

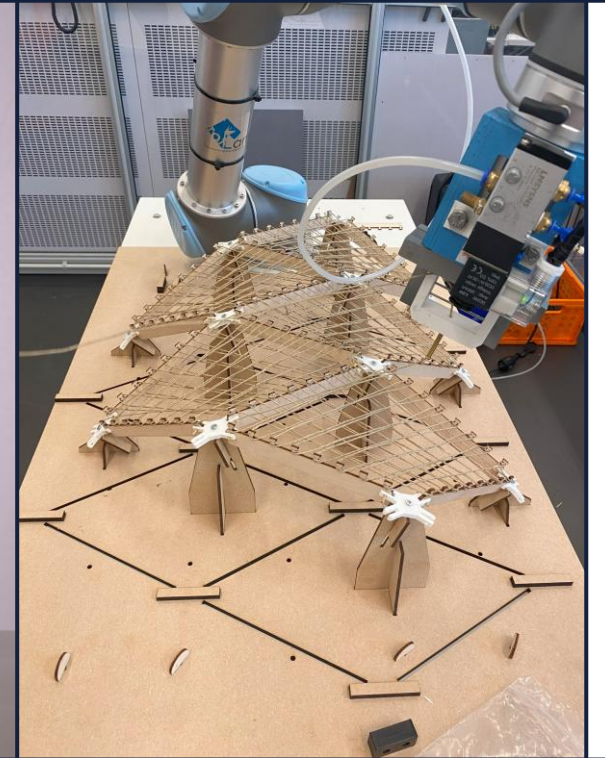
PROGRAMMABLE DEPLOYABLE STRUCTURES

Exploring the Potential of Mechanical Metamaterials and Large-Scale 3D Printing for Fast Production and Assembly of Deployable Structures.

Pepijn Feijen
Thesis Project
Building Technology

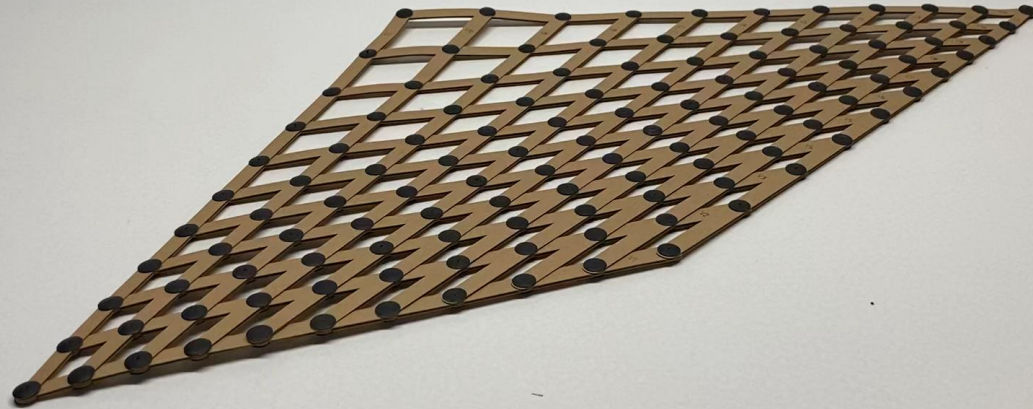
INTRODUCTION

PREVIOUS PROJECTS



INTRODUCTION

PREVIOUS PROJECTS



METAMATERIAL

MECHANICAL METAMATERIAL

PROGRAMMABLE METAMATERIAL

INTRODUCTION

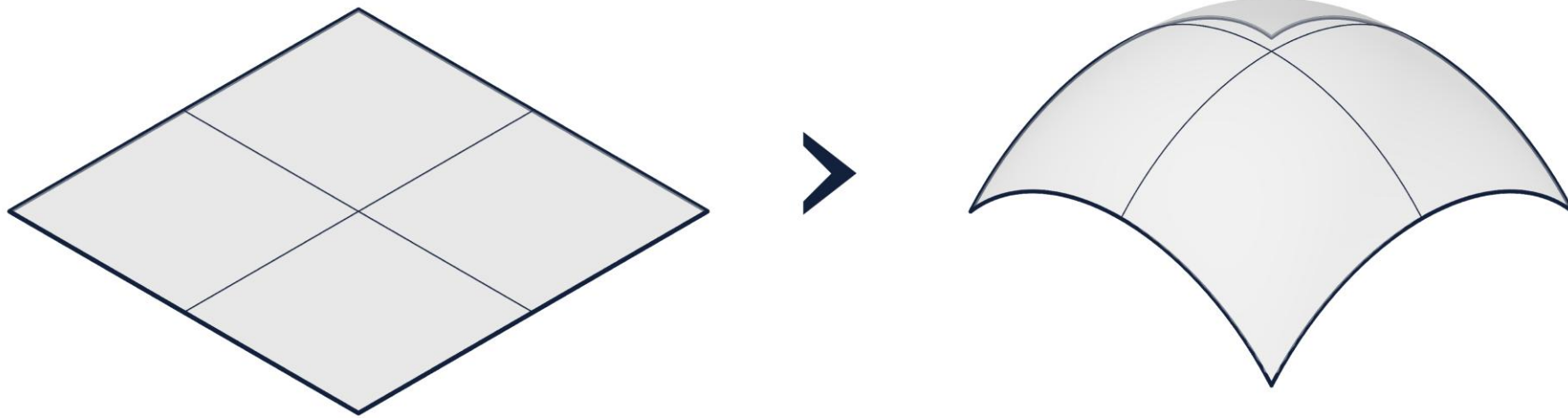
**ADAPTIVE
STRUCTURES**

**ENGINEERING
CHALLENGE**

INTERACTIVE

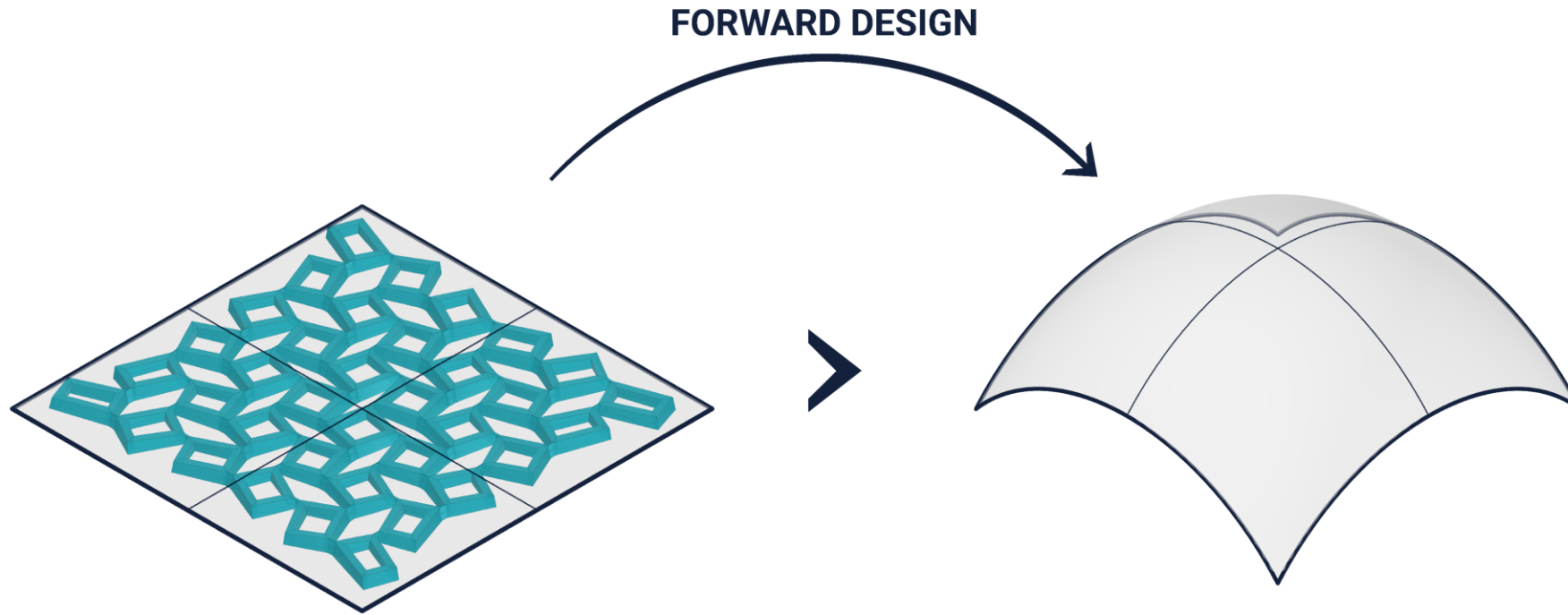
INTRODUCTION

INVERSE DESIGN PROBLEM



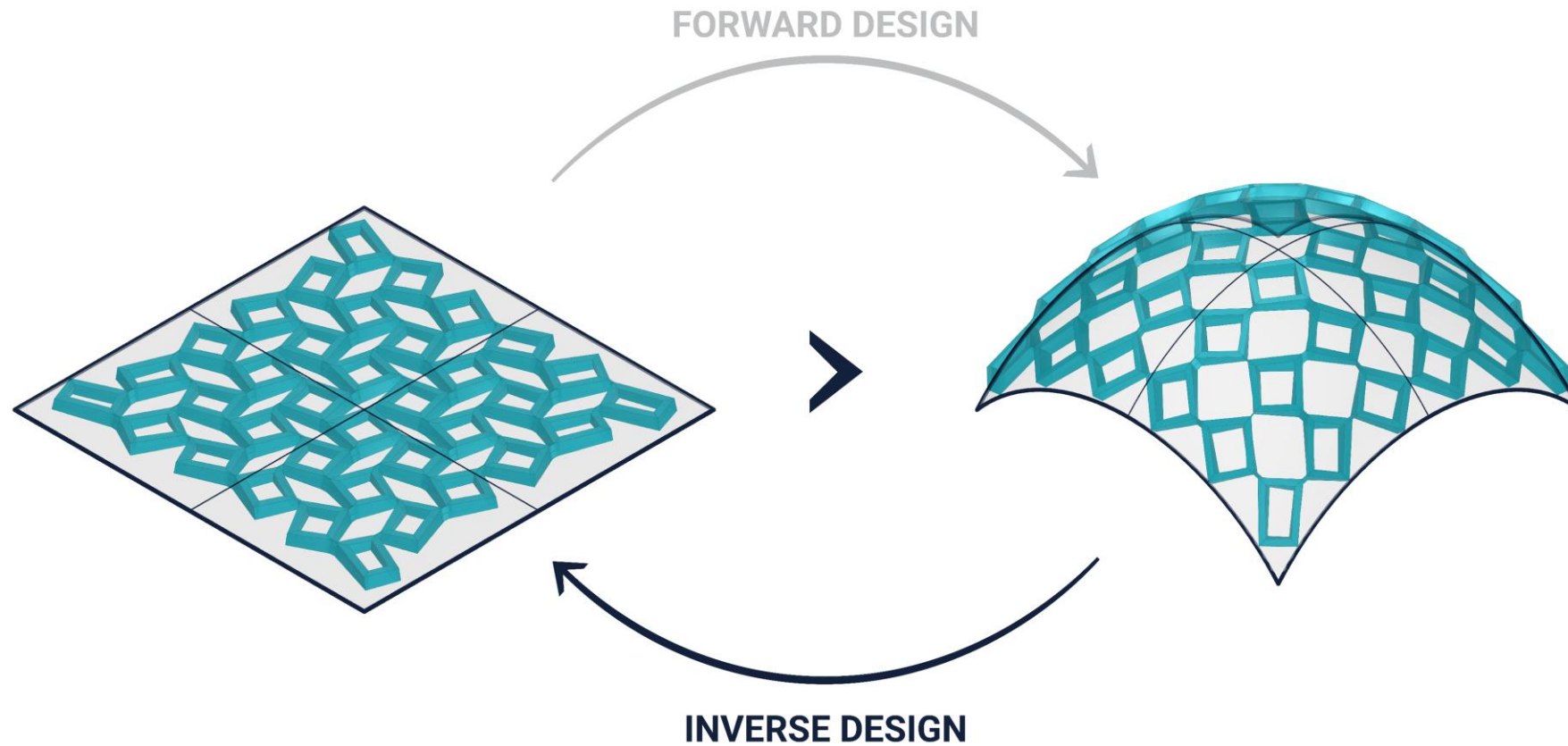
INTRODUCTION

INVERSE DESIGN PROBLEM



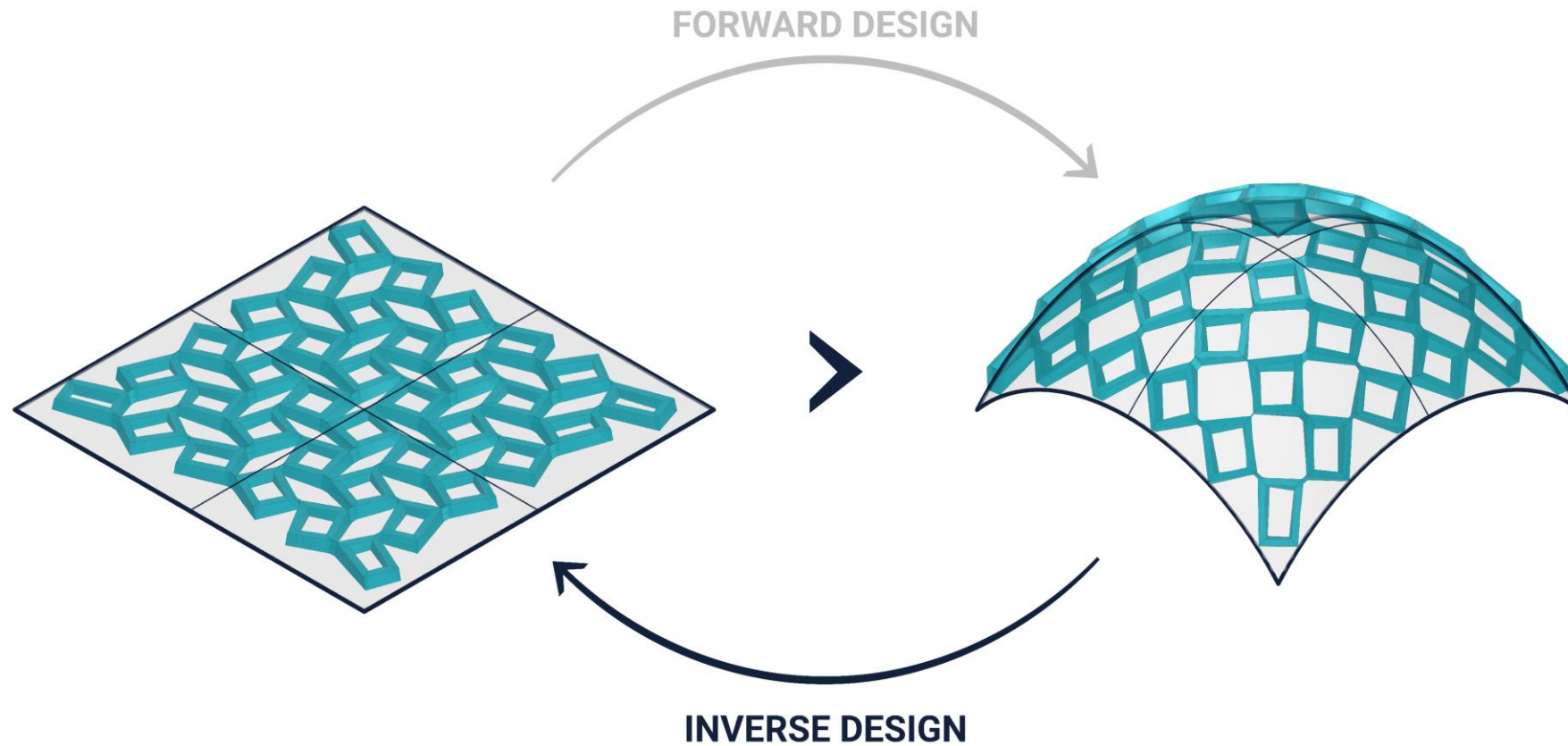
INTRODUCTION

INVERSE DESIGN PROBLEM



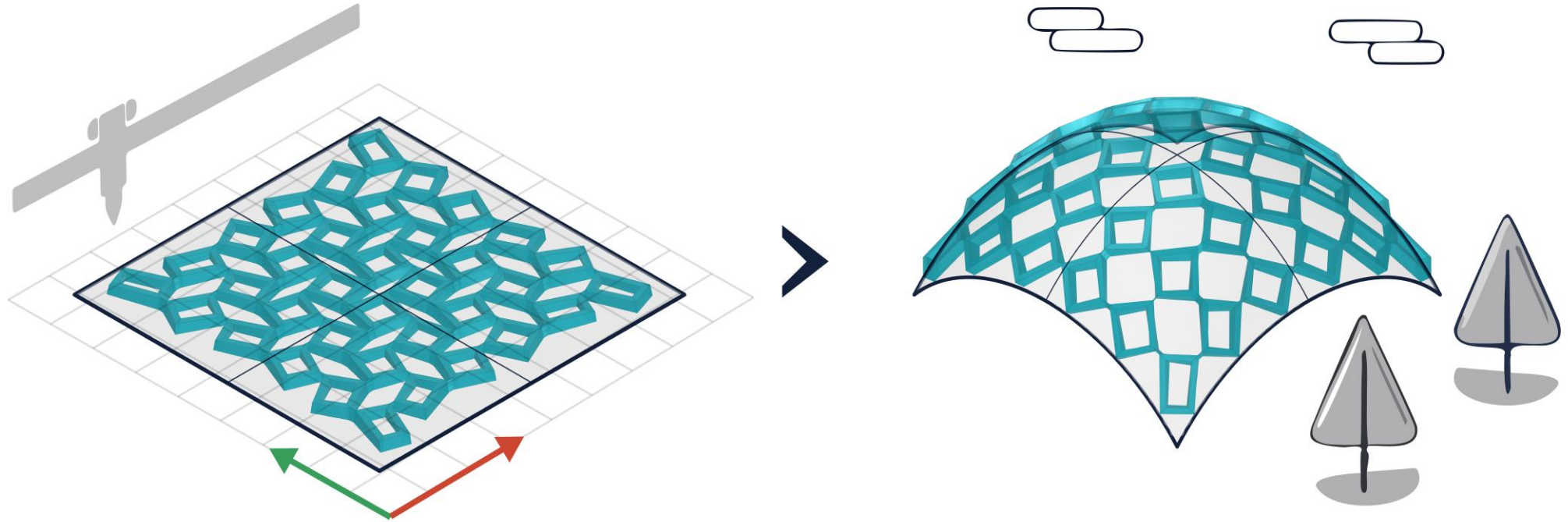
INTRODUCTION

INVERSE DESIGN PROBLEM



INTRODUCTION

INVERSE DESIGN PROBLEM



INTRODUCTION

MAIN RESEARCH QUESTION

- **How can the principles of mechanical-metamaterial be scaled up to design deployable structures?**

INTRODUCTION

SUB-QUESTIONS

- **How can a computational tool be developed to support the design of adaptive deployable structures?**

INTRODUCTION

SUB-QUESTIONS

- How can a computational tool be developed to support the design of adaptive deployable structures?
- **To what extent can programmable mechanical metamaterials be scaled to enable the creation of large, safe, and stable structures?**

INTRODUCTION

SUB-QUESTIONS

- How can a computational tool be developed to support the design of adaptive deployable structures?
- To what extent can programmable mechanical metamaterials be scaled to enable the creation of large, safe, and stable structures?
- **In what ways can the developed computational tool be applied to design a functional, deployable shelter for events?**

PROCESS

- Approach
- Kinematic system

FORM-FINDING

- Development of the tool
- Computational part

LARGE SCALE PRINTING

- Challenges
manufacturing
Scaling up

CASESTUDY

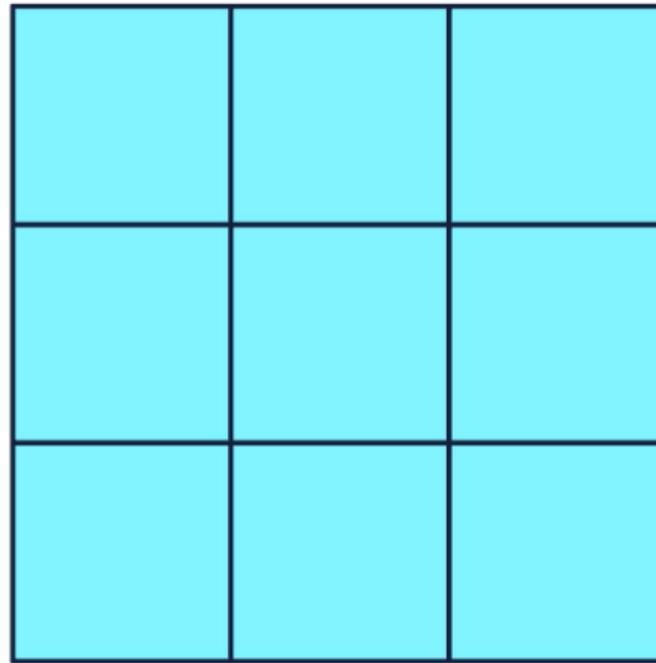
- Application
- Structural analysis

PROCESS

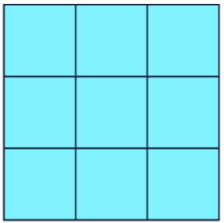
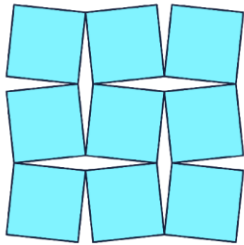
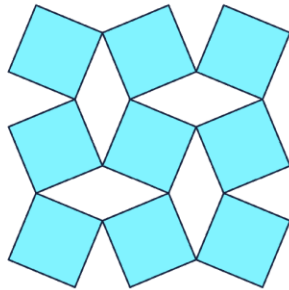
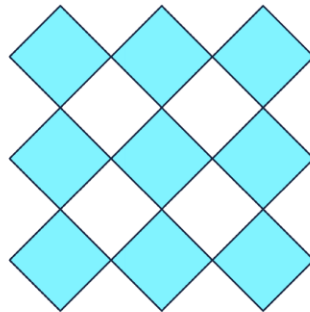
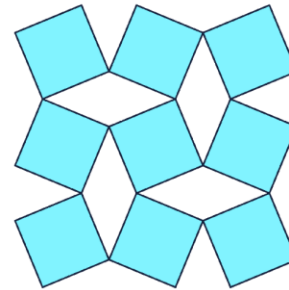
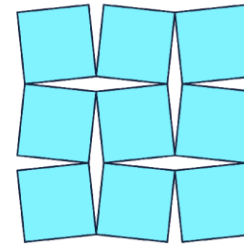
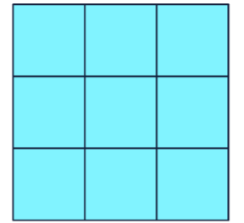
EXPLORATION PHASE



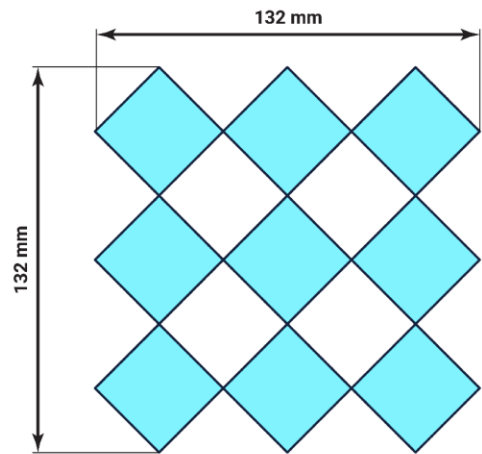
ROTATING POLYGON STRUCTURE



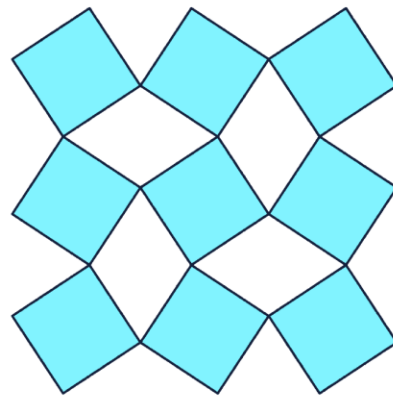
DoD: -90

DEGREE OF DEPLOYMENT:**-90°****-60°****-30°****0°****30°****60°****90°**

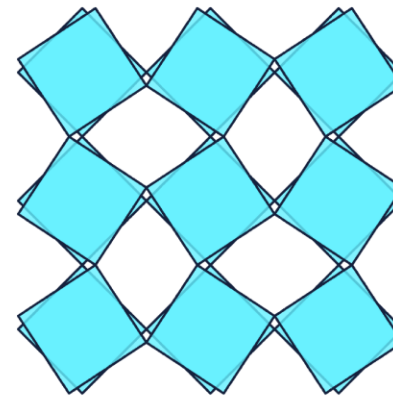
KINEMATIC SYSTEM



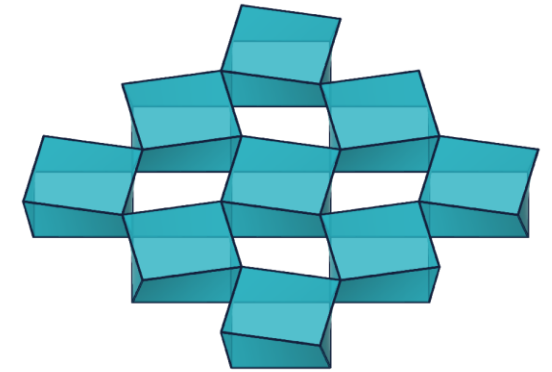
Rotating polygon
structure with
DoD: 80°



Rotating polygon
structure with
DoD: 30°



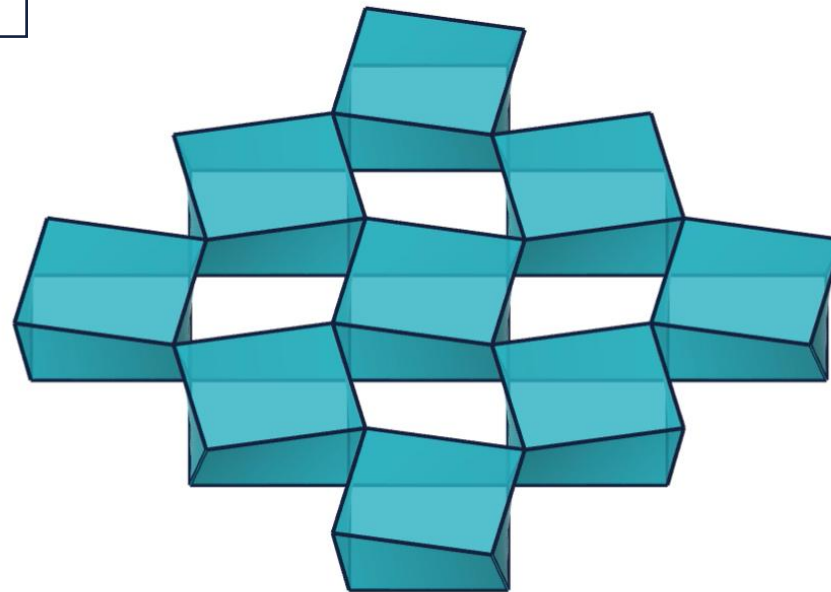
Two structure are stacked
on top of each other with
distance = 12 mm



Two layers are lofted
together, locking them
in rotation

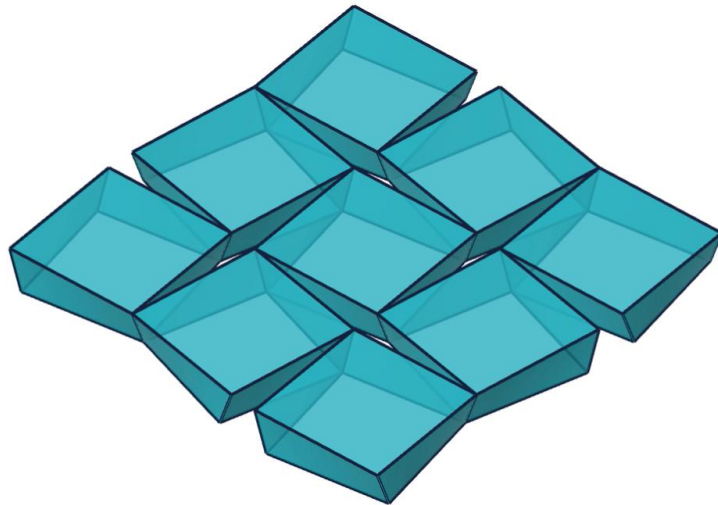
KINEMATIC SYSTEM

DoD top layer:	15°
DoD bottom layer:	0°
Distance between Layers:	12 mm
DoD structure:	0° - 32°



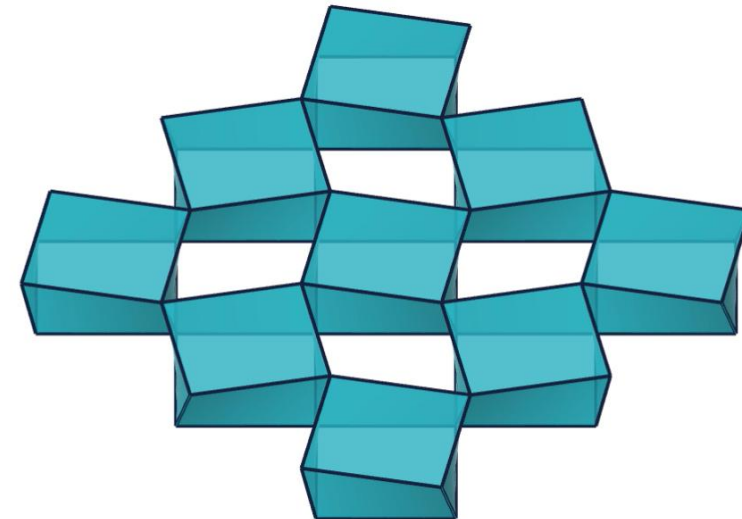
KINEMATIC SYSTEM

DoD top layer:	60°
DoD bottom layer:	45°
Distance between Layers:	12 mm
DoD structure:	0° - 39°



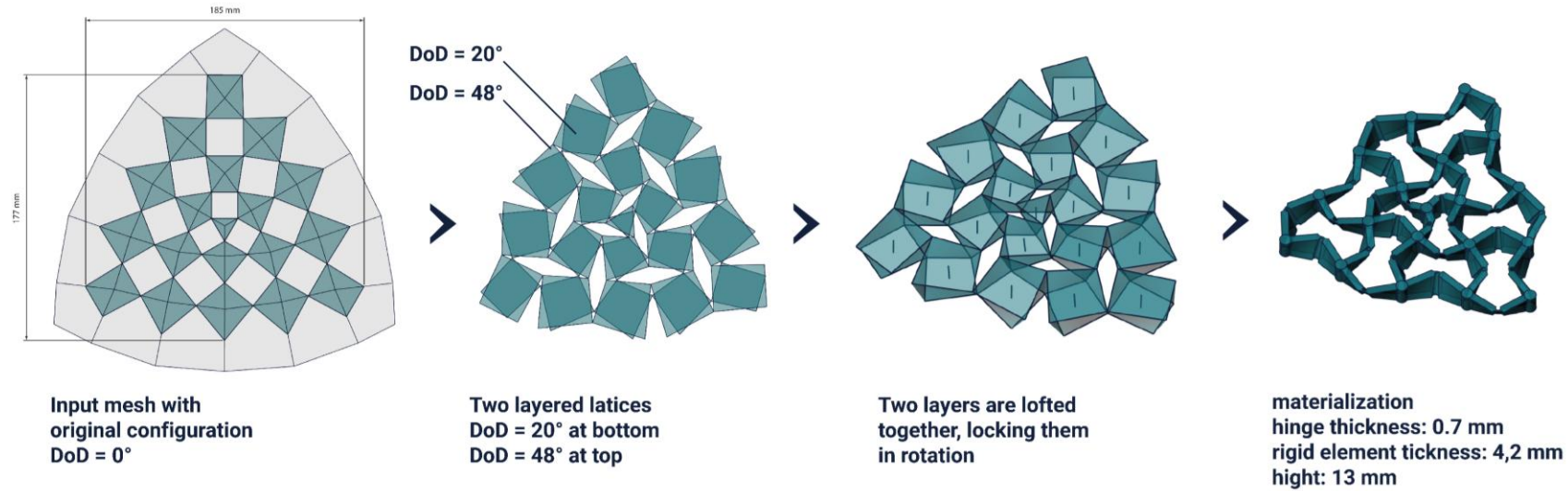
EXPANDING STRUCTURE

DoD top layer:	15°
DoD bottom layer:	0°
Distance between Layers:	12 mm
DoD structure:	0° - 32°

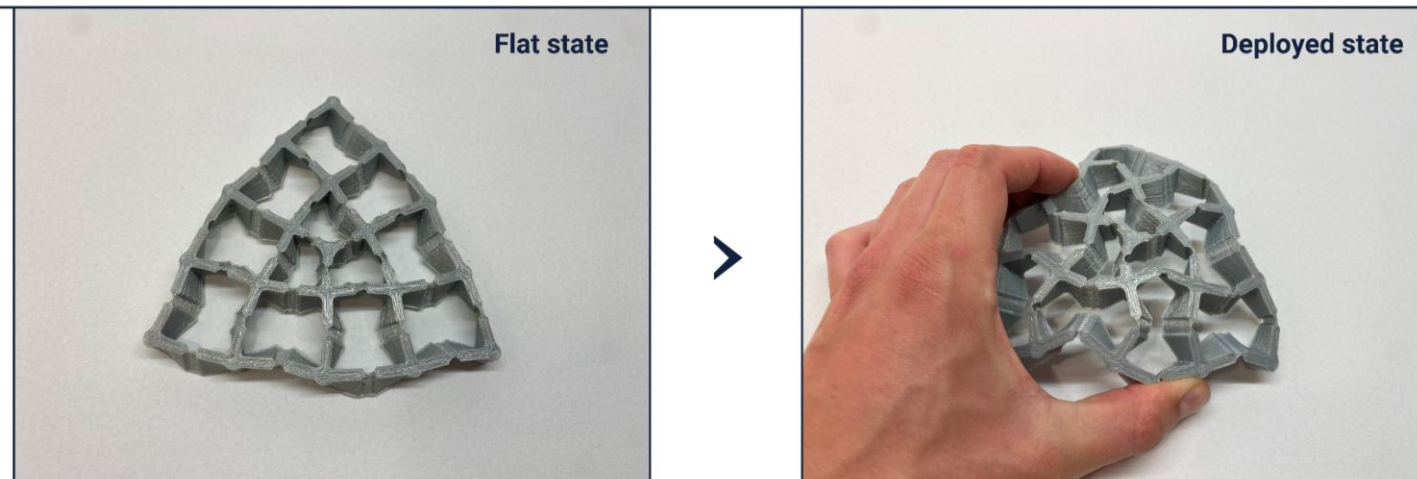


CONTRACTING STRUCTURE

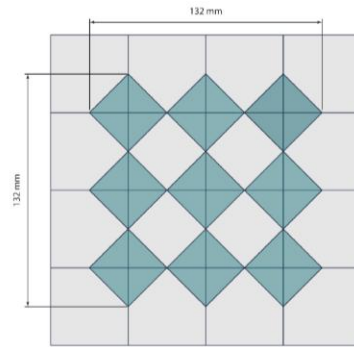
PROCESS MODEL 4



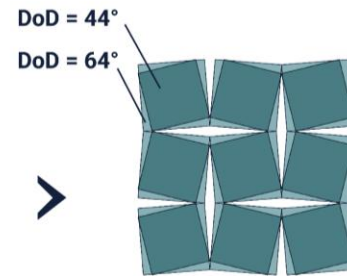
PHYSICAL MODEL 4



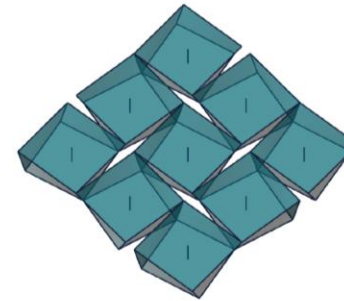
PROCESS MODEL 5



Input mesh with
original configuration
DoD = 0°



Two layered lattices
DoD = 44° at bottom
DoD = 64° at top



In this model the
panels are not locked
and free to rotate



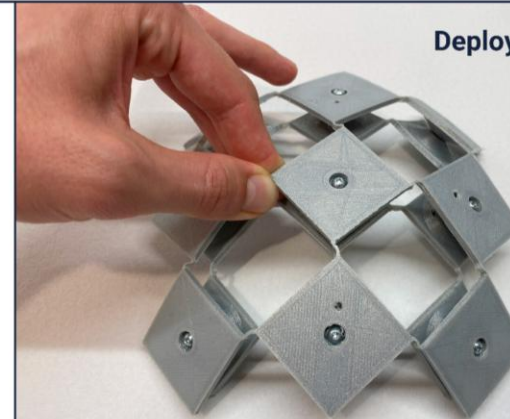
materialization
hinge thickness: 0.7 mm
rigid panels thickness: 3 mm
height: 13 mm

PHYSICAL MODEL 5

Flat state



Deployed state

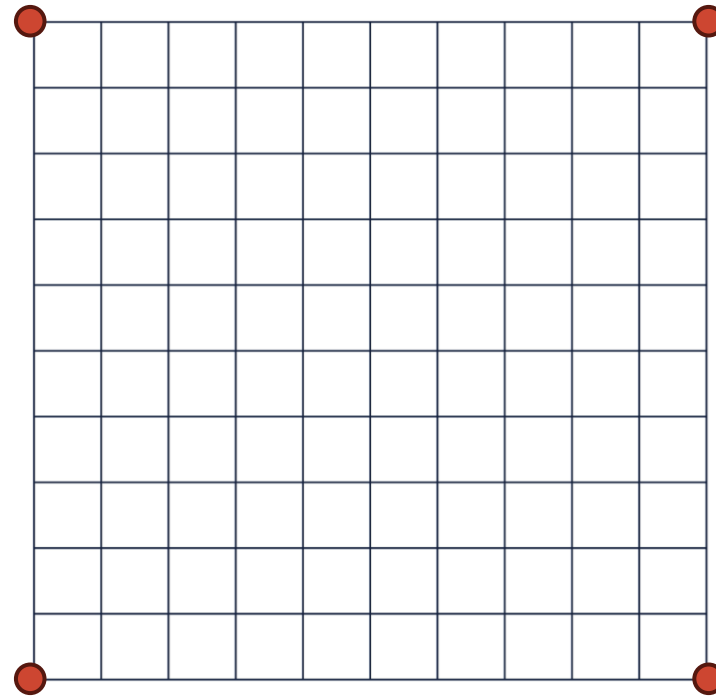


FORM-FINDING

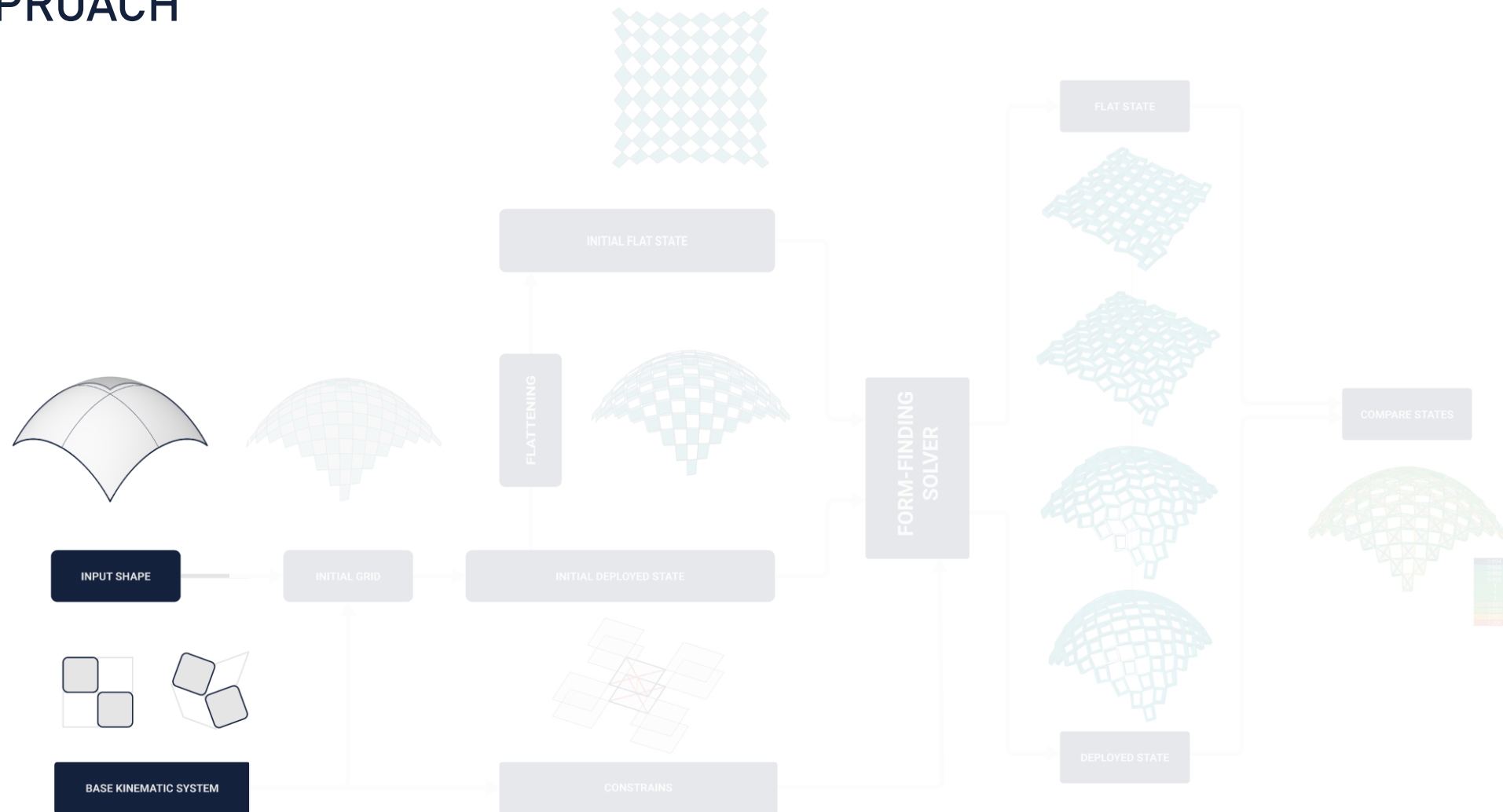
DYNAMIC RELAXATION

FORM-FINDING

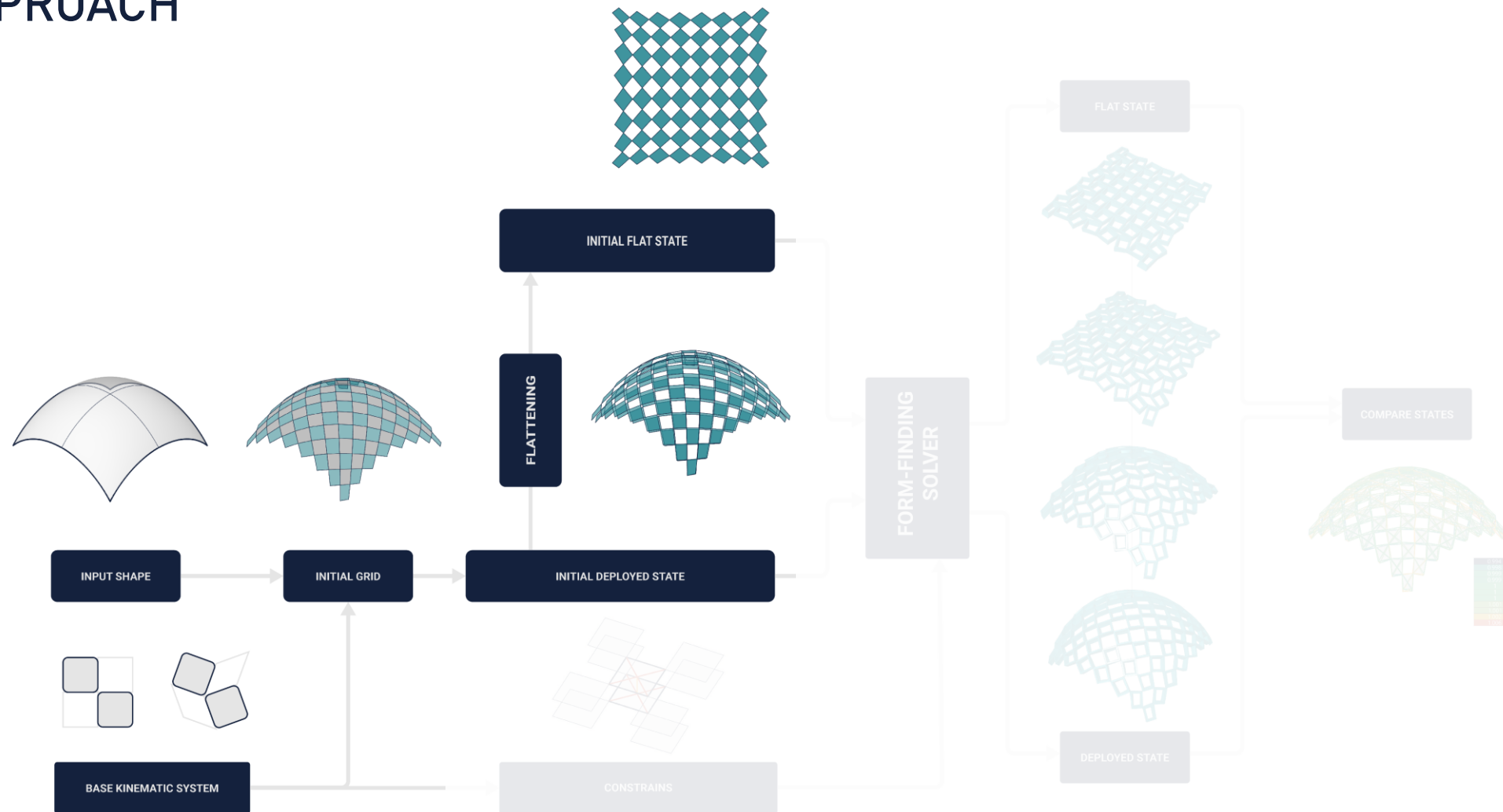
CONSTRAIN BASED SOLVER



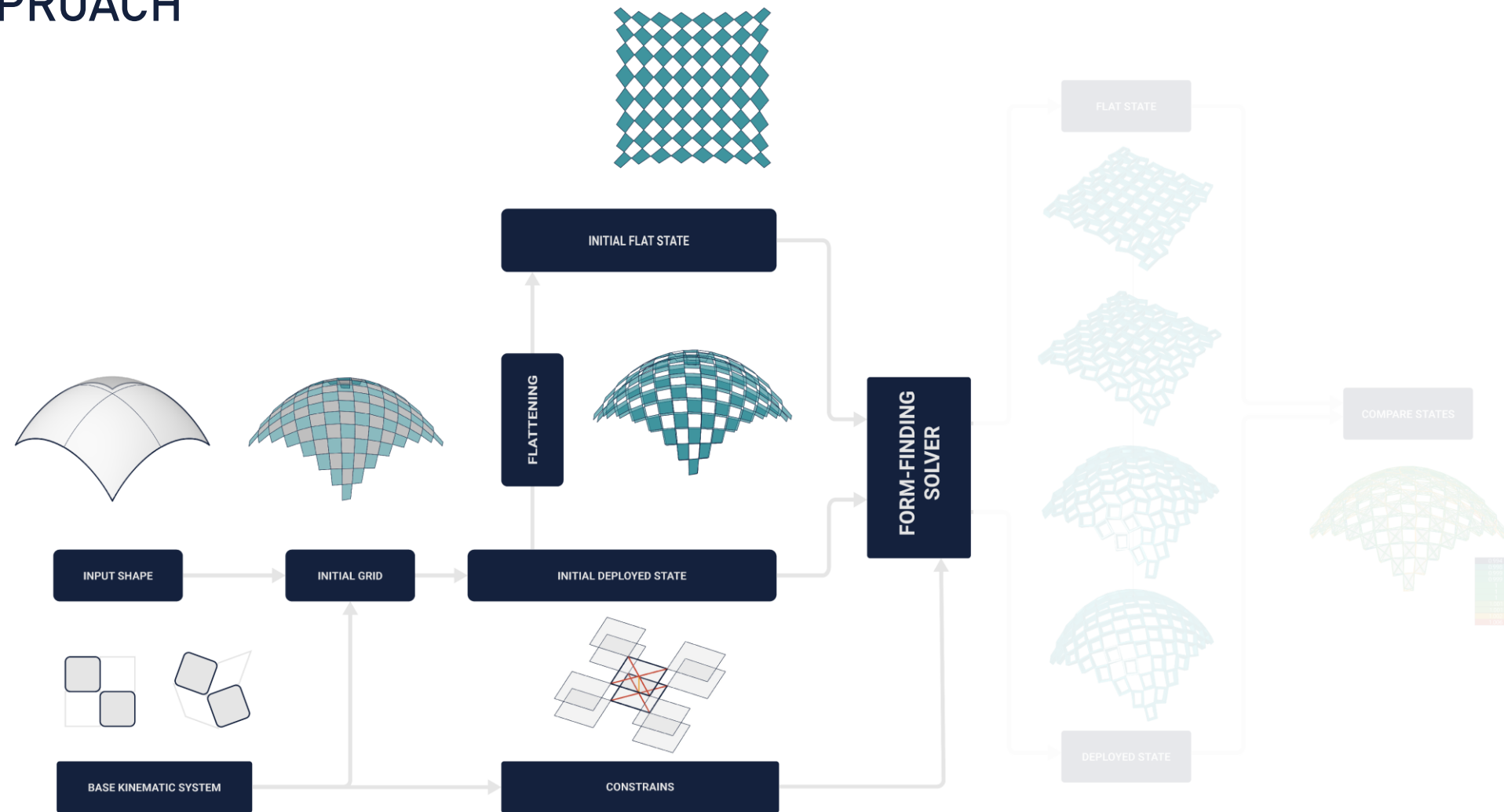
APPROACH



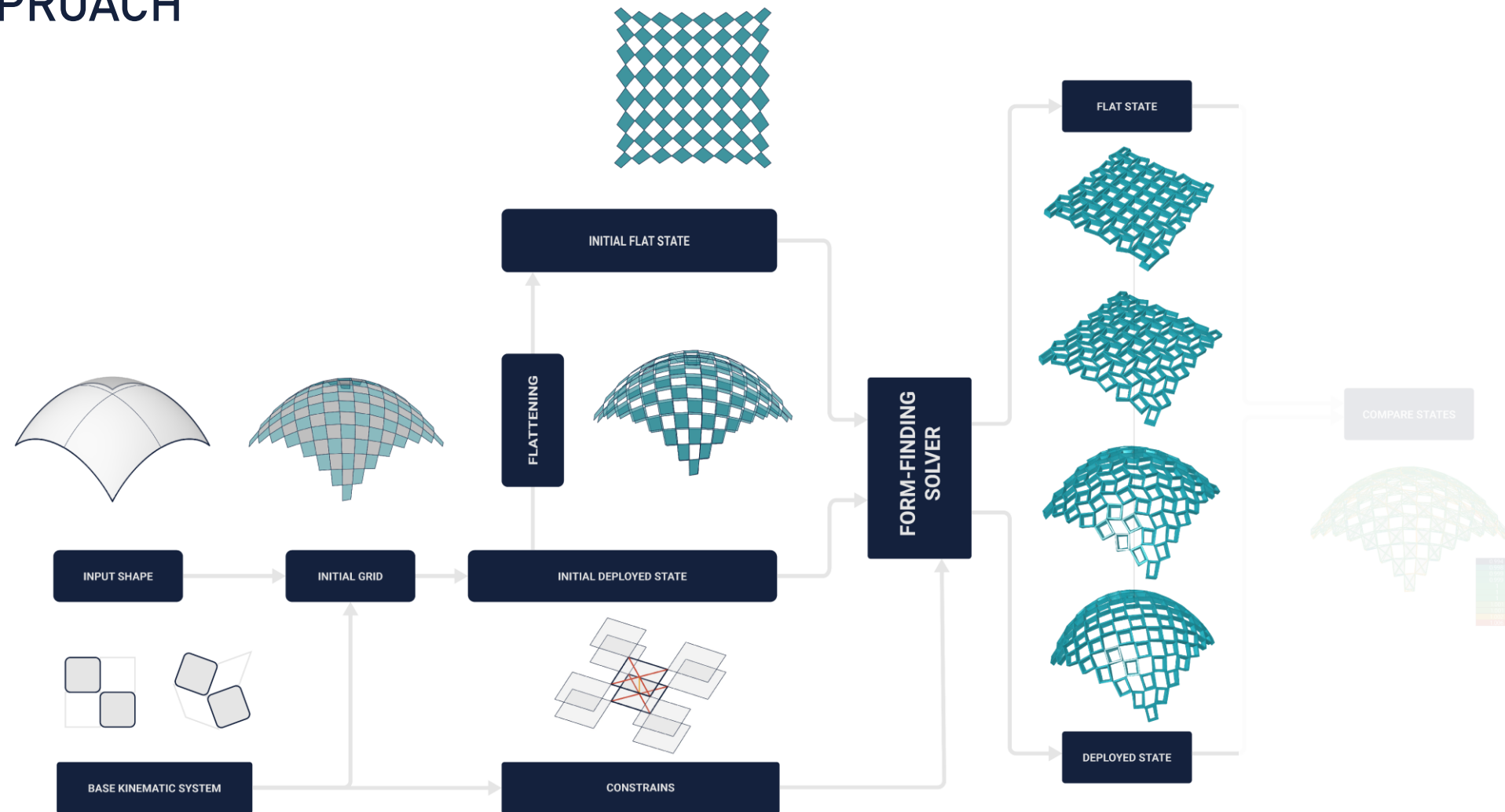
APPROACH



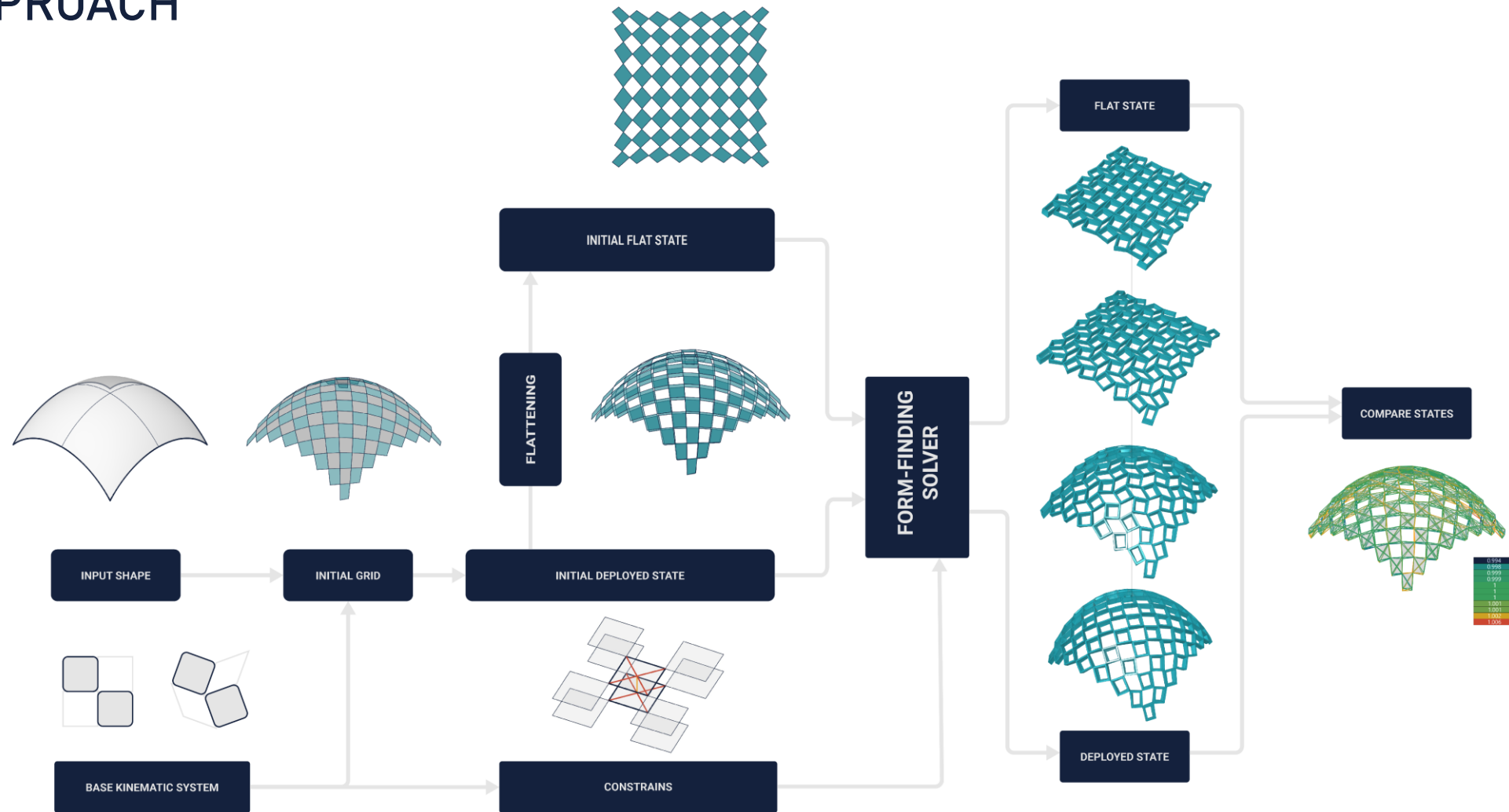
APPROACH



APPROACH

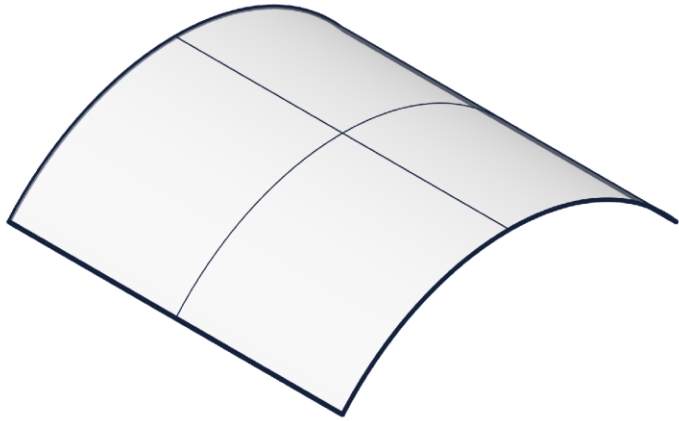


APPROACH

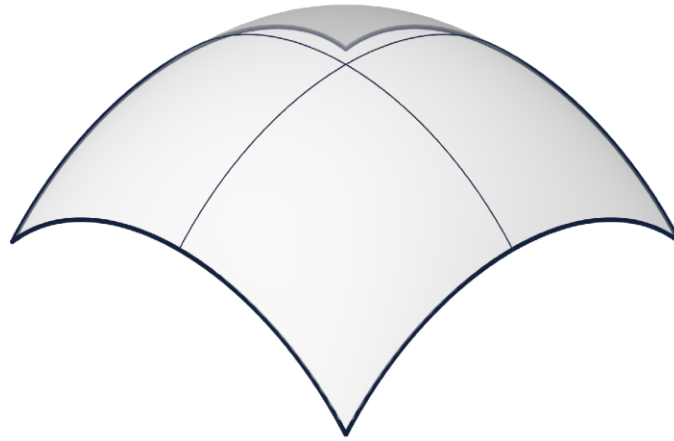


FORM-FINDING

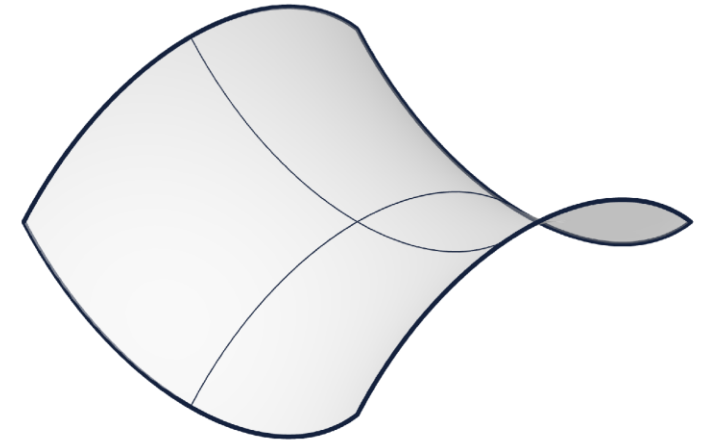
TEST SHAPES



MONOCLASTIC



SYNCLASTIC



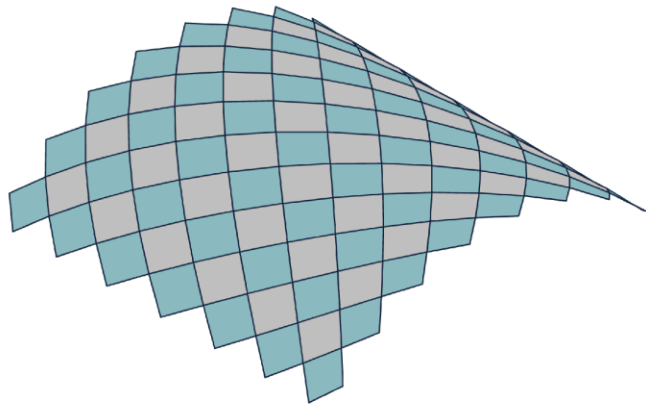
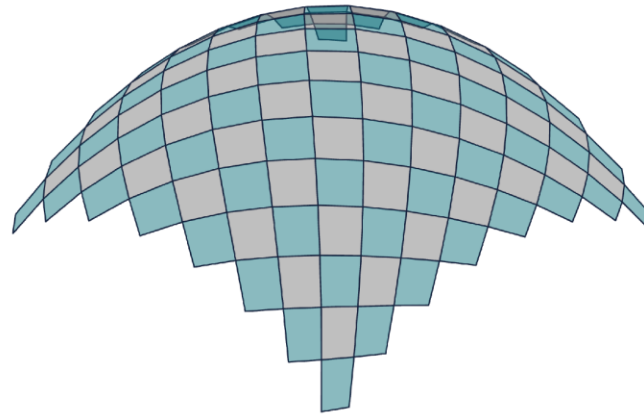
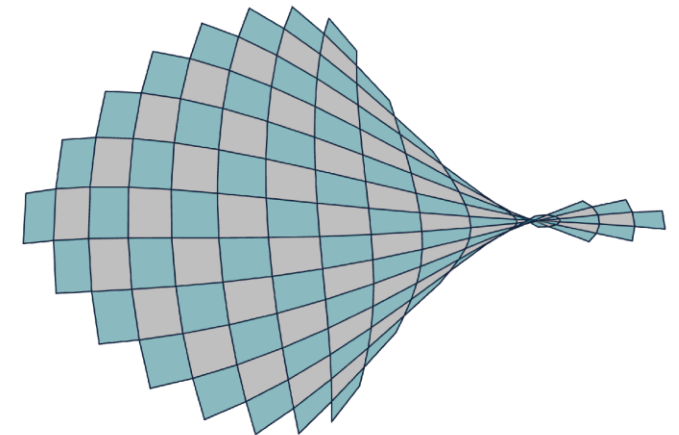
ANTICLASTIC

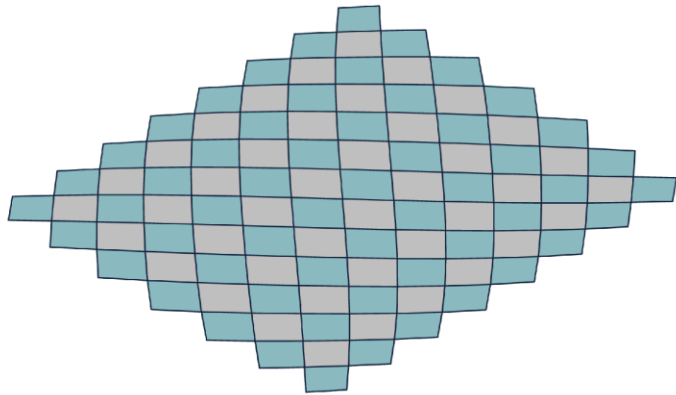
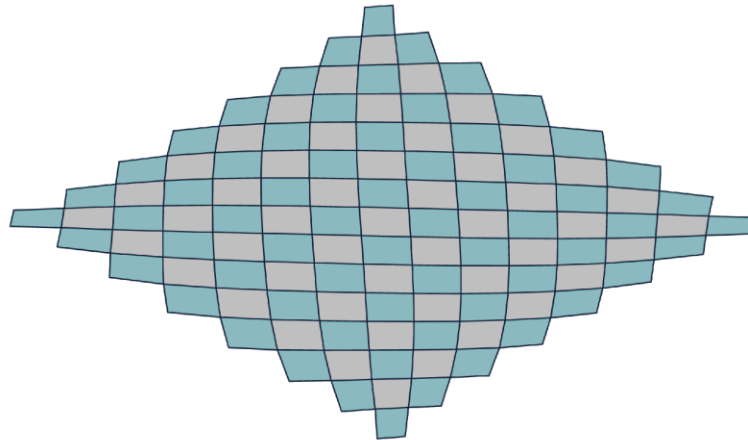
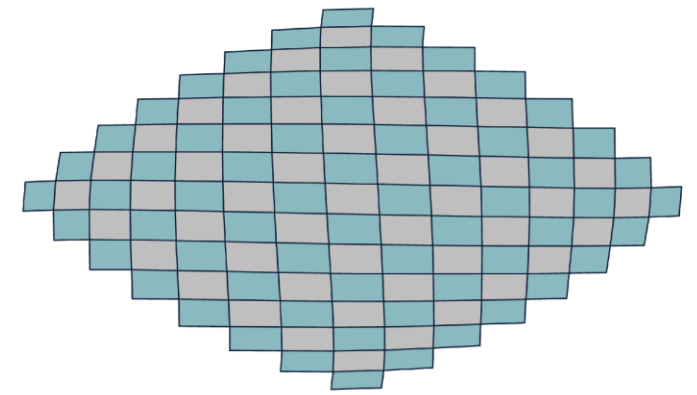
INITIAL GRID

PANELS



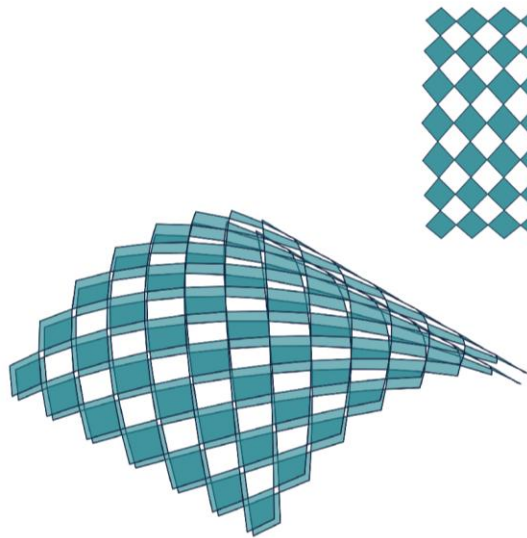
NEGATIVE SPACE

**MONOCLASTIC****SYNCLASTIC****ANTICLASTIC**

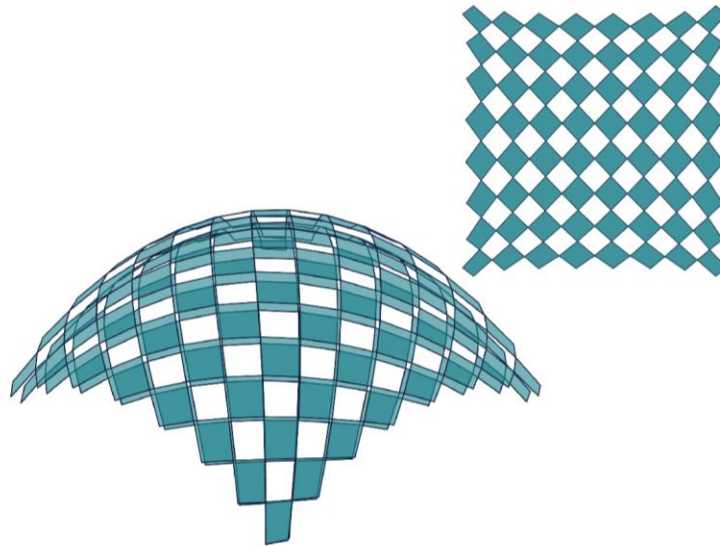
FLATTEN GRID**PANELS****NEGATIVE SPACE****MONOCLASTIC****SYNCLASTIC****ANTICLASTIC**

Form-Finding

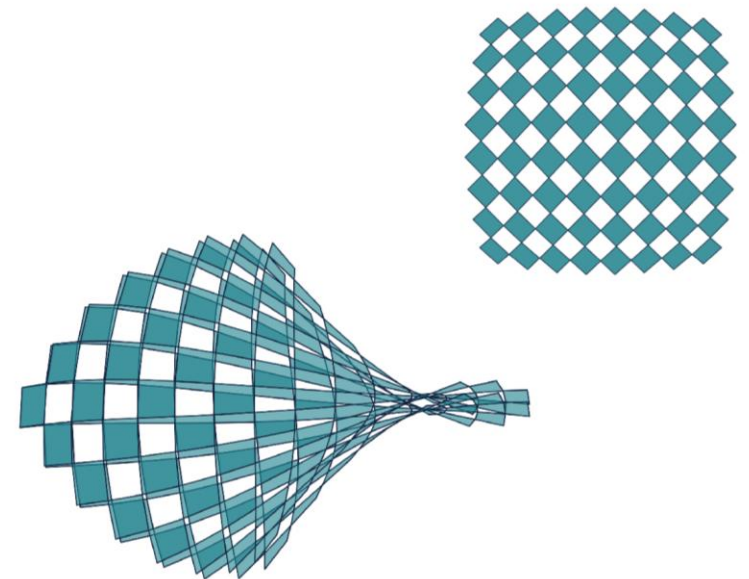
INPUT



MONOCLASTIC



SYNCLASTIC



ANTICLASTIC

CONSTRAINTS

**GLOBAL EQUAL
LENGTH
CONSTRAIN**

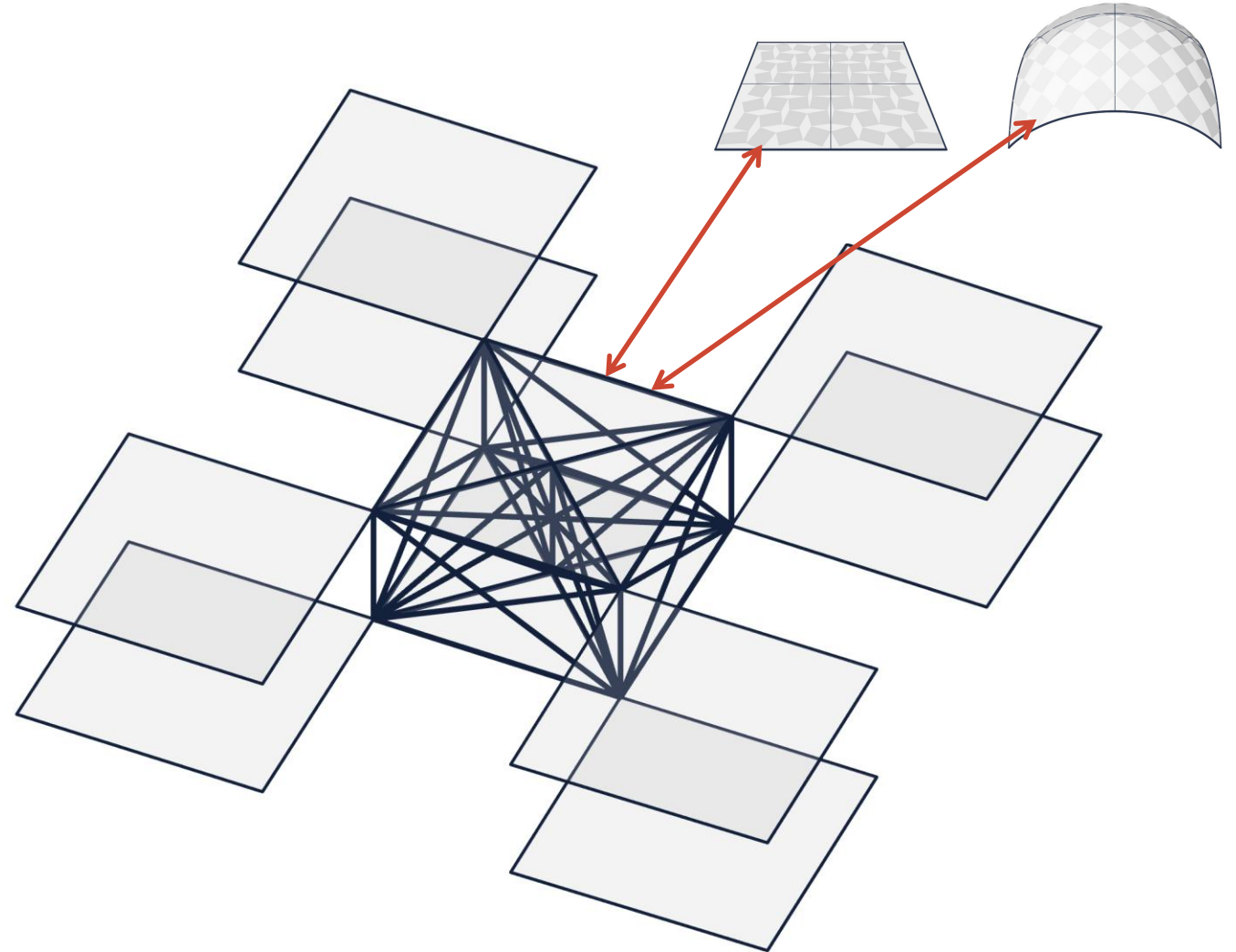
**INTERNAL
CONSTRAINS**

**KINEMATIC
CONSTRAINS**

**STATE
CONSTRAINS**

EQUAL LENGTE

- **Global Equal length Constraint**
- every node-to-node link
- Between two states
- Ensure elements to stay similar
- While allowing deformation



INTERNAL CONSTRAINTS

- **Centre axis**

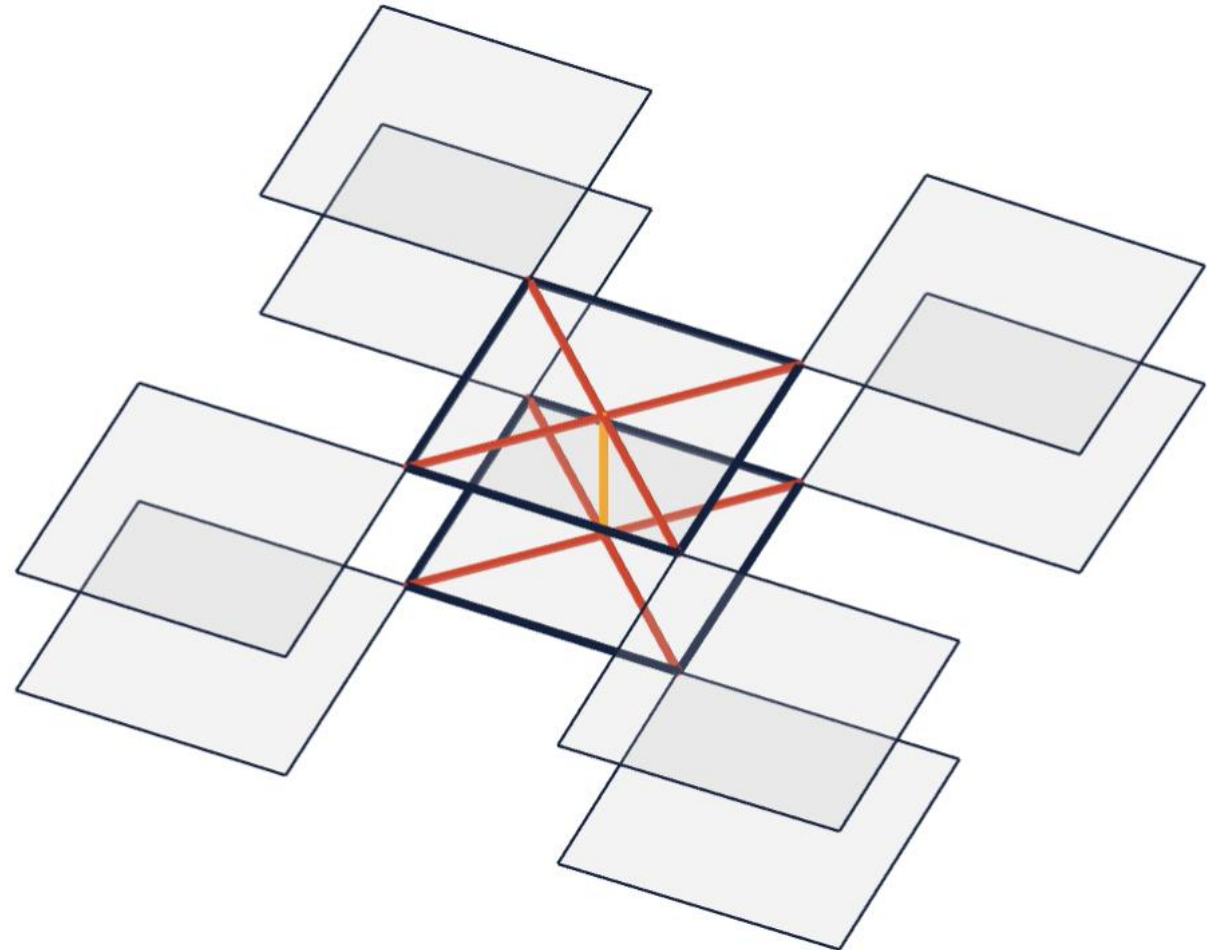
Ensuring rotation between panels

- **Diagonals**

Keep the center in its place, the panels square

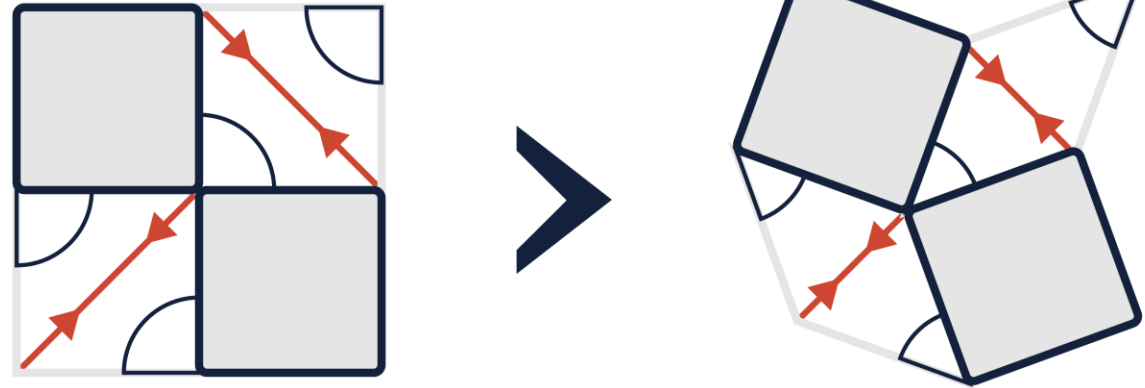
- **Panel edges**

Are allowed limited deformation



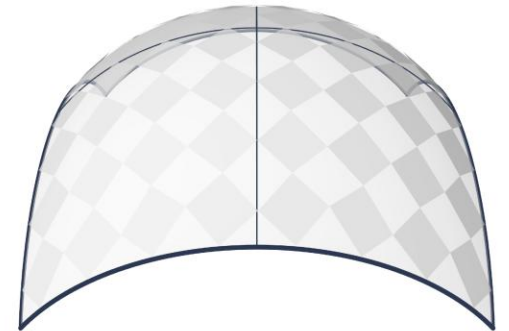
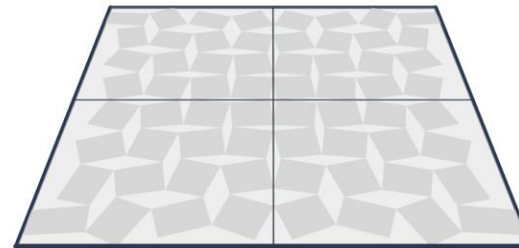
KINEMATIC CONSTRAINTS

- Constrains between the elements
- Applied on negative area
- Encourage the kinematic system from rotating
- Prevent collision

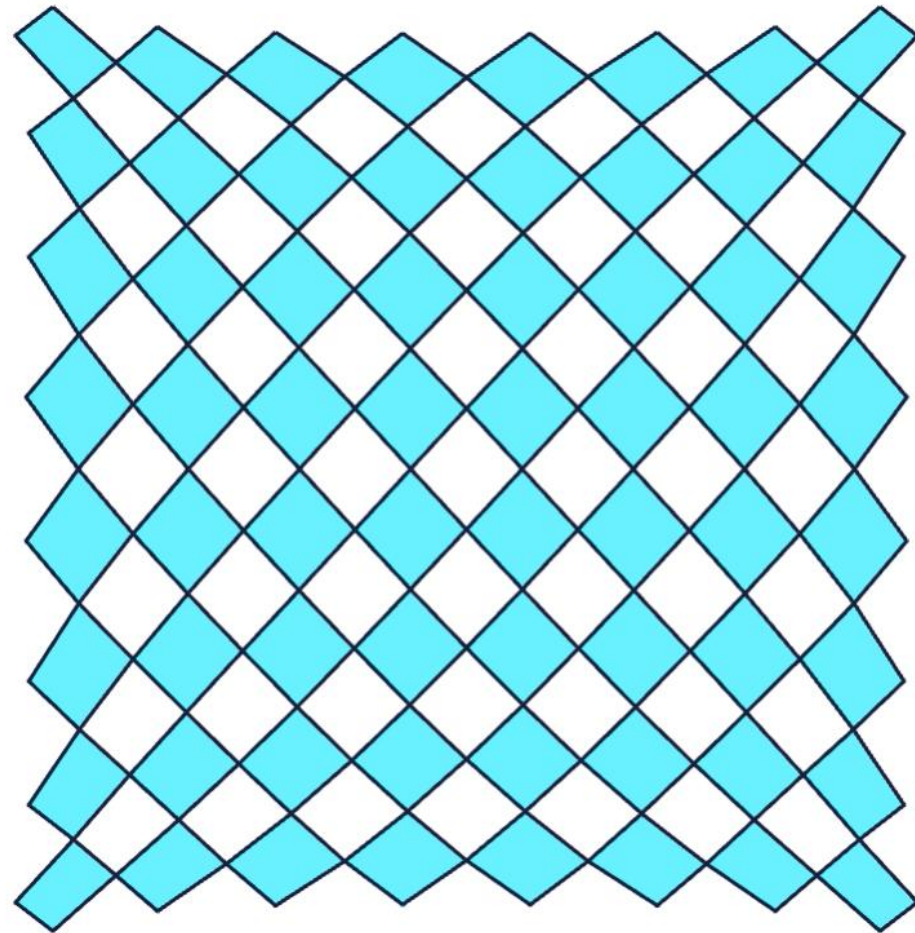
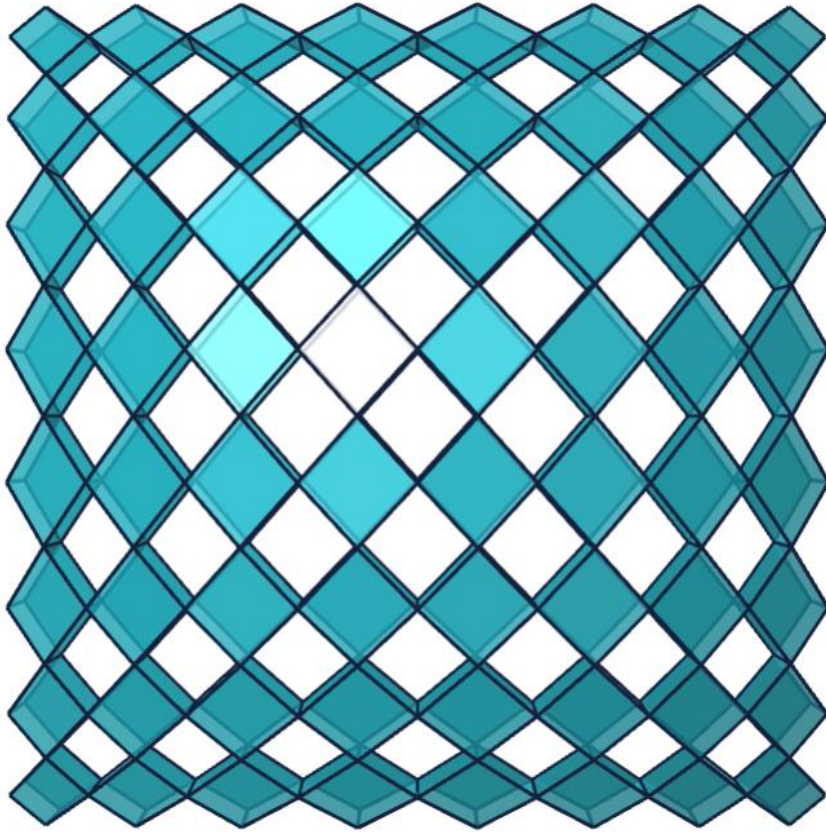


STATE CONSTRAINTS

- Constrain that differ per state
- Ensures panels stays flat/ stays on input surface
- Fix boundary points



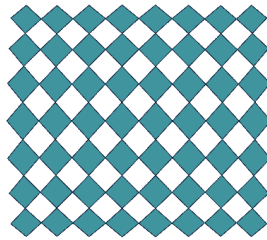
FORMFINDING PROCESS



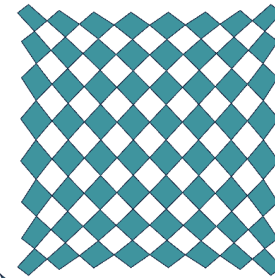
SOLVER

INPUT

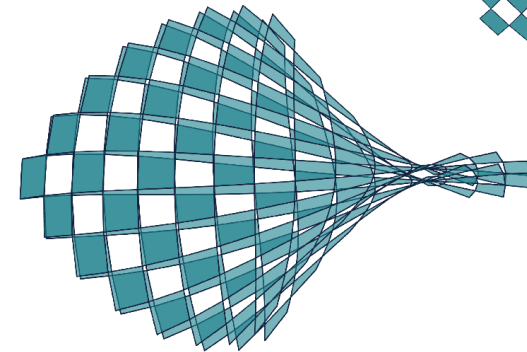
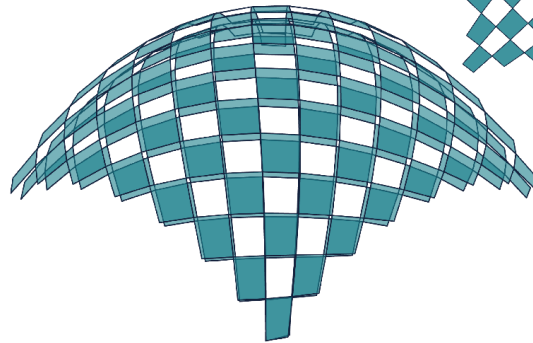
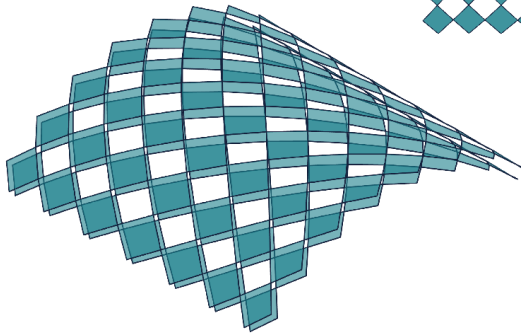
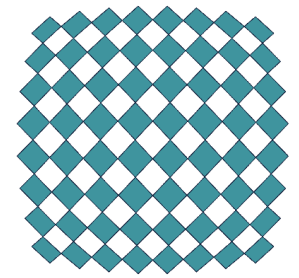
Grid:
8 x 7



Grid:
8 x 8



Grid:
8 x 8



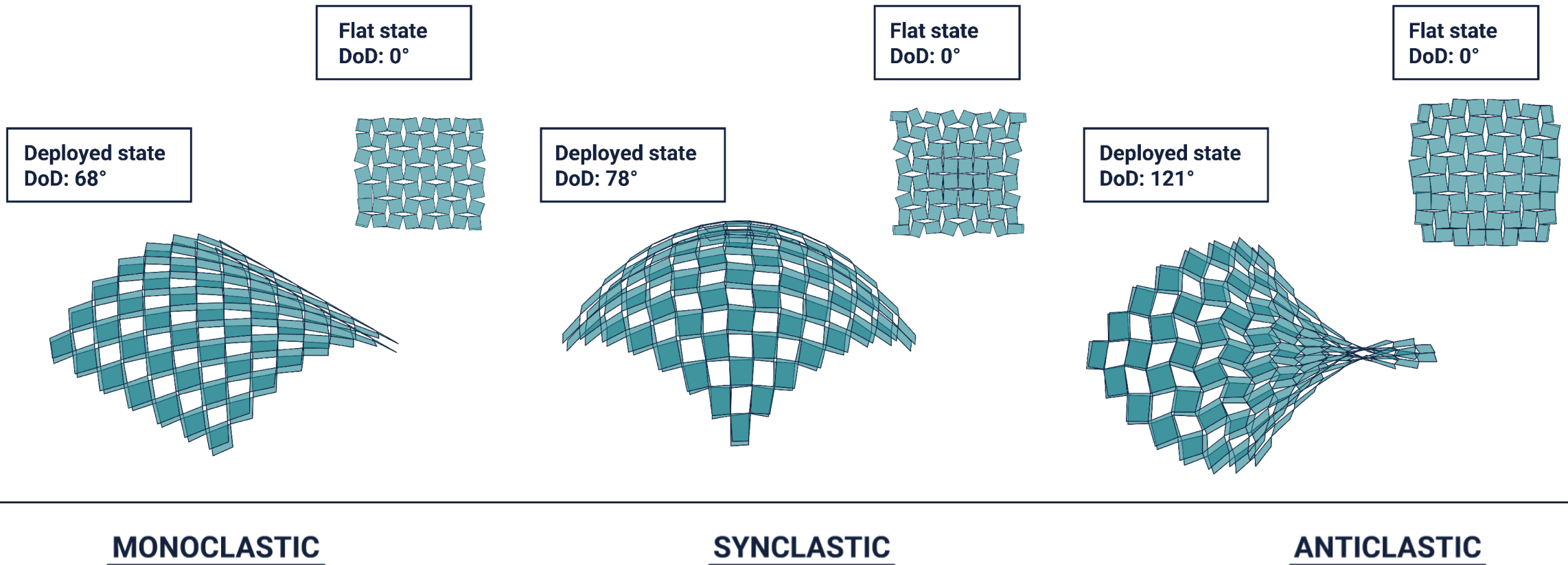
MONOCLASTIC

SYNCLASTIC

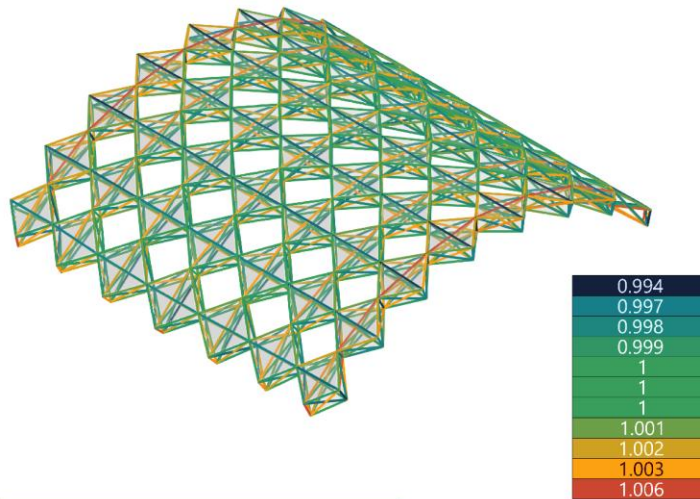
ANTICLASTIC

SOLVER

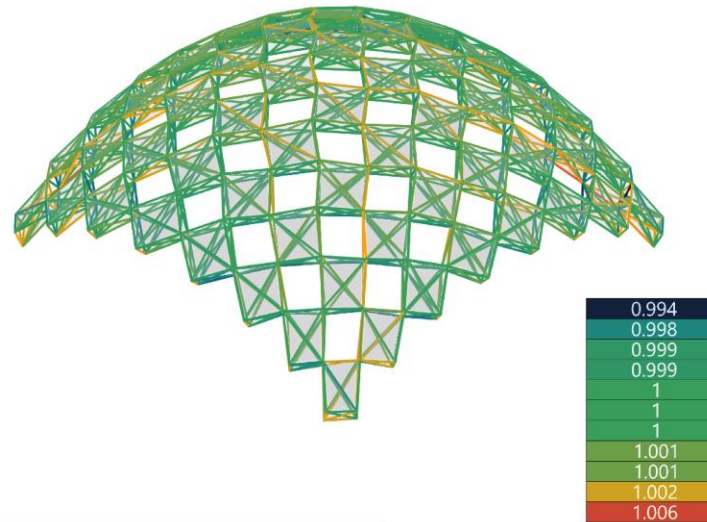
OUTPUT



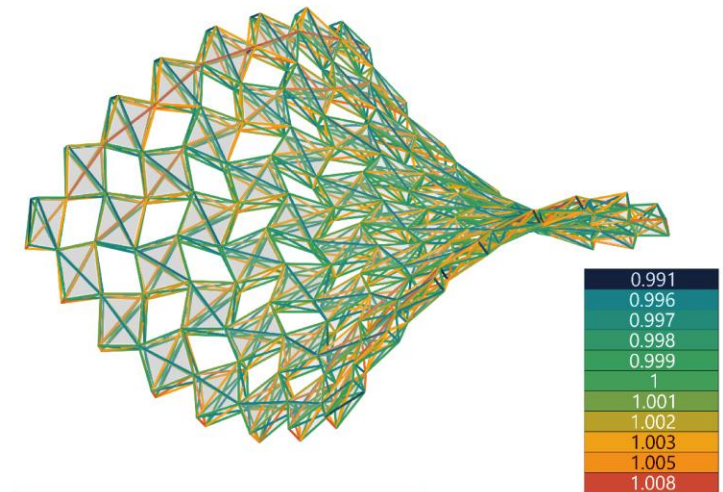
VALIDATION



largest scale differences
0,6%



largest scale differences
0,6%



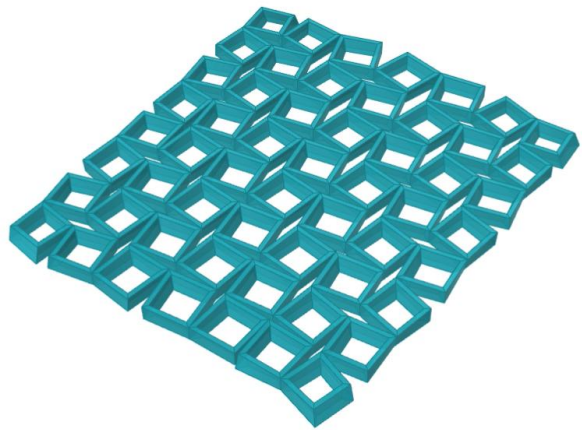
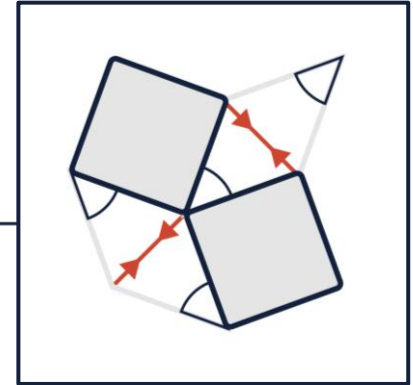
largest scale differences
0,9%

MONOCLASTIC

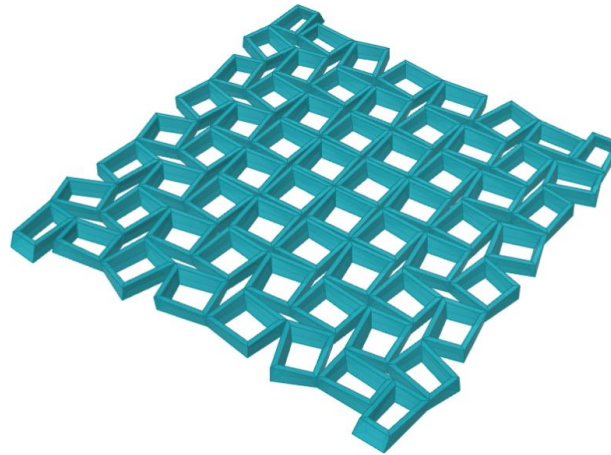
SYNCLASTIC

ANTICLASTIC

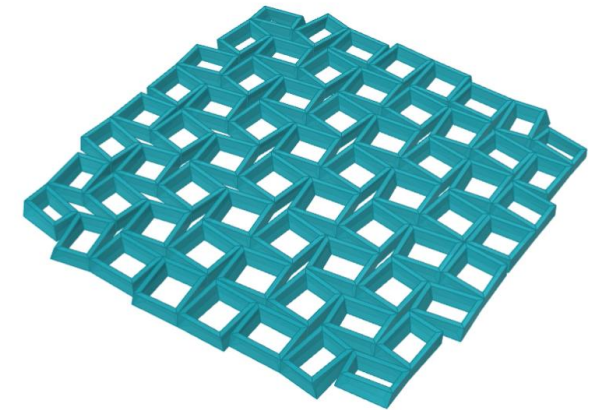
DEPLOYMENT



MONOCLASTIC

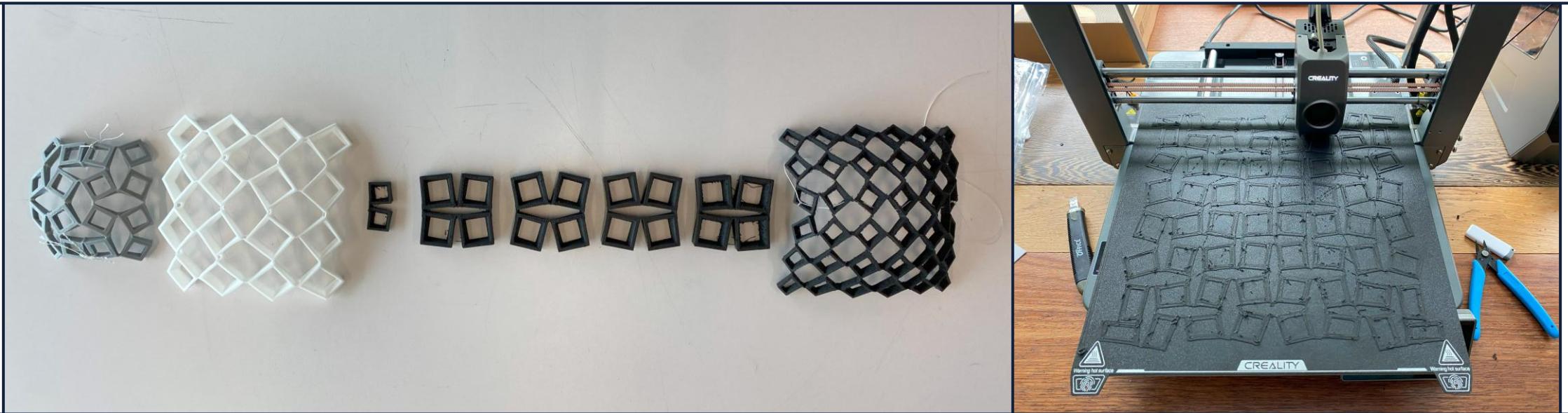


SYNCLASTIC

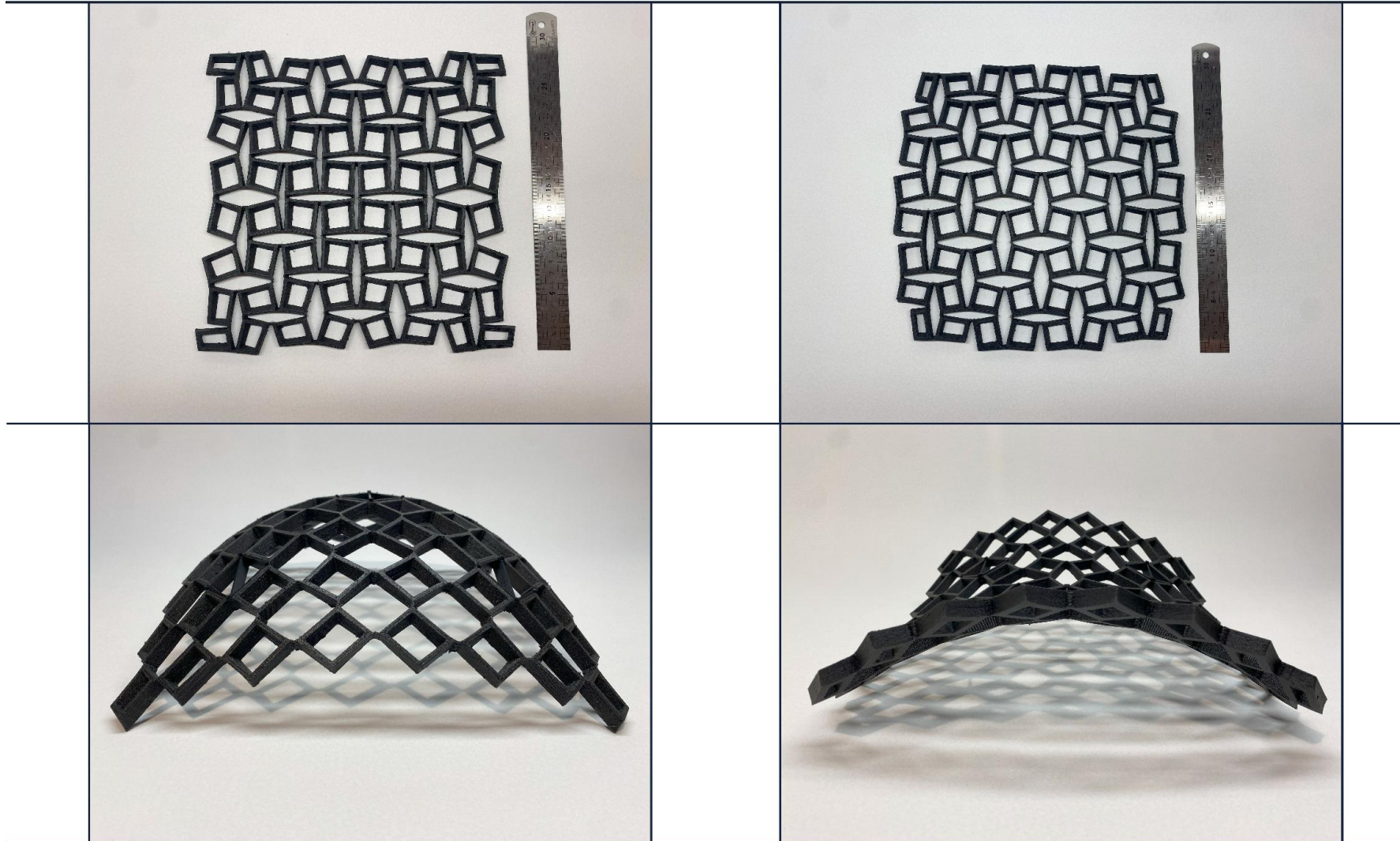


ANTICLASTIC

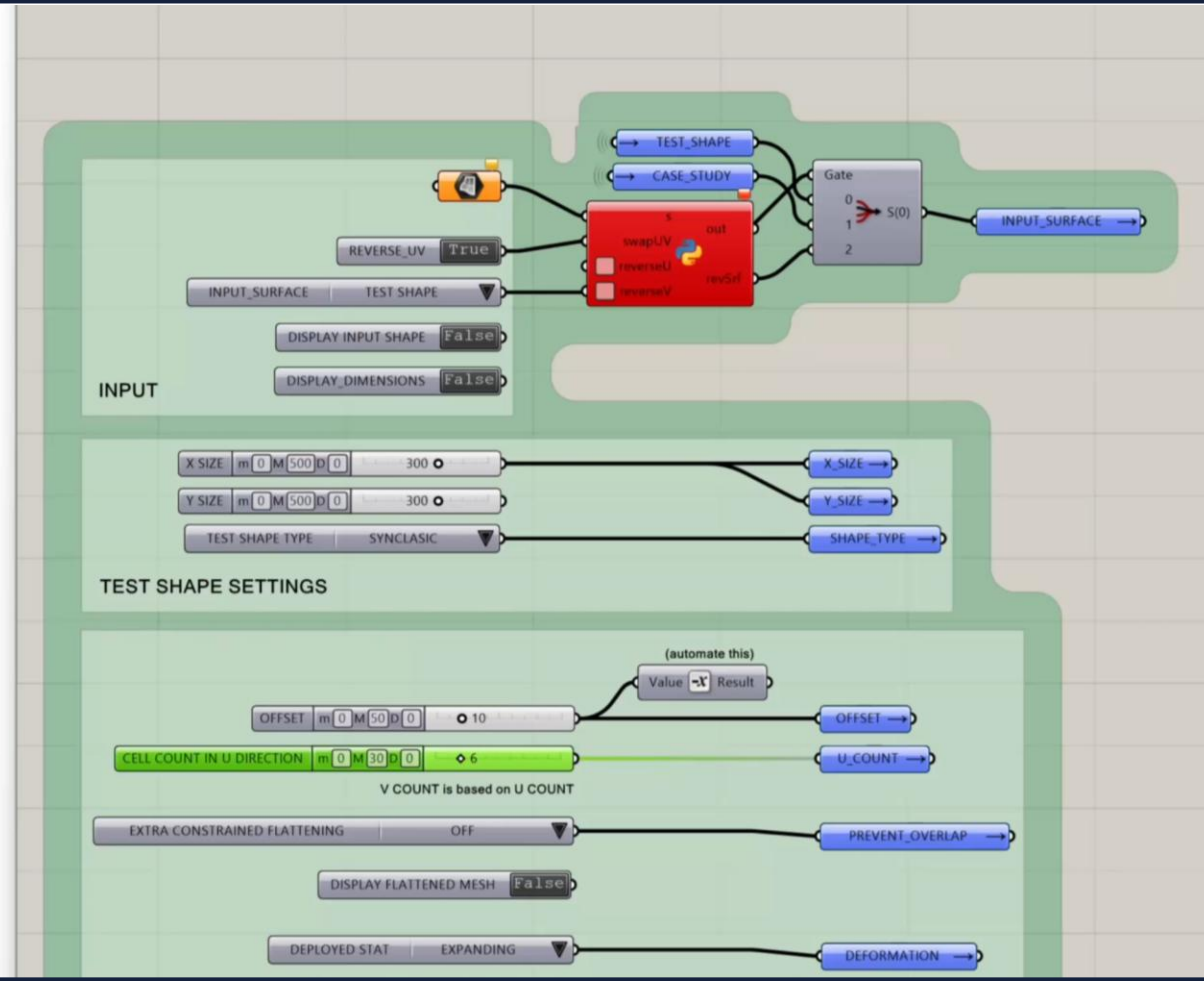
PRINTING PROCESS



PROTOTYPES TEST SHAPES



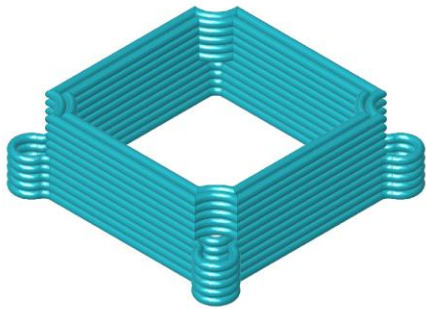
THE TOOL



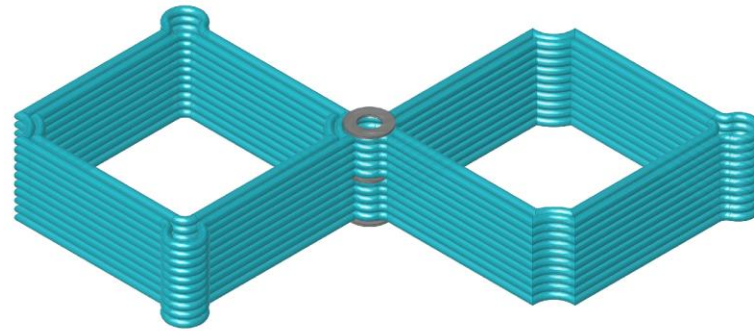
LARGE SCALE PRINTING

HINGES

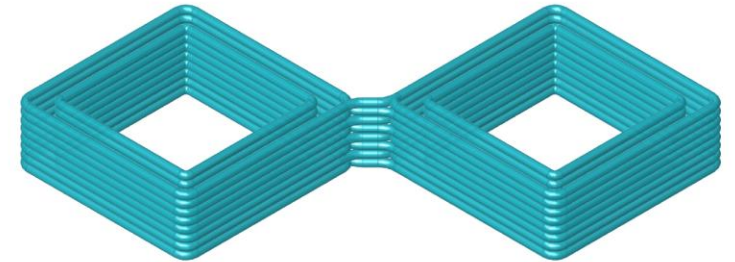
**SEPARATE
COMPONENTS**



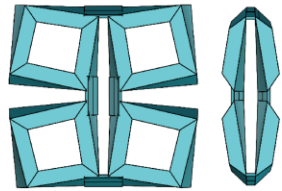
**MECHANICAL
HINGES INPLACE**



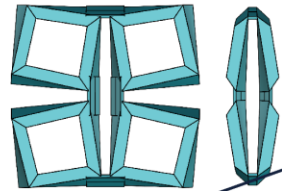
**SINGLE MATERIAL
COMPLIANT HINGES**



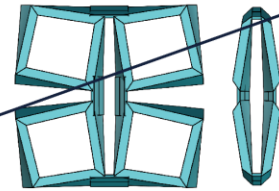
PROTOTYPE CONFIGURATIONS



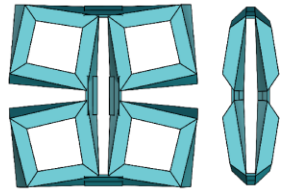
HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 1,0 mm
ELEMENT THICKNESS = 5,0 mm



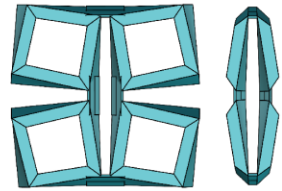
HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 1,0 mm
ELEMENT THICKNESS = 4,0 mm



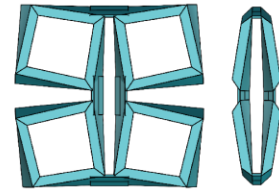
HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 1,0 mm
ELEMENT THICKNESS = 3,0 mm



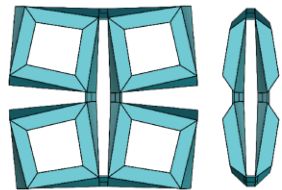
HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 0,8 mm
ELEMENT THICKNESS = 4,8 mm



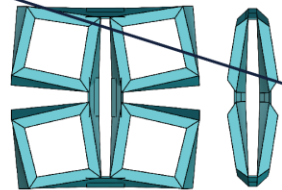
HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 0,8 mm
ELEMENT THICKNESS = 4,0 mm



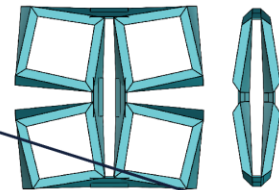
HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 0,8 mm
ELEMENT THICKNESS = 3,2 mm



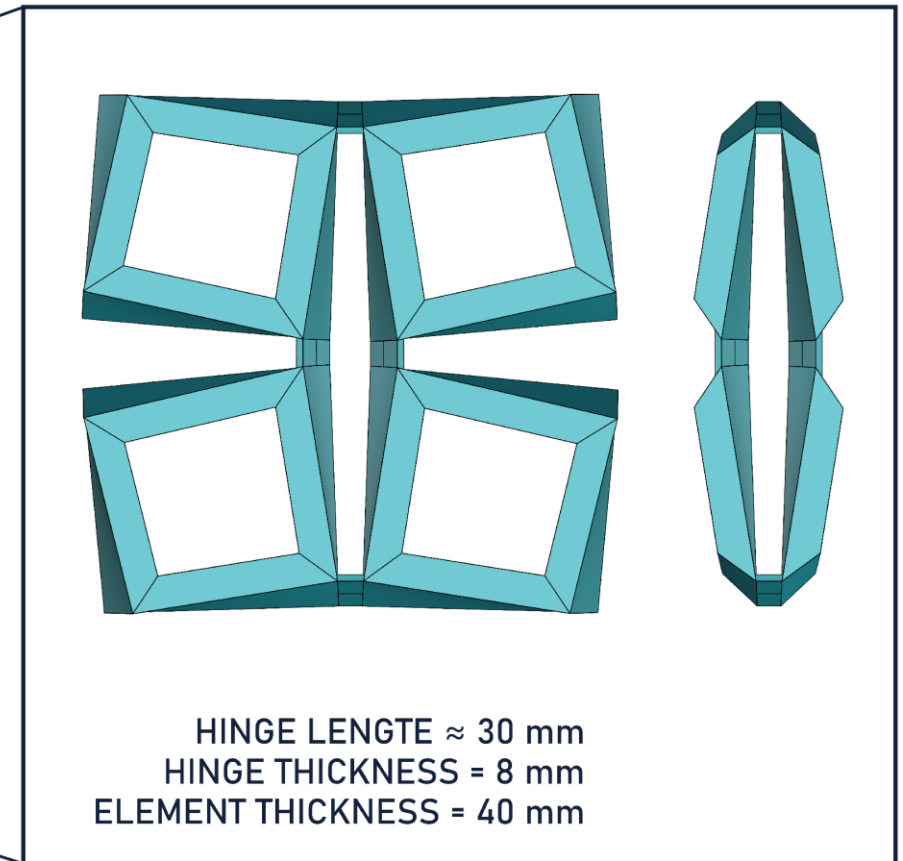
HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 0,6 mm
ELEMENT THICKNESS = 4,8 mm



HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 0,6 mm
ELEMENT THICKNESS = 3,6 mm



HINGE LENGTE \approx 3,0 mm
HINGE THICKNESS = 0,6 mm
ELEMENT THICKNESS = 3,0 mm

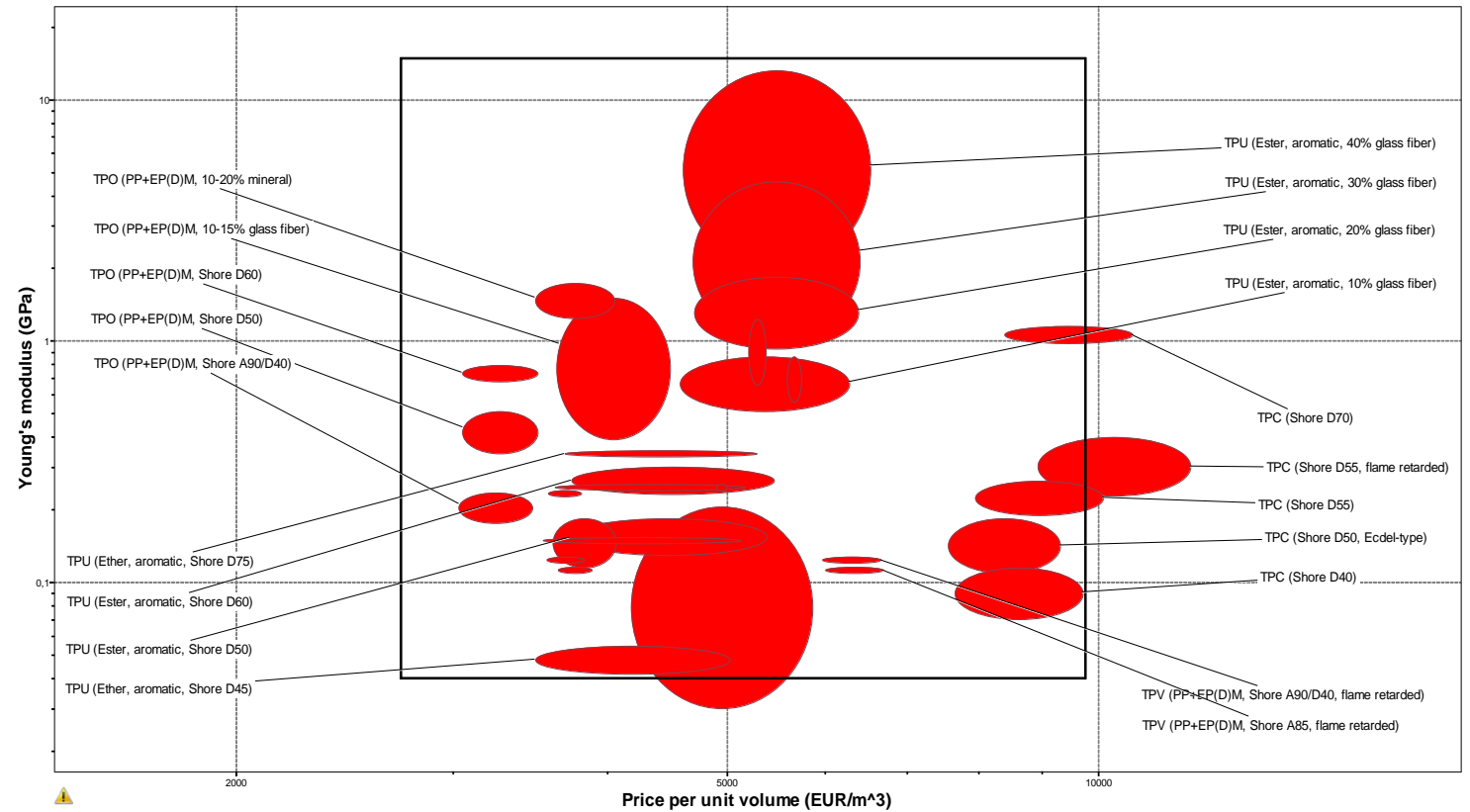


HINGE LENGTE \approx 30 mm
HINGE THICKNESS = 8 mm
ELEMENT THICKNESS = 40 mm

SELECTED CONFIGURATIONS

MATERIAL

- Thermoplastic elastomers
- Young's modulus: 0.01 - 5 GPa
- Adding glass fiber can make TPU stiffer
- Alternative = TPC 70D

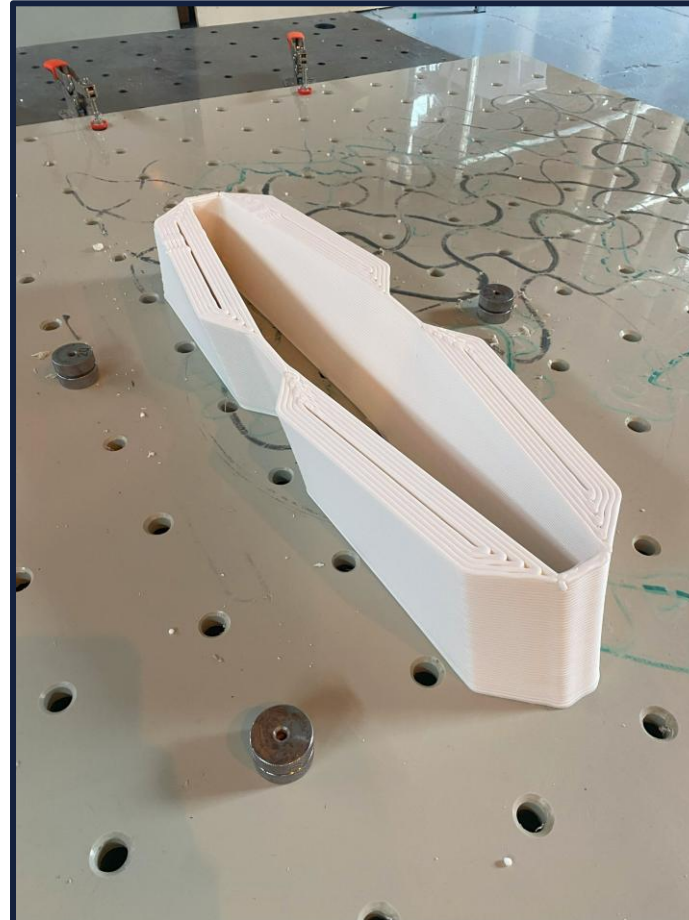


MATERIAL

ESTANE® 3D TPU F98A-030 CR HC PL

Property	Value
Young's (Elastic) Modulus	35 kN / cm ² ≈ 350 MPa
In-plane Shear Modulus	12 kN / cm ² ≈ 120 MPa
Transverse Shear Modulus	4.8 kN / cm ² ≈ 48 MPa
Specific Weight γ	10.7 kN / m ³
Coefficient of Thermal Expansion (linear)	$0.9 - 1.1 \times 10^{-4} / ^\circ\text{C}$
Tensile Strength (XY build)	2.8 kN / cm ² ≈ 28 MPa
Compressive Strength (yield)	1.5 kN / cm ² ≈ 15 MPa est.

Ensoft SO-161-70A



PRINTING PROCESS

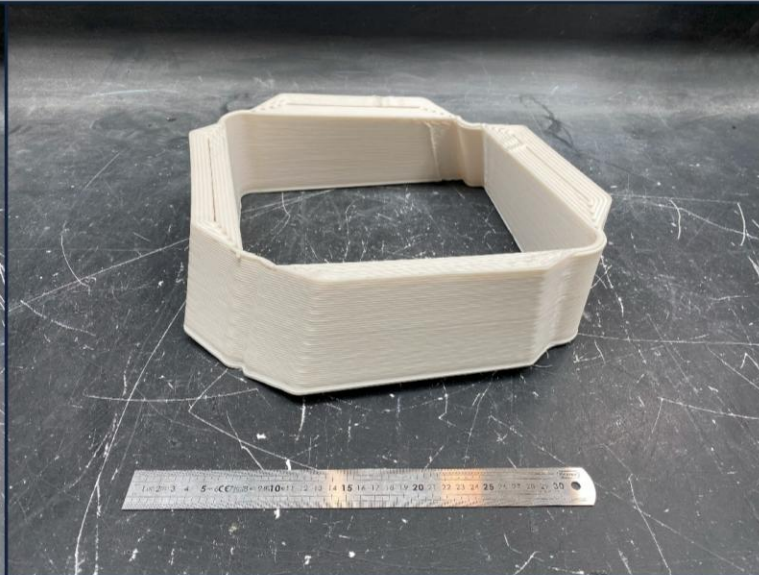


1:1 PROTOTYPES

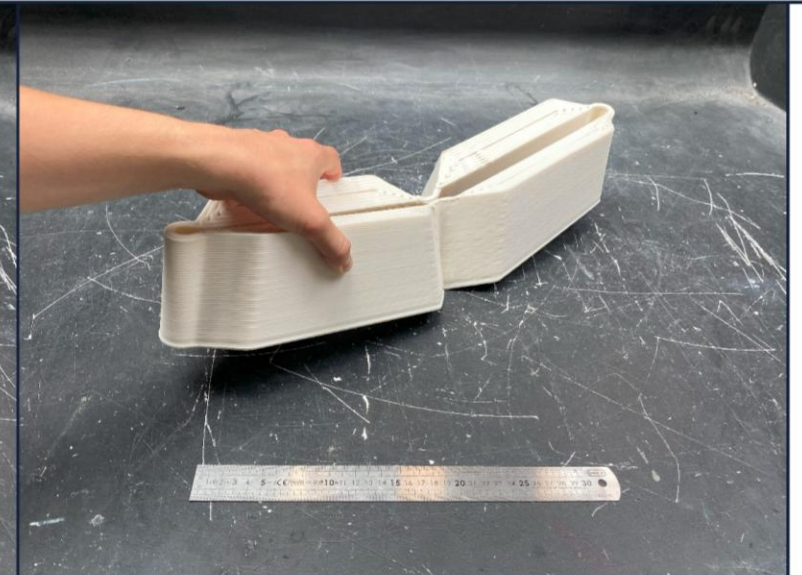
PROTOTYPE 1



FLAT STATE DoD $\approx 0^\circ$



DEPLOYED STATE DoD $\approx 90^\circ$



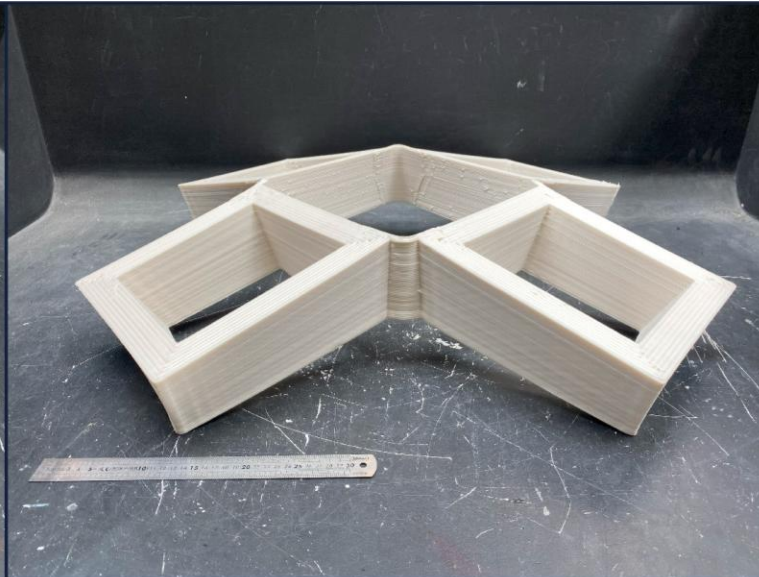
OVER DEPLOYED DoD $\approx 180^\circ$

1:1 PROTOTYPES

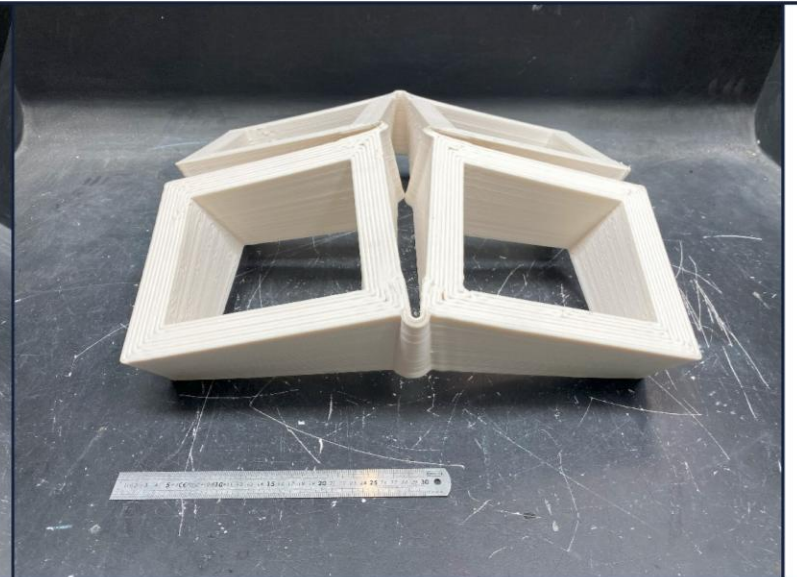
PROTOTYPE 2



FLAT STATE DoD $\approx 0^\circ$



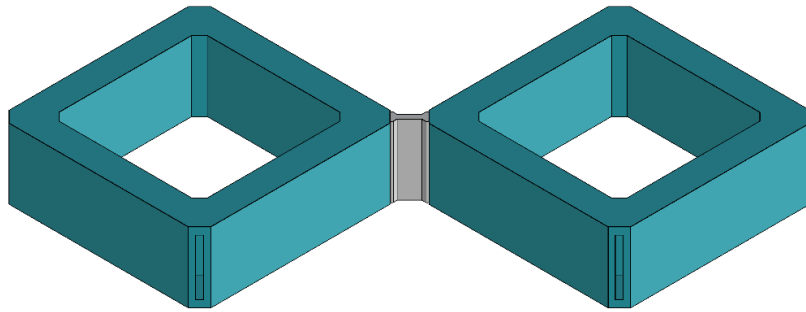
DEPLOYED STATE DoD $\approx 90^\circ$



OVER DEPLOYED DoD $\approx 180^\circ$

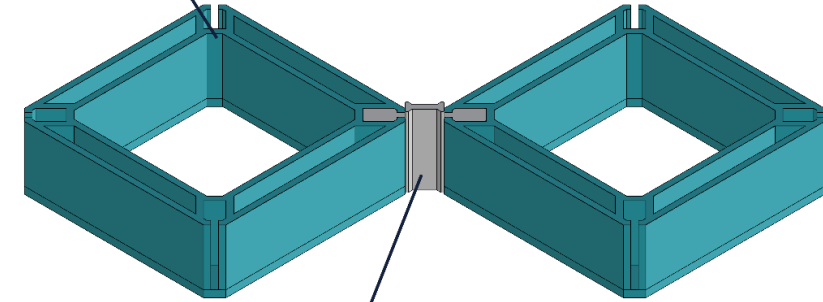
IMPROVEMENTS

DUAL MATERIAL 3D PRINTING



Structure 3D printed in two different materials. Requires dual nozzle

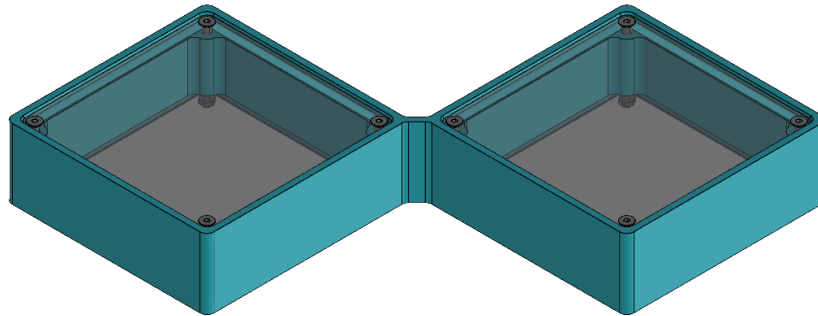
Stiff hollow elements
(PLA or PETG)



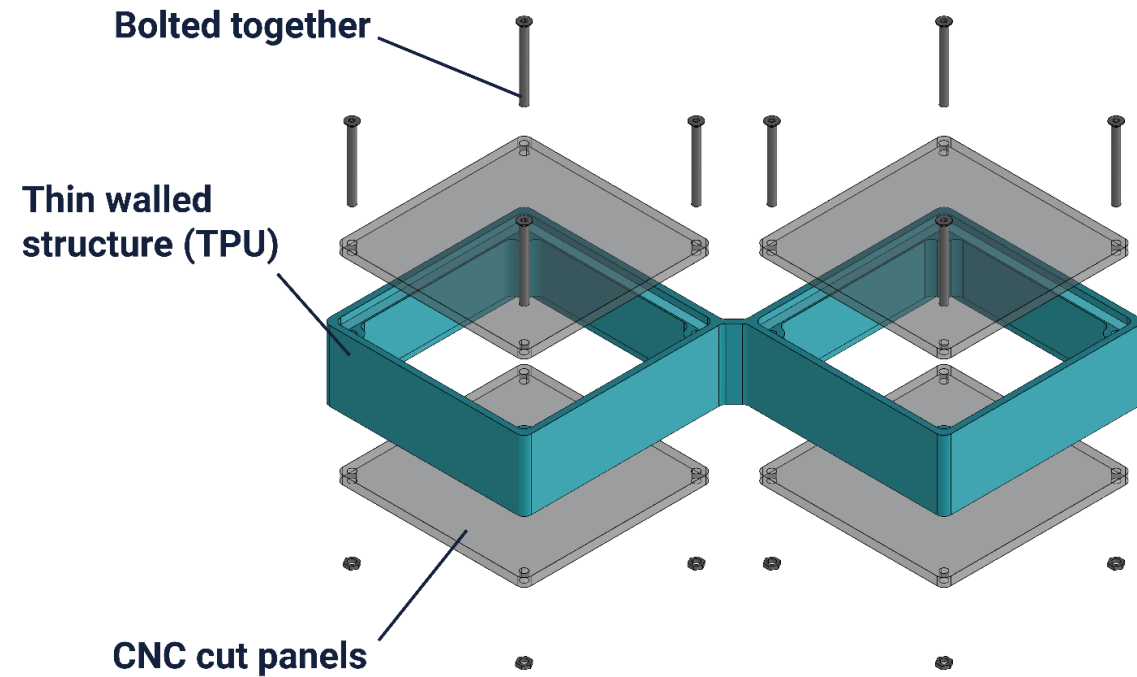
Compliant hinges
(TPU)

IMPROVEMENTS

STIFFENING BY PANELS



Thin printed elements stiffened by panels of a stiff light weight material



CASE STUDY

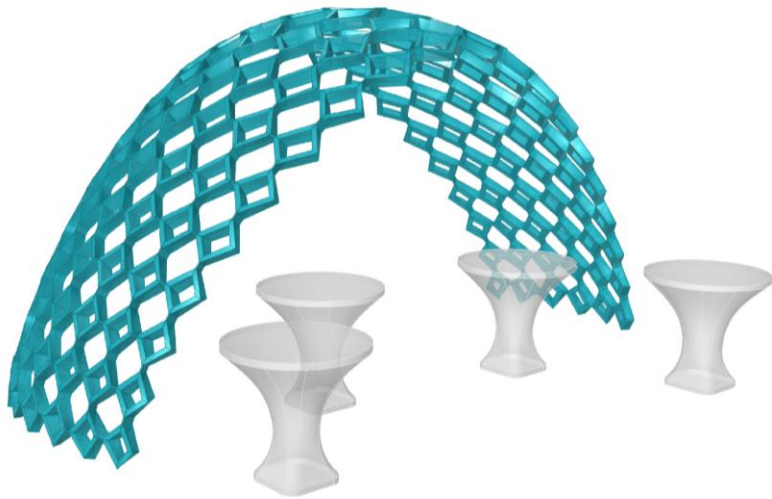
CASE STUDY

Small structure for festival events

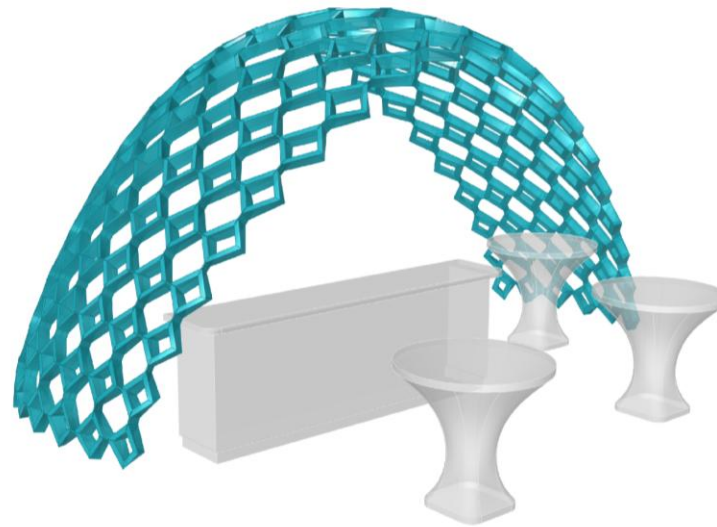
- Build with the knowledge and technology right now
- Demonstrate the ability to span a small distance
- Evaluate whether the system can be deployed quickly and easily



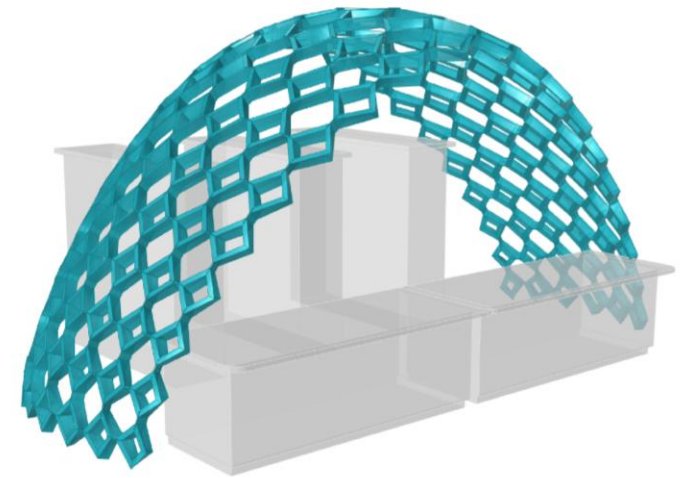
APPLICATION



NETWORKING STAND



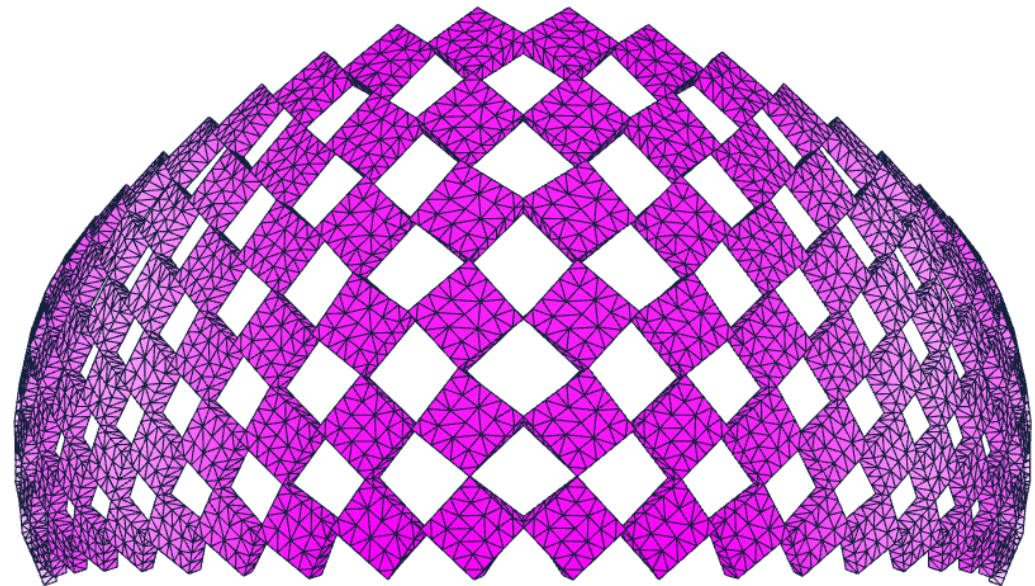
BAR



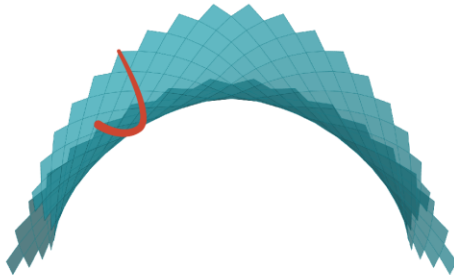
WARDROBE

OPTIMIZATION

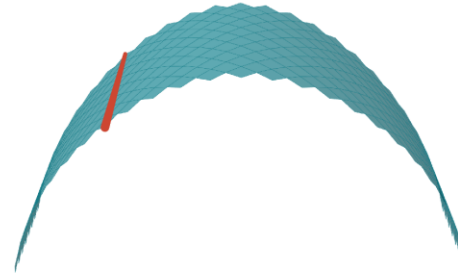
- Optimization Geometry
- Optimization mesh/grid density
- Fitness landscape
 - 2 parameters for x and y
- Deformation as objective
 - Simplified Karamba model
 - Gravity and a vertical load



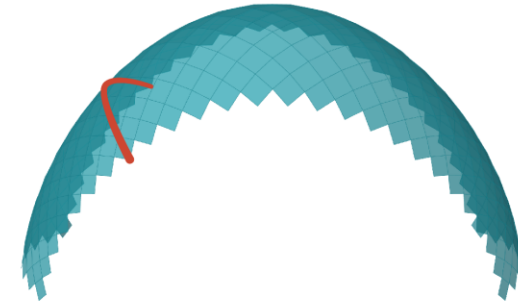
OPTIMISATION PARAMETERS



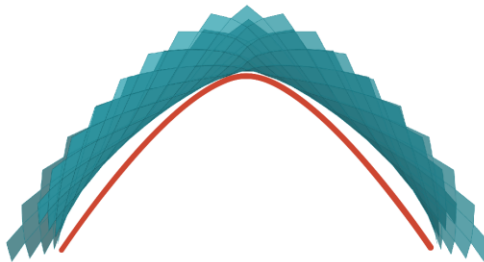
CURVATURE U: 0
CURVATURE V: 0.5



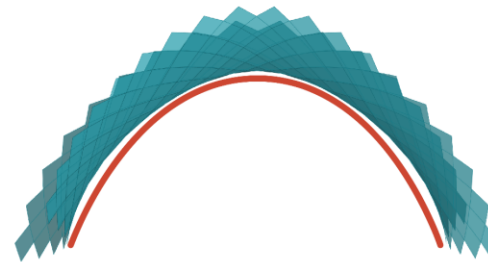
CURVATURE U: 0.5
CURVATURE V: 0.5



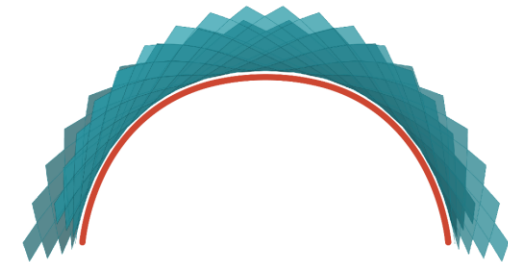
CURVATURE U: 1
CURVATURE V: 0.5



CURVATURE U: 0
CURVATURE V: 0

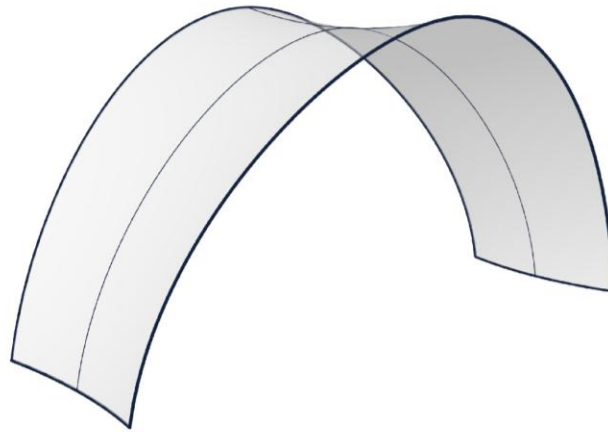
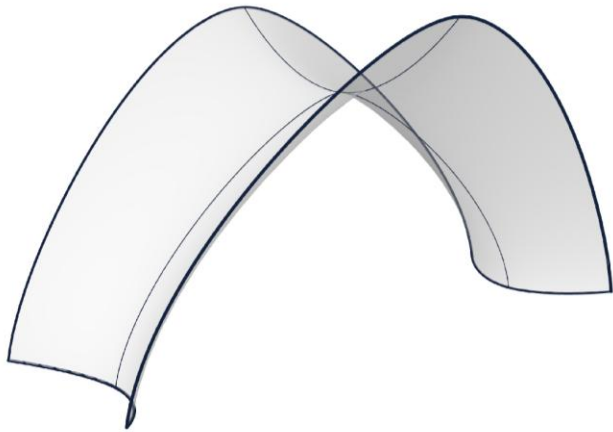


CURVATURE U: 0
CURVATURE V: 0.5

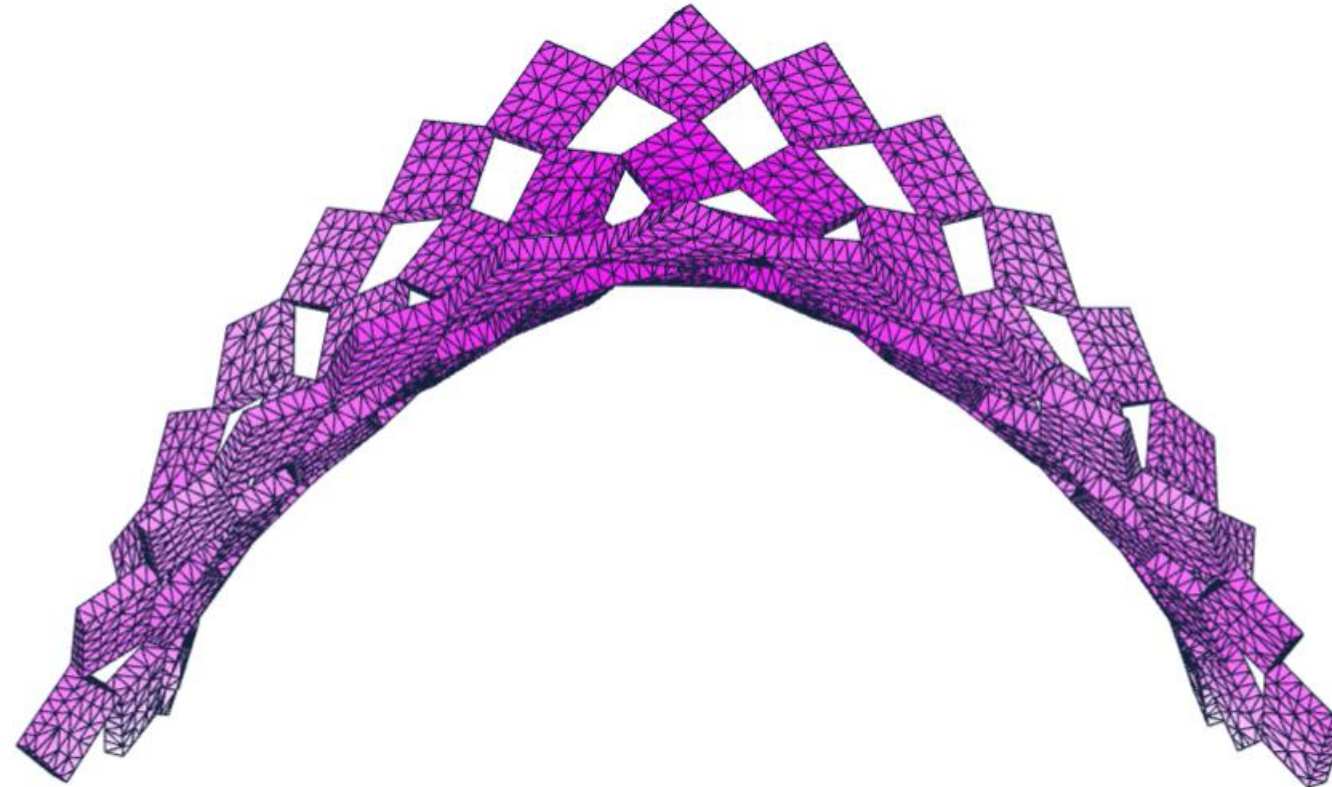


CURVATURE U: 0
CURVATURE V: 1

CASE STUDY GEOMETRY



Optimization Geometry

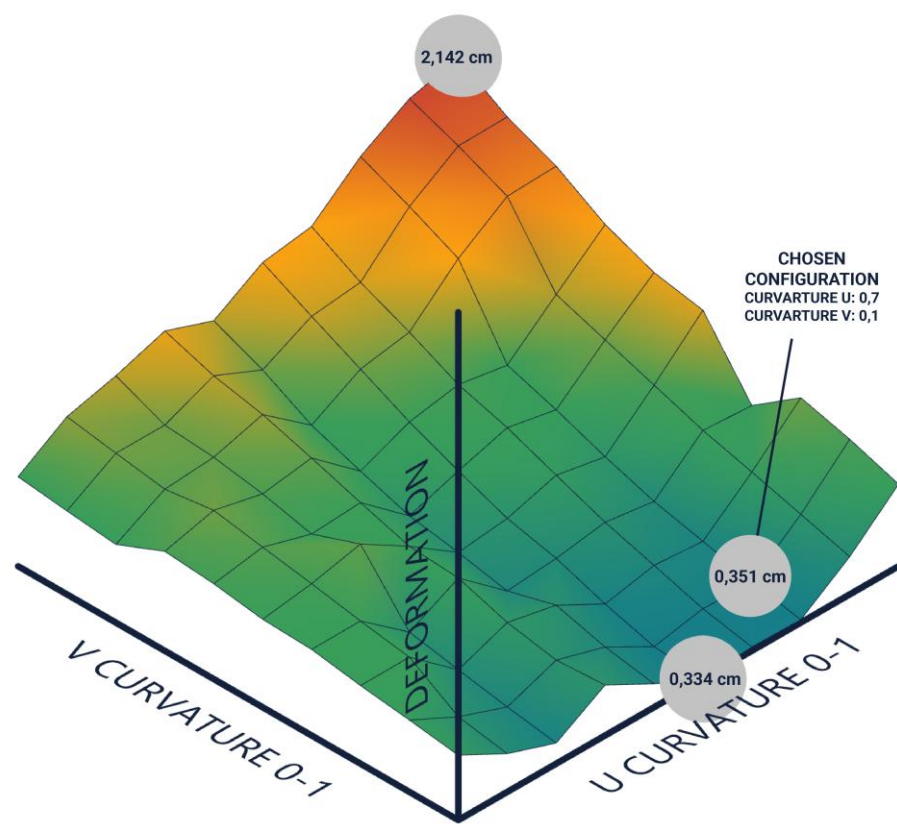


CASESTUDY
CURVATURE U: 0
CURVATURE V: 0

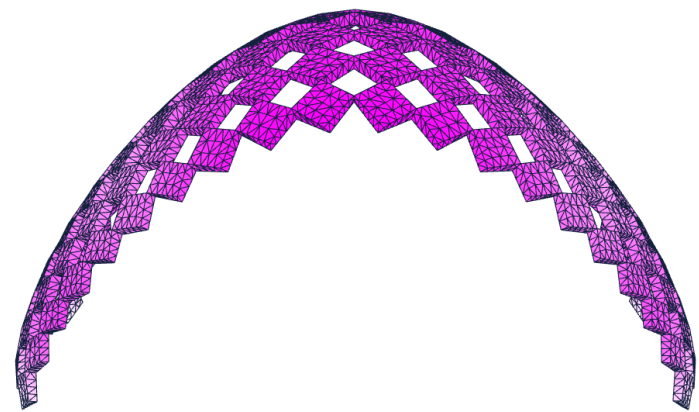
MESH DENSITY
DENSITY U: 0
DENSITY V: 0

MAX DISPLACEMENT
0.795583 cm

Optimization Geometry



CHOSEN CONFIGURATION

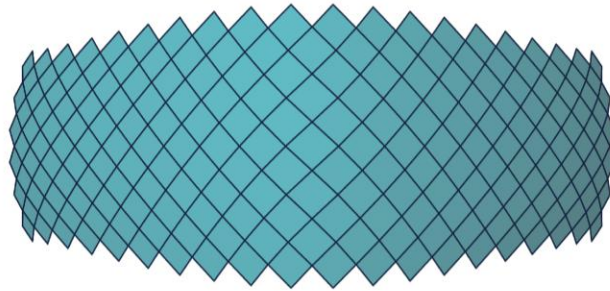


CASESTUDY
CURVATURE U: 0,7
CURVATURE V: 0,1

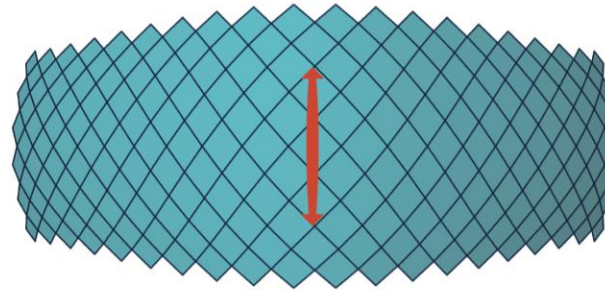
MESH DENSITY
DISTORTION U: 0
DISTORTION V: 0

MAX DISPLACEMENT
0,351 cm

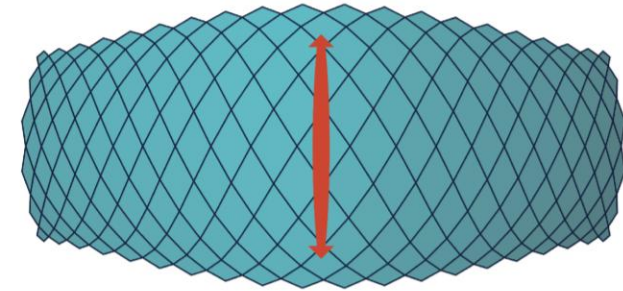
Optimization Mesh Density



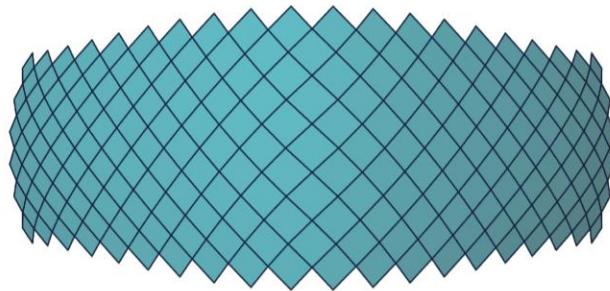
U DISTORTION: 0
V DISTORTION: 0



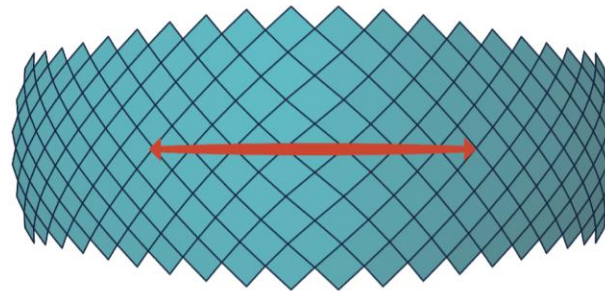
U DISTORTION: 0,5
V DISTORTION: 0



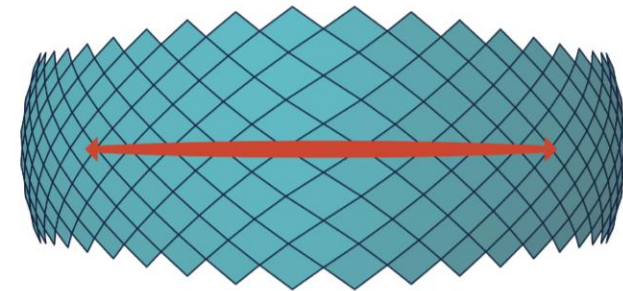
U DISTORTION: 1
V DISTORTION: 0



U DISTORTION: 0
V DISTORTION: 0

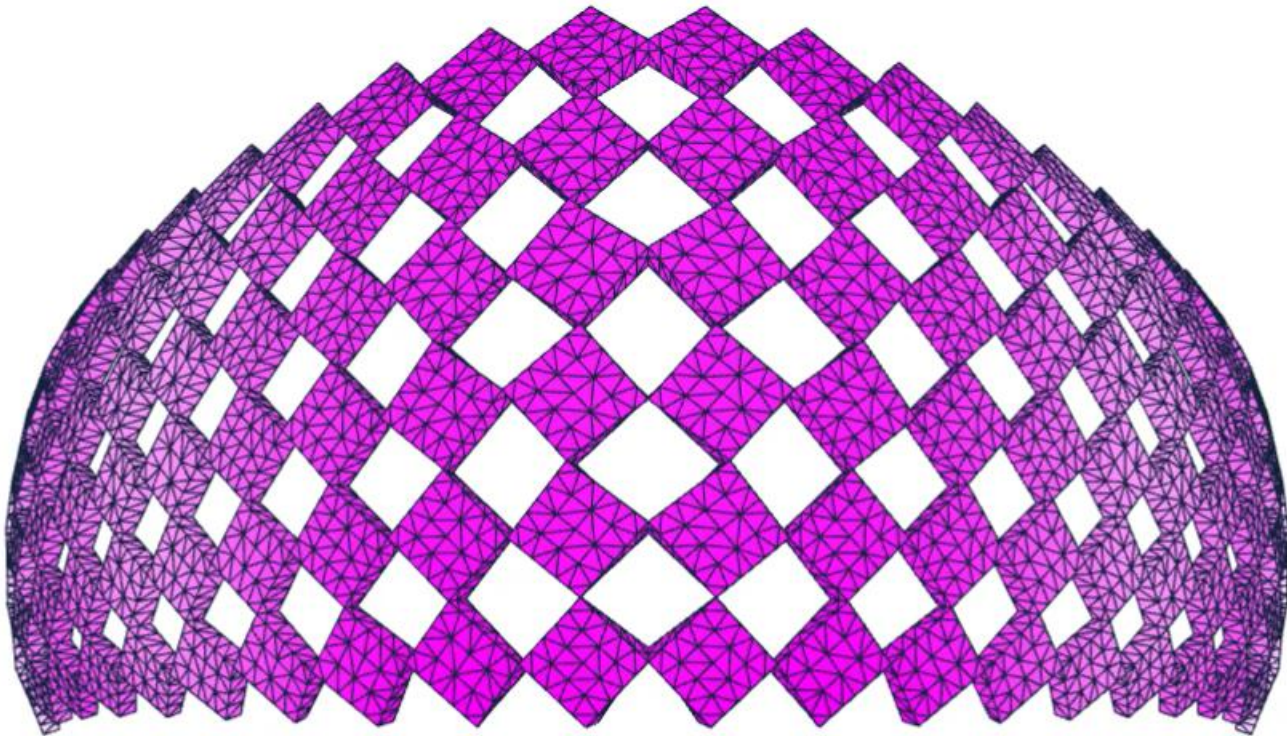


U DISTORTION: 0
V DISTORTION: 0,5



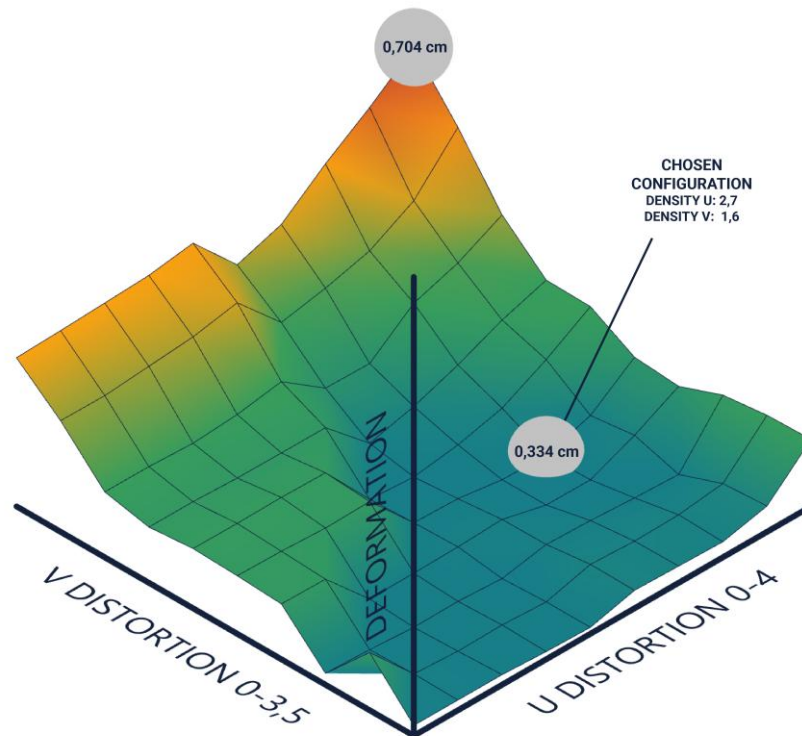
U DISTORTION: 0
V DISTORTION: 1

Optimization Mesh Density

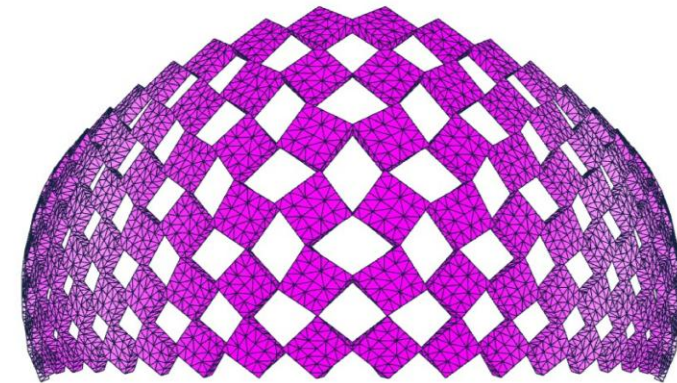


CASESTUDY	MESH DENSITY	MAX DISPLACEMENT
CURVATURE U: 0.7	DENSITY U: 0	0.351222 cm
CURVATURE V: 0.1	DENSITY V: 0	

Optimization Mesh Density



LOWEST MAX DISPLACEMENT



CASESTUDY
CURVATURE U: 0,7
CURVATURE V: 0,1

MESH DENSITY
DENSITY U: 2,722
DENSITY V: 1,556

MAX DISPLACEMENT
0,334 cm

STRUCTURAL ANALYSIS

LARGE DISPLACEMENT ANALYSIS

- Simulate deployment
- Effect of compliant hinges and other material deformation during deployment

STATIC ANALYSIS

- On the deformed model
- Load cases
- Stress/Deformation

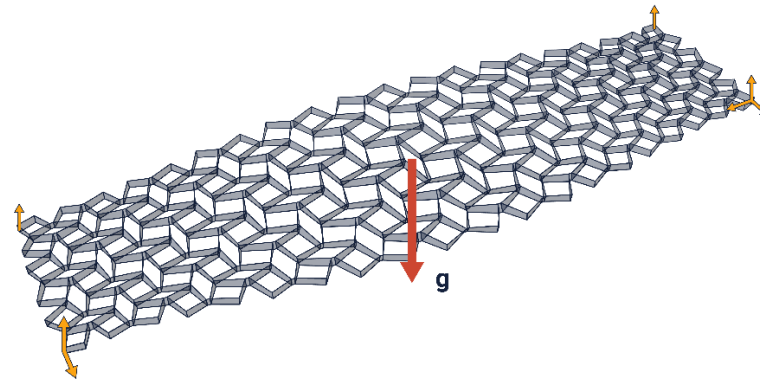
LARGE DISPLACMENT ANALYSIS

DETAILS

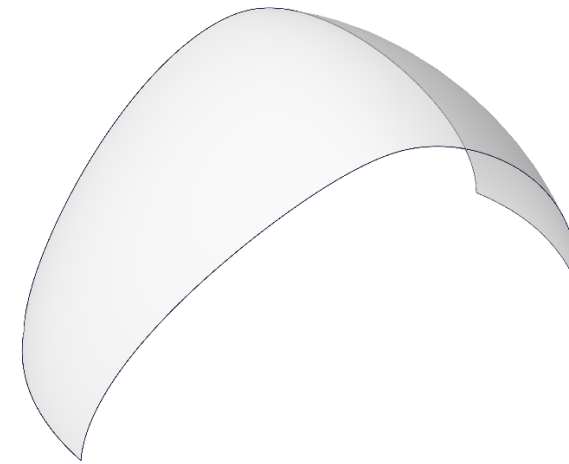
Material:
ESTANE® 3D TPU F98A-030 CR

THK rigid elements: 60 mm
THK hinge: 10 mm
length hinge: 30 mm

FLAT STATE



TARGET SHAPE



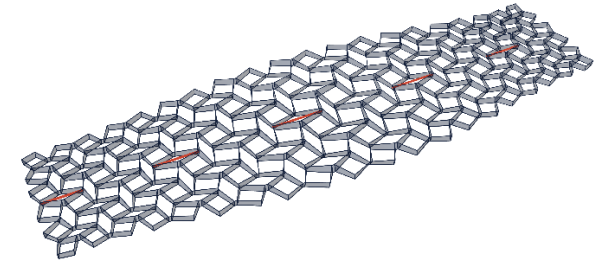
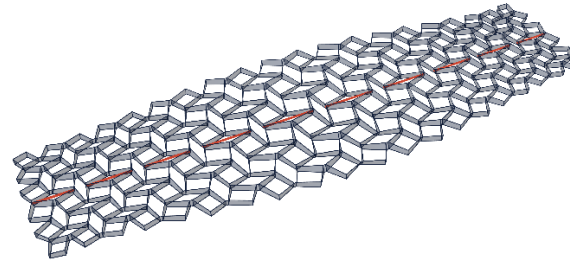
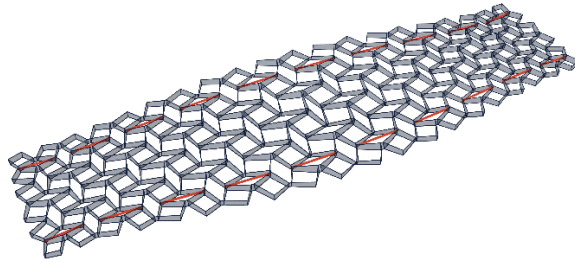
LARGE DISPLACEMENT ANALYSIS

20 TENSION POINTS

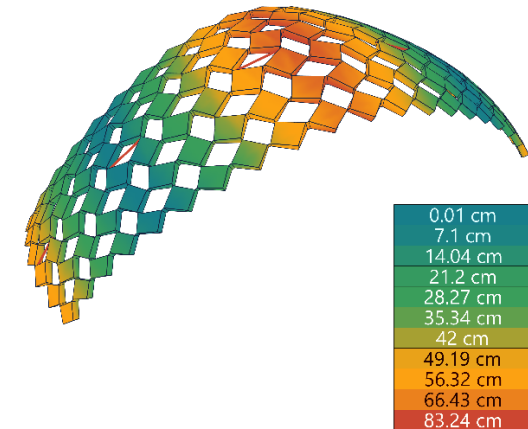
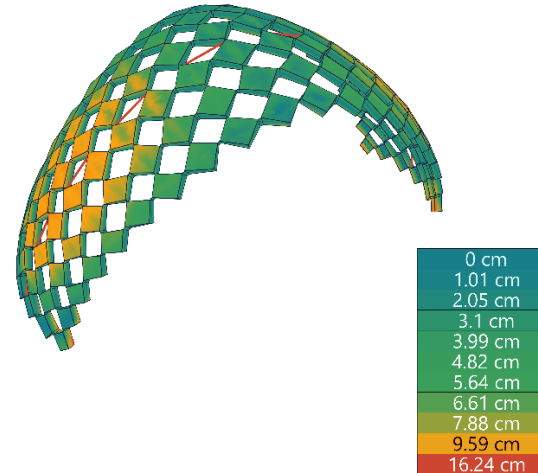
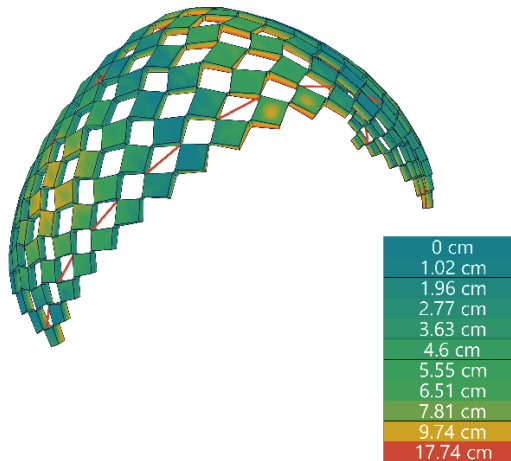
10 TENSION POINTS

5 TENSION POINTS

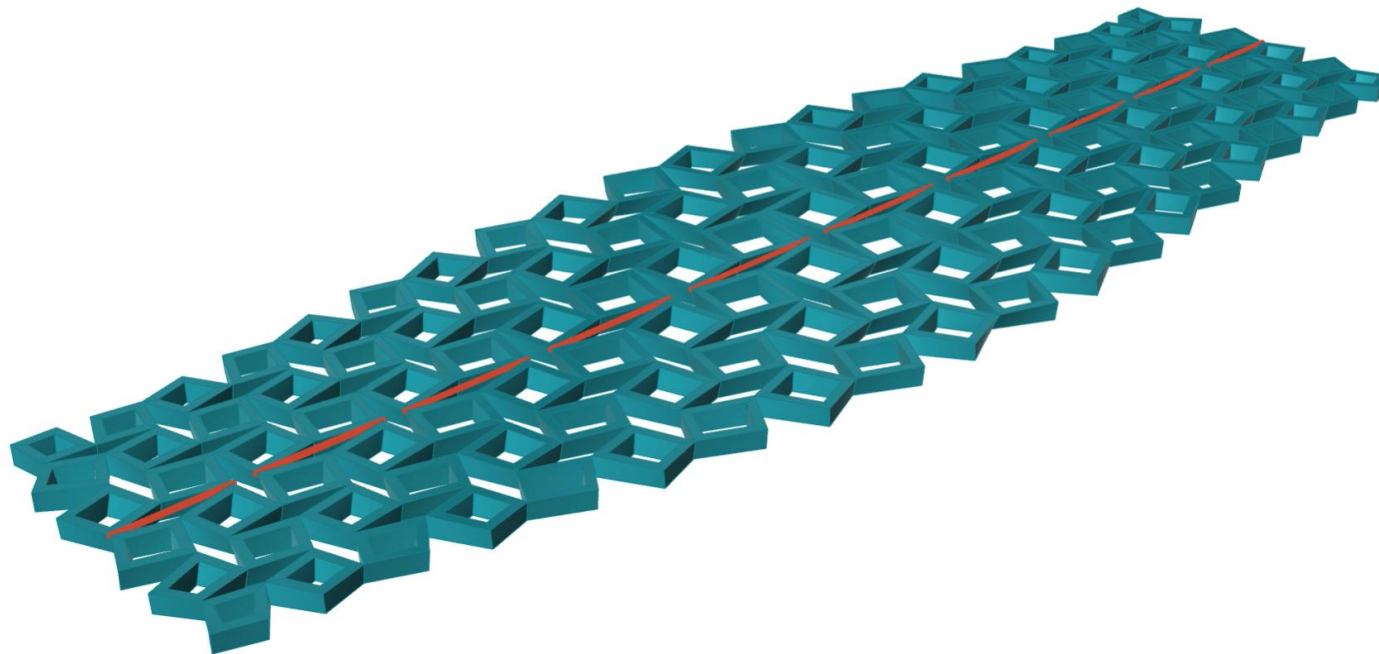
TENSION SYSTEM



CONTOUR PLOT

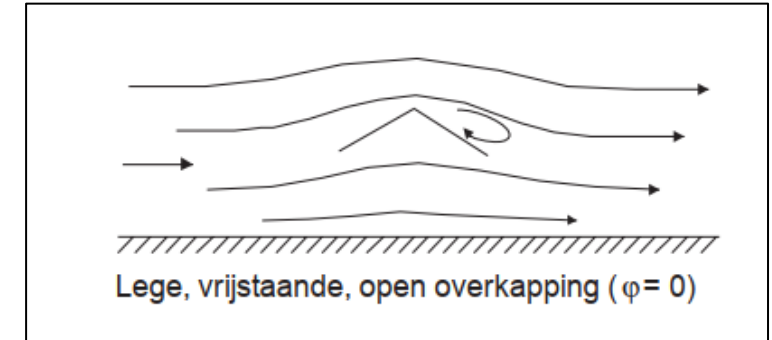


DEPLOYMENT CASE STUDY



STATIC ANALYSIS

- Wind load simplified to a two-sided sloping canopy with free airflow



Ultimate limit state:					
ULS 1	1,22	Selfweight			
ULS 2	1,08	Selfweight	+	1,35	Wind 0°
ULS 3	1,08	Selfweight	+	1,35	Wind 90°
Serviceability limit state:					
SLS 1	1,0	Selfweight			
SLS 2	1,0	Selfweight	+	1,0	Wind 0°
SLS 3	1,0	Selfweight	+	1,0	Wind 90°

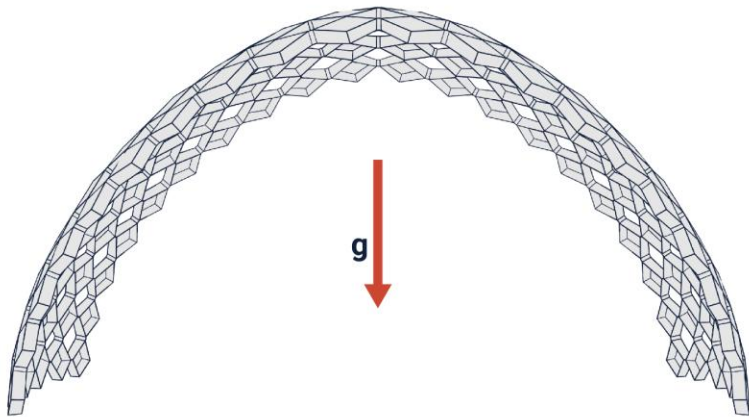
Tabel NB.5 — Extreme stuwdruk in kN/m² als functie van de hoogte

Hoogte m	Gebied I			Gebied II			Gebied III	
	Kust	Onbebouwd	Bebouwd	Kust	Onbebouwd	Bebouwd	Onbebouwd	Bebouwd
1	0,93	0,71	0,69	0,78	0,60	0,58	0,49	0,48
2	1,11	0,71	0,69	0,93	0,60	0,58	0,49	0,48
3	1,22	0,71	0,69	1,02	0,60	0,58	0,49	0,48
4	1,30	0,71	0,69	1,09	0,60	0,58	0,49	0,48
5	1,37	0,78	0,69	1,14	0,66	0,58	0,54	0,48
6	1,42	0,84	0,69	1,19	0,71	0,58	0,58	0,48
7	1,47	0,89	0,69	1,23	0,75	0,58	0,62	0,48
8	1,51	0,94	0,73	1,26	0,79	0,62	0,65	0,51
9	1,55	0,98	0,77	1,29	0,82	0,65	0,68	0,53
10	1,58	1,02	0,81	1,32	0,85	0,68	0,70	0,56
15	1,71	1,16	0,96	1,43	0,98	0,80	0,80	0,66
20	1,80	1,27	1,07	1,51	1,07	0,90	0,88	0,74

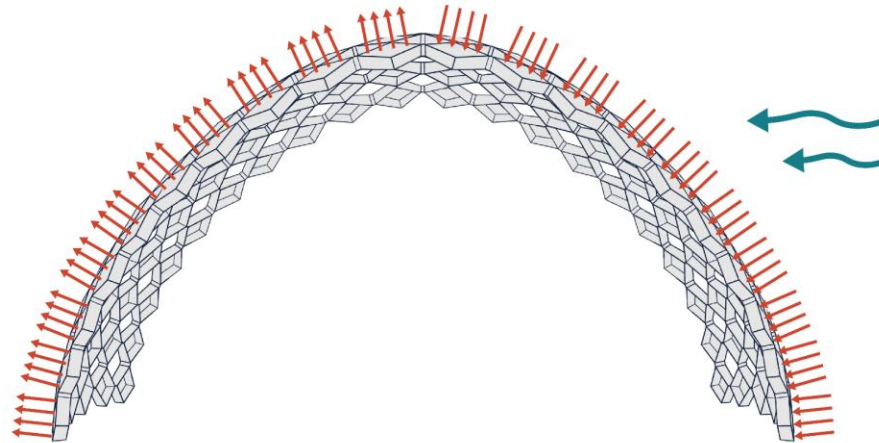
STATIC ANALYSIS

**LOAD CASE 1
GRAVITY**

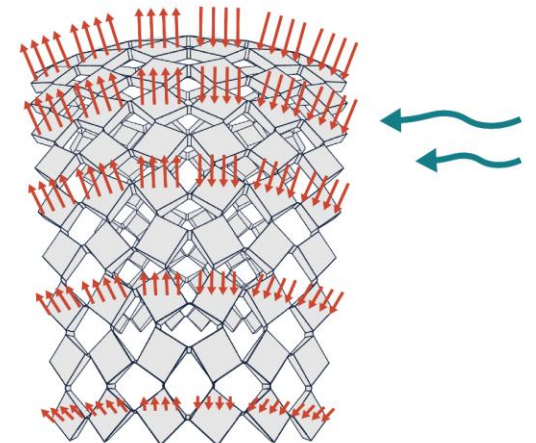
LOADS



**LOAD CASE 2
WIND 90°**



**LOAD CASE 3
WIND 0°**



MATERIAL: ESTANE® 3D TPU F98A-030 CR

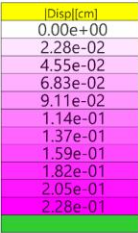
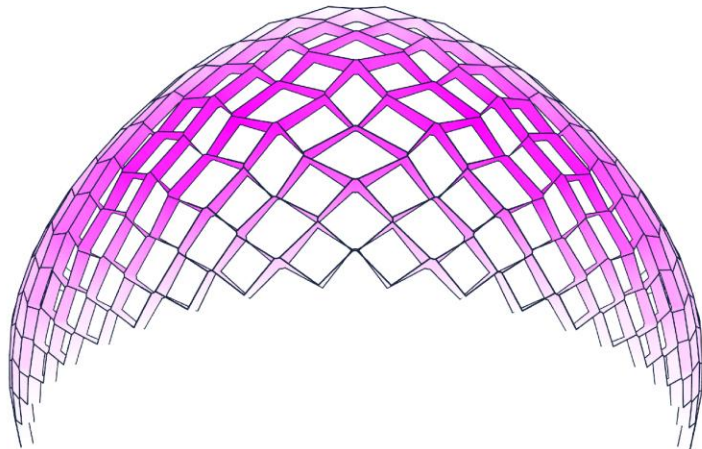
STATIC ANALYSIS

LOAD CASE 1
GRAVITY

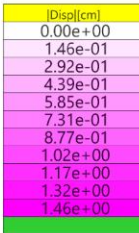
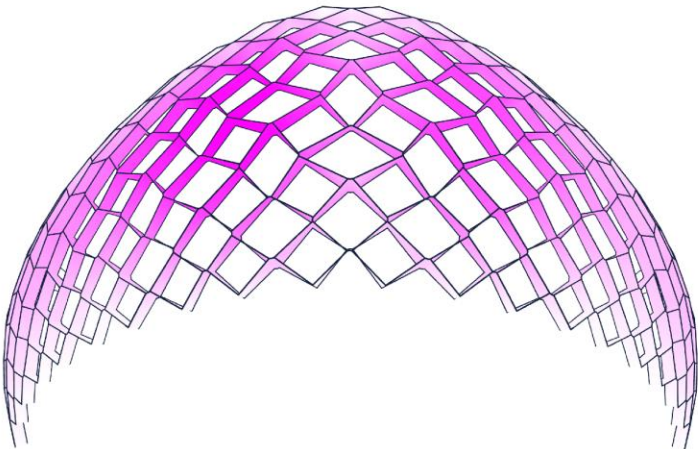
LOAD CASE 2
WIND 90°

LOAD CASE 3
WIND 0°

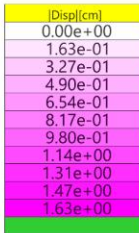
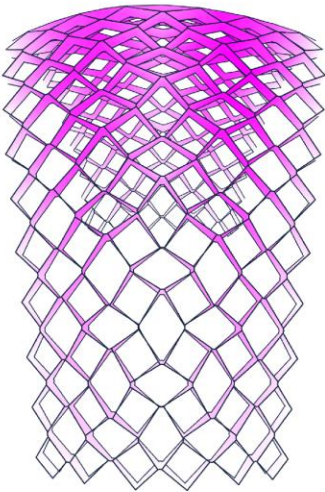
DISPLACEMENT SLS



UNITY DISP. : 2,28 mm



UNITY DISP. : 14,6 mm

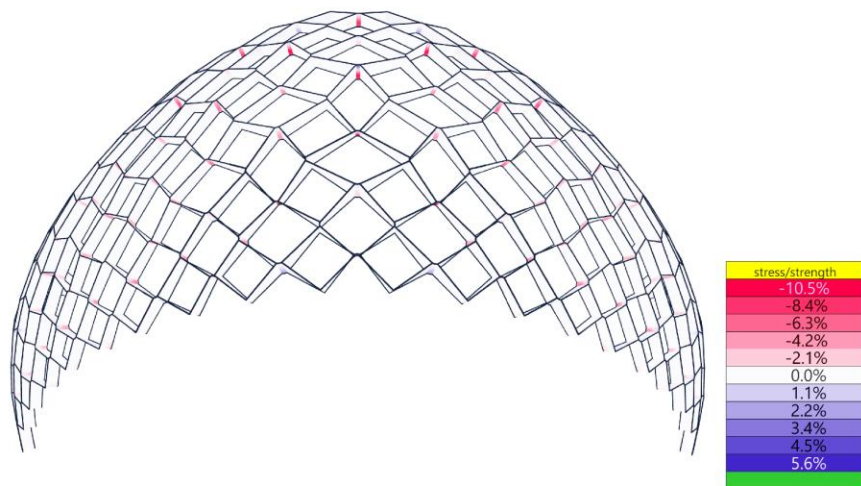


UNITY DISP. : 16,3 mm

MATERIAL: ESTANE® 3D TPU F98A-030 CR

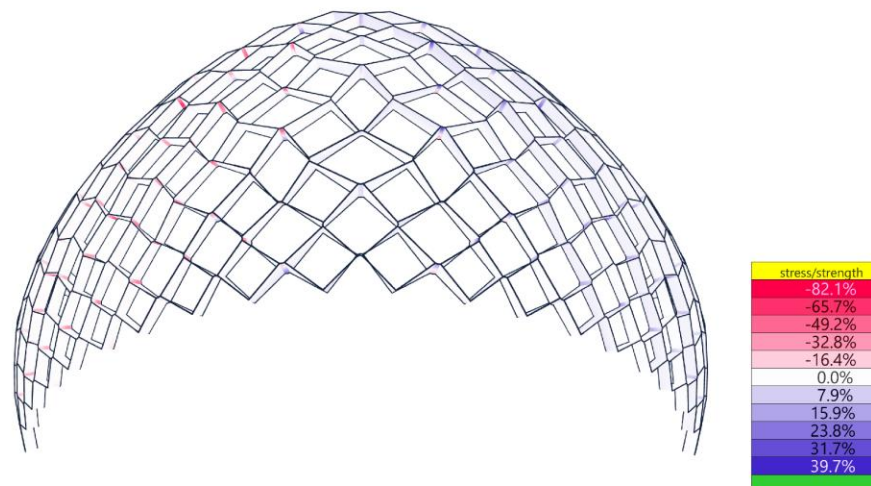
STATIC ANALYSIS

**LOAD CASE 1
GRAVITY**



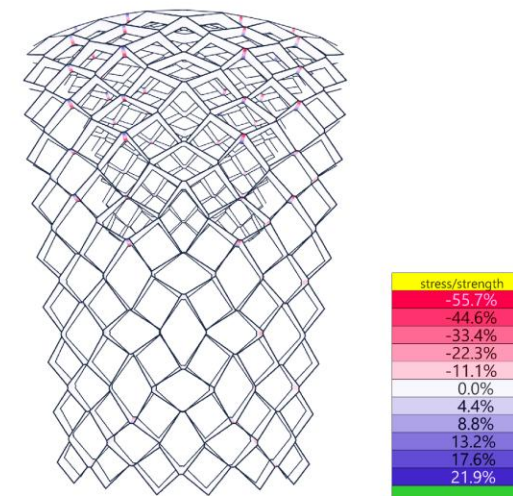
UNITY MAX: 5.6% MIN: -10.5%

**LOAD CASE 2
WIND 90°**



UNITY MAX: 39.7% MIN: -82.1%

**LOAD CASE 3
WIND 0°**



UNITY MAX: 21.9% MIN: -55.7%

MATERIAL: ESTANE® 3D TPU F98A-030 CR

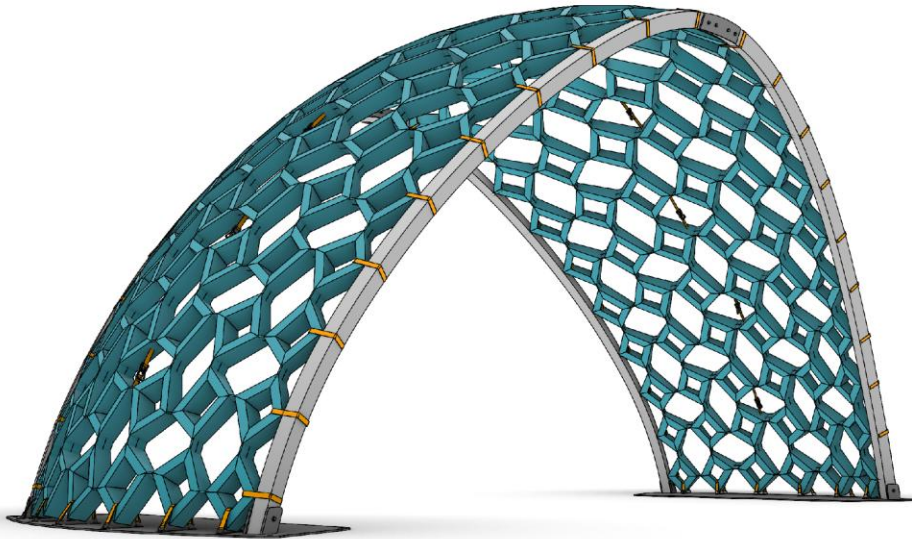
NOTES

Important notes/further research:

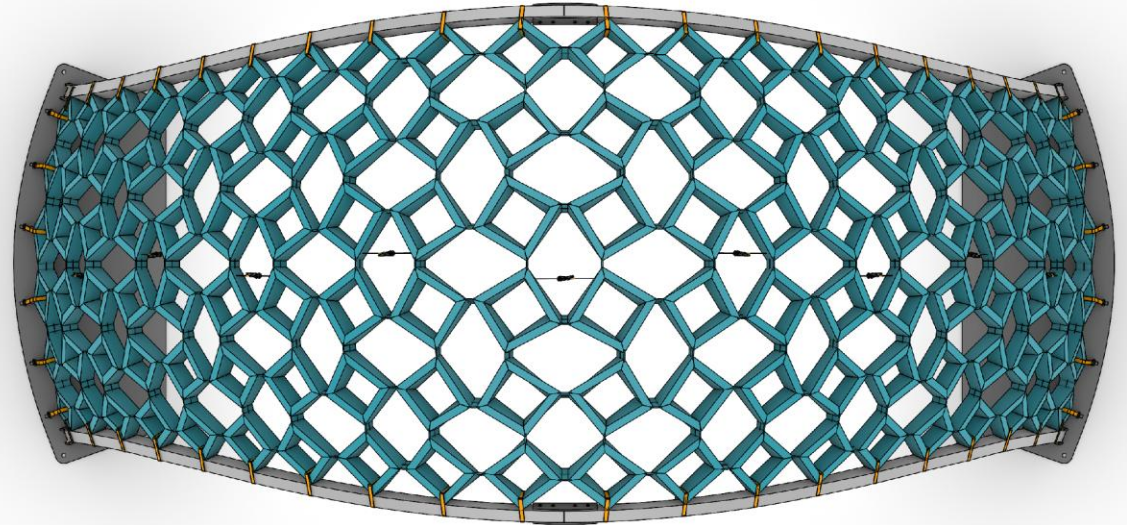
- Stresses from the deployment process (during load cases).
- No material safety factors have been applied (none exist for the selected material).
- Material is assumed isotropic, despite 3D printed parts typically being anisotropic.

MODEL DETAILS

A. CASE STUDY DESIGN

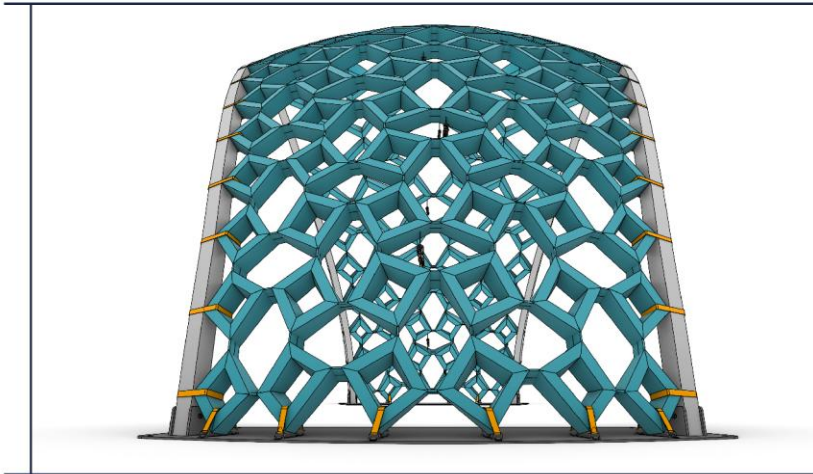


D. TOP VIEW

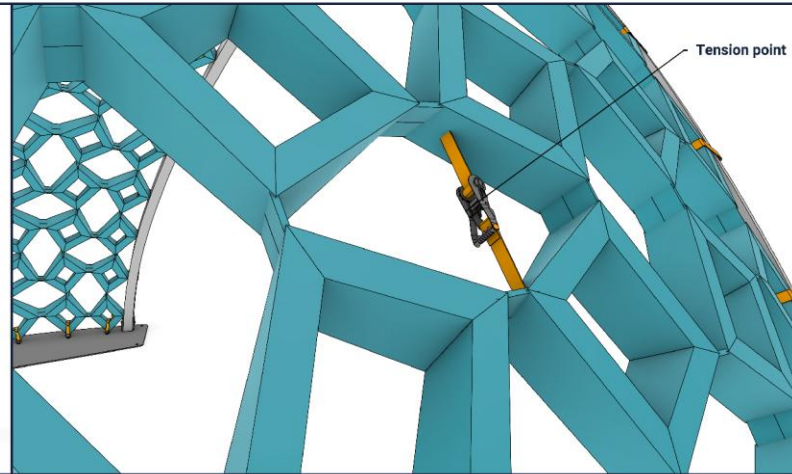


MODEL DETAILS

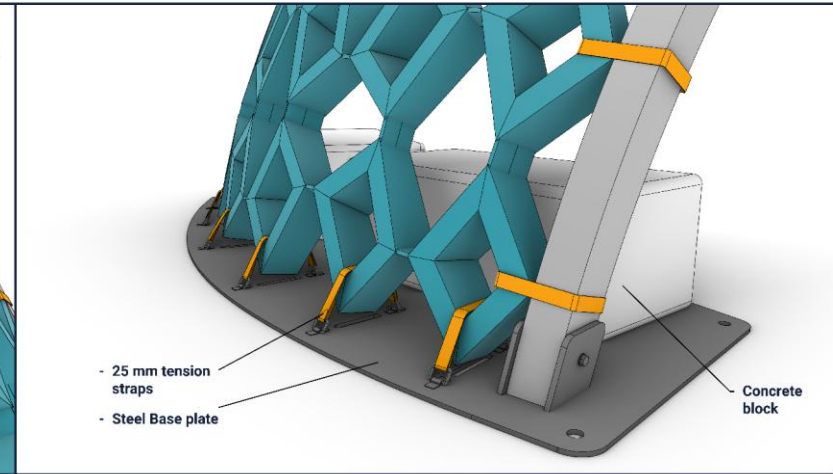
B. SIDE VIEW



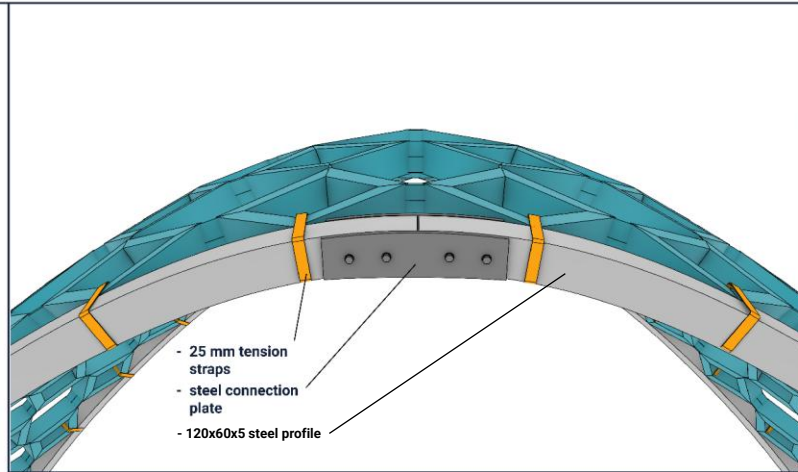
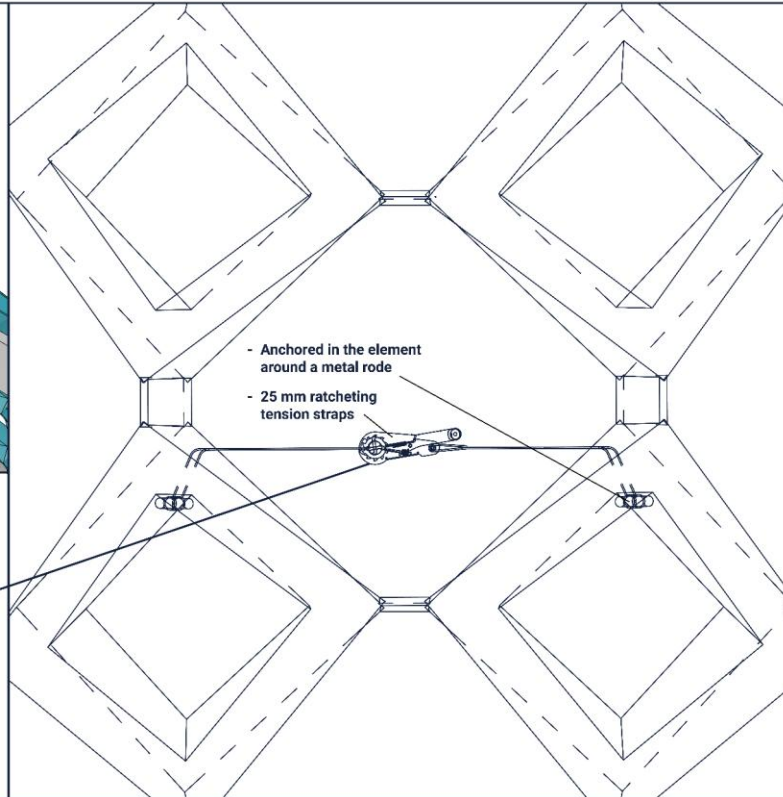
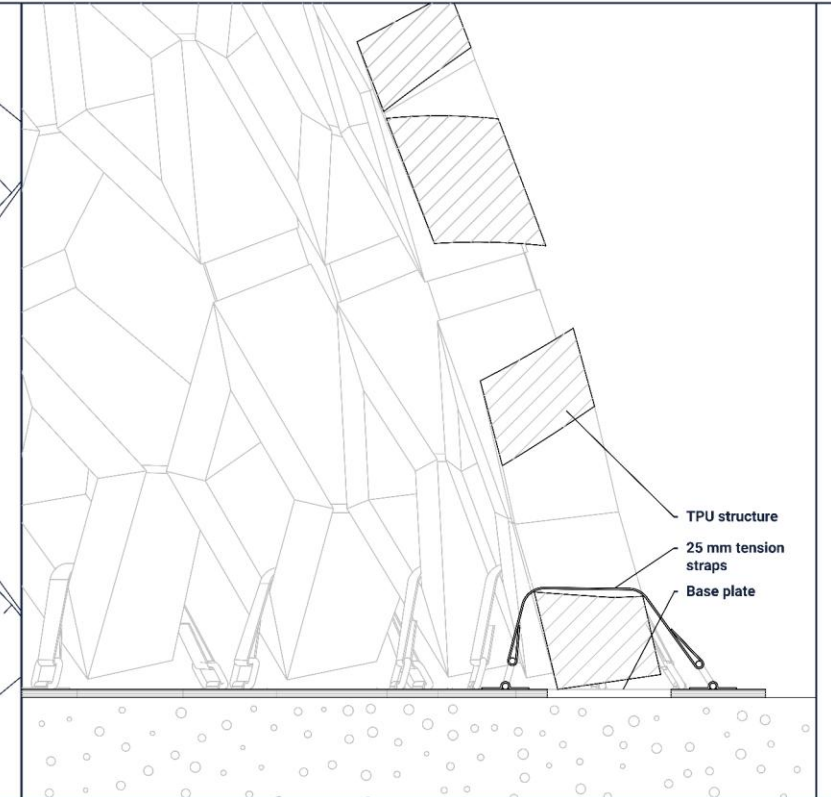
E. TENSION POINT



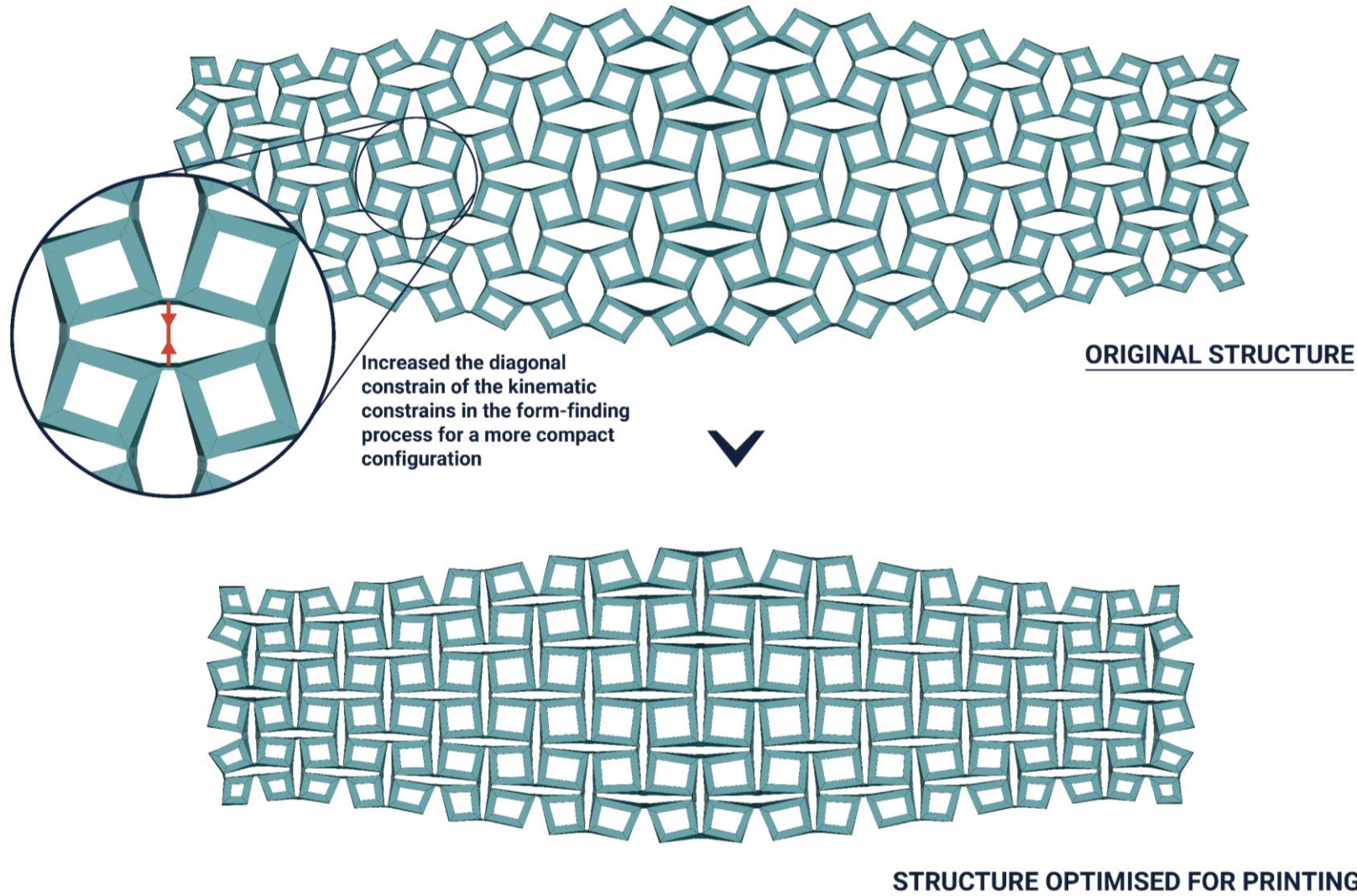
F. BASE PLATE



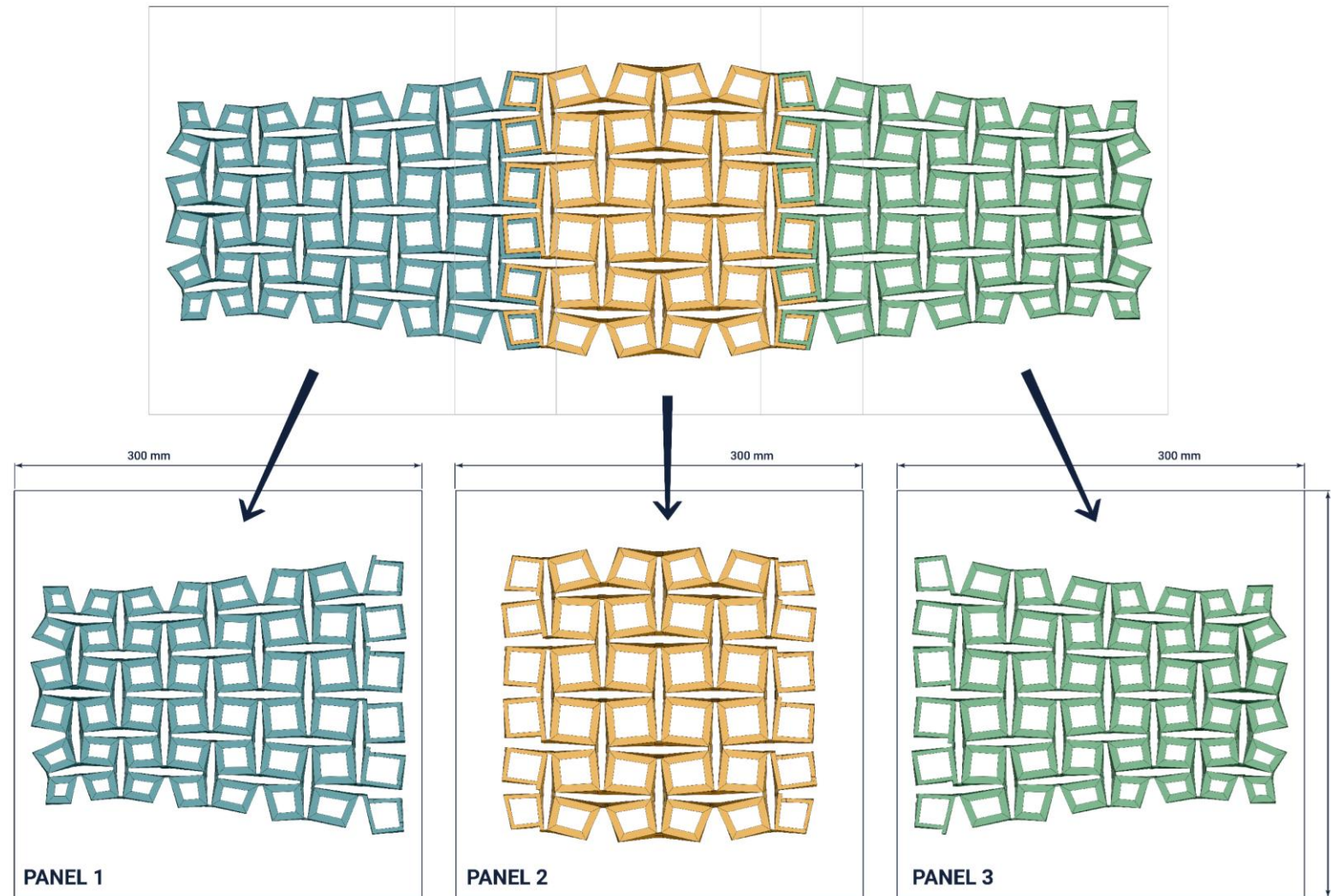
MODEL DETAILS

C. BEAM CONNECTION**TENSION STRAP****G. 1:2 DETAIL TENSION****H. 1:2 DETAIL BASE**

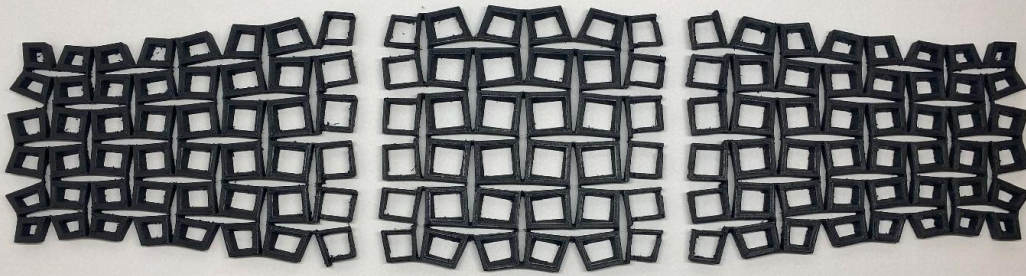
ADJUST FLAT STATE



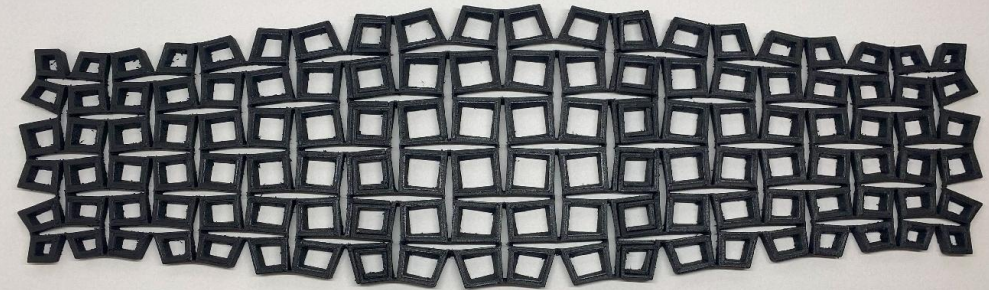
DIVISION IN PANELS



CONNECTION ELEMENTS

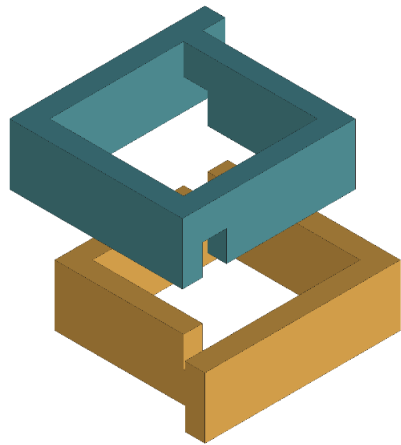


PANELS SEGMENTS

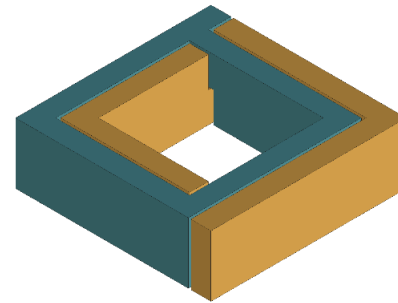


ASSEMBLED STRUCTURE

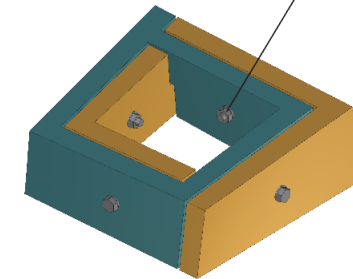
CONNECTION ELEMENTS



2 Interlocking parts



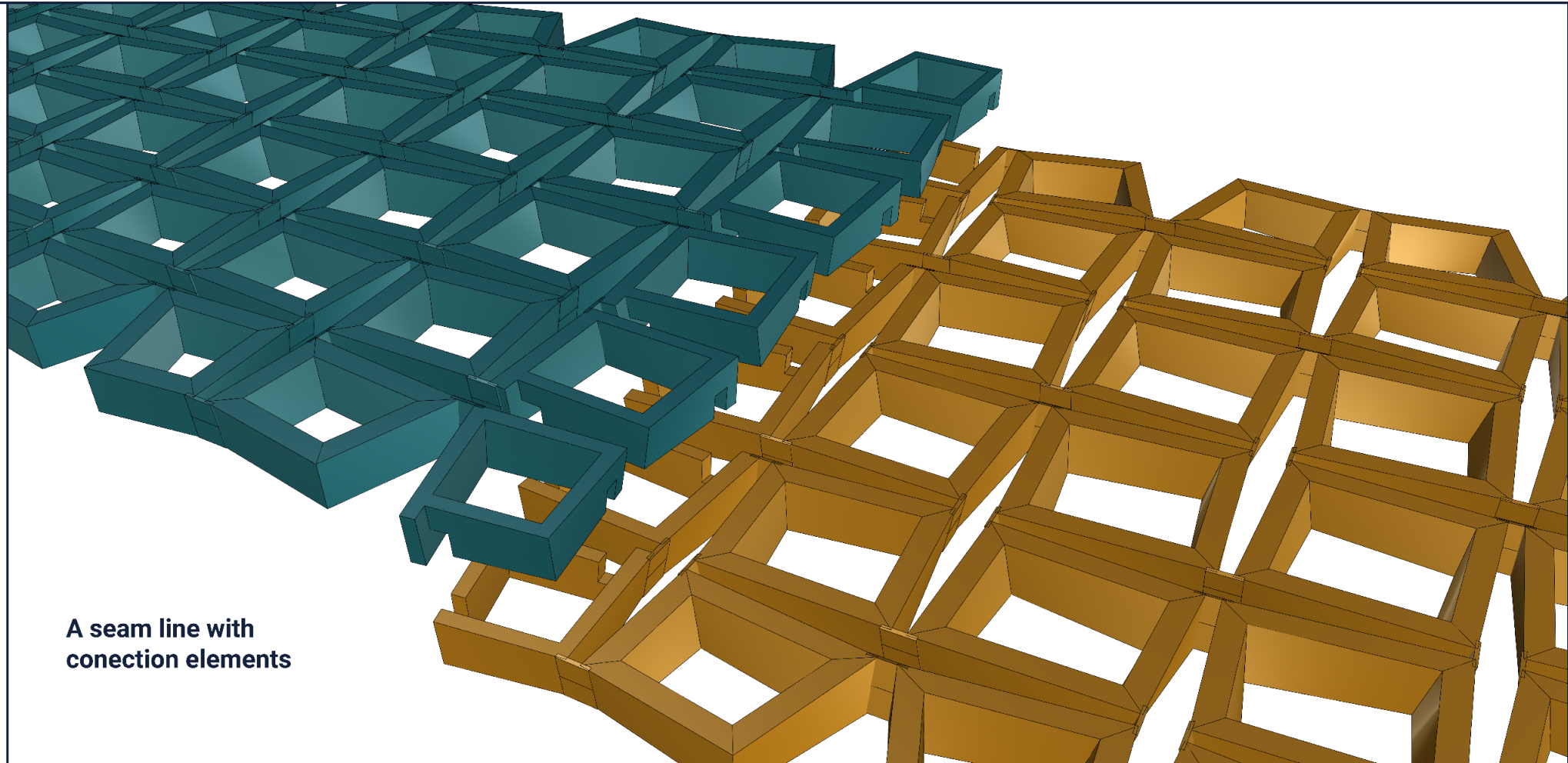
Join together into one element



Morphed into the required geometry

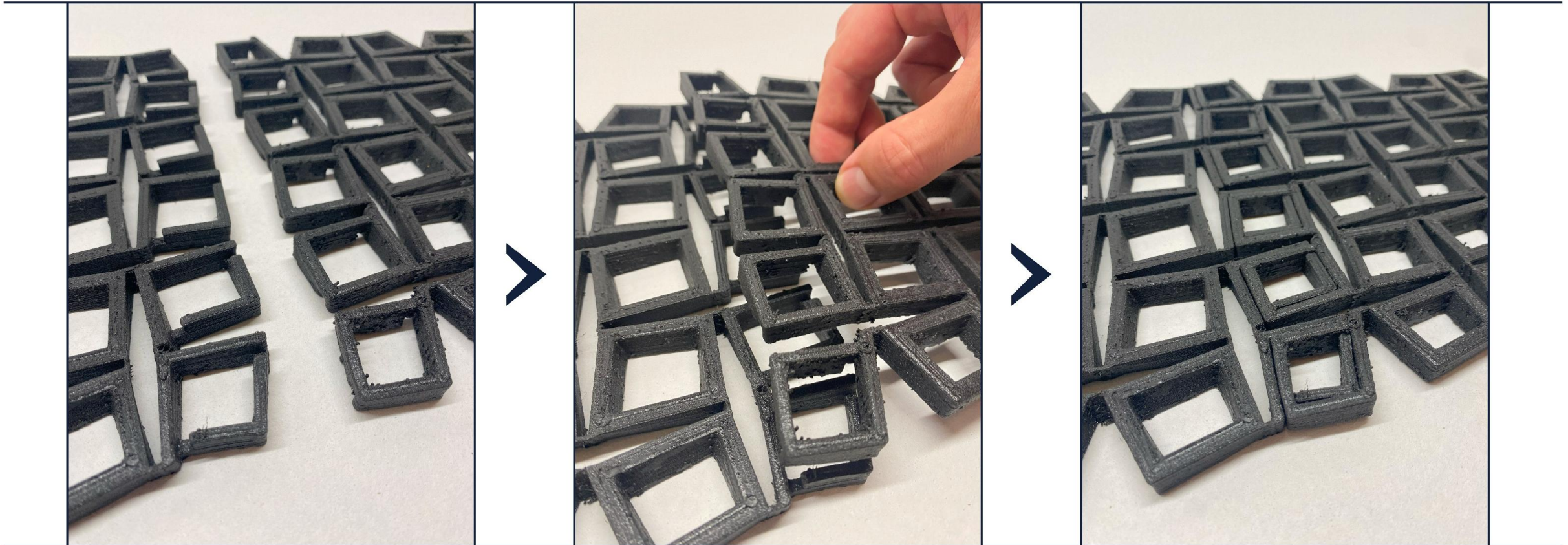
Can secured with bolts from the side

CONNECTION ELEMENTS



A seam line with
conection elements

CONNECTION ELEMENTS



CONCLUSION

THE TOOL

- Predictive design tool/ inverse design
- Tested on a variety of surfaces
- Methode shows potential for other kinematic structures/patterns

SCALING UP

- Possible to manufacture on a larger scale
- Weight issue/ alternative designs
- Further research in materials balance structural stiffness and complaint hinges

CASE STUDY

- Design shelter on a architectural scale
- More research on stresses during deployment

END

Questions?