper aircraft-based measurements have been assimilated in a sequential data assimilation system to provide volcanic ash transport forecast. The LOTOS-EUROS model has been adapted to model volcanic ash transport. The experimental results show that the LOTOS-EUROS model is capable as a reliable Volcanic Ash Transport and Diffusion Model. Our goals were to improve ash transport forecast accuracy through assimilating aircraft-based measurements, and to study the impact of measurements from different flight heights and the influence of uncertainties in PH, MER and
 measurements.

Twin experiments were carried out to evaluate the assimilation results. The results showed through assimilating aircraft-based measurements, the forecast of volcanic ash transport can be significantly improved. The accurate advice of aeroplane flying safety was made through the assimilation forecast, whereas the simulation result gave a wrong advice.

Another experiment revealed that for the interested advice level 9 km, 658 the aircraft-based measurements should be taken at this level. However, 659 when at this level measurements can not be provided, through assimilating 660 measurements from close levels, an acceptable advice can still be obtained. 661 Through comparing assimilation result of correct PH, MER and measurement 662 uncertainty with those of wrong uncertainty information, we found that as-663 similating aircraft-based measurements only can perform well with sufficient 664 knowledge of the statistics of the uncertainties. Otherwise, accurate assimi-665 lation results cannot be guaranteed. One thing needs to be mentioned that 666 from this study, we investigated measurement is the most sensitive parameter 667 for our assimilation system, which only means for assimilating aircraft-based 668 measurements, not for all the other assimilation systems such as assimilat-669 ing satellite measurements. For these assimilation systems, the sensitivity of 670 parameters still needs to be investigated. 671

In this paper, we applied an off-line approach for model running and simply used meteorological input data as deterministic. Actually these data also contain uncertainties which have an influence on ash cloud transport. In future work, for more accurate ash forecasting, it is better to take also

uncertainties of meteorological data like wind speed into account. This paper only used aircraft-based measurements from one aircraft and showed that ash forecasts were significantly improved through this one aircraft setup. we may expect that with more measuring aircrafts the results will be better than with one aircraft if all the aircrafts can enter outskirt of ash plume to get appropriate measurements.

Aircraft-based type of measurements can perform well in data assimilation system. Therefore based on this study, the real aircraft-based data experiment, which is in process, will be in a follow-up paper which also considers to increase the efficiency of the data assimilation system.

686 8. Acknowledgements

In this paper, OpenDA software (van Velzen and Verlaan, 2007), (van Velzen and Segers, 2010) was used to perform EnKF data assimilation algorithm.

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Research Highlights:

- Aircraft-based measurements are assimilated into a volcanic ash transport model.
- Assimilation can significantly improve the state of ash clouds and ash forecast.
- Measurements should be preferably taken from the same level with aeroplane flying.
- Uncertainty knowledge of ESPs and measurements is important for aviation advice.